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Space Station

Mission Planning System Development Study

Final Report *Volume II - Technical Report*

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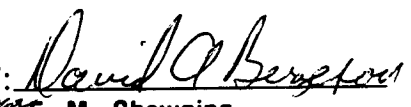
Mission Planning System (MPS) Development Study

Final Report Volume II - Technical Report

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Section 1

INTRODUCTION

1.1 PURPOSE

The purpose of this volume is to provide a detailed description of the results of the Space Station (SS) Mission Planning System (MPS) Development Study.

1.2 SCOPE

This volume includes a description of the overall Study objectives and approach in Section 2, a programmatic summary of Study activities and accomplishments in Section 3, a detailed presentation of individual task activities, methods and accomplishments in Sections 4 through 8, and a presentation of Study conclusions and recommendations in Section 9. Major products of the Study are contained in this volume and in Volume III, SS MPS Software Development Plan.

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Section 2

STUDY OBJECTIVES AND APPROACH

2.1 OBJECTIVES

The basic objective of the SS MPS Development Study was to define a baseline Space Station mission planning concept and the associated hardware and software requirements for the system. Specific objectives in support of the basic objective were the following:

a. Develop a mission planning concept which is consistent with the overall Space Station operations philosophy.

b. Define and assess the capability of the Spacelab mission planning system software for use in Space Station mission planning consistent with the concept developed under objective a.

c. Determine and recommend where Artificial Intelligence (AI) concepts and techniques can be effectively utilized for Space Station mission planning. AI areas to be investigated for application to the specific requirements of mission planning include natural language interfaces, expert systems, and automatic programming.

d. Construct a software development plan for a phased development of a Space Station mission planning system. The plan shall consider the modifications identified in Objective b, and the implementation of any AI concepts recommended in Objective c. The plan shall include a schedule and a manpower estimate.

2.2 TECHNICAL APPROACH

The SS MPS Development Study included the following tasks to accomplish the study objectives:

- Task 1 - Orientation
- Task 2 - Review Spacelab Mission Planning Process and Software
- Task 3 - Space Station Mission Planning Software Requirements
- Task 4 - Investigate Artificial Intelligence Applications to Mission Planning
- Task 5 - Mission Planning Software Development Plan

The flow of these tasks is reflected in Figure 2.2-1.

Task 1 was intended for the study team to obtain an initial familiarization with the process and existing software used for Spacelab payload mission planning at MSFC and to travel to other NASA centers to obtain a general familiarization with the processes and software in use for mission planning at those centers.

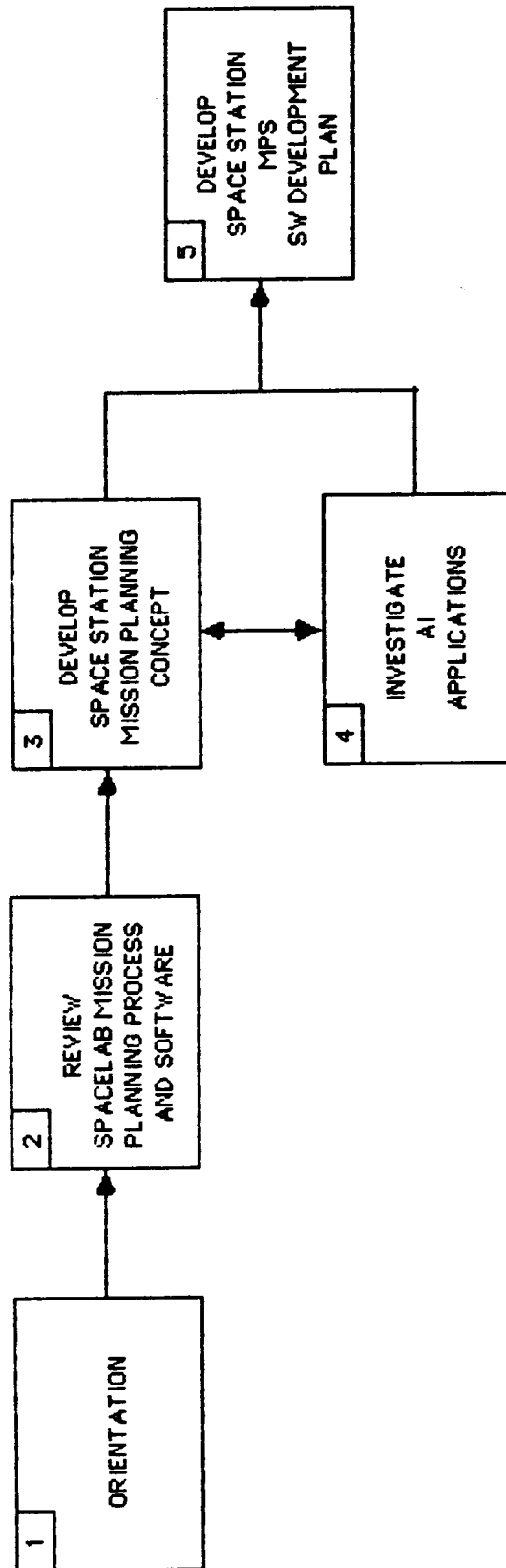


FIGURE 2.2-1 SS MPS DEVELOPMENT STUDY TASK FLOW

The objective of Task 2 was to establish a complete baseline definition of the Spacelab payload mission planning process, along with a definition of existing software capabilities for potential extrapolation to the Space Station era. Areas to be included were orbital mechanics analysis and planning, mission timeline generation, data flow analysis and planning, onboard computer timelines generation and implementation, experiments command planning and implementation, and planning for Payload Operations Control Center (POCC) support. Pre-flight planning and real-time planning and replanning activities were also to be defined. The process definition was required to be defined using detailed functional flow diagrams, and individual software module functions were to be defined.

Task 3 was to use the information developed in Task 2 for the Spacelab payload mission planning process and software as the basis for defining system requirements to support Space Station mission planning. The system was required to include the capability to permit the mission planning function to be centralized or distributed, and to be performed by non-expert mission planners as well as experts. The role of mission planning onboard the Space Station and the interfaces with the ground were required to be assessed. Initially, five Space Station mission planning concepts were identified for assessment; these ranged from all mission planning done on the ground to all mission planning done on-board the Space Station. Subsequent MSFC guidance narrowed the possible concepts to one in which mission planning was to be done on the ground with minor real-time replanning capability to be provided on-board. Comparable to the Spacelab process, detailed flow diagrams of the Space Station mission planning concept were to be developed, including the flow of planning data. Also, software functions were to be identified, and modifications/additions to the Spacelab payload mission planning system software to support the Space Station mission planning concept were to be defined.

In Task 4, the Space Station mission planning concept (developed in Task 3) was to be reviewed for the purpose of identifying areas where Artificial Intelligence (AI) concepts might offer substantially improved capability. Three specific AI concepts were to be investigated for applicability: natural language interfaces, expert systems, and automatic programming. The advantages and disadvantages of interfacing an AI language with existing FORTRAN programs or of converting totally to a new programming language were to be identified.

Task 5 was intended to integrate the outputs of Task 3 and 4 to produce the primary product of the Study, a Space Station mission planning system software development plan. The plan was required to include:

- A detailed description of modifications and additions to the Spacelab mission planning system which are required in order to make this system suitable for use in Space Station mission planning.

- Recommendations on the use of AI as means of improving the overall mission planning process, including identification of specific areas where AI may be beneficial.

- A development schedule compatible with the overall Space Station schedules, and the manpower required.

The development plan was also required to include a description of the Space Station mission planning concept, a review of the functions to be performed, and a description of the modules required for each function. Module development standards, such as language used for coding, were also required to be defined.

Section 3

PROGRAMMATIC SUMMARY OF STUDY ACTIVITIES AND ACCOMPLISHMENTS

The SS MPS Development Study, as depicted in Figure 3-1, was originally intended to be an eight-month Study; however, six (6) months into the Study, the overall schedule was extended two (2) months to accommodate the longer (than originally anticipated) time to complete Task 2. Also, the extension provided the opportunity to support MSFC inputs to the NASA Space Station Operations Task Force and to incorporate appropriate Task Force concepts and conclusions into the Study.

An interim review of Study activities and accomplishments, originally planned for approximately four (4) months into the Study, was waived by MSFC in favor of weekly progress meetings. However, a formal presentation of the Spacelab payload mission planning process functional flow diagrams was made on 20 October 1986. Monthly progress reports were prepared and submitted as required and a final review of the Study was presented as required on 4 March 1987.

Study activities concluded with the submittal of a final report (of which this volume is a part) and a SS MPS Software Development Plan on 20 March 1987.

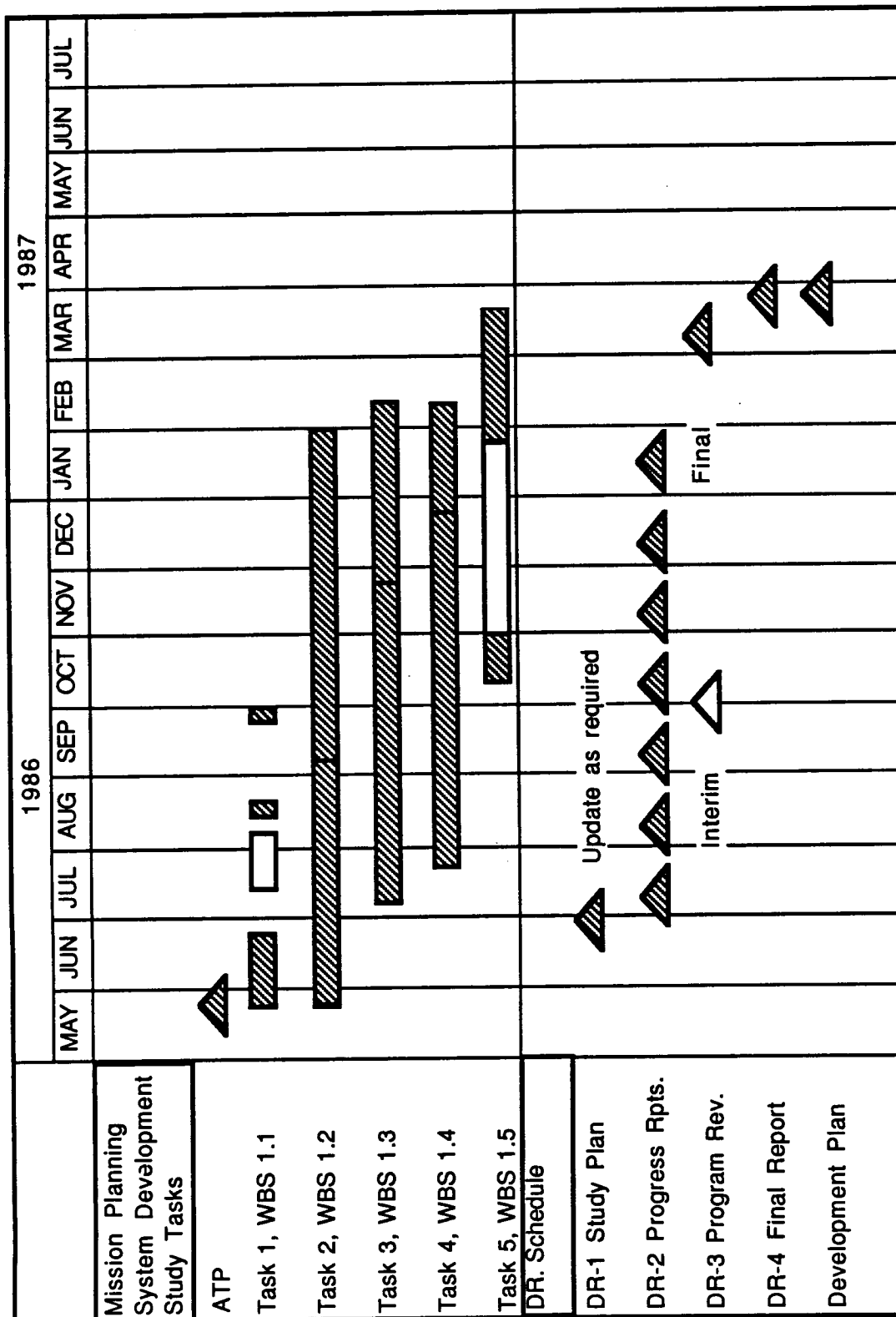


FIGURE 3-1. SS MPS DEVELOPMENT STUDY SCHEDULE

Section 4

TASK 1 - ORIENTATION

Task 1 activities first included orientation meetings with MSFC from 30 May 1986 through early June 1988. These orientation meetings primarily consisted of MSFC briefings and demonstrations of the Spacelab payload mission planning process and software and a tour of the MSFC Payload Operations Control Center (POCC). The knowledge gained from these meetings, plus handout materials and reference documents, equipped the Study team to commence its activities on Task 2. Of no less significance, these meetings permitted the establishment of working relationships with MSFC mission planning personnel whose inputs to all subsequent Study tasks were invaluable.

Task 1 activities also included MSFC briefings on 9 July 1986. These briefings provided the study team MSFC concepts and considerations as inputs to development of the Space Station mission planning concept in Task 3.

Final Task 1 activities consisted of travel to other NASA centers to investigate mission planning methods and tools (including AI applications) in use or under development at those centers, especially methods/tools oriented toward Space Station. On 13-15 August 1986, a trip to Johnson Space Center was accomplished. Subsequently, on 30 September through 2 October 1986, a trip to Ames Research Center and the Jet Propulsion Laboratory was accomplished. Appropriate reference documents and key contacts on the Space Station Program were obtained at JSC as subsequent reference sources for Task 3. Both trips provided information and contacts on potentially applicable AI concepts and technologies for Task 4.

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Section 5

TASK 2 - REVIEW SPACELAB MISSION PLANNING PROCESS AND SOFTWARE

5.1

ACTIVITIES AND ACCOMPLISHMENTS

As previously stated, the purpose of Task 2 was to review the current Spacelab (SL) payload mission planning process and software and to develop a complete definition and understanding of the process and Mission Integration Planning (software) System (MIPS). The approach taken to this task was first to develop an upper level Spacelab functional flow diagram, then to group the major activities from the overall diagram into major functional areas of activity (which tended to correspond to MSFC mission planning organizational elements), and, finally, for each functional area, to develop detailed flows to a level sufficient to acquire a thorough understanding of the mission planning activities and to be able to correlate the capability of a SL MIPS software module to the objective of a specific activity. Based on knowledge gained, a data base of mission planning activities, activity descriptions, and resource data was also developed.

The major inputs to the task were MSFC briefings, demonstrations and handout materials, Spacelab mission planning process and software documentation, and personal interviews with Spacelab mission planning personnel. By far the most valuable of these inputs were the interviews/working sessions with mission planning personnel for development of the upper level functional flow and detailed flows. Mission planning personnel also made certain inputs to the data base which could only be provided by people who were experienced in the SL mission planning process. The support of these NASA personnel was essential in accomplishing this task.

The major products of this task were the Spacelab mission planning process functional flow diagrams and data base. These products, and the knowledge gained from their development, served as a significant input to Task 3, because they identified not only the SL Payload MIPS software modules of potential applicability to Space Station, but also a detailed understanding of the scope, nature, and sequence of activities and inputs/outputs that are required for the planning of payload on-orbit operations in general.

Finally, this task revealed certain characteristics and lessons learned from Spacelab payload mission planning that served as important considerations in the establishment of the fundamental objectives and approach toward Space Station mission planning in Task 3. These characteristics and lessons learned are presented below:

- Spacelab mission planning activities are centralized.
- Payload activities are scheduled down to the minute to make maximum utilization of resources during a short-duration mission.
- The collection of principal investigator experiment operations requirements is a very sizable manual effort which continues through all planning cycles.

- Spacelab mission planning employs a system of 58 actively used computer programs which have evolved over a ten-year period without the benefit of a rigidly controlled, structured process of development. (Upgrading of capabilities is still underway.)
- Though employing computer software, the Spacelab mission planning process involves considerable manual effort of highly skilled personnel.
- User-friendly interactive and automated software is considered of key importance to reducing mission planning manpower requirements.

5.2 SPACELAB MISSION PLANNING PROCESS

This subsection contains and provides introductory explanations of the Spacelab mission planning process functional flow diagrams and data base produced by Task 2 of the SS MPS Development Study. Together, the flow diagrams and data base constitute a complete and thorough definition of the sequence and nature of Spacelab mission planning process activities, and the associated Spacelab MIPS software capabilities and resource requirements.

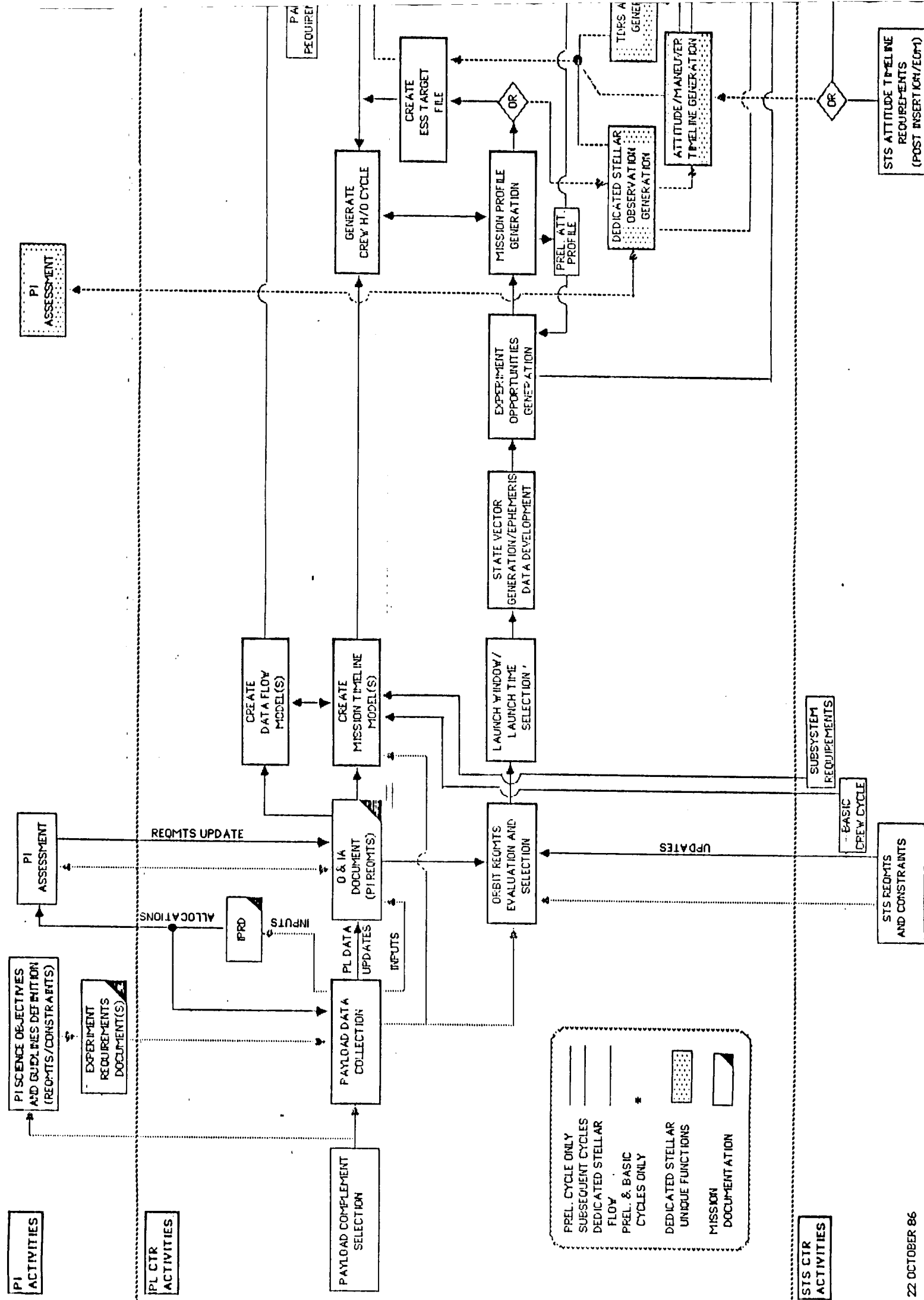
5.2.1 Spacelab Functional Flow

The Spacelab Functional Flow diagram, presented in Figure 5.2.1-1, was developed in order to identify all major activities of the Spacelab payload mission planning process. The diagram shows interfaces required by the planning center (MSFC) with the principal Investigators (PI's) and with the STS center (JSC). The PI interfaces are indicated at the top of the diagram and STS center interfaces are shown at the bottom of the diagram.

The Spacelab Functional Flow diagram includes activities ranging from payload data collection, through the required analyses, to preparation of payload mission execution documentation. The activities for three (3) planning cycles (preliminary, basic, update) are encompassed by the flow except where noted by the diagram legend. Real-time replanning activities are also encompassed by the flow. All activities may not be performed, or may be significantly reduced in a planning cycle based on the changes/updates required from a previous cycle. The flow accommodates a multidiscipline payload complement but includes a unique path for a payload complement of co-aligned IPS-mounted stellar observation experiments.

The SL mission planning process activities shown in the Spacelab Functional Flow diagram are grouped into nine (9) major functions. These functions and the subfunctions which comprise each are identified in Table 5.2.1-1.

Table 5.2.1-2 is a listing of acronyms and abbreviations used in the Spacelab Functional Flow diagram, and subsequently in the detailed flow diagrams and data base.



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SPACELAB FUNCTIONAL FLOW



TABLE 5.2.1-1
SPACELAB FUNCTIONAL FLOW ACTIVITY GROUPINGS

| <u>FUNCTION</u> | <u>SUBFUNCTIONS</u> |
|---------------------------|---|
| Payload Data Collection | N/A |
| Orbit Analysis | Orbit Requirements Evaluation and Selection |
| | Launch Window/Launch Time Selection |
| | State Vector Generation/Ephemeris Data Development |
| | Experiment Opportunities Generation |
| | Mission Profile Generation |
| | Dedicated Stellar Observation Generation |
| | Attitude/Maneuver Timeline Generation (Multidiscipline) |
| | Attitude/Maneuver Timeline Generation (Dedicated Stellar) |
| | Orbiter Pointing Data Generation |
| | TDRS Acquisition/Loss Generation |
| | POCC MMU Data Set Generation |
| | Objective Loads Generation |
| | Joint Operations Target File Generation (Dedicated Stellar) |
| Mission Timeline Analysis | Create Mission Timeline Models |
| | Generate Crew Handover Cycle |
| | Create ESS Target File |
| | Mission Timeline Generation |
| | Payload Crew Activity Plan Development |

TABLE 5.2.1-1
SPACELAB FUNCTIONAL FLOW ACTIVITY GROUPINGS (CONT'D)

| <u>FUNCTION</u> | <u>SUBFUNCTIONS</u> |
|--|--|
| Flight Definition Document Development | N/A |
| Flight Planning Annex Input Development | N/A |
| Crew Procedures Development | <ul style="list-style-type: none"> * Develop Stowage Book * Develop TV, Photo Procedures * Develop Experiment Crew procedures * Develop Payload Systems Handbook * Develop CDMS Dictionary * Build PFDF Documents |
| Data Flow Analysis | Create Data Flow Models Generate Mission Data Requirements Profile Schedule Onboard Data Management and Downlink Schedule POCC Data Capture/Management/Distribution Verification of Data Flow Schedules Data Flow and System Configuration Document Development Update or Enhance Existing Schedules |
| MMU Load Input Development | <ul style="list-style-type: none"> * Create ECOS Subordinate Timelines * Create ECOS Master Timeline * Build ECOS Timeline Tape * MMU Optimization |

* These subfunctions do not appear in the upper level functional flow but are defined in the detailed flow diagrams

TABLE 5.2.1-1
SPACELAB FUNCTIONAL FLOW ACTIVITY GROUPINGS (CONT'D)

| <u>FUNCTION</u> | <u>SUBFUNCTIONS</u> |
|--|--|
| Experiment Command Planning Development | <ul style="list-style-type: none">* Generate Command List* Check Command Syntax* Produce Command Timetags* Generate Command Timeline* Create POCC Checklist* Check Activity Syntax* Produce Activity Timetags* Generate POCC Checklist and Command Timeline |

* These subfunctions do not appear in the upper level functional flow but are defined in the detailed flow diagrams

TABLE 5.2.1-2

ACRONYMS AND ABBREVIATIONS

| | | |
|--------------|---|--|
| ACT/DEACT | - | Activation/Deactivation (Spacelab) |
| ASCII | - | American Standard Code For Information Interchange |
| AT PHY | - | Atmospheric Physics |
| CAP | - | Crew Activity Plan |
| CDMS | - | Command and Data Management Subsystem (Spacelab) |
| CEL | - | Celestial |
| CMNDS | - | Commands |
| COO | - | Coobservation (File) |
| DDU | - | Data Display Unit (Spacelab) |
| DEFN | - | Definition |
| DEP | - | Dedicated Experiment Processor |
| DFA | - | Data Flow Analyst |
| DS | - | Dedicated Stellar (Mission) |
| EBCDC | - | Extended Binary Coded Decimal |
| ECAS | - | Experiment Computer Applications Software (Spacelab) |
| ECOS | - | Experiment Computer Operating System (Spacelab) |
| ERD | - | Experiment Requirements Document |
| EDT | - | VAX Editor |
| ESS | - | Experiment Scheduling System |
| FDD | - | Flight Definition Document |
| FO's | - | Functional Objectives (Experiments) |
| FPA | - | Flight Planning Annex |
| H/O | - | Handover (Crew Handover Cycle) |
| HDRR | - | High Data Rate Recorder (Spacelab) |
| HEX | - | Hexidecimal |
| HRM | - | High Rate Multiplexer (Spacelab) |
| IPRD | - | Integrated Payload Requirements Document |
| IPS | - | Instrument Pointing Subsystem (Spacelab) |
| IWG | - | Investigators Working Group |
| JSC | - | Johnson Space Center |
| LDF | - | List Directed File |
| MDP'S | - | Mission Dependent Parameters |
| MGMT, MANGMT | - | Management |
| MMU | - | Mass Memory Unit |
| MMUM | - | Mass Memory Unit Manager |
| MPE | - | Mission Peculiar Equipment |
| MSFC | - | Marshall Space Flight Center |
| MSN | - | Mission |
| MSN IND | - | Mission Independent |
| MTL | - | Master Timeline |
| MVR | - | Maneuver |
| NDF | - | Name Directed File |
| O&IA | - | Operations and Integration Agreements |
| O/O | - | On/Off (File) |
| OCCULT | - | Occultation (Orbiter) |
| OPS | - | Operations |
| PAO | - | Public Affairs Office |
| PCAP | - | Payload Crew Activity Plan |

TABLE 5.1.1-2

ACRONYMS AND ABBREVIATIONS

| | | |
|---------|---|---|
| PFDf | - | Payload Flight Data File |
| PI | - | Principal Investigator |
| PL PHY | - | Plasma Physics |
| PL, P/L | - | Payload |
| POCC | - | Payload Operations Control Center |
| POH | - | Payload Operations Handbook |
| PTS | - | Payload Timeline Summary |
| SAA | - | South Atlantic Anomaly |
| SCAS | - | Subsystem Computer Applications Software (Spacelab) |
| SCOS | - | Subsystem Computer Operating System (Spacelab) |
| SL | - | Spacelab |
| SOPG | - | Science Operations Planning Group |
| SPAH | - | Spacelab Payload Accommodations Handbook |
| STL | - | Subordinate Timeline |
| STO | - | Storage (File) |
| STS | - | Space Transportation System |
| T/L, TL | - | Timeline |
| TDRS | - | Tracking and Data Relay Satellite |
| VAX | - | Digital Equipment Corporation Computer |

5.2.2

Detailed Flow Diagrams

The SL mission planning process detailed flows provide, as necessary, a breakdown of functions and/or subfunctions to a task/subtask level necessary to understand the mission planning activities, or to a level to correlate a particular software module to an activity. For example, "Payload Data Collection", which is a manual activity, is detailed at the function level, whereas the Orbital Analysis subfunction "Experiment Opportunities Generation" is broken down to tasks and subtasks - e.g., "Generate Solar Targets" (task) and "Generate Sun Rise/Set" (subtask).

Activities may be manual, automated, or a combination of manual and automated. Manual activities normally include the collection of information (verbal inputs, informal or formal documentation), the evaluation and assessment of this information, and the publication of the results (informal or formal documentation). However, some manual activities produce a computerized input for a subsequent activity - e.g., use of the VAX editor to create a computerized file for use by a software module in a subsequent automated activity.

Automated activities include a software module, based on some fixed algorithm, which reads a computerized input file(s) (fixed format), performs specific operations on the input data, and then outputs the results as either a computerized output file(s) or as a printout. Some automated activities require, or permit, manual inputs to the software module via a keyboard.

A legend for the detailed flows is presented in Figure 5.2.2-1 which shows the conventions utilized in their development. The set of flows which represent the breakdown of the upper level Spacelab Functional Flow diagram to lower level detailed flows (subfunctions, tasks, subtasks) is presented on page 5-12 through page 5-59.

5.2.3

SL MIPS Data Base

The SL MIPS data base was developed in order to provide activity summary data, software description and requirements data, and activity time and skill requirements data. The level of detail of the data base is consistent with the level of detail in the Spacelab mission planning process detailed flow diagrams; that is, entries exist in the data base corresponding to each lowest hierarchical level activity (function, subfunction, task or subtask) identified for every function in the flow diagrams. When assessed in conjunction with the detailed flows, the data base provides a comprehensive definition of the Spacelab payload mission planning process.

The data base consists of eight (8) interrelated tables of data:

- o Activity Summary Data
- o Activity Time and Skill Requirements
- o Software Used by Activity
- o Software Description
- o Software Peripherals Required
- o Activity Input/Outputs
- o Computer Input/Output Summary
- o Manual Input/Output Summary

The complete data base, including an introductory explanation of each table, is contained in Appendix A of this volume.

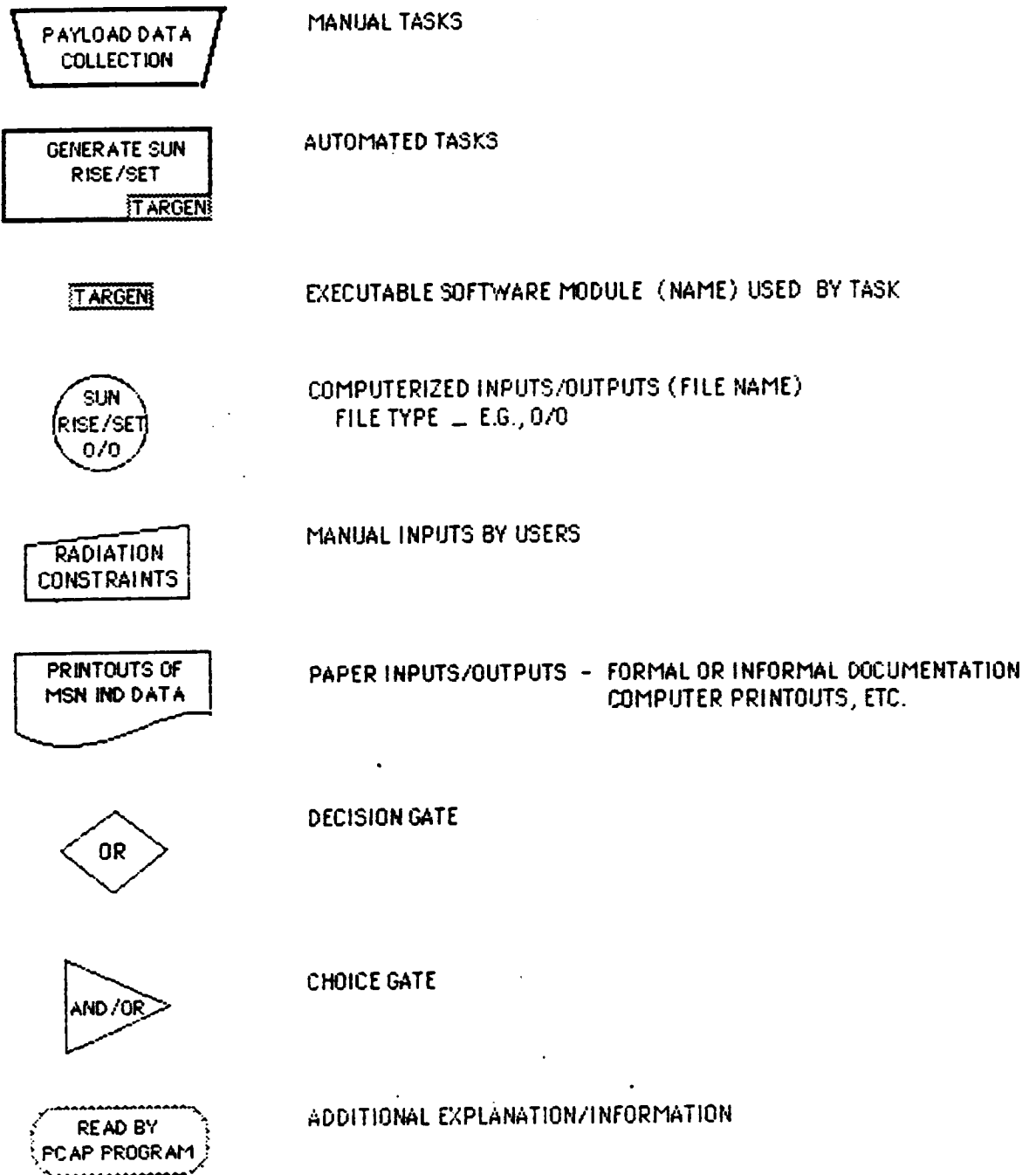
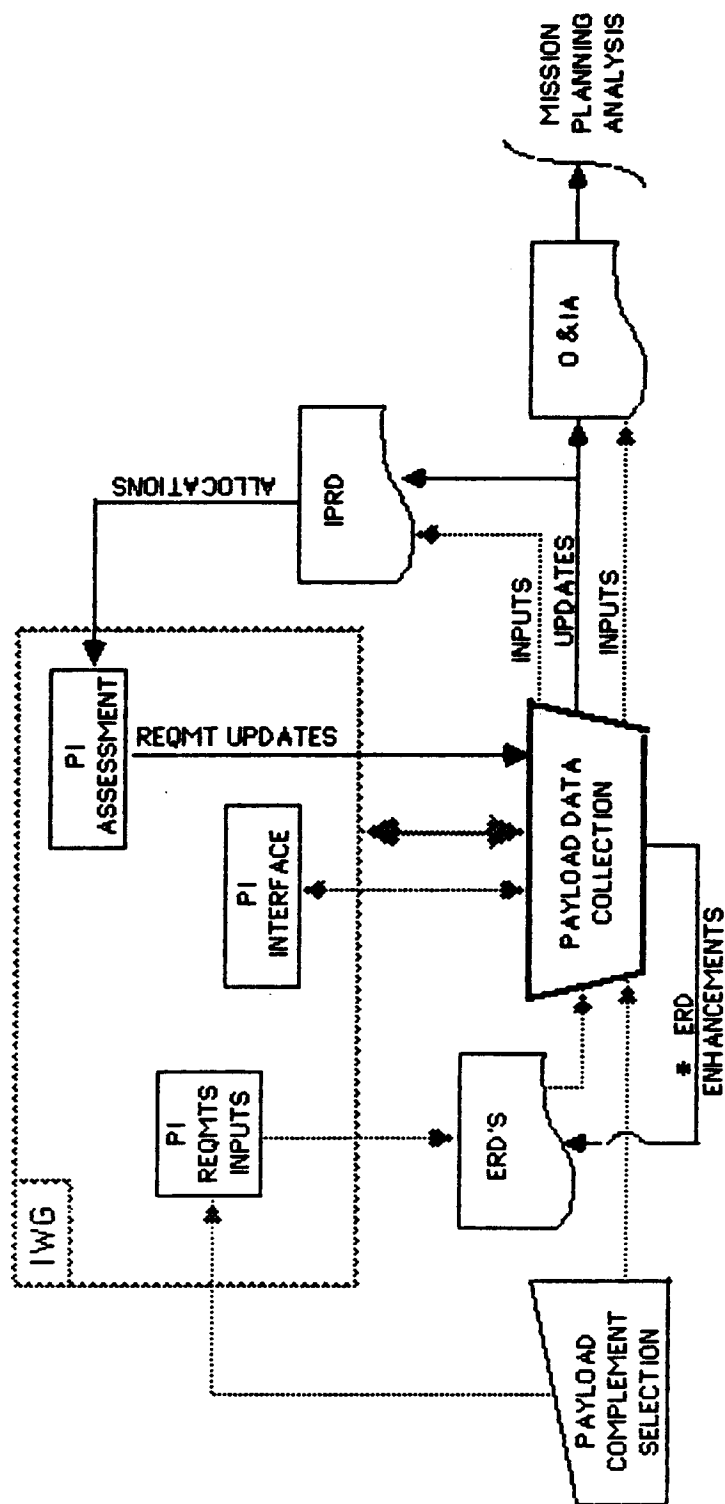


FIGURE 5.2.2-1. SPACELAB FLOW DIAGRAM CONVENTIONS

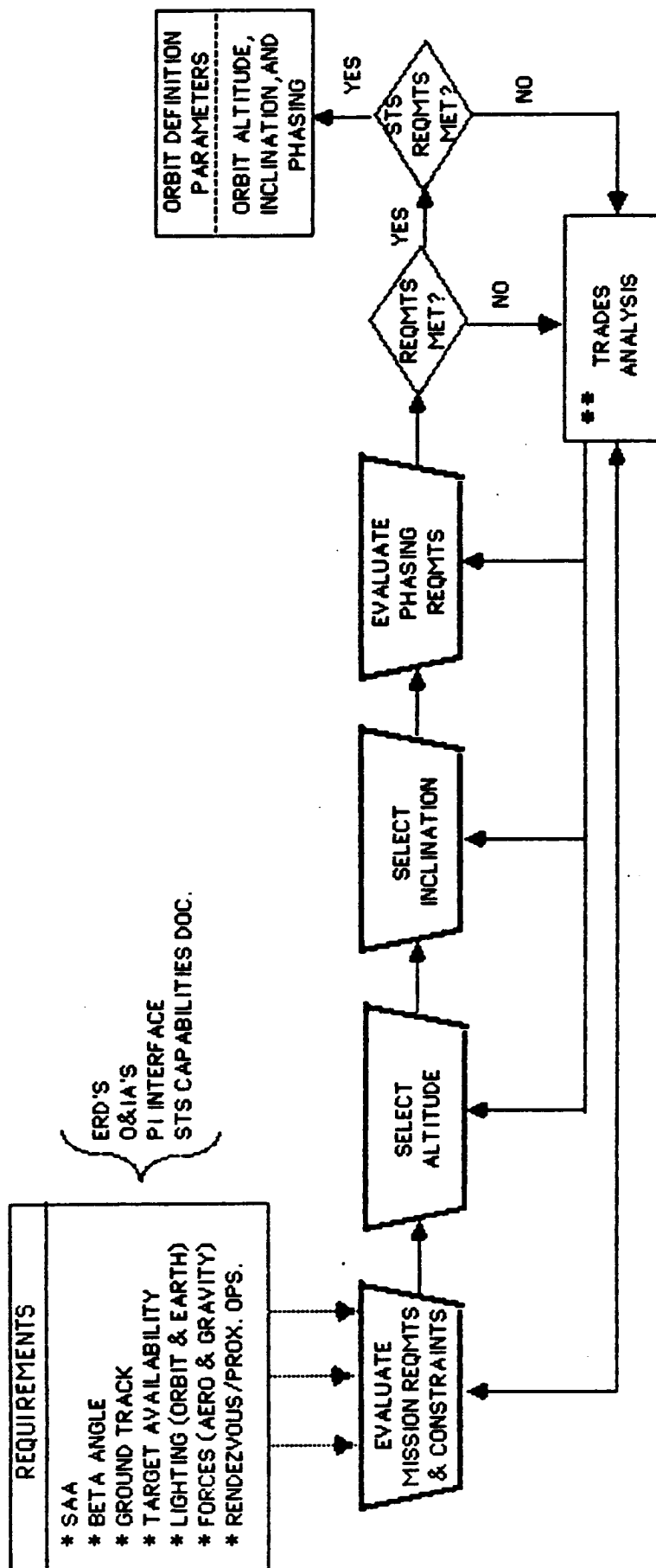
FUNCTION: PAYLOAD DATA COLLECTION



* PREL. CYCLE ONLY

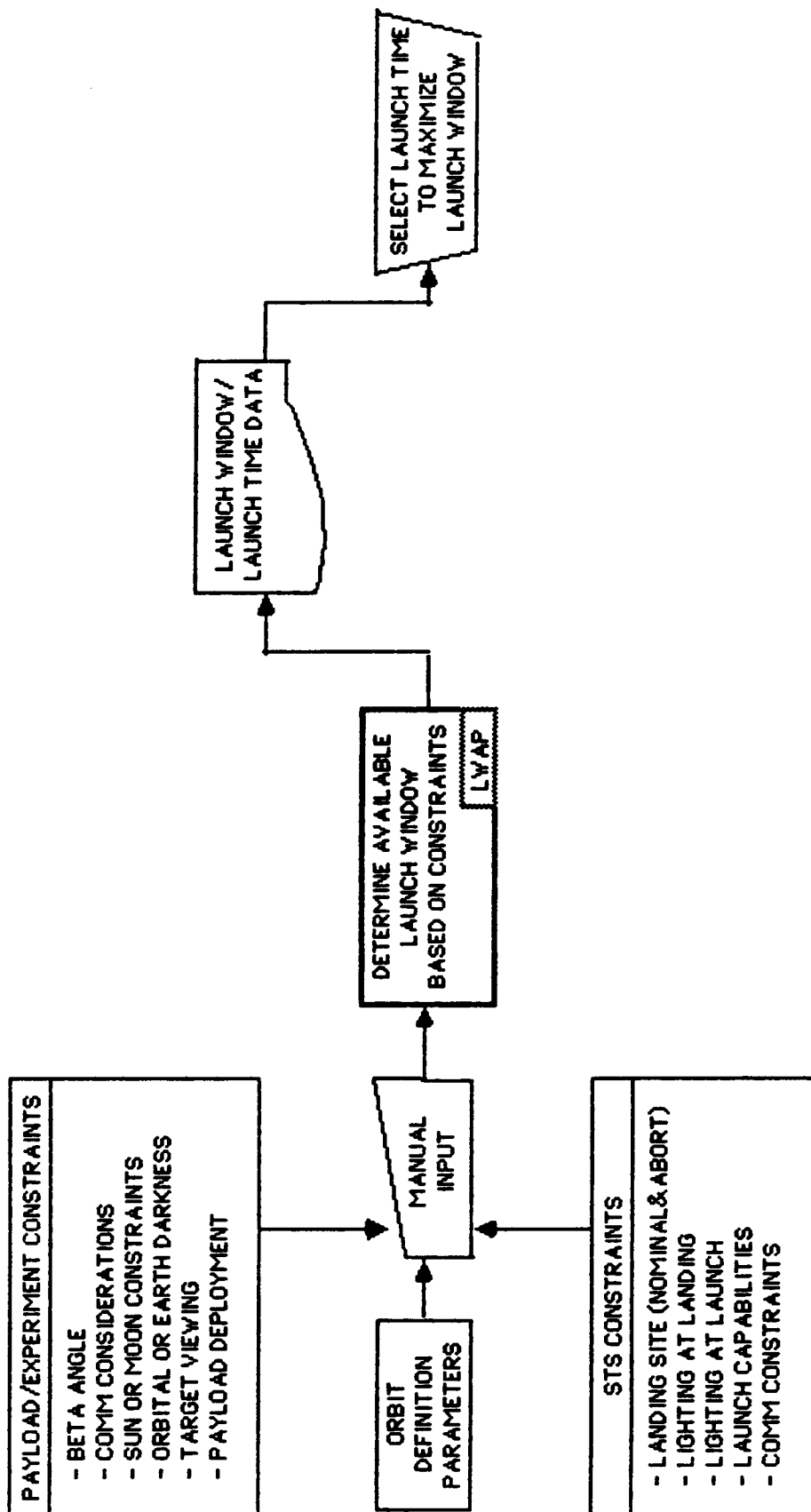
PREL. CYCLE ONLY
SUBSEQUENT CYCLES

FUNCTION: ORBITAL ANALYSIS SUBFUNCTION: ORBIT REQUIREMENTS EVALUATION AND SELECTION

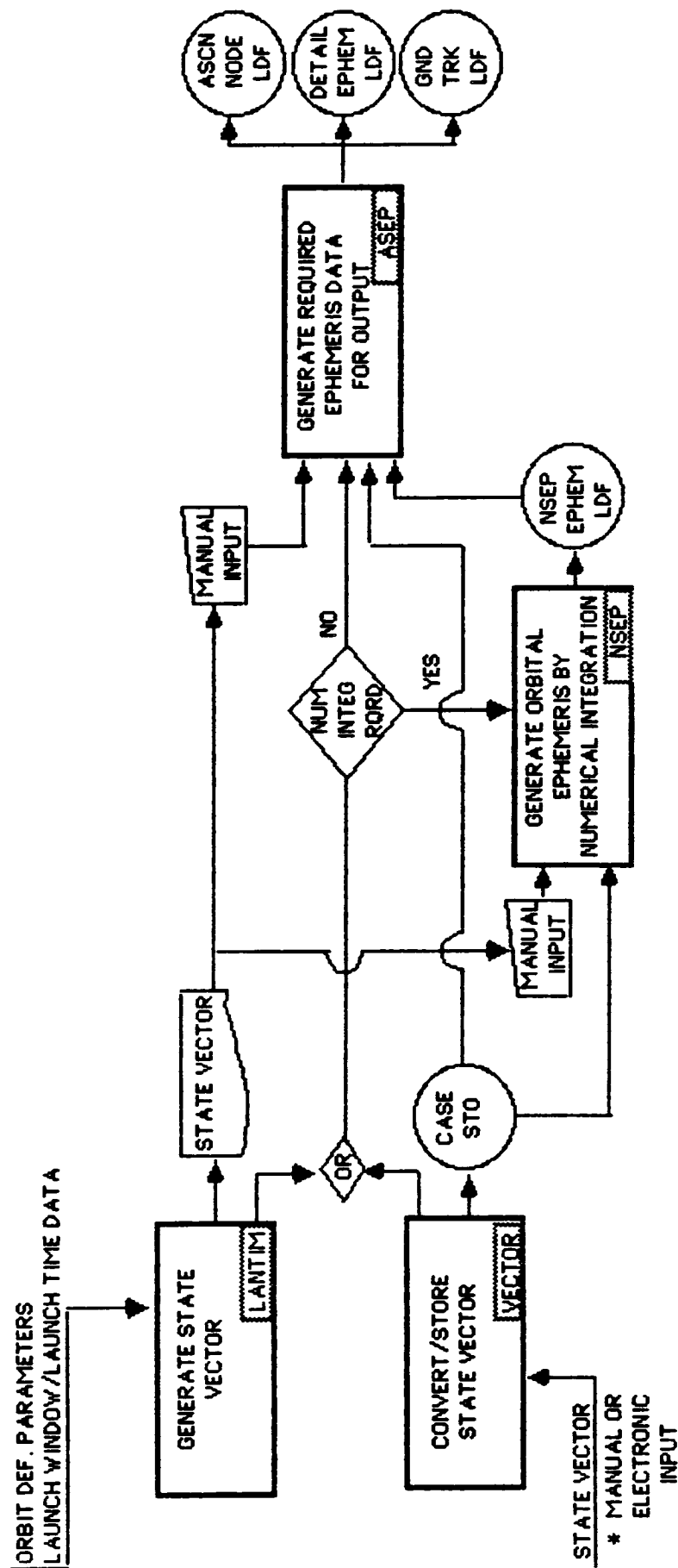


** THIS IS NORMALLY A MANUAL ACTIVITY.
 HOWEVER, IF A TRADES ANALYSIS IS
 REQUIRED THE FOLLOWING PROGRAMS
 MAY BE UTILIZED: ASEP/NSEP, ESAL,
 ESDATA, RAD12, BORB, OTHERS.

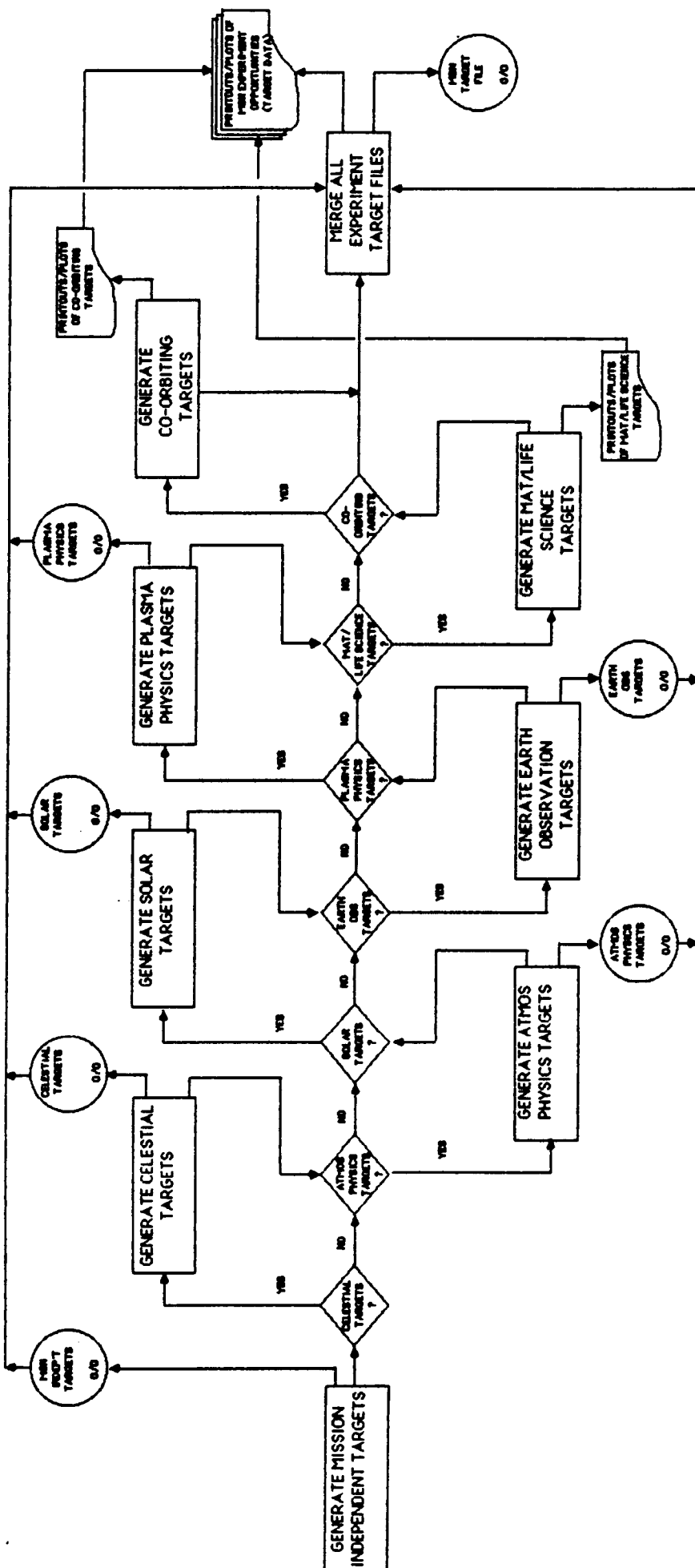
FUNCTION: ORBITAL ANALYSIS SUBFUNCTION: LAUNCH WINDOW/LAUNCH TIME SELECTION



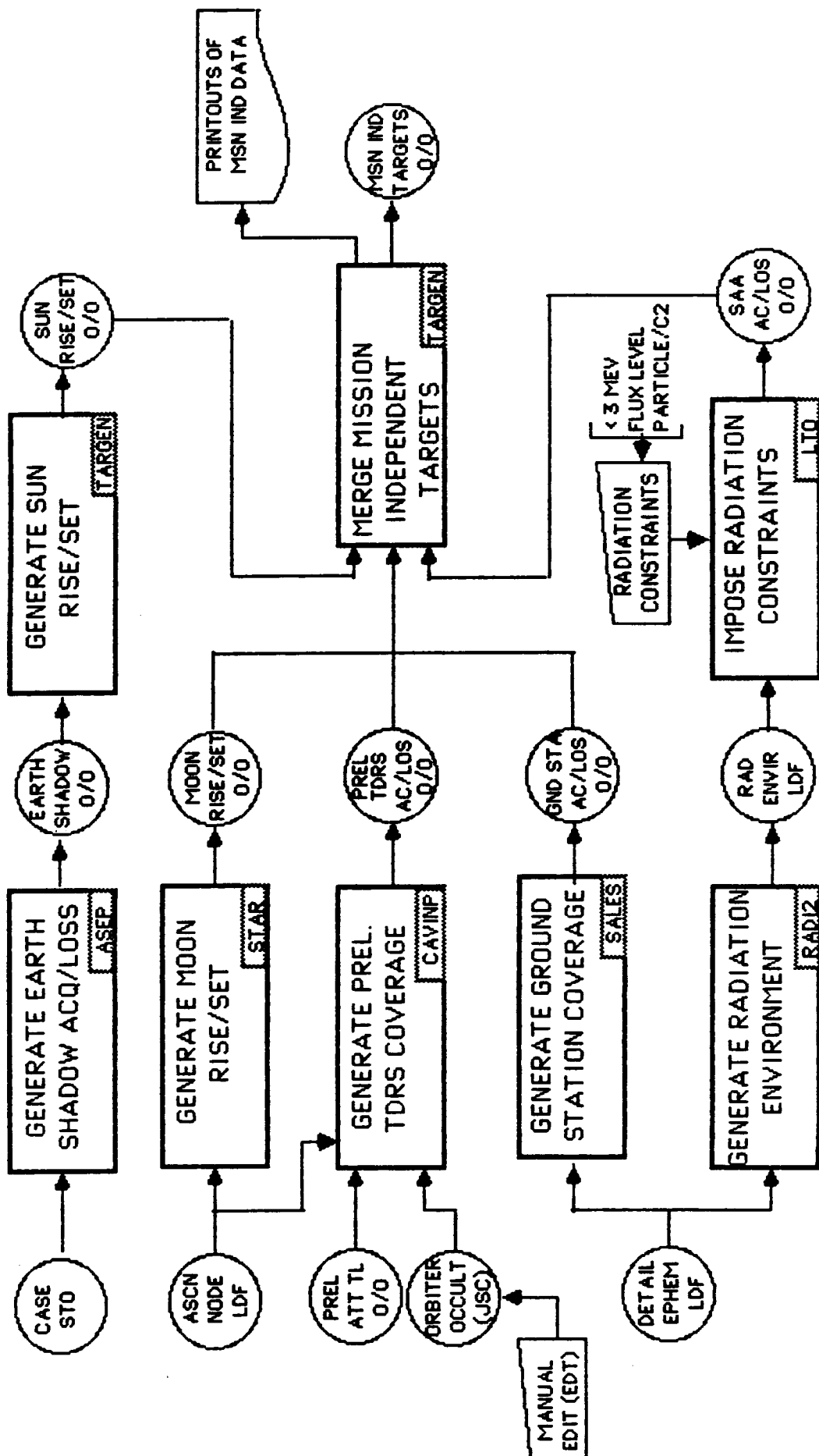
FUNCTION: ORBITAL ANALYSIS
 SUBFUNCTION: STATE VECTOR GENERATION/EPHEMERIS DATA DEVELOPMENT



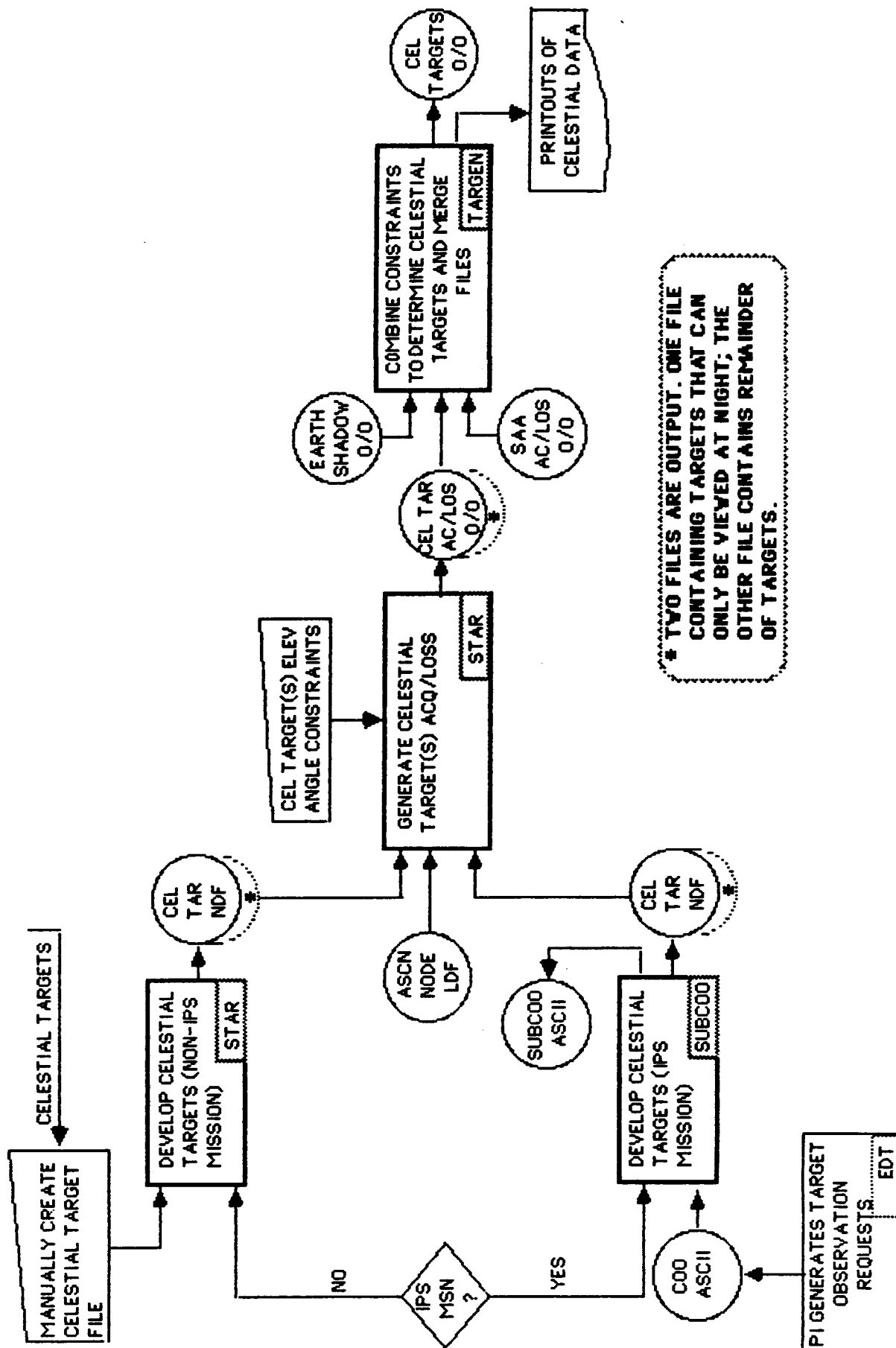
SUBFUNCTION: EXPERIMENT OPPORTUNITIES GENERATION



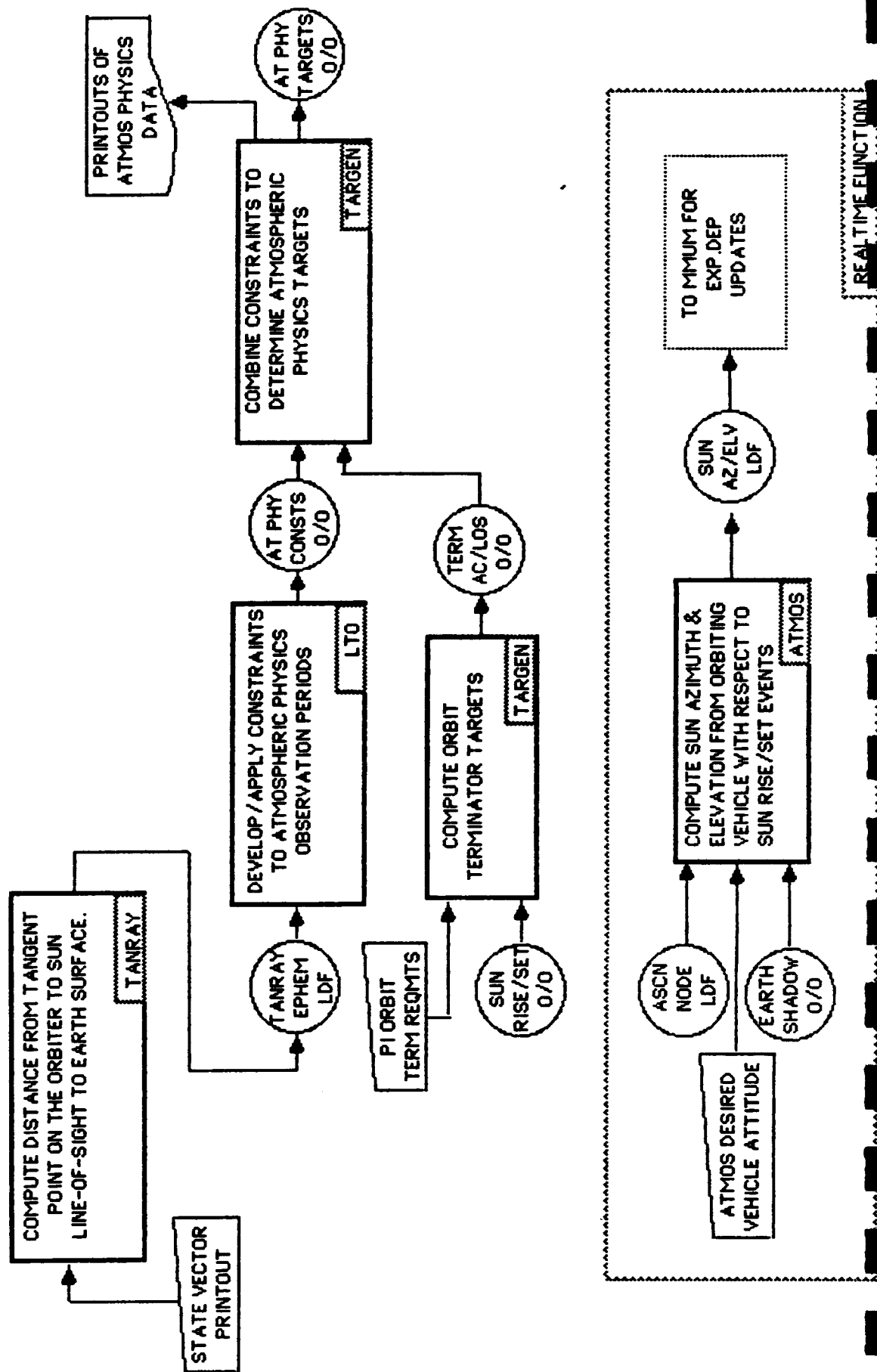
FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: EXPERIMENT OPPORTUNITIES GENERATION
TASK: GENERATE MISSION INDEPENDENT TARGETS



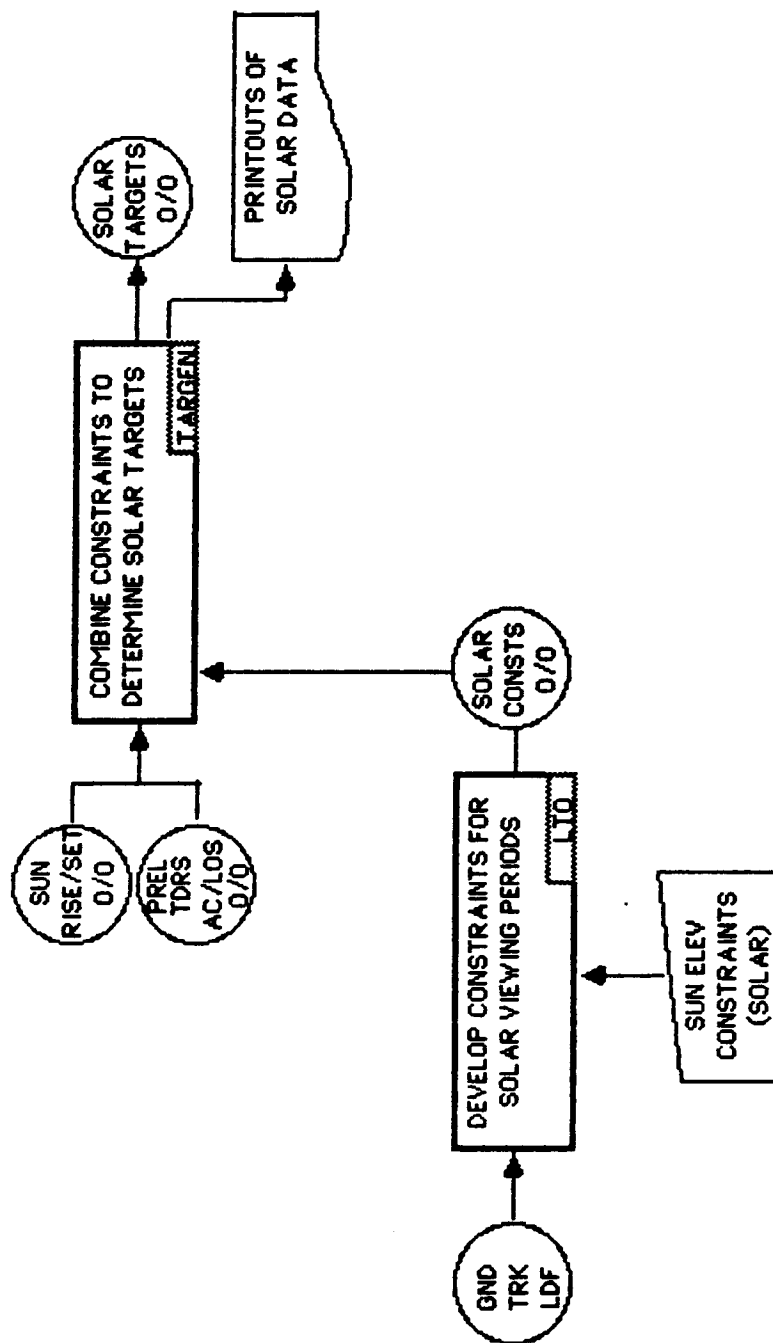
FUNCTION: ORBITAL ANALYSIS
 SUBFUNCTION: EXPERIMENT OPPORTUNITIES GENERATION
 TASK: GENERATE CELESTIAL TARGETS



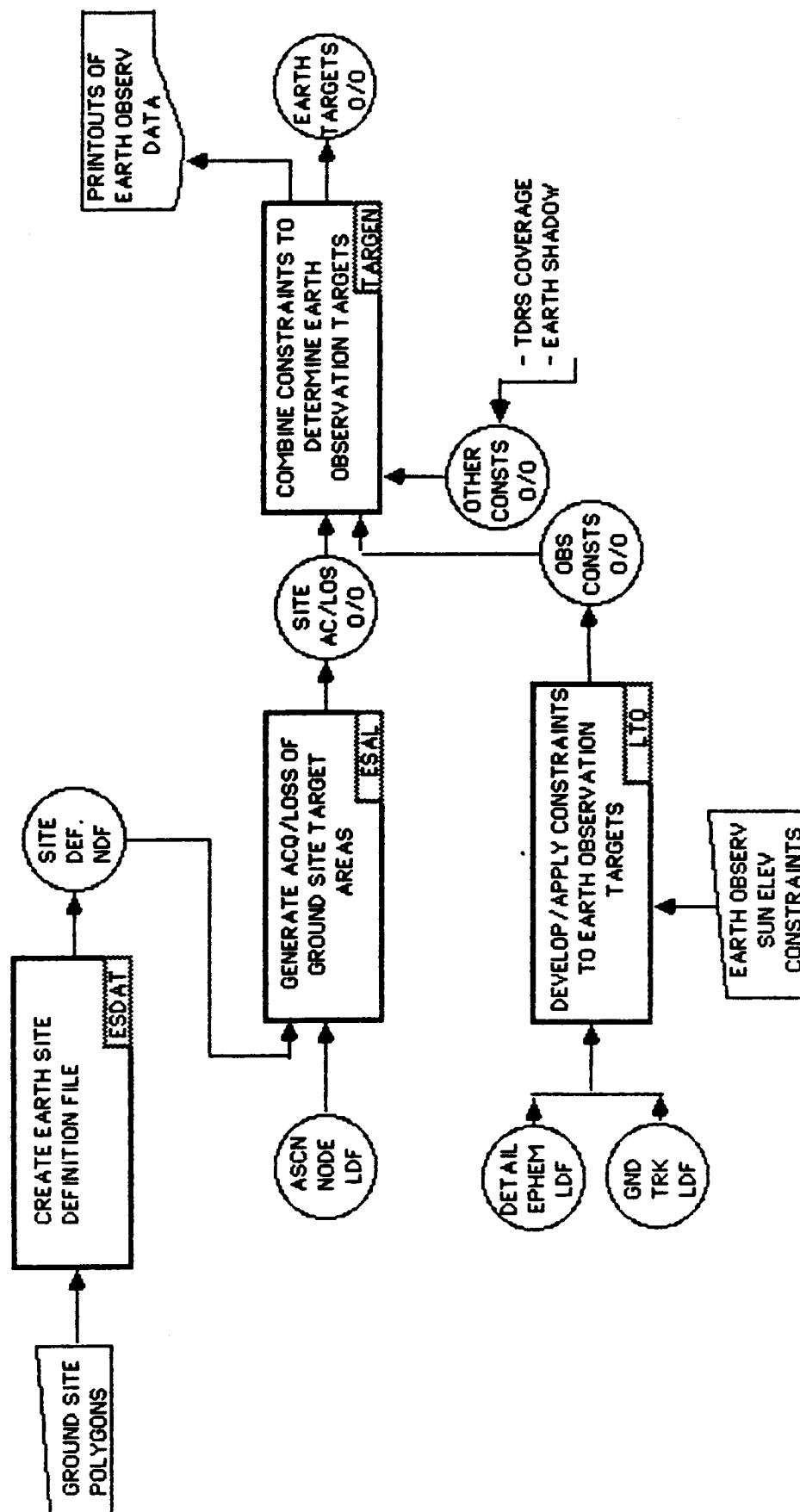
FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: EXPERIMENT OPPORTUNITIES GENERATION
TASK: GENERATE ATMOSPHERIC PHYSICS TARGETS



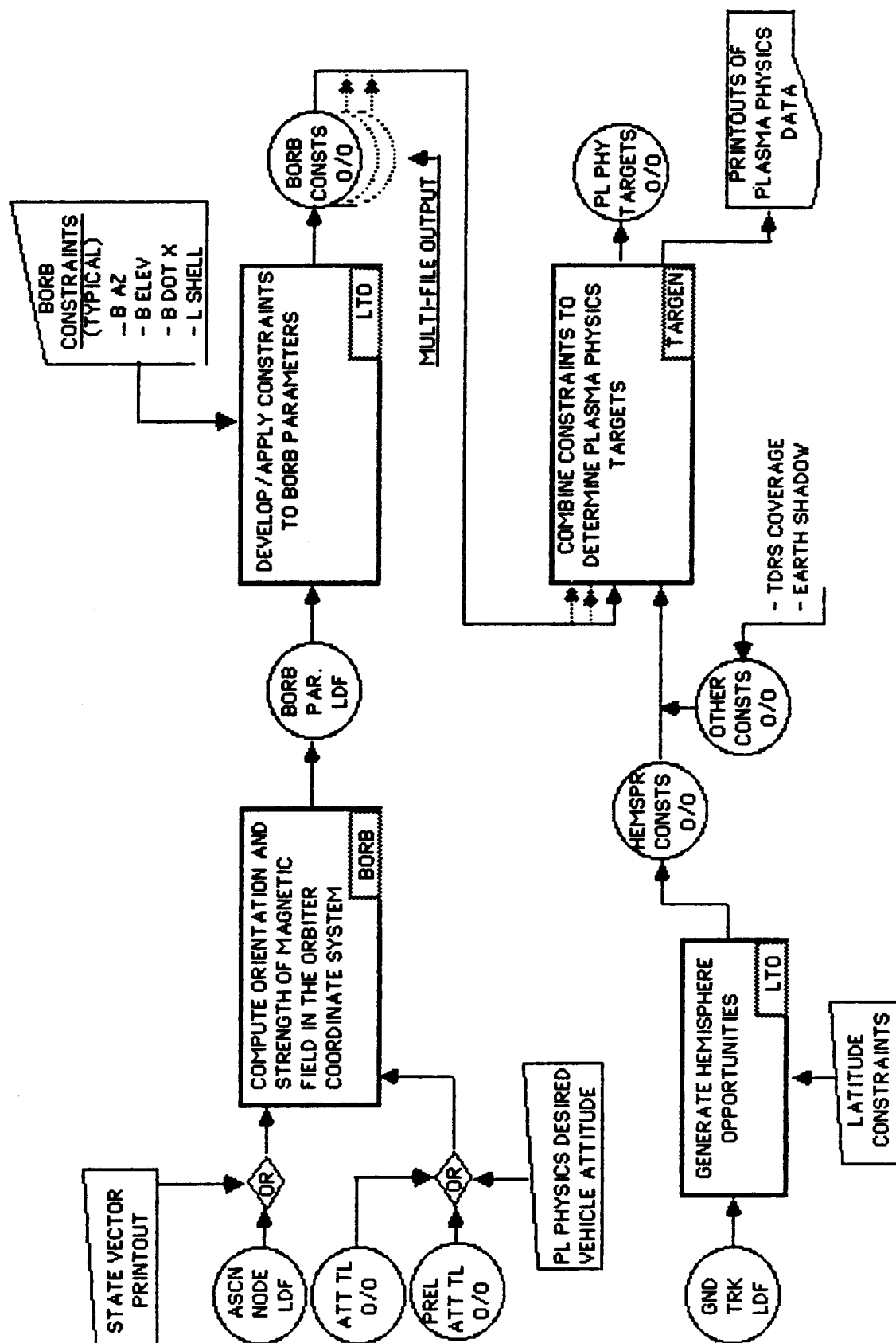
FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: EXPERIMENT OPPORTUNITIES GENERATION
TASK: GENERATE SOLAR TARGETS



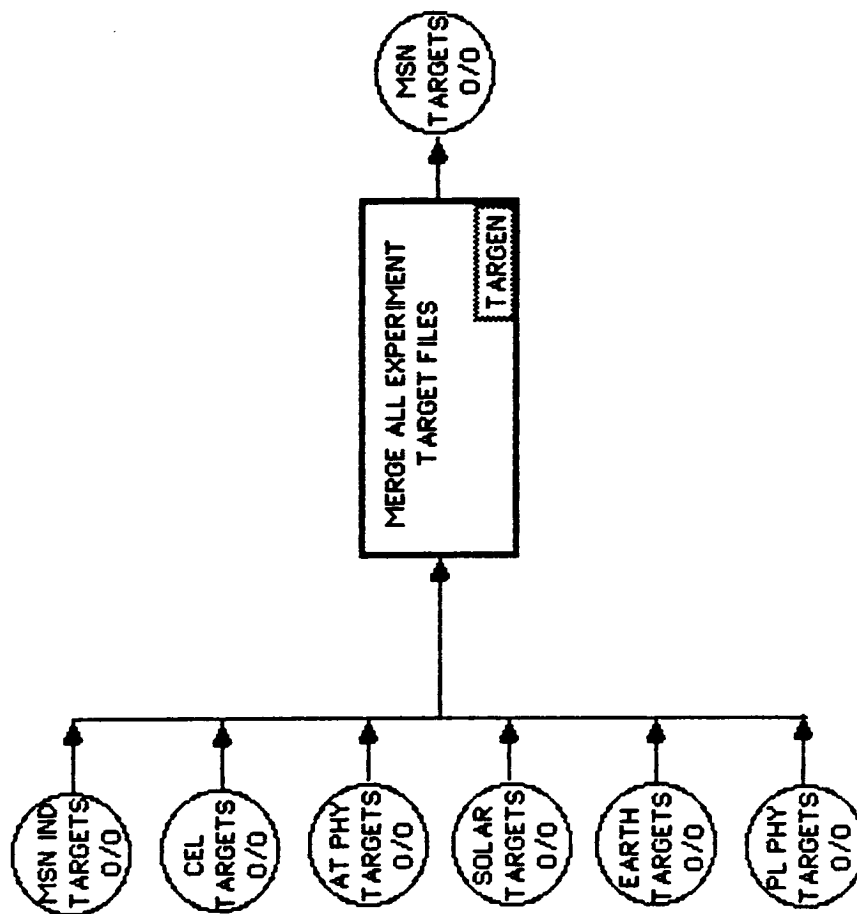
FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: EXPERIMENT OPPORTUNITIES GENERATION
TASK: GENERATE EARTH OBSERVATION TARGETS



FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: EXPERIMENT OPPORTUNITIES GENERATION
TASK: GENERATE PLASMA PHYSICS TARGETS



FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: EXPERIMENT OPPORTUNITIES GENERATION
TASK: MERGE ALL EXPERIMENT TARGET FILES

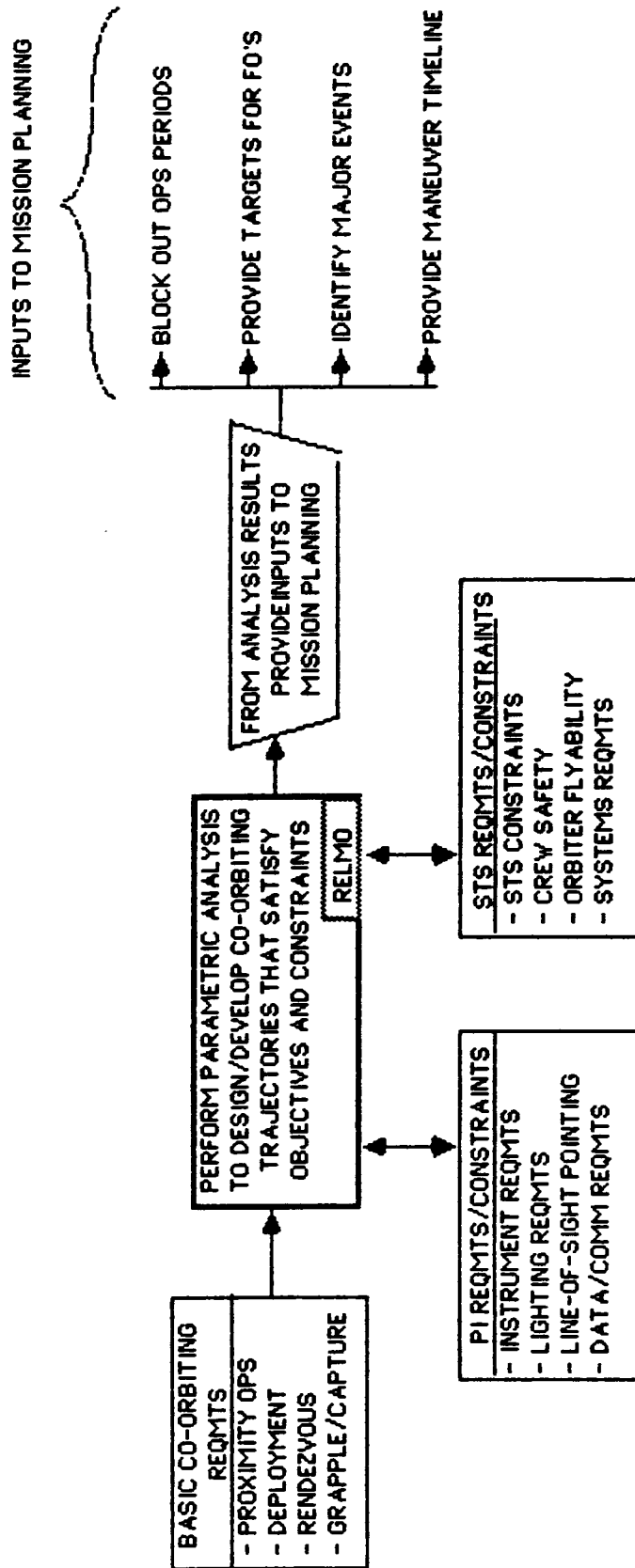


FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: EXPERIMENT OPPORTUNITIES GENERATION
TASK: GENERATE MATERIAL AND/OR LIFE SCIENCE TARGETS

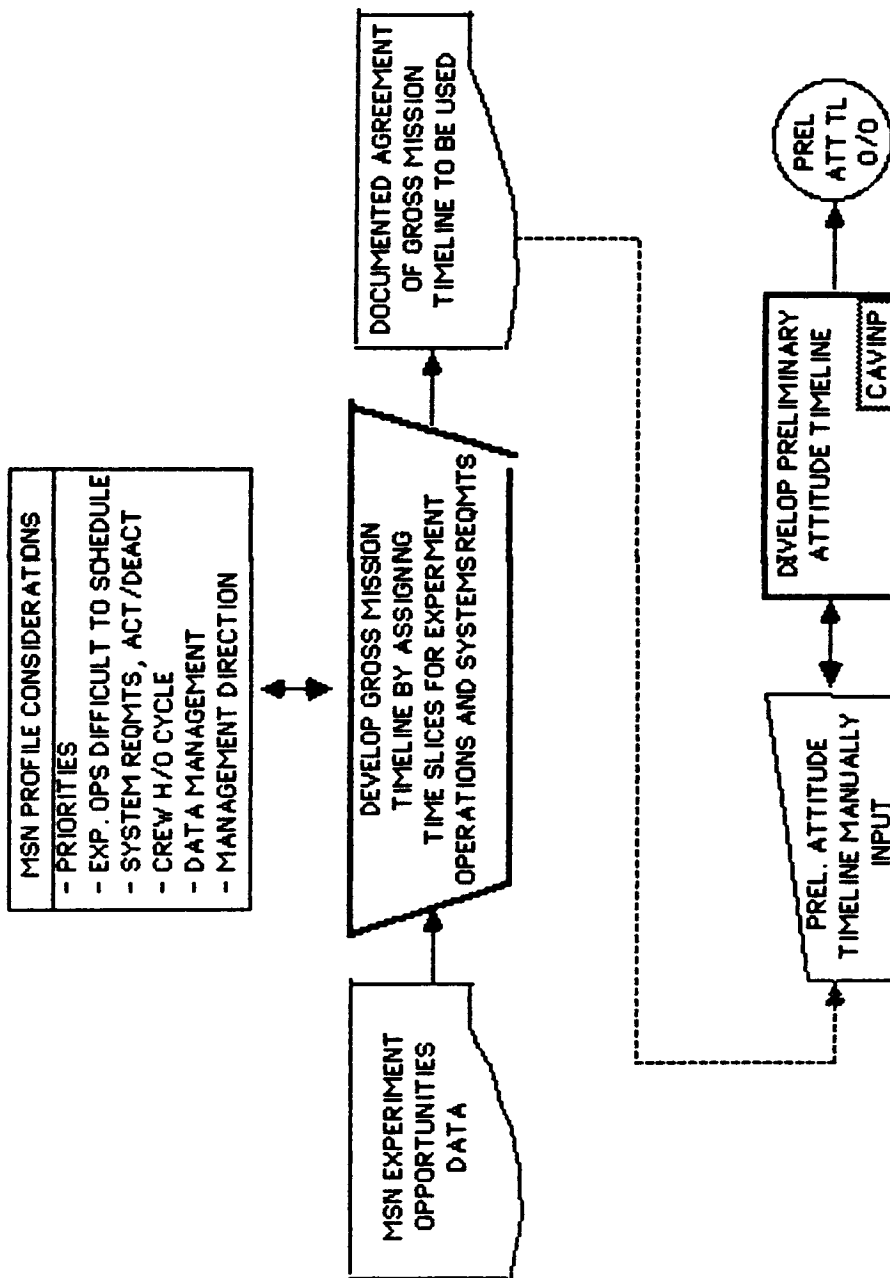
MATERIAL/LIFE SCIENCE TARGET PERIODS ARE DEVELOPED/ASSIGNED BASED ON EXPERIMENT OPERATIONS AND SYSTEMS REQUIREMENTS IN THE " MISSION PROFILE GENERATION " SUBFUNCTION.

MISSION PLANNERS WOULD MAXIMIZE TDRS COVERAGE (WITHIN CONSTRAINTS) FOR EXPERIMENTS OF THIS TYPE. TDRS COVERAGE WILL BE DEVELOPED IN THE " ATTITUDE/MANEUVER TIMELINE GENERATION " SUBFUNCTION IN CONJUNCTION WITH THE " MISSION TIMELINE GENERATION " SUBFUNCTION.

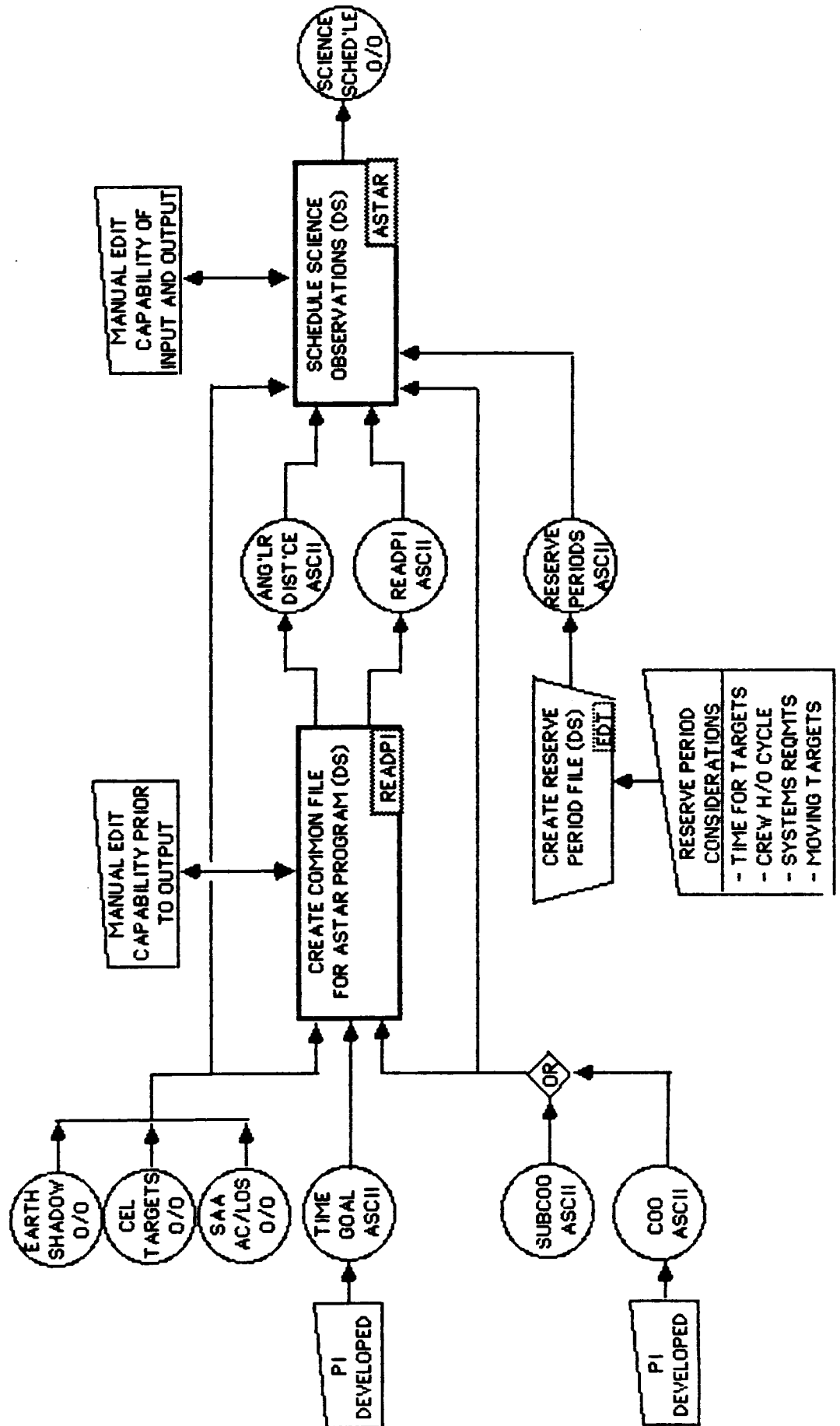
FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: EXPERIMENT OPPORTUNITIES GENERATION
TASK: GENERATE CO-ORBITING TARGETS



FUNCTION: ORBITAL ANALYSIS SUBFUNCTION: MISSION PROFILE GENERATION

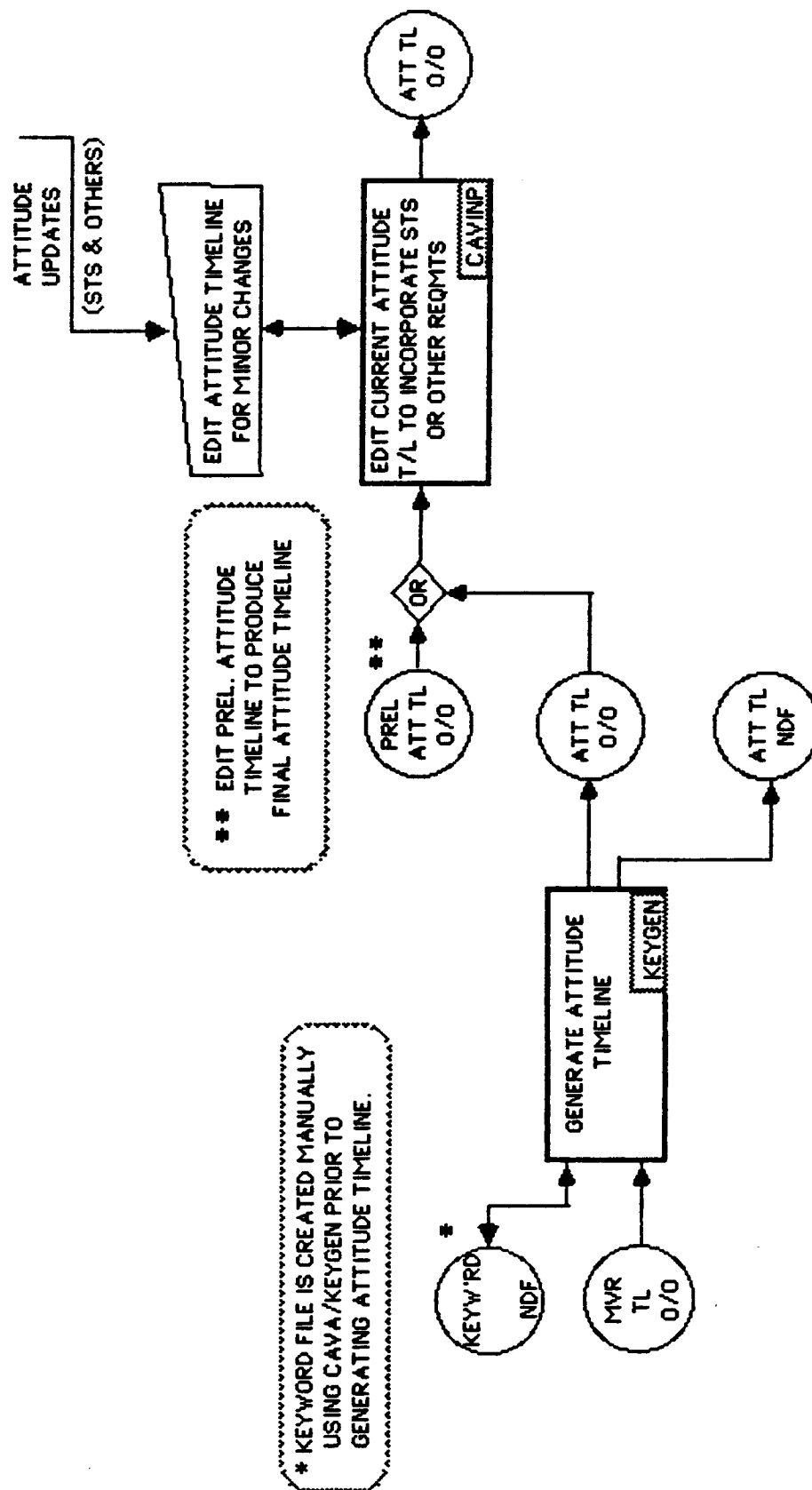


FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: DEDICATED STELLAR OBSERVATION GENERATION



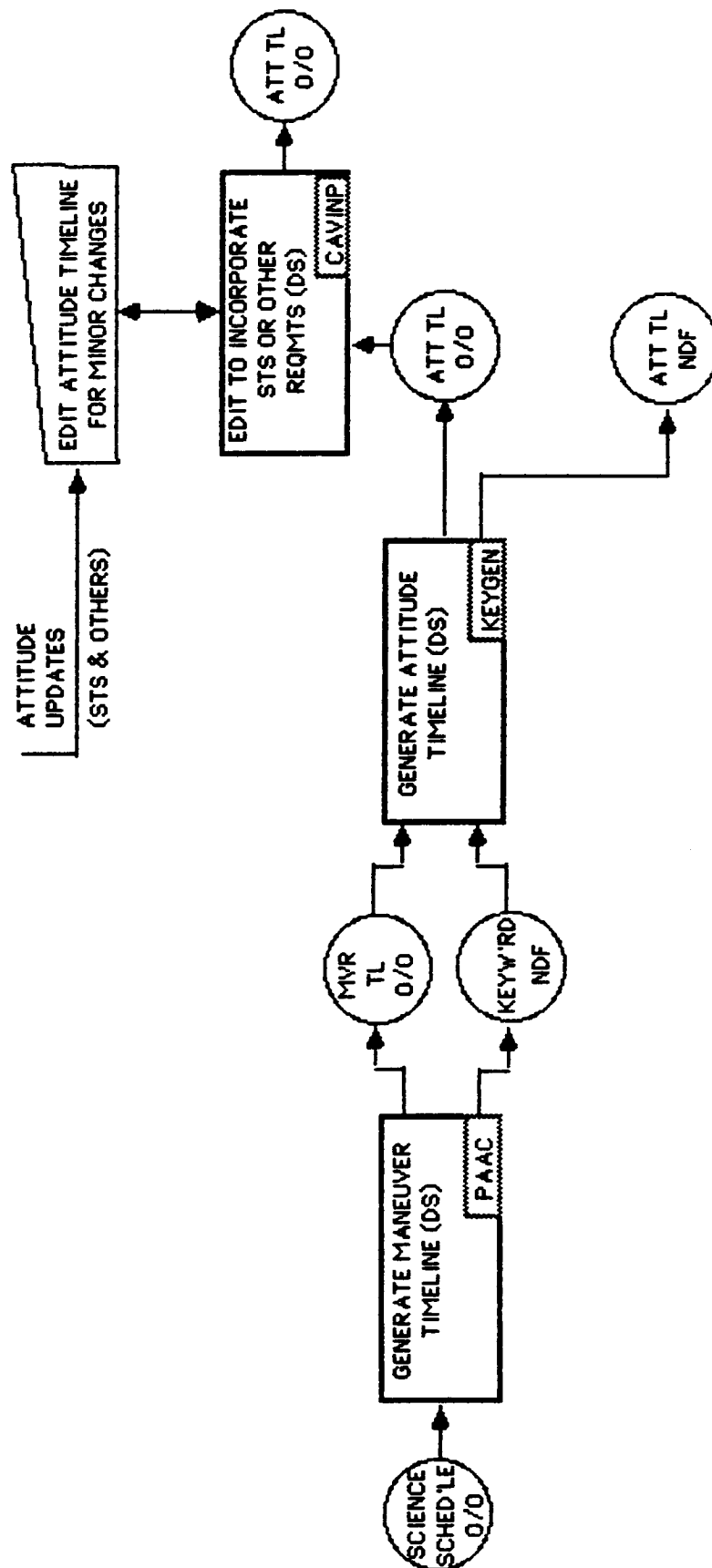
FUNCTION: ORBITAL ANALYSIS

SUBFUNCTION: ATTITUDE/MANEUVER TIMELINE GENERATION (MULTIDISCIPLINE)

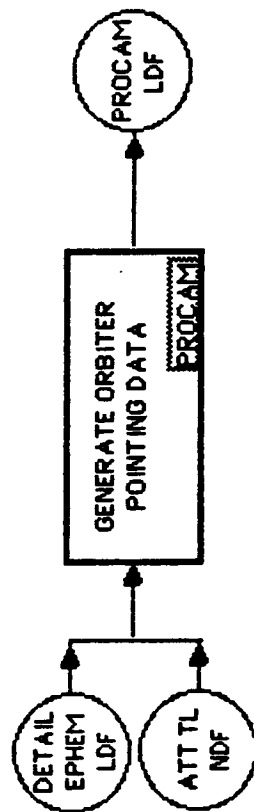


FUNCTION: ORBITAL ANALYSIS

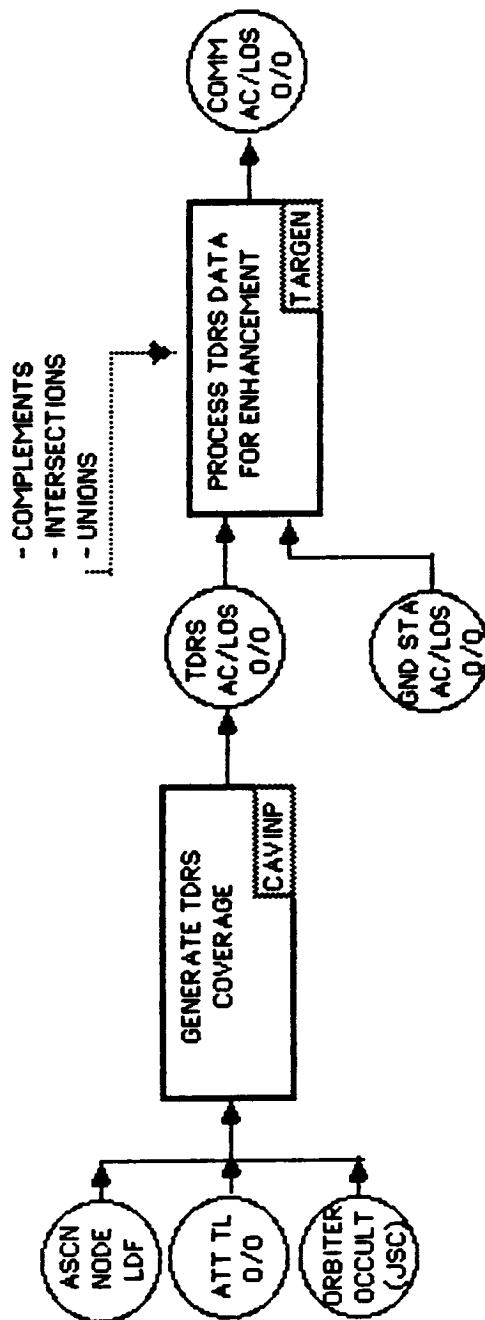
SUBFUNCTION: ATTITUDE/MANEUVER TIMELINE GENERATION (DEDICATED STELLAR)



FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: ORBITER POINTING DATA GENERATION

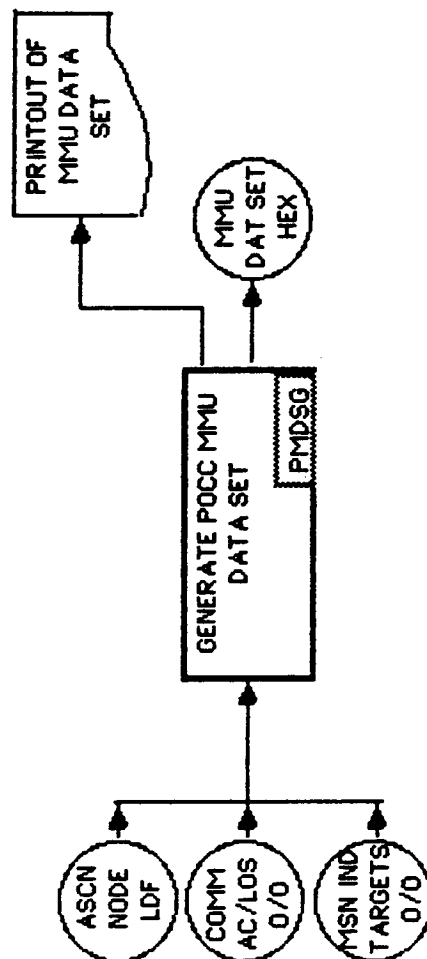


FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: TDRS ACQ/LOSS GENERATION

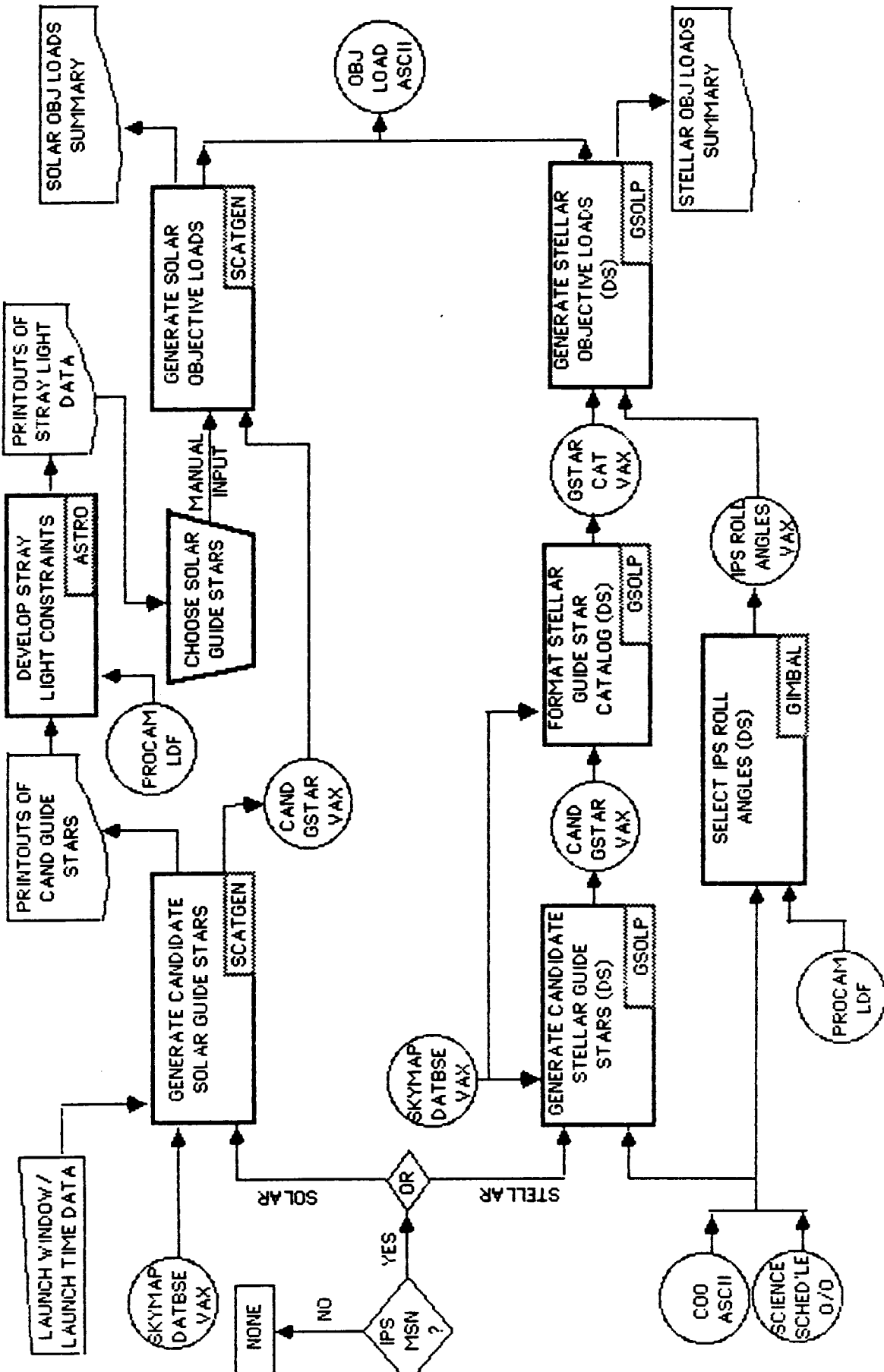


NOTE: THIS PROCESS IS APPLICABLE TO EITHER
MULTIDISCIPLINE OR DEDICATED STELLAR
MISSIONS

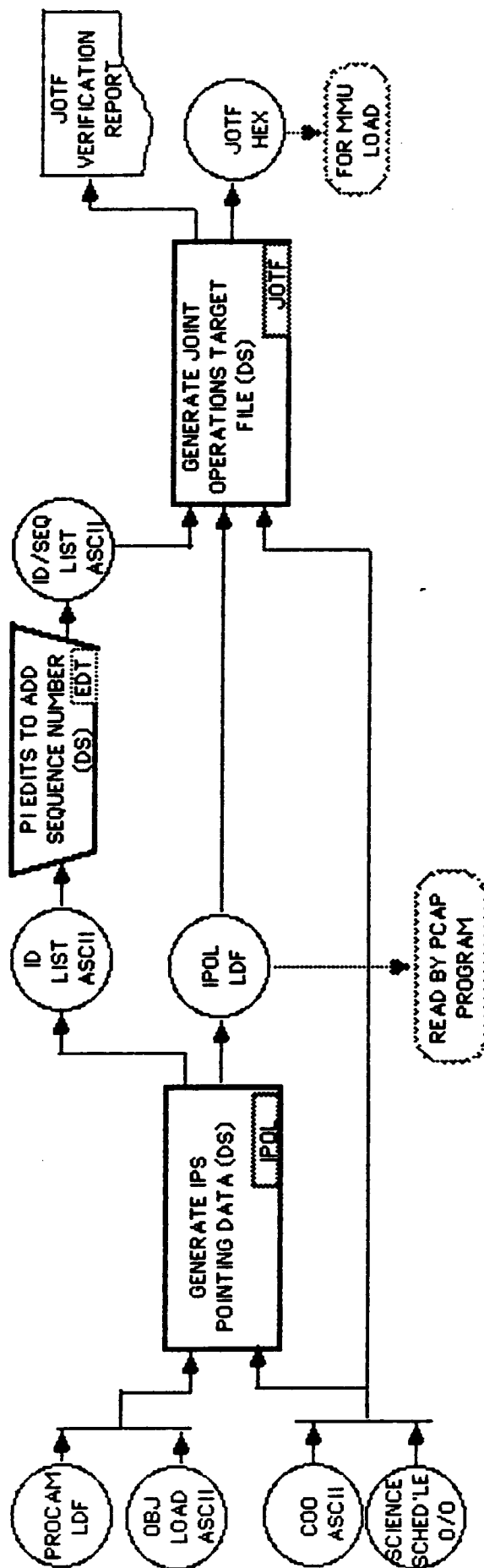
FUNCTION: ORBITAL ANALYSIS
SUBFUNCTION: POCC MMU DATA SET GENERATION



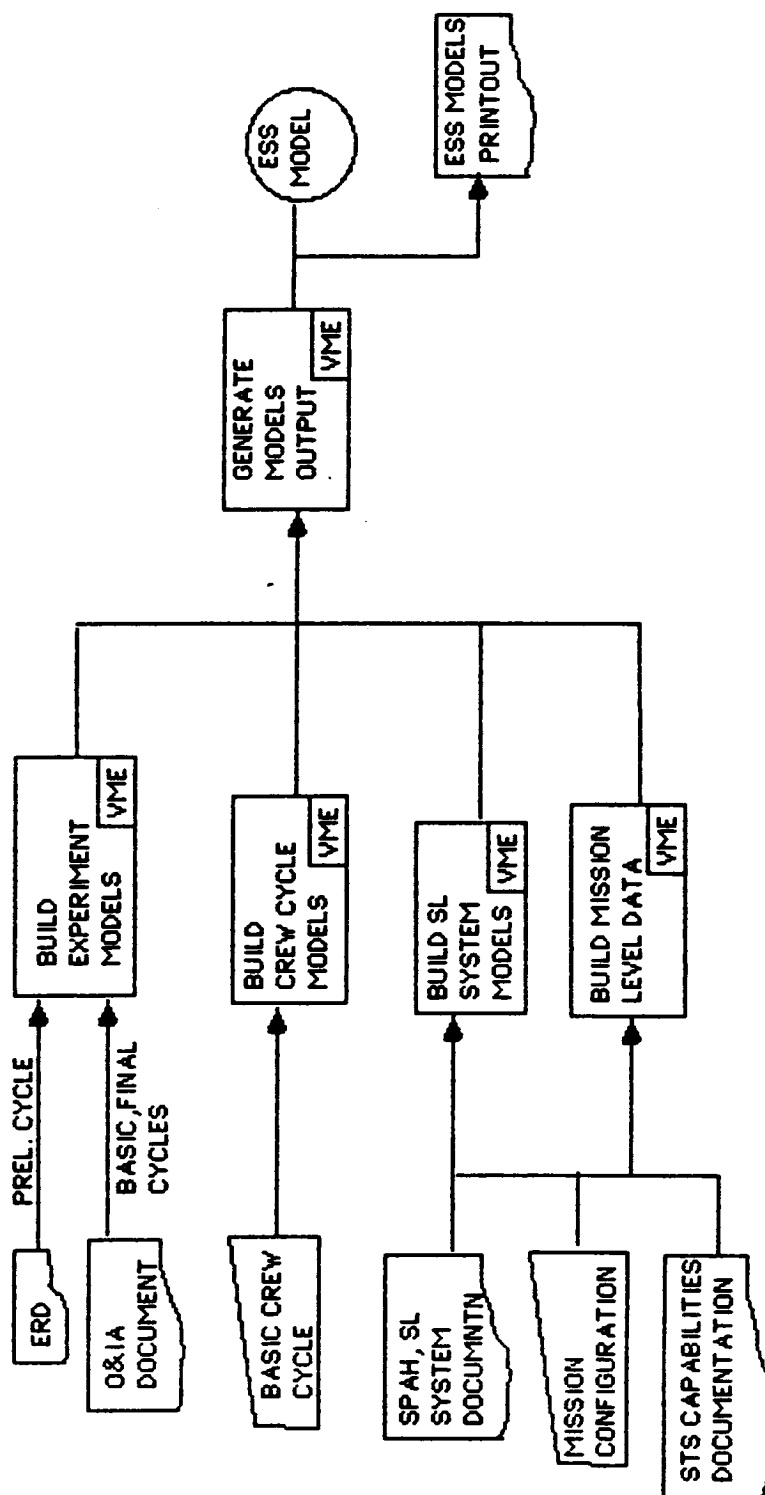
SUBFUNCTION: OBJECTIVE LOADS GENERATION



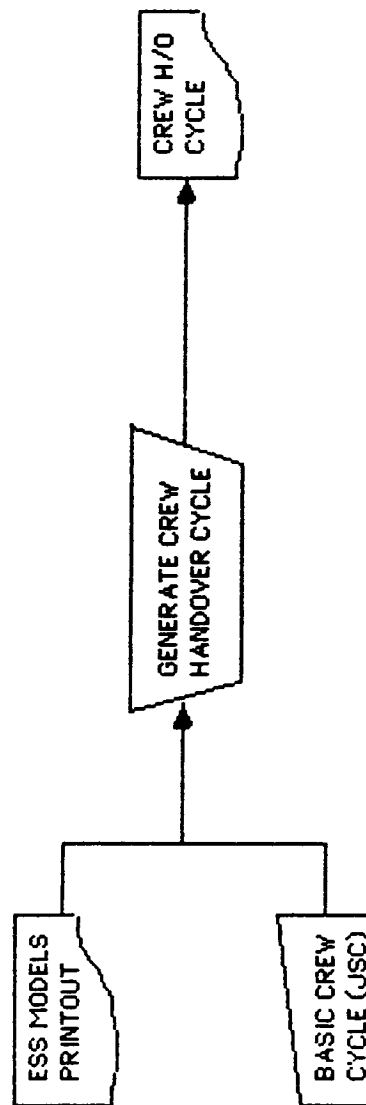
FUNCTION: ORBITAL ANALYSIS
 SUBFUNCTION: JOINT OPERATIONS TARGET FILE GENERATION (DEDICATED STELLAR)



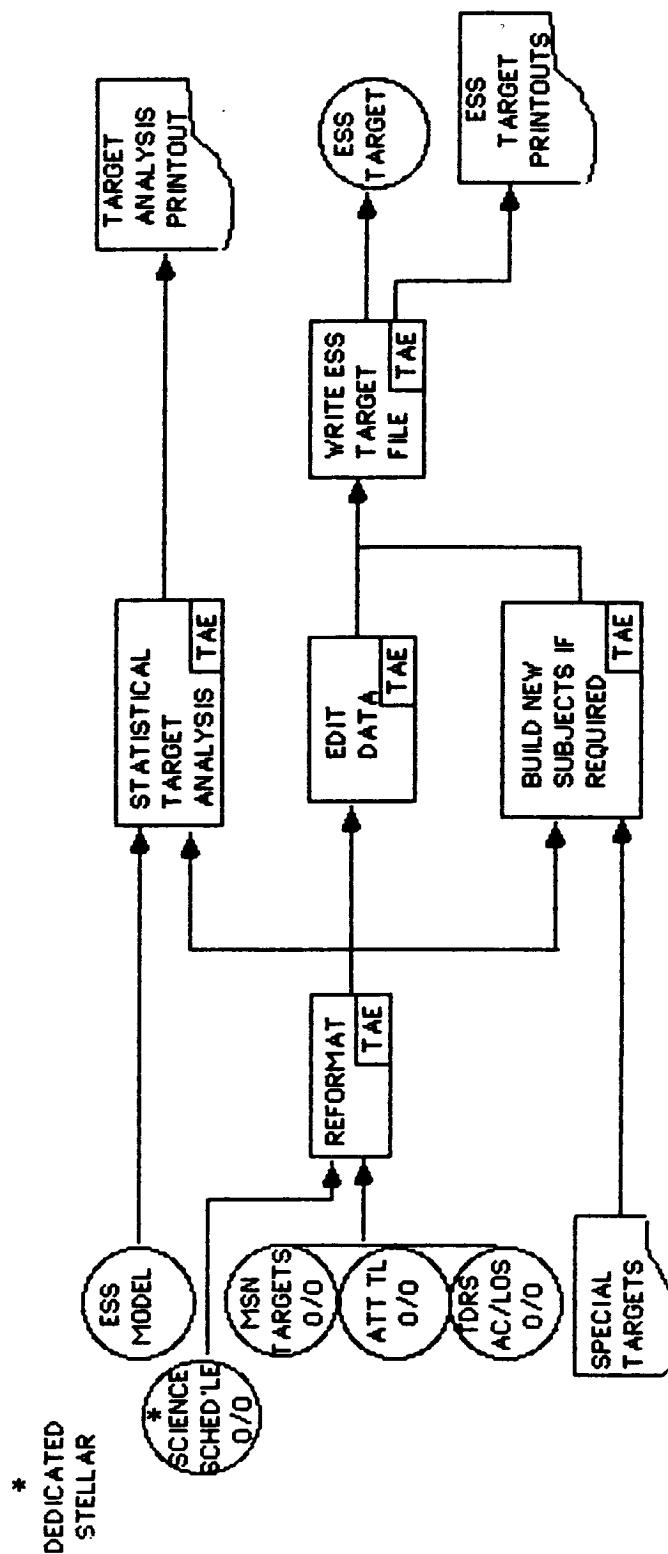
FUNCTION: MISSION TIMELINE ANALYSIS
SUBFUNCTION: CREATE MISSION TIMELINE MODELS



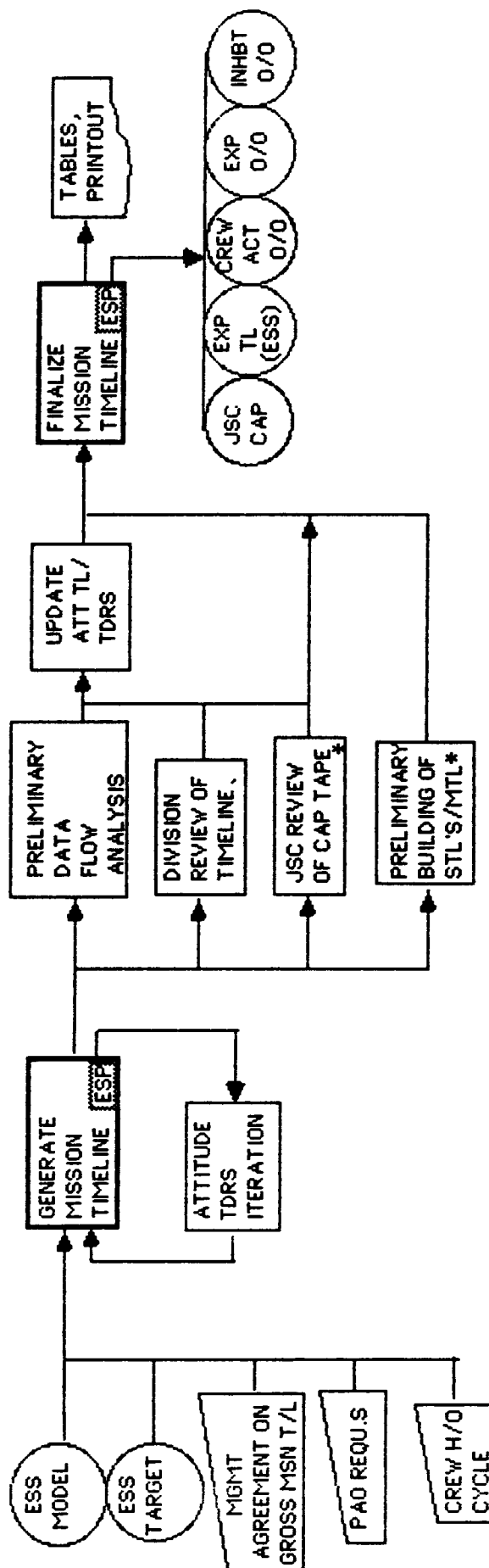
FUNCTION: MISSION TIMELINE ANALYSIS
SUBFUNCTION: GENERATE CREW H/O CYCLE



FUNCTION: MISSION TIMELINE ANALYSIS
SUBFUNCTION: CREATE ESS TARGET FILE

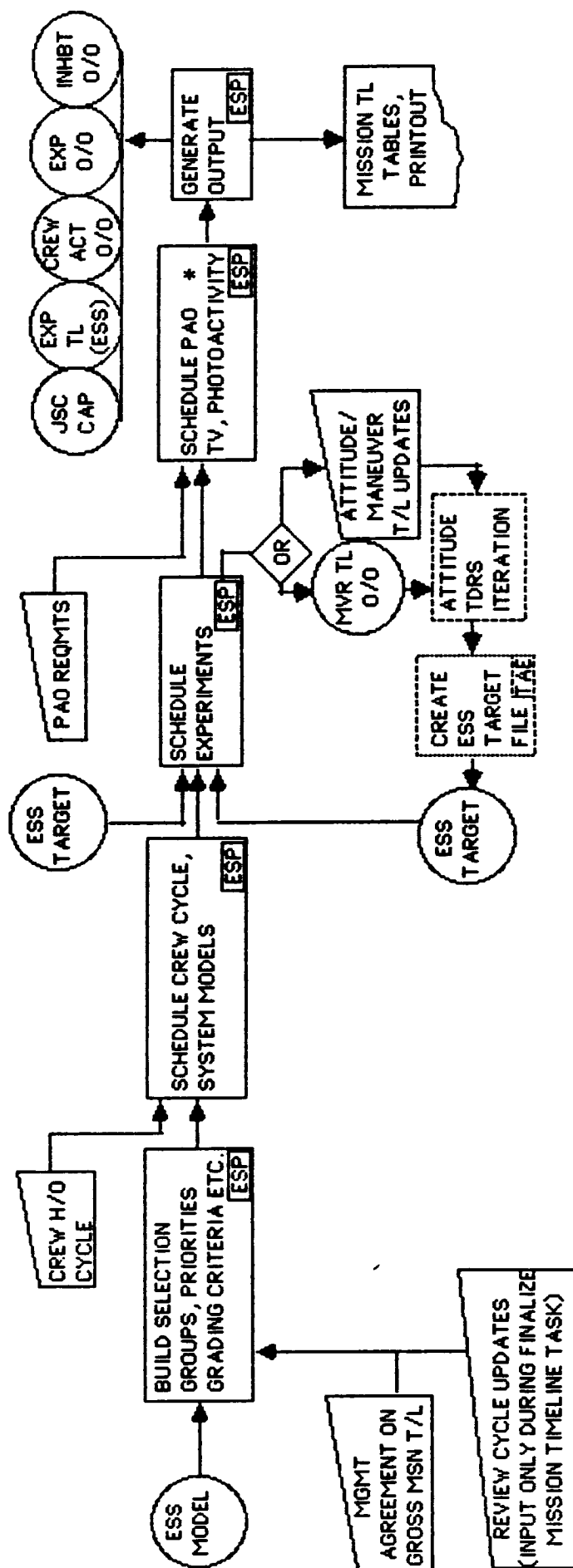


FUNCTION: MISSION TIMELINE ANALYSIS
SUBFUNCTION: MISSION TIMELINE GENERATION



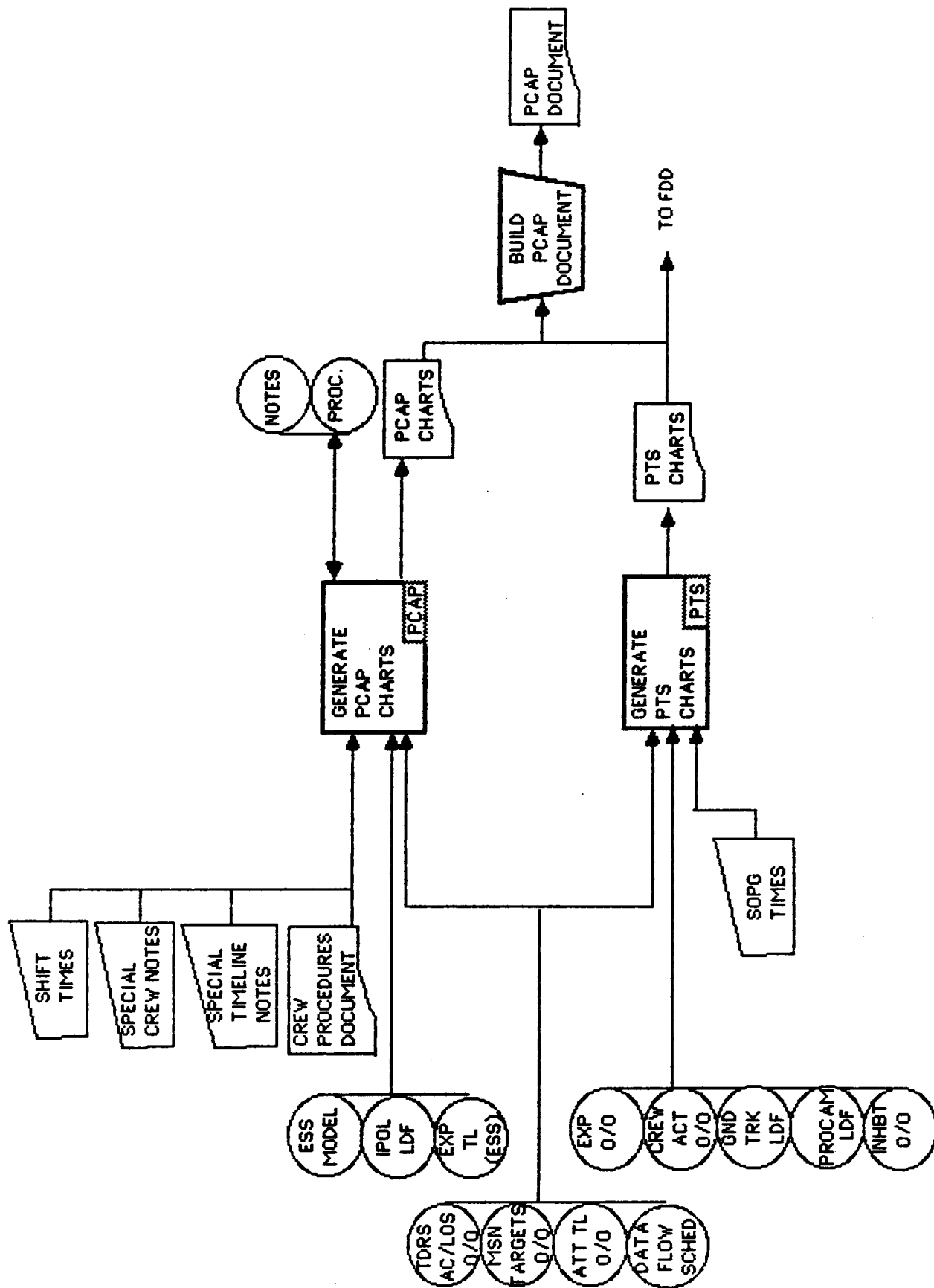
* NOT PERFORMED IN PRELIMINARY CYCLE
NOTE : BOLD LINES INDICATE MISSION TIMELINE GENERATION TASKS.
REMAINING TASKS ARE SHOWN FOR CLARITY.

FUNCTION: MISSION TIMELINE ANALYSIS
SUBFUNCTION: MISSION TIMELINE GENERATION
TASK: GENERATE MISSION TIMELINE
 (FINALIZE MISSION TIMELINE TASK INCLUDES THE NECESSARY
 SUBTASKS OF THE GENERATE MISSION TIMELINE TASK)

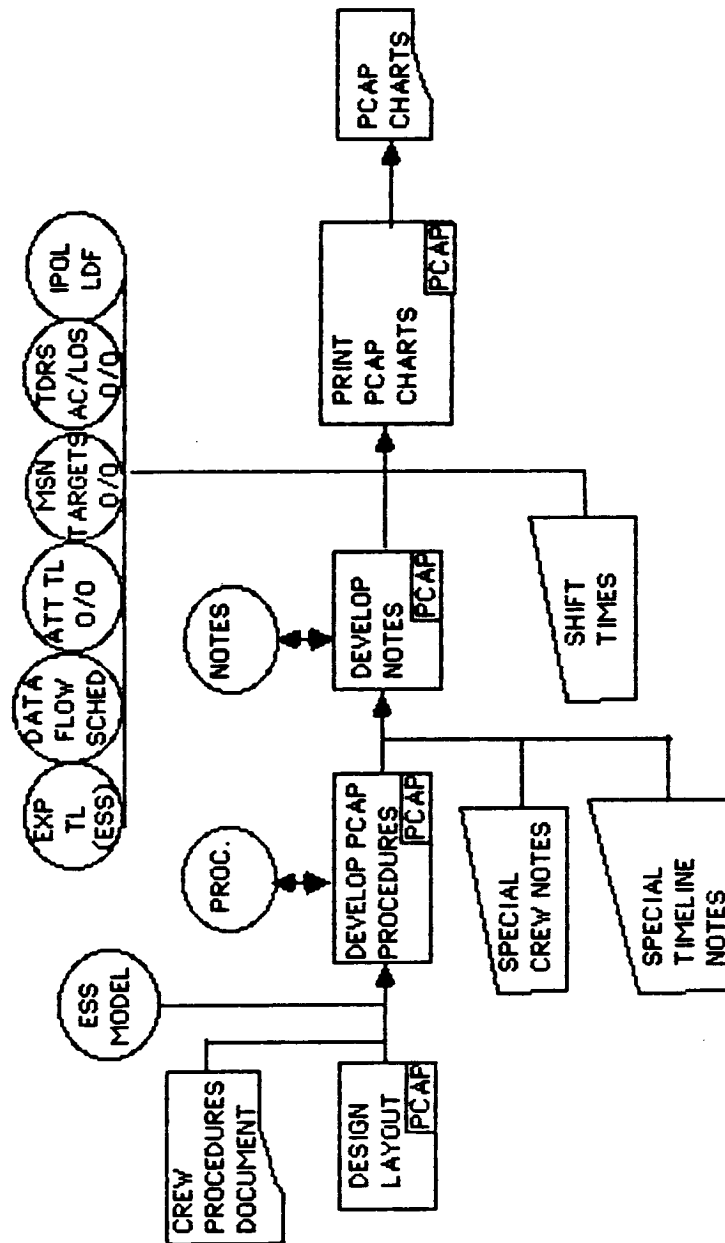


* NOT PERFORMED IN PRELIMINARY CYCLE
 NOTE: DOTTED LINES INDICATE TASKS SHOWN FOR CLARITY
 AND ARE NOT MISSION TL GENERATION SUBTASKS

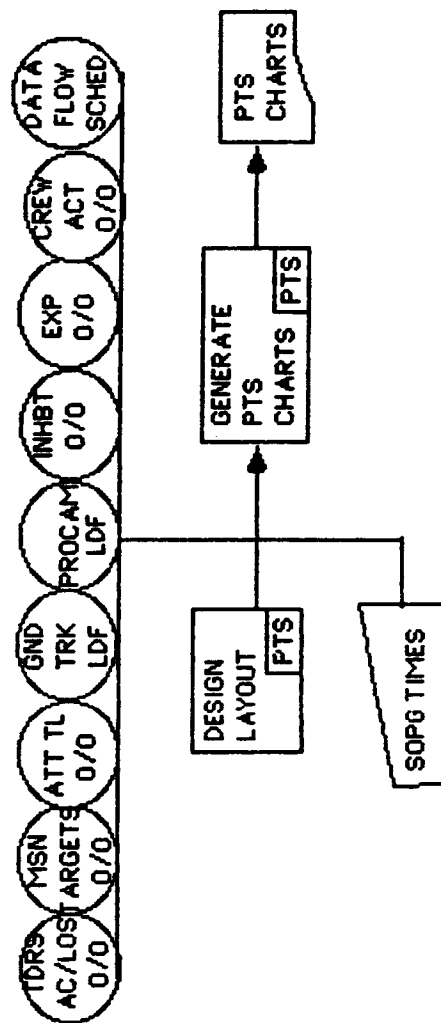
FUNCTION: MISSION TIMELINE ANALYSIS
SUBFUNCTION: PAYLOAD CREW ACTIVITY PLAN DEVELOPMENT



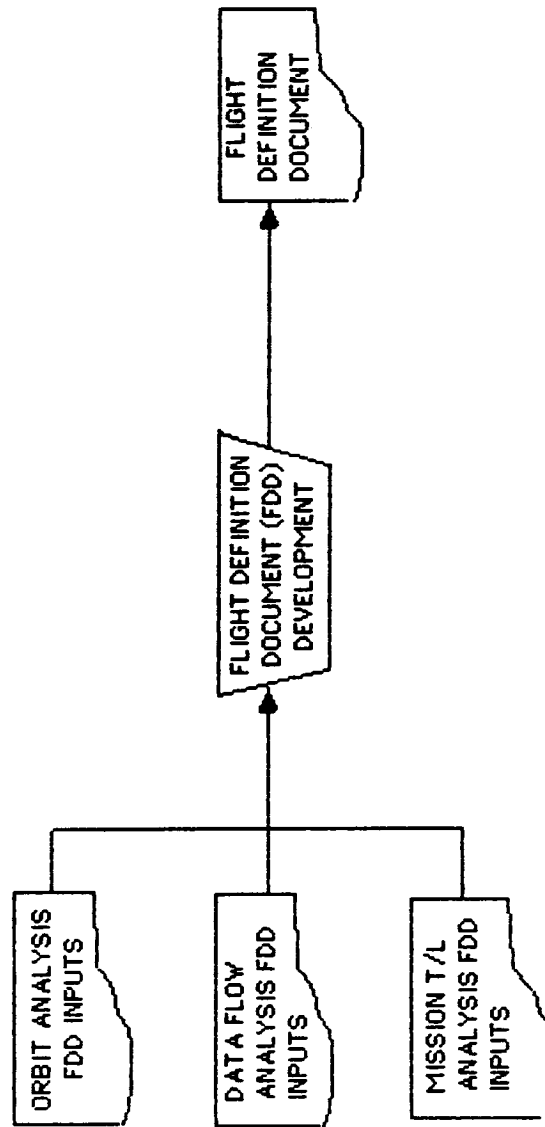
FUNCTION: MISSION TIMELINE ANALYSIS
SUBFUNCTION: PAYLOAD CREW ACTIVITY PLAN DEVELOPMENT
TASK: GENERATE PCAP CHARTS



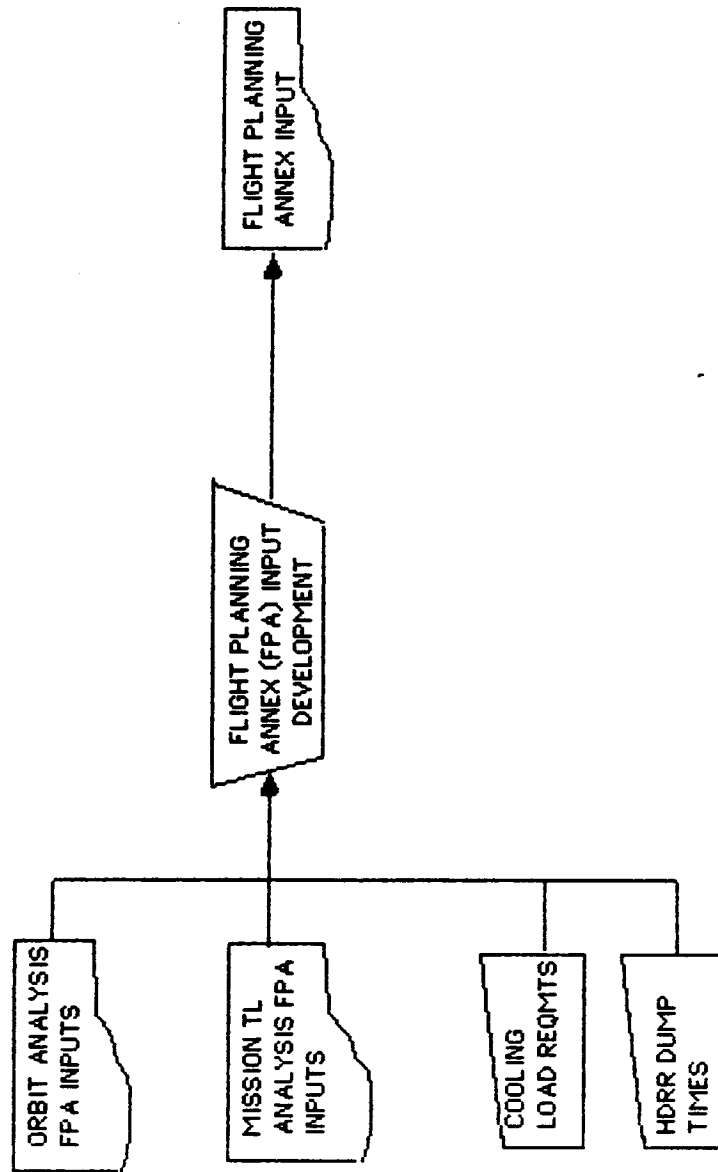
FUNCTION: MISSION TIMELINE ANALYSIS
SUBFUNCTION: PAYLOAD CREW ACTIVITY PLAN DEVELOPMENT
TASK: GENERATE PTS CHARTS



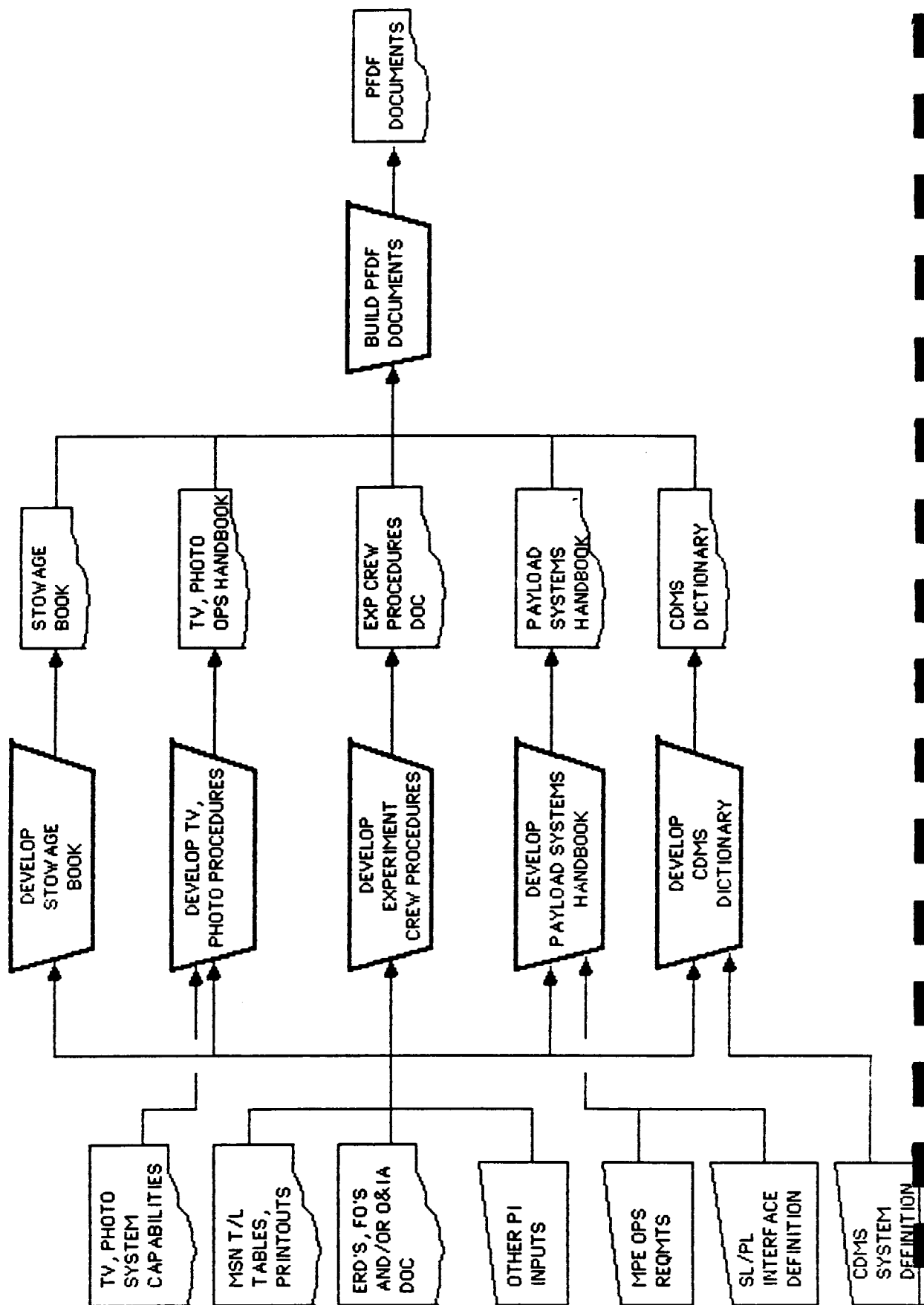
FUNCTION: FLIGHT DEFINITION DOCUMENT DEVELOPMENT



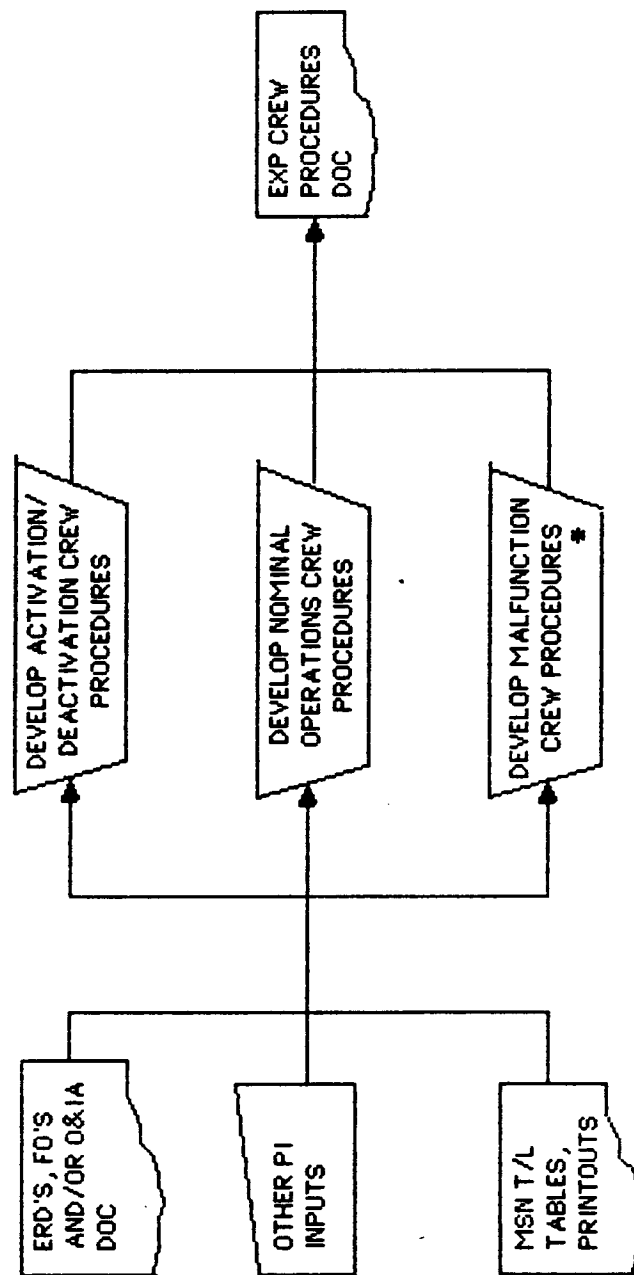
FUNCTION: FLIGHT PLANNING ANNEX INPUT DEVELOPMENT



FUNCTION: CREW PROCEDURES DEVELOPMENT

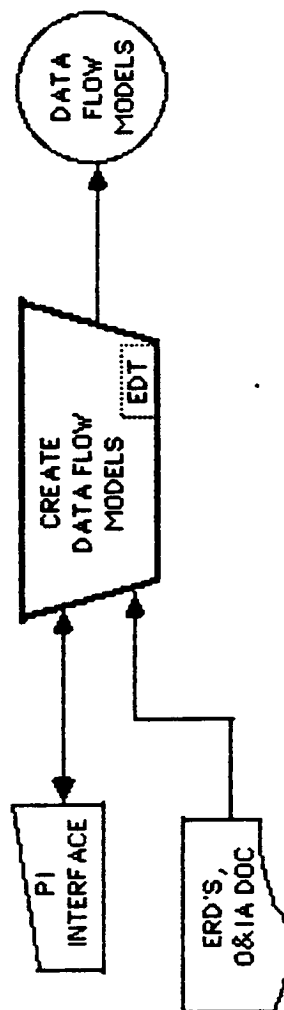


FUNCTION: CREW PROCEDURES DEVELOPMENT
SUBFUNCTION: DEVELOP EXPERIMENT CREW PROCEDURES

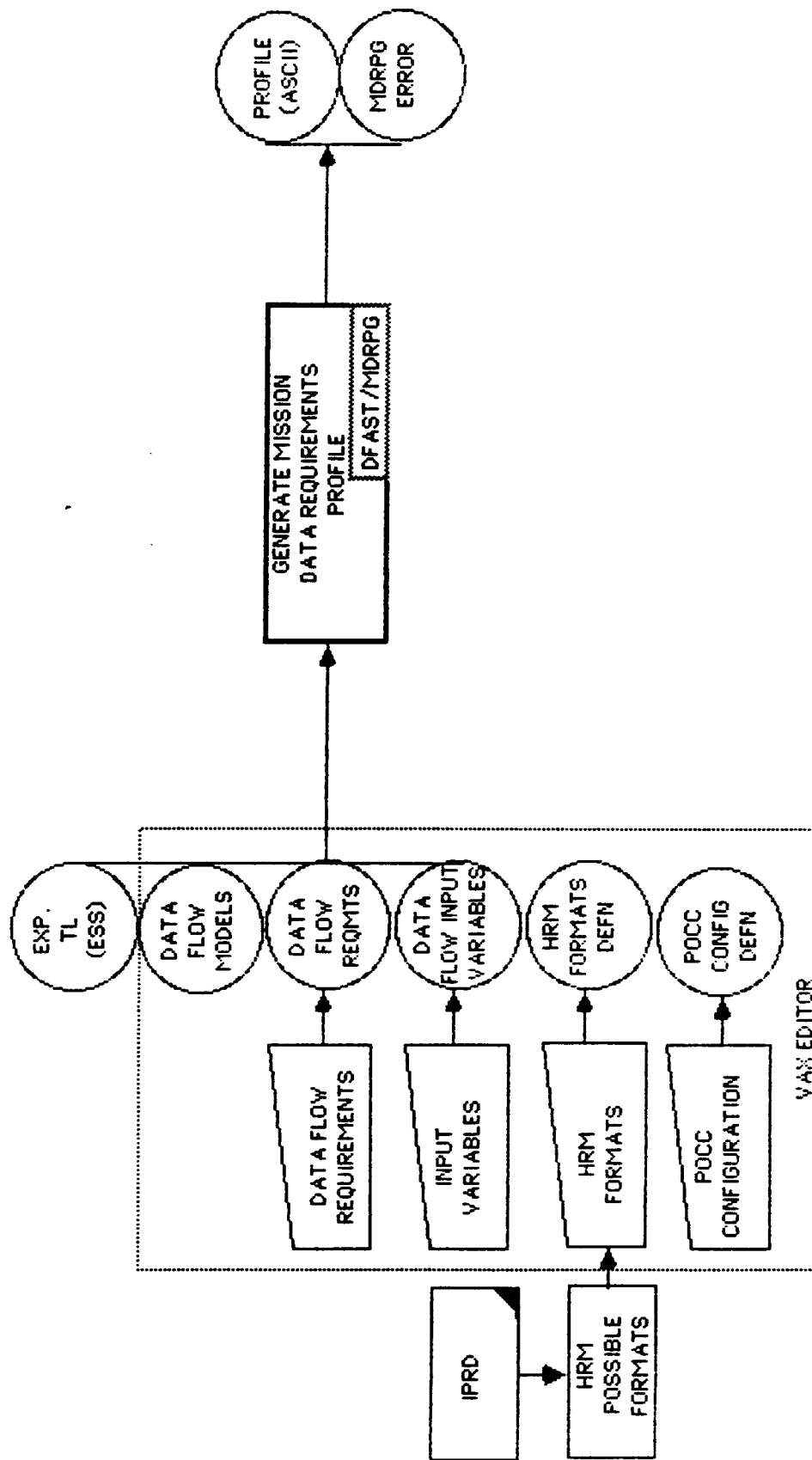


* NOT PERFORMED IN PRELIMINARY CYCLE

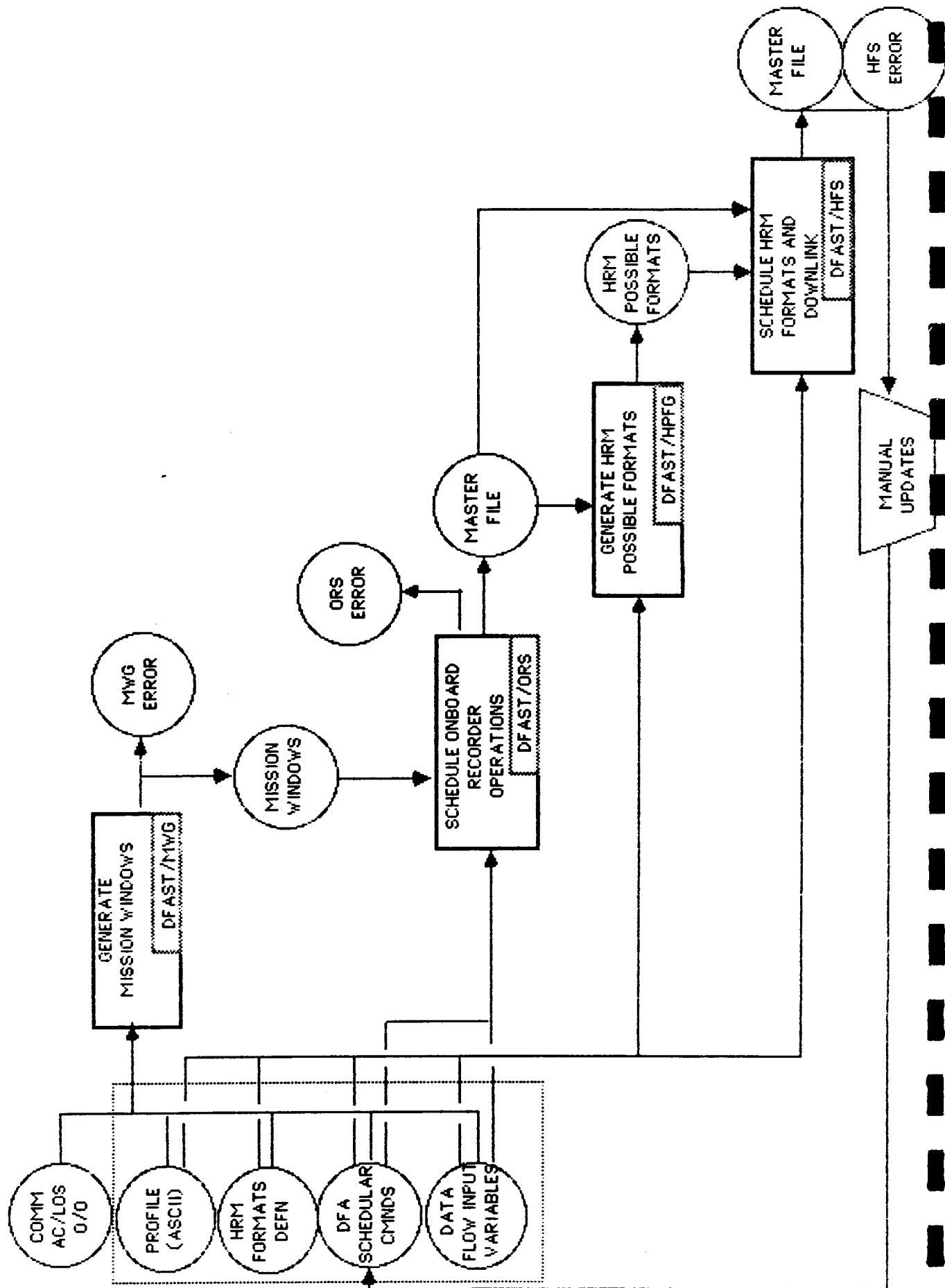
FUNCTION: DATA FLOW ANALYSIS
SUBFUNCTION: CREATE DATA FLOW MODELS



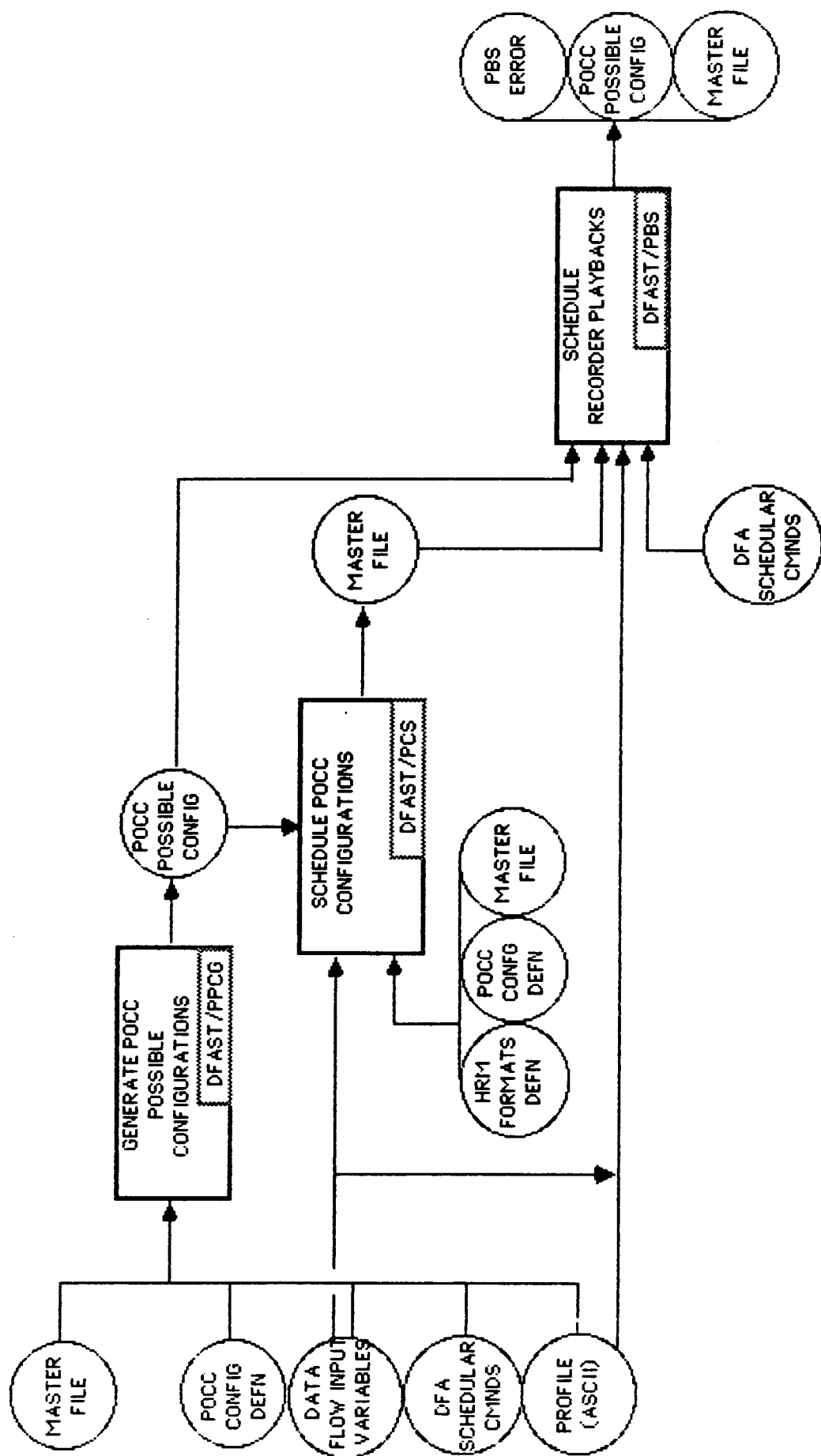
FUNCTION: DATA FLOW ANALYSIS
 SUBFUNCTION: GENERATE MISSION DATA REQUIREMENTS PROFILE



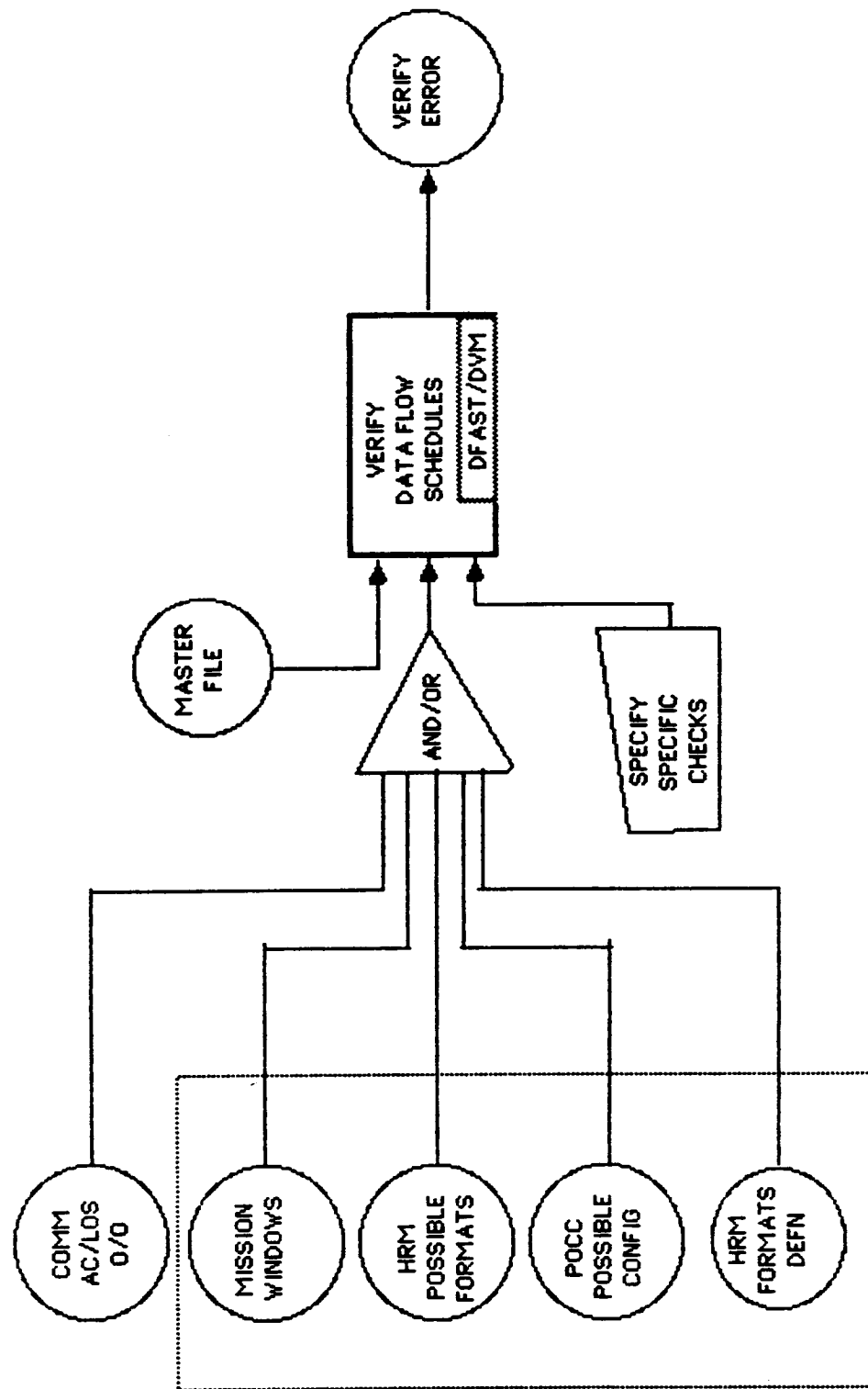
FUNCTION: DATA FLOW ANALYSIS
SUBFUNCTION: SCHEDULE ONBOARD DATA MANAGEMENT AND DOWNLINK



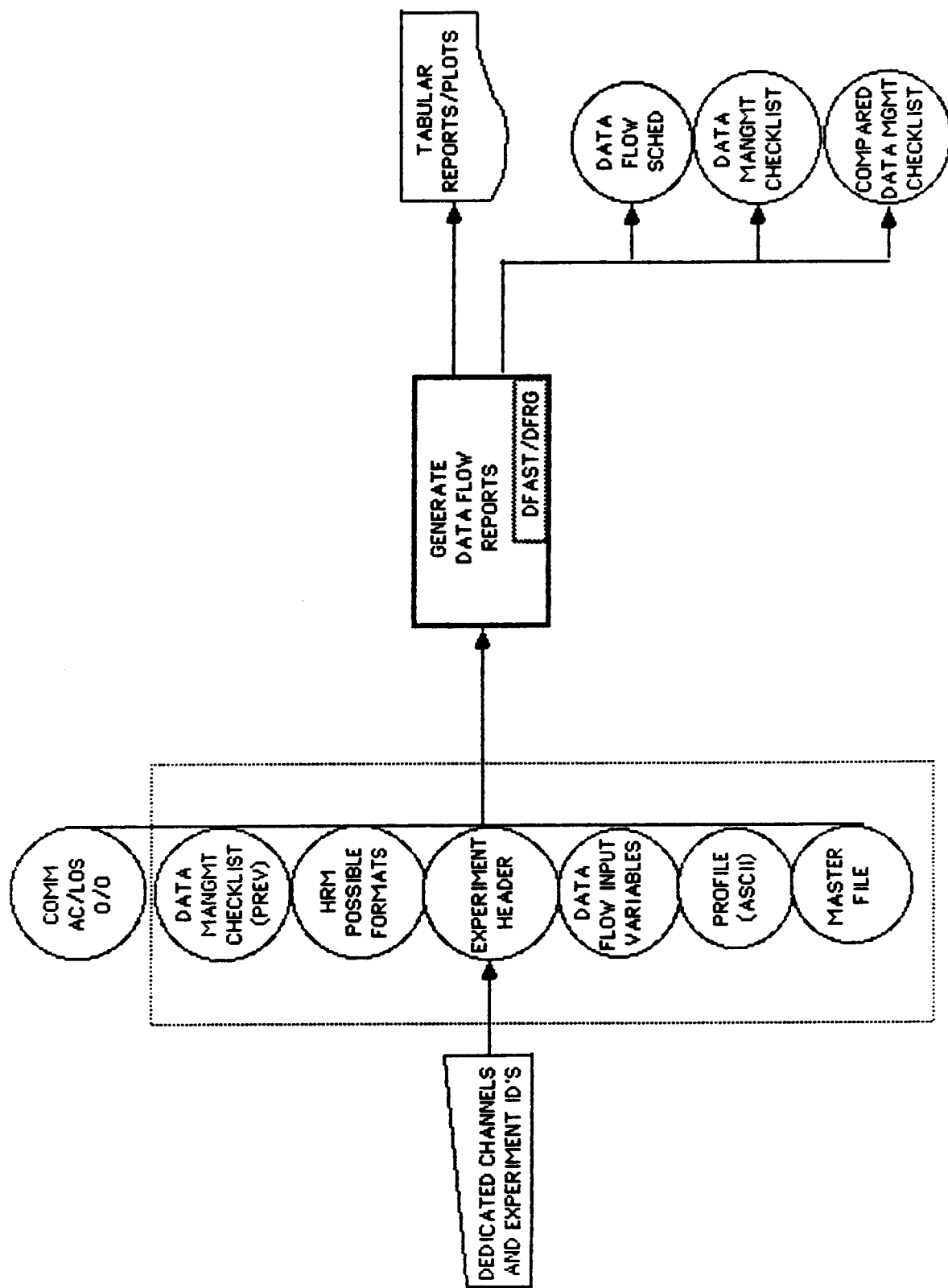
FUNCTION: DATA FLOW ANALYSIS
 SUBFUNCTION: SCHEDULE POCC DATA CAPTURE/MANAGEMENT/DISTRIBUTION



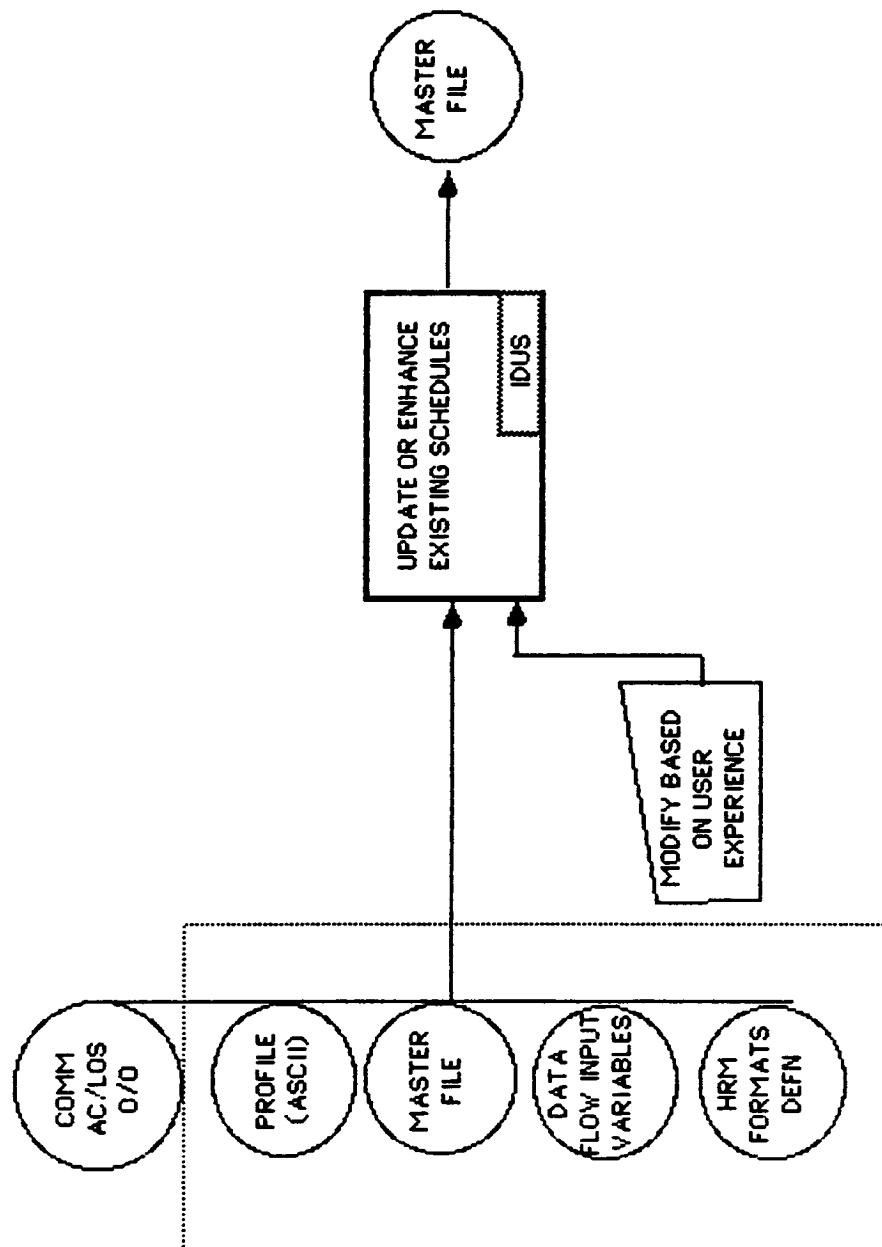
FUNCTION: DATA FLOW ANALYSIS
SUBFUNCTION: VERIFICATION OF DATA FLOW SCHEDULES



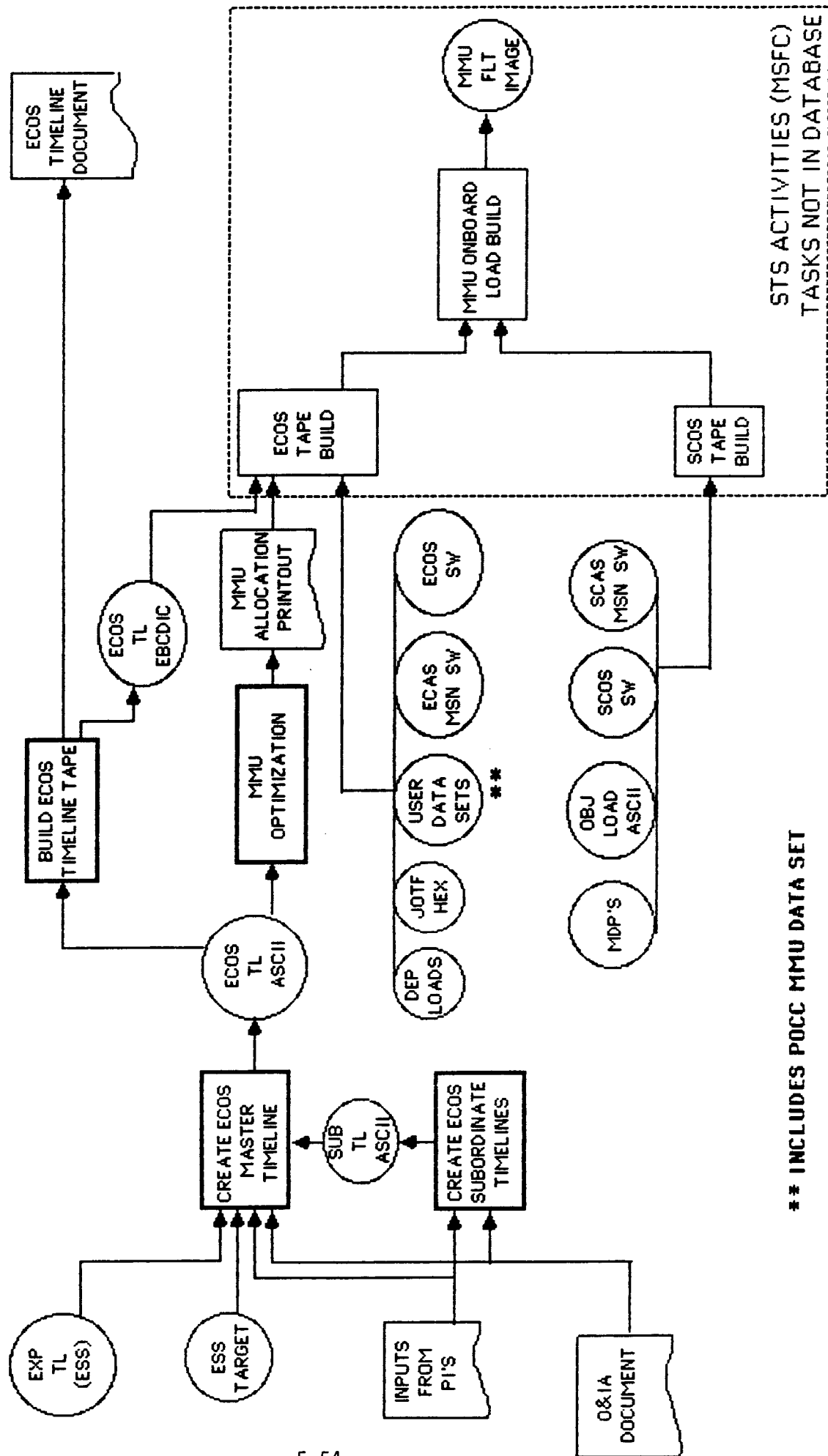
FUNCTION: DATA FLOW ANALYSIS
 SUBFUNCTION: DATA FLOW AND SYSTEMS CONFIGURATION DOCUMENT DEVELOPMENT



FUNCTION: DATA FLOW ANALYSIS
SUBFUNCTION: UPDATE OR ENHANCE EXISTING SCHEDULE

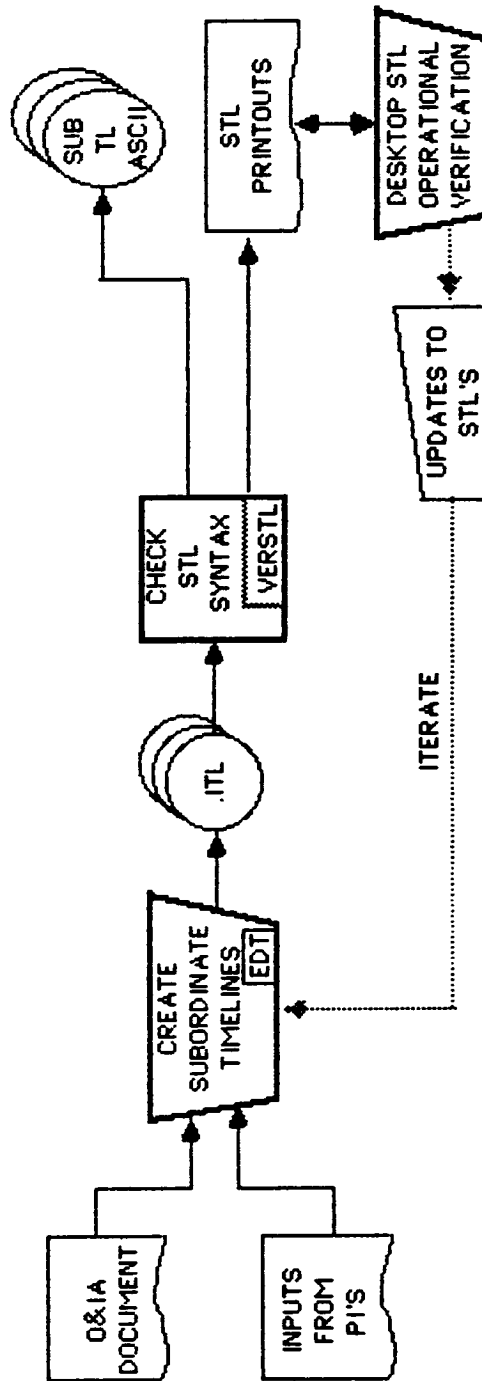


FUNCTION: MMU LOAD INPUT DEVELOPMENT

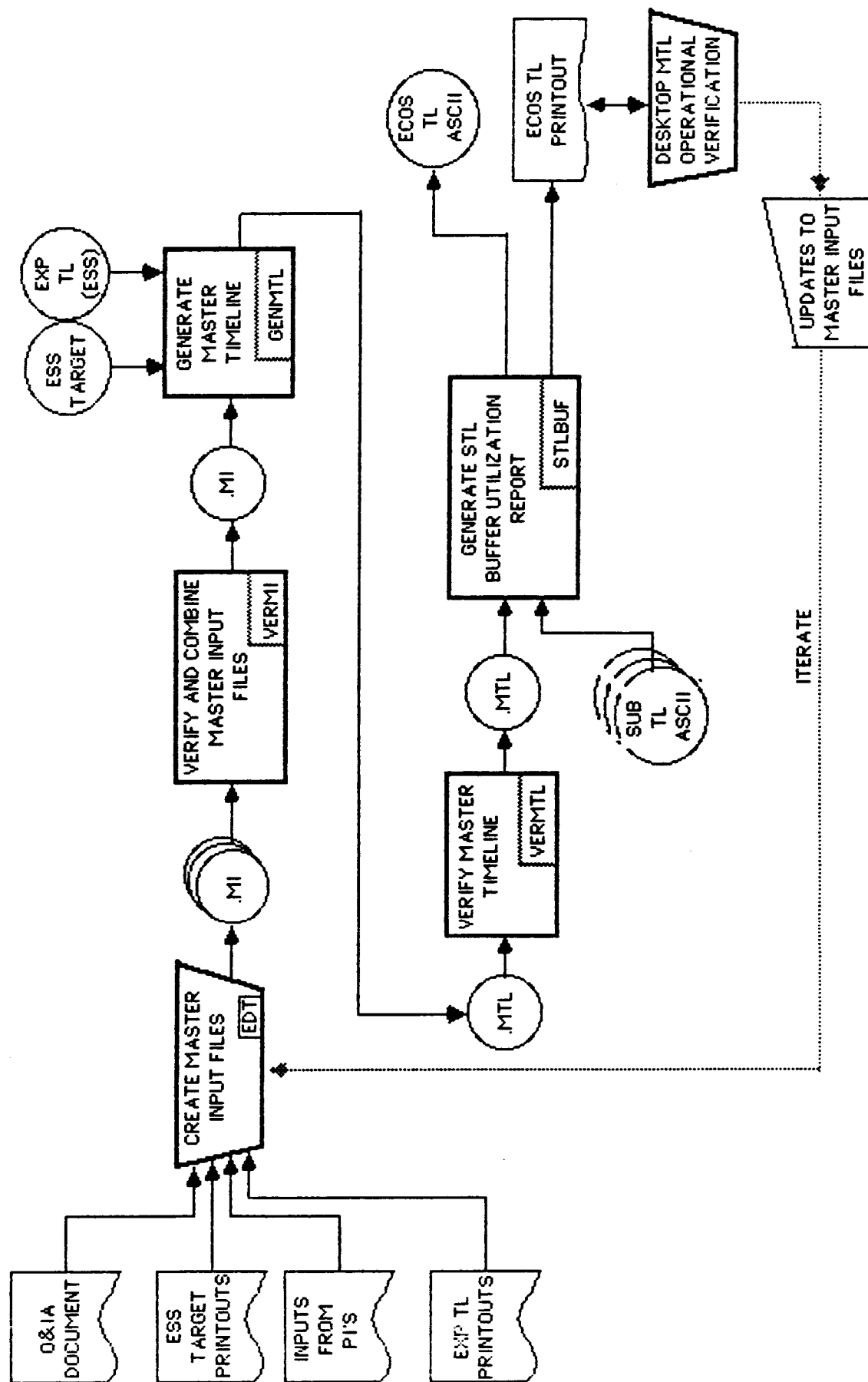


** INCLUDES POCC MMU DATA SET

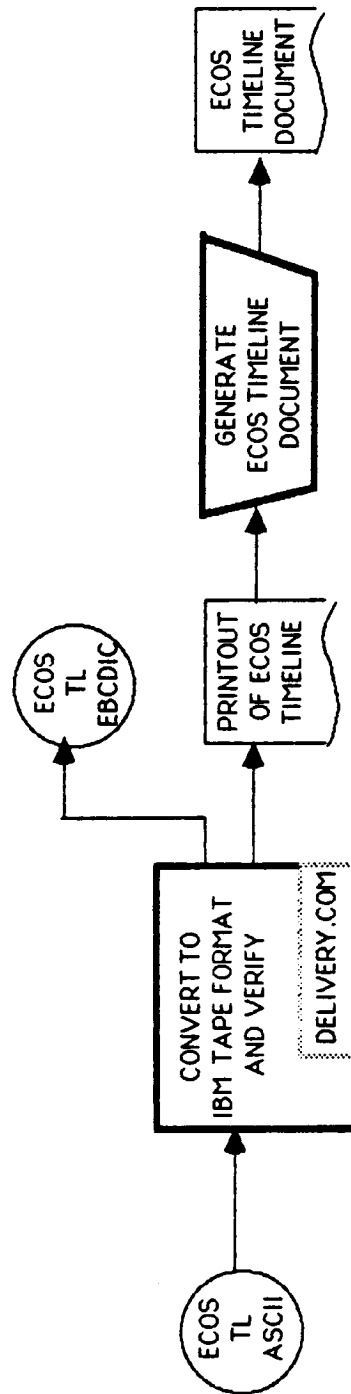
FUNCTION : MMU LOAD INPUT DEVELOPMENT
 SUBFUNCTION : CREATE ECOS SUBORDINATE TIMELINES



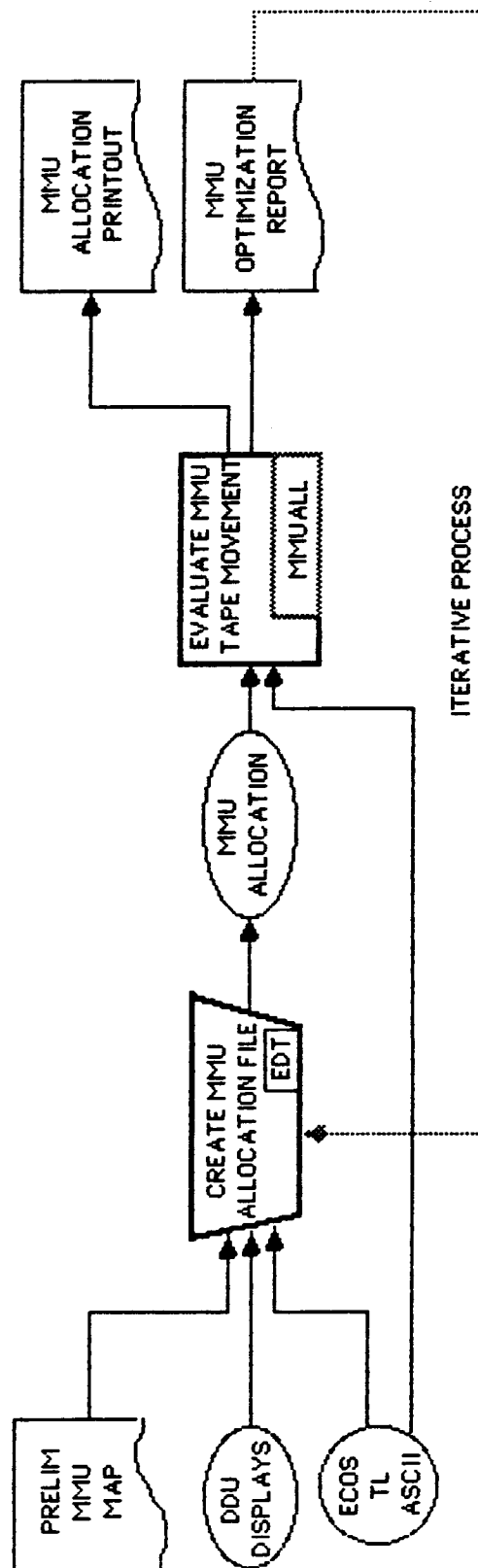
FUNCTION : MMU LOAD INPUT DEVELOPMENT
SUBFUNCTION : CREATE ECOS MASTER TIMELINE



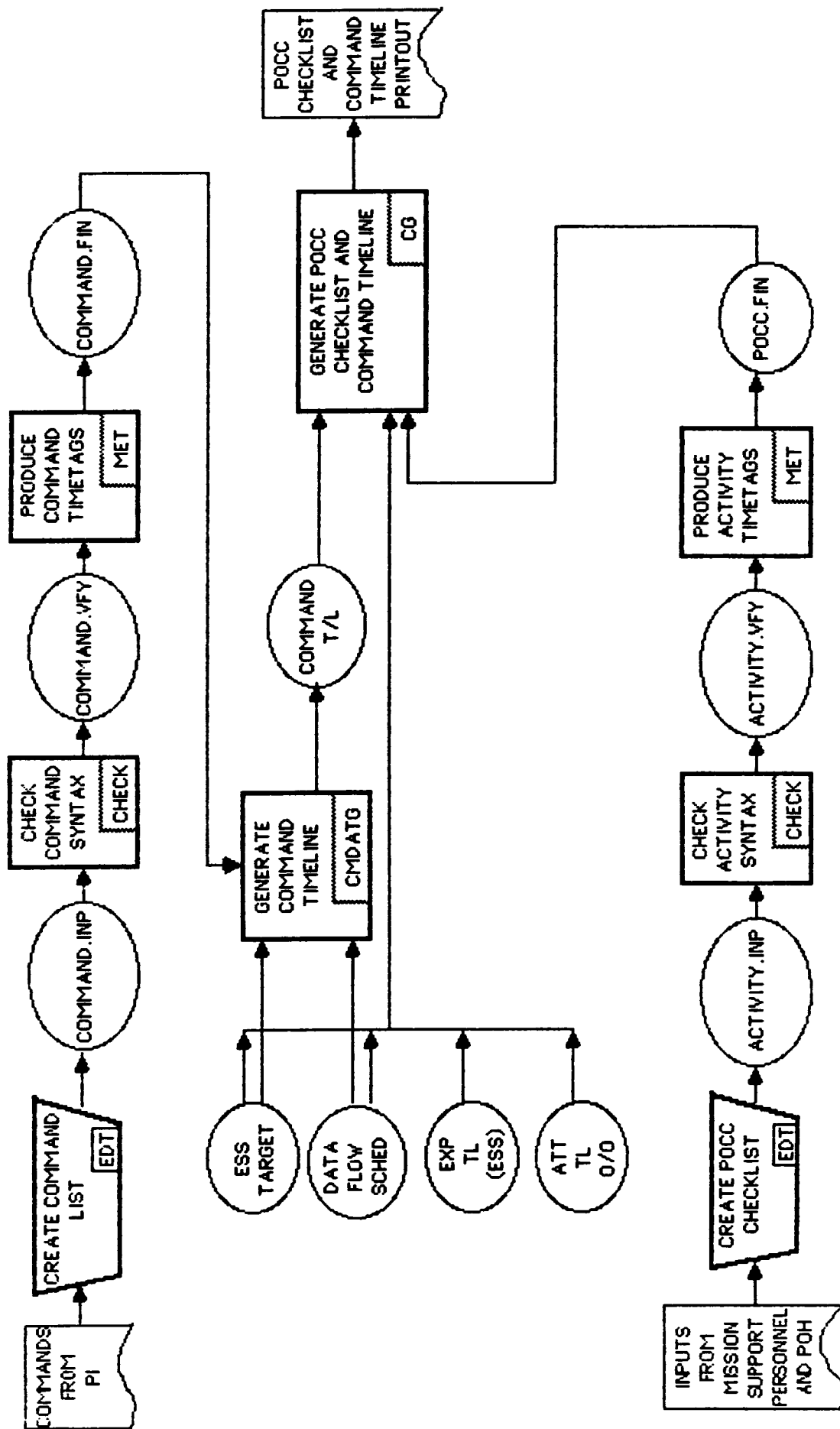
FUNCTION : MMU LOAD INPUT DEVELOPMENT
SUBFUNCTION : BUILD ECOS TIMELINE TAPE



FUNCTION : MMU LOAD INPUT DEVELOPMENT
 SUBFUNCTION : MMU OPTIMIZATION



FUNCTION : EXPERIMENT COMMAND PLANNING DEVELOPMENT



Section 6

TASK 3 - DEVELOP SPACE STATION MISSION PLANNING CONCEPT AND SOFTWARE REQUIREMENTS

6.1 ACTIVITIES AND ACCOMPLISHMENTS

As indicated previously, the objective of this task was to develop a payload mission planning concept consistent with the overall Space Station operations philosophy and to define a system of software requirements maximizing use of SL MIPS software modules (modified as necessary) to implement the concept.

The approach taken to this task consisted of four subtasks. First, basic definitions, groundrules, and assumptions were established; these pertained to the current Space Station design and operations concepts and philosophies, the scope of mission planning for Space Station, objectives/requirements to be achieved/satisfied by the approach to mission planning, the structure of organizations/personnel involved in mission planning, the number, purpose, and nature of planning cycles for Space Station, and the degree of allocation of mission planning functions between ground-based organizations and the on-board crew. The second subtask involved the construction of a set of functional flow diagrams defining the Space Station payload mission planning concept to a level of detail equivalent to the Spacelab functional flow diagrams. The third subtask then involved the identification of modified SL MIPS software modules or new computer programs to automate individual mission planning activities identified in the flow diagrams. The fourth and final subtask involved the summarization and systemization into a hierarchical structure of the new or modified SL MIPS software programs as the basis for preparation of a software development plan in Task 5.

Inputs to this study task were derived from a variety of sources:

- Space Station Program reference documents
- Space Station plans, study reports, white papers, briefings, meeting minutes, etc., published by NASA organizations, contractors, and working groups, including the NASA Space Station Operations Task Force and its panels
- Task 2 products and knowledge pertaining to the Spacelab mission planning process

The products of this task consist of the Space Station payload mission planning concept functional flow diagrams, a summary table describing the new and modified SL MIPS software modules required to implement the SS MPS concept, and the hierarchical structure of software for the SS MPS. The SS mission planning concept functional flow diagrams, including an explanation of the fundamental definitions, groundrules, and assumptions supporting the mission planning approach, as well as an explanation of the flow diagrams themselves, are contained in Subsection 6.2. The summary table of required new and modified SL MIPS software modules and the hierarchical structure of required software modules are presented in Subsection 6.3.

6.2 SPACE STATION MISSION PLANNING CONCEPT

6.2.1 Introduction

This section presents and explains the functional flow diagrams representative of the Space Station (SS) payload mission planning concept developed under Task 3 of the SS Mission Planning System (MPS) Development Study.

Prior to presenting the functional flow diagrams in Section 6.2.3 below, the following section provides the fundamental definitions, groundrules, and assumptions which support the approach to SS mission planning reflected in the flow diagrams.

6.2.2 Definitions, Groundrules and Assumptions

6.2.2.1 Space Station Physical Configuration

The Initial Operations Capability (IOC) configuration of the International Space Station was used as the baseline for this Study. It will be built up over a four year period by about 30 assembly missions and is designed for future growth and enhancements. It consists of the following elements:

- | | |
|---|------------------------|
| o U.S. Laboratory Module | o Airlock |
| o ESA Laboratory Module | o Hyperbaric Airlock |
| o JEM Laboratory and Exposed Facility | o Telerobotic Servicer |
| o Habit Module | o Solar Power Module |
| o Mobile Servicing Center (MSC) | o Truss Assembly |
| o MSC Maintenance Depot | o Propulsion Assembly |
| o Mobile Transporter | o Resource Node 1 |
| o Servicing Facility | o Resource Node 2 |
| o Attached Payloads Platforms and Accommodation Equipment | o Resource Node 3 |
| o Pressurized Logistics Carrier | o Resource Node 4 |
| o Unpressurized Logistics Carrier | |
| o JEM Experiment Logistics Module | |

In addition to the manned base described above, the current definition for the Space Station system includes co-orbiting and polar platforms.

Primary physical accommodations to payloads on the manned base will be provided by the laboratories and the attached payload platforms and accommodation equipment.

Additional detailed definition of the Space Station physical configuration may be found in JSC 30000, the Space Station Program Definition and Requirements Document (PDRD).

6.2.2.2 Space Station Flight Operations Scenario

The Space Station orbit will be nominally circular with a normal operative altitude of 463 km and an inclination of 28.5° (JSC 30000). Orbit characteristics will not be adjusted to accommodate particular payload requirements.

The Space Transportation System (STS) Space Shuttle will be the primary support vehicle to the Space Station. As the Space Station orbit decays, the STS will be planned to rendezvous with the Space Station for the purpose of accomplishing logistics resupply, payload equipment and crew changeout. Space Station reboost will be nominally performed after each STS visit.

6.2.2.3 Scope of Payload Mission Planning

For the purpose of this Study, the scope of payload mission planning was assumed to encompass the operations of multi-discipline payloads contained within or attached to the Space Station manned base elements. Excluded, therefore, were the operations of payloads on co-orbiting or polar platforms; it was assumed that the influences of these platforms on manned base payload operations will be input to the planning process in the form of Space Station operations constraints.

The various payload disciplines considered, particularly those whose operational requirements would include specific orbital environmental conditions, fields of view, or targets, were the same as those currently accommodated by the Spacelab mission planning process - namely,

- Astrophysics
- Solar physics
- Plasma physics
- Earth sciences
- Life sciences
- Material Science

The scope of mission planning was further assumed to apply to a "mission increment", the period (up to 90 days) of Space Station orbital operations bounded by STS visits (i.e., fixed payload complement).

The payload mission planning process was assumed to commence with definition of the payload complement and the corresponding accommodating Space Station elements for the mission increment.

The payload mission planning process activities were assumed to range from the collection of payload operations requirements data through the preparation of mission execution plans and procedures. It was further assumed that the process must accommodate real-time replanning, as well as pre-flight planning. (This is similar in scope to the Spacelab process, the definition of which provided an excellent foundation for identifying required Space Station planning activities.)

6.2.2.4 Approach to Payload Mission Planning

A. Objectives

The following objectives were established for the Space Station mission planning process, many of which were based on an assessment of the characteristics of, or lessons learned from the Spacelab mission planning process:

- decentralize planning; specifically maximize direct Space Station user involvement via user-friendly interfaces
- to ensure the use and production of common data, and to facilitate the integration of planning data, provide common capabilities at common planning levels (from the users up to an assumed payload operations integration center)
- automate to the maximum extent possible
- eliminate paper; emphasize readily accessible data bases between geographically dispersed locations of planning activity
- provide SS user flexibility within allocated resource constraints
- minimize the intensity (labor and computer) of planning activities

The final two objectives were especially encouraged by lessons learned from the Spacelab payload mission planning process, in which the relatively short duration mission (7-10 days) forced the planning of payload activities down to the minute to maximize the utilization of resources. To achieve these two objectives, an approach of using "resource allocation envelopes" was assumed, where such an envelope is a prescribed period of time with an associated vector of average resource utilization levels. In addition to achieving the aforementioned two objectives, this approach was justified by the longer duration mission for Space Station (compared to Spacelab). Also, the obvious disadvantage of this approach - the inefficient management and use of resources - can be overcome, if necessary, since the scheduling software can deal with scheduling data to a finer granularity.

B. Planning Organizations

Based primarily on concepts and definitions in use by the NASA Space Station Operations Task Force, the following mission planning organizations were defined:

- o Users - Principal Investigators (PI's)

- o Planning Center - An organization which integrates the requirements, planning, and operations of a particular science/engineering discipline or of a particular Space Station physical element (e.g., US Lab)
- o Payload Operations Integration Center (POIC) - The organization responsible for integrating payload operations plans from all the planning centers and for providing the primary interface between the user community and the Space Station Systems Operations organization.
- o Space Station Systems Operations - the organization responsible for overall management and integration of Space Station operations
- o Investigator Working Group (IWG) - An organization of users headed by a mission scientist to encourage cooperative science operations and to resolve conflicts among users.

The IWG is an organization successfully employed in the Spacelab process. IWG's have therefore been recommended to be established for Space Station at each planning center and at the POIC planning levels.

Whether planning centers will be organized around science/engineering disciplines (discipline centers) or around Space Station physical elements (element centers), is a matter to be decided by NASA and its international partners. The matter has been discussed at length by the NASA Space Station Operations Task Force. The two approaches are depicted graphically in Figures 6.2.2-1 and 6.2.2-2. A third hybrid approach which employs both discipline centers and element centers in-line in the planning process is depicted in Figure 6.2.2-3, but has been discarded because of the complex network of planning interfaces.

The overall advantage to the discipline center approach is the enhancement of cooperative science, while the advantage to the element center approach is that it directly supports the analytical integration process for SS elements and allows resource allocation/utilization planning to be controlled/verified for compatibility with element design/operational capabilities as planning is integrated. Under the discipline center approach, control/verification of resource allocation/utilization planning versus SS element capabilities must be centralized at the POIC.

For this Study, the mission planning concept has been derived to accommodate either discipline centers or element centers.

C. On-Board Crew Mission Planning

Based on MSFC guidance for this Study, the provision of planning capabilities to the on-board crew has been limited to a minor real-time replanning capability. Space Station Phase B studies have shown the crew to be the most critical Space Station resource. Also, astronaut corps inputs to

the NASA Space Station Operations Task Force have indicated a preference for no mission planning responsibility. Therefore, crew utilization should be restricted to activities which must be performed on-board in order to maximize crew availability for experimentation. Furthermore, on-board planning will place a significant demand on on-board resources (e.g., mass storage). For these reasons, providing the on-board crew any mission planning capability is subject to reconsideration, in favor of increasing the automation of mission planning activities on the ground to minimize manpower requirements.

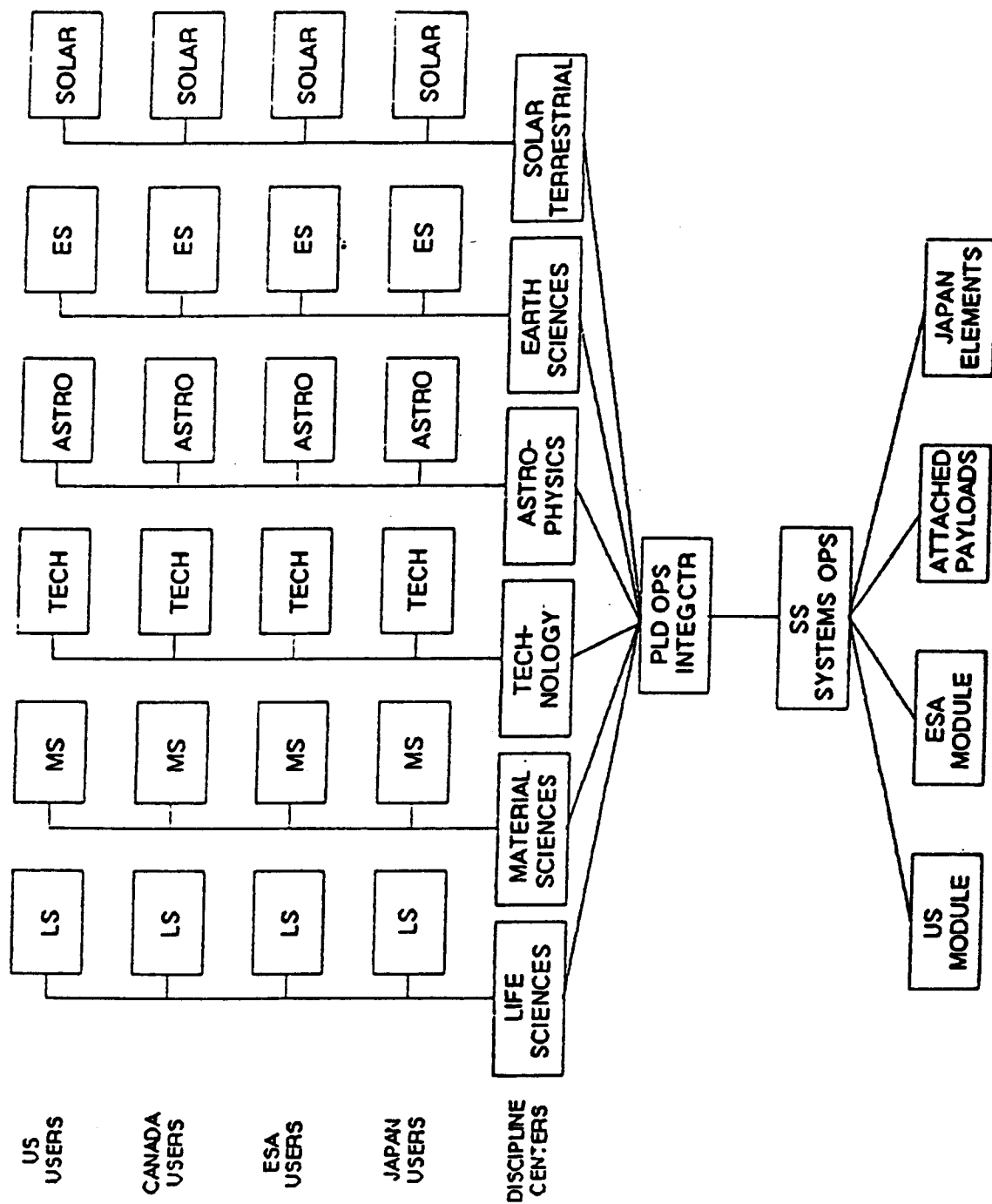


FIGURE 6.2.2-1. DISCIPLINE CENTERS INTEGRATION

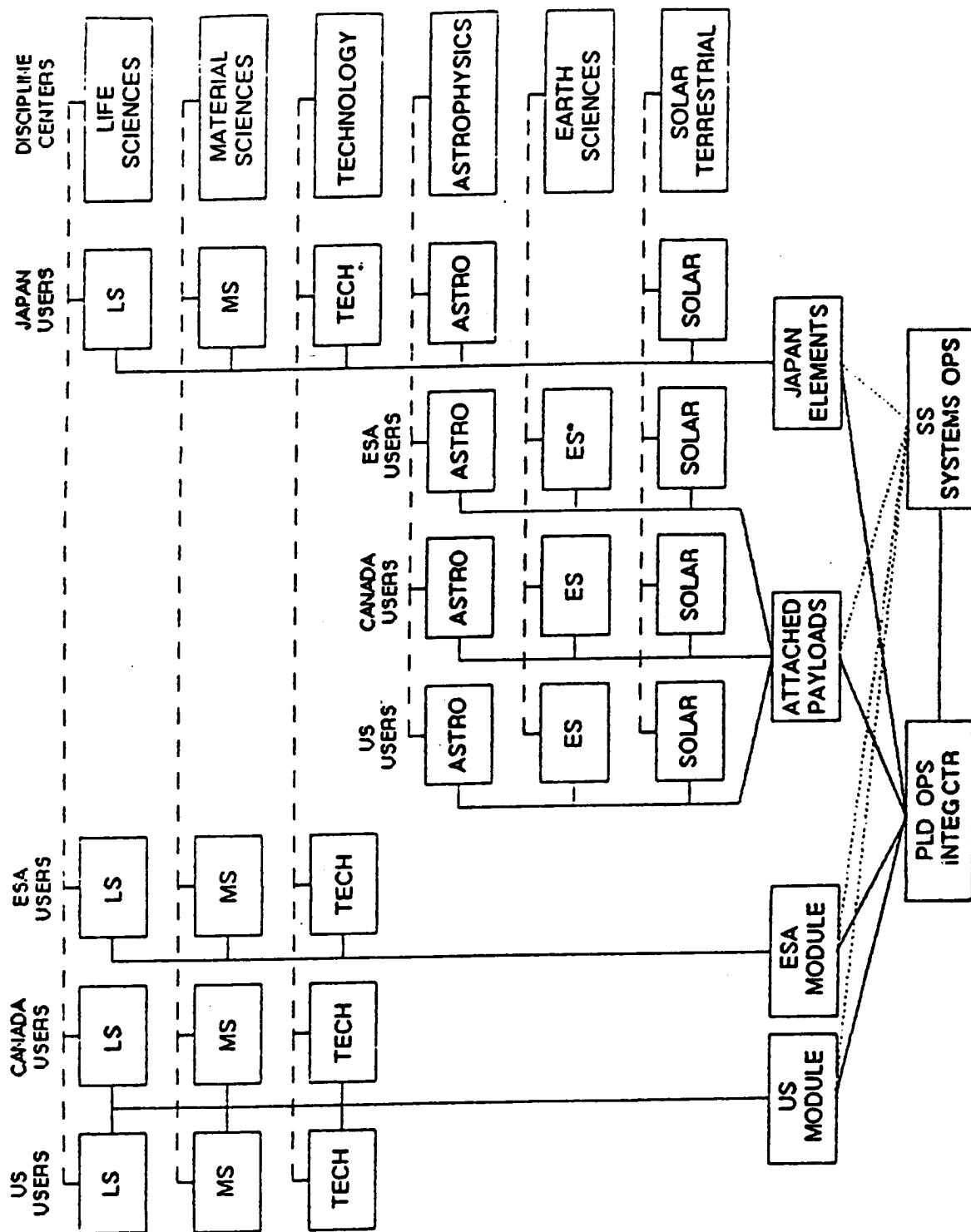


FIGURE 6.2.2-2. ELEMENT CENTERS PRIME INTEGRATION/DISCIPLINE CENTERS INDIRECT

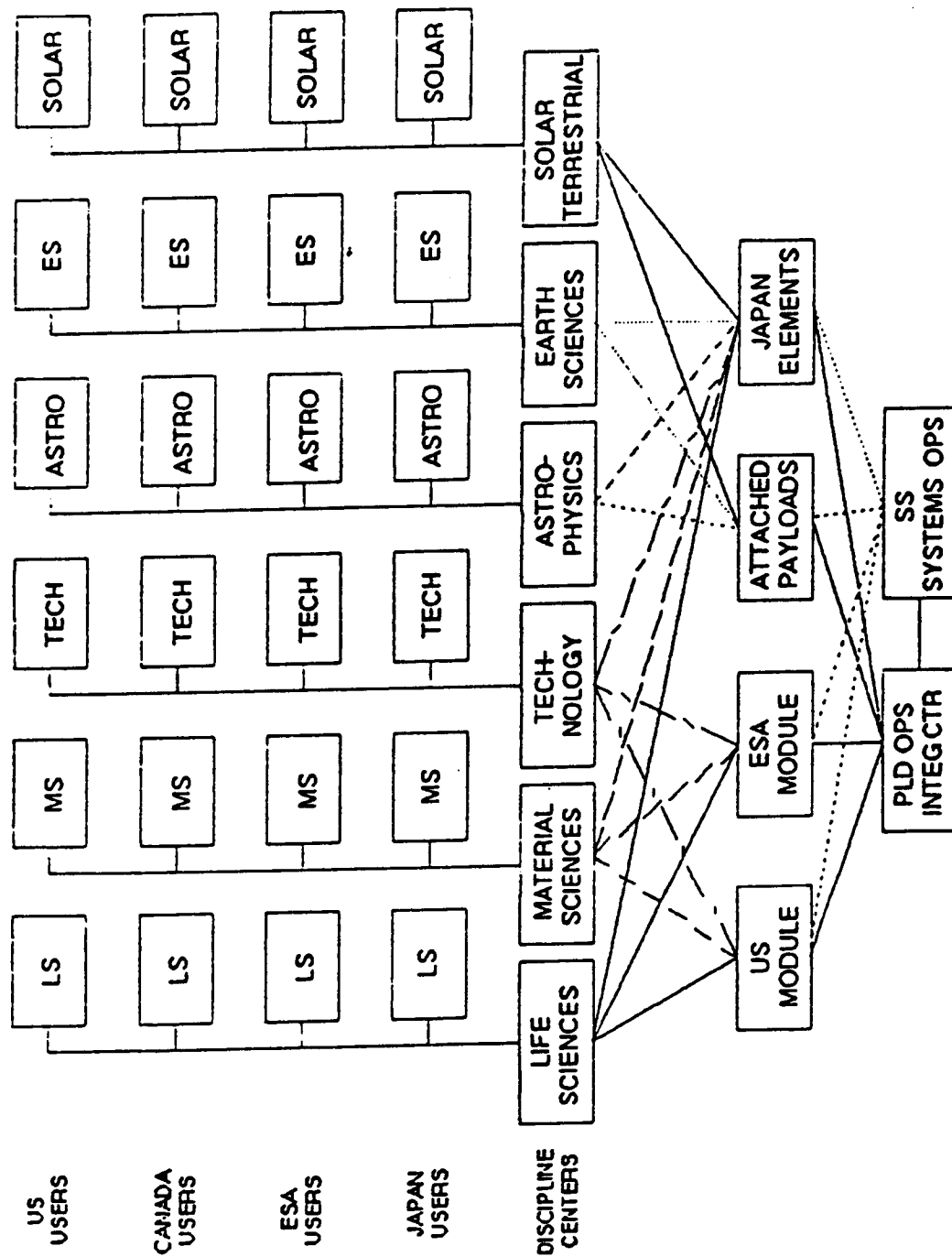


FIGURE 6.2.2-3. DISCIPLINE CENTERS PLUS ELEMENT CENTERS INTEGRATION

6.2.3 SS Payload Mission Planning Process Functional Description

Four distinct high level payload mission planning cycles have been identified as depicted in the Space Station (SS) Payload Mission Planning System (MPS) Top Level Functional Flow, Figure 6.2.3-1.

Cycle A, Define Resource Allocation Envelopes, is necessary for preliminary definition of user resource requirements and integration/approval of these requirements to arrive at agreed upon resource allocations for each experiment entity (single experiment or group of experiments) as well as resource allocations for each planning center.

Cycle B, Generate Tactical Operations Plan (TOP), results in a resource allocation plan for the mission increment that assigns time blocks (resource allocation envelopes) available for each experiment entity within which to schedule detailed operations.

Cycle C, Generate Execution Plans, includes the user activity involved in generating detailed activity and command plans for resource allocation envelopes, plus the activities to integrated those plans. This cycle also results in an integrated payload data flow plan.

Cycle D, Perform Mission Increment Replanning, encompasses similar activities to Functions B and C. Resources are reallocated; users change detailed activity/command plans; the crew replans activities over which they have control; and all changes are integrated. The data flow plan is finalized and the detailed payload crew activity plans are generated.

A flow of subfunctions for each planning cycle is presented in the flow diagrams designated A, B, C and D. Each of the unique subfunctions appearing on these charts is identified by the number in the upper right hand corner of the flow diagram block. A corresponding flow diagram at the subfunction level is included with the detailed description of each subfunction presented in the following sections.

6.2.3.1 Subfunction 1 - SS Projected Orbit Ephemeris

This subfunction will likely be performed by the Space Station Systems Operations organization. It is included here because the software required to perform this function could be easily derived from the SL MIPS. The basic activity is to generate detailed ephemeris data, such as ascending node data, ground track data and earth shadow on/off times to serve as a basis for subsequent mission planning activities.

6.2.3.2 Subfunction 2 - Standard Orbit Opportunities Generation

This activity may also be performed by the SS Systems Operations organization, but is included here because again SL MIPS software could be utilized with modifications to perform the activity. The basic activity is the generation of standard orbit observation opportunities. An observation opportunity (obs opp) is a particular object or condition that is available as a function of time (on or off). The designation "standard" is made because the obs opps generated during this activity are those that are used by a wide variety of mission planning organizations and scientific disciplines. Grouping these into one subfunction performed by the same organization insures use of common data across all Space Station users and planning centers.

SPACE STATION (SS) PAYLOAD MISSION PLANNING SYSTEM (MPS) TOP LEVEL FUNCTIONAL FLOW

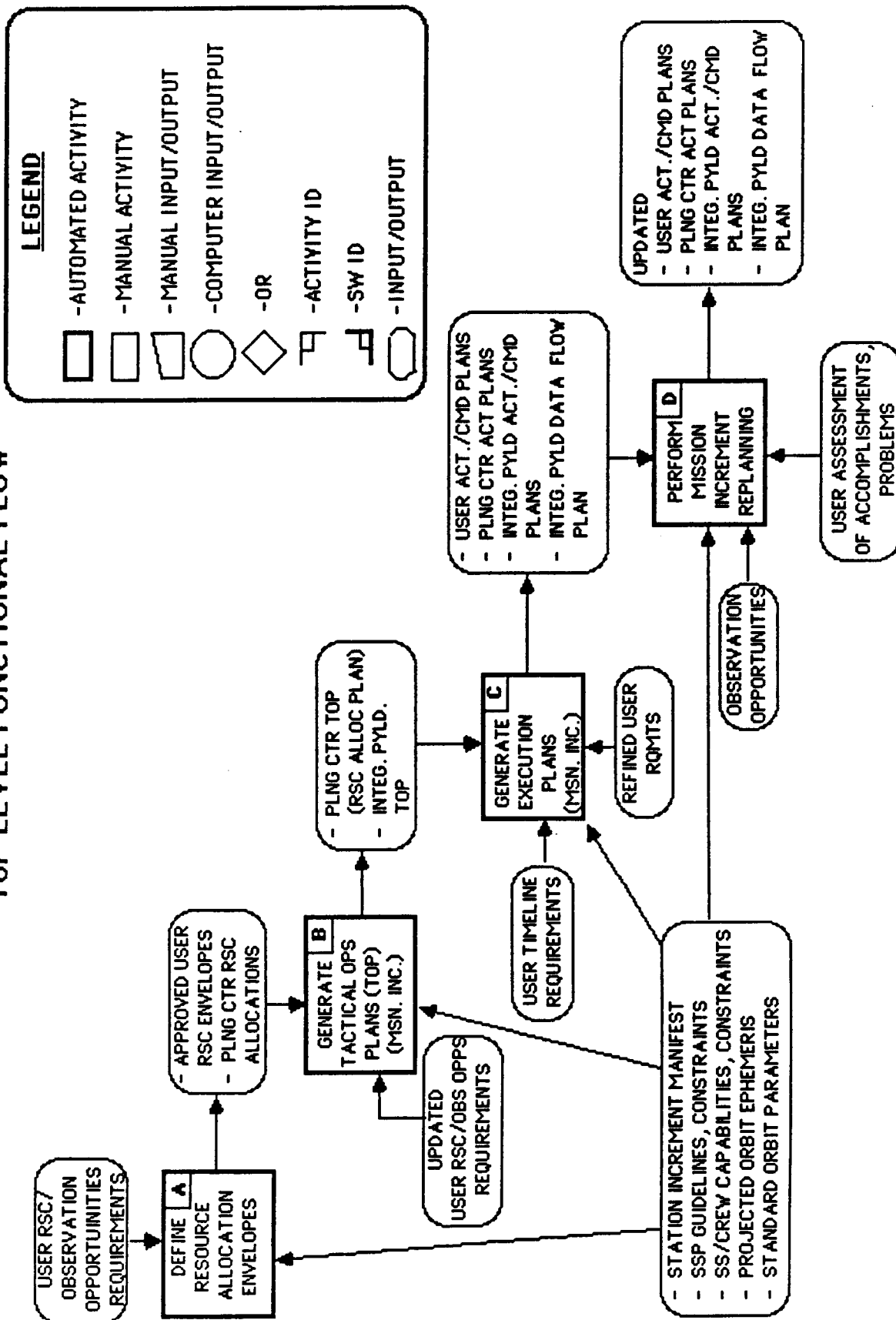
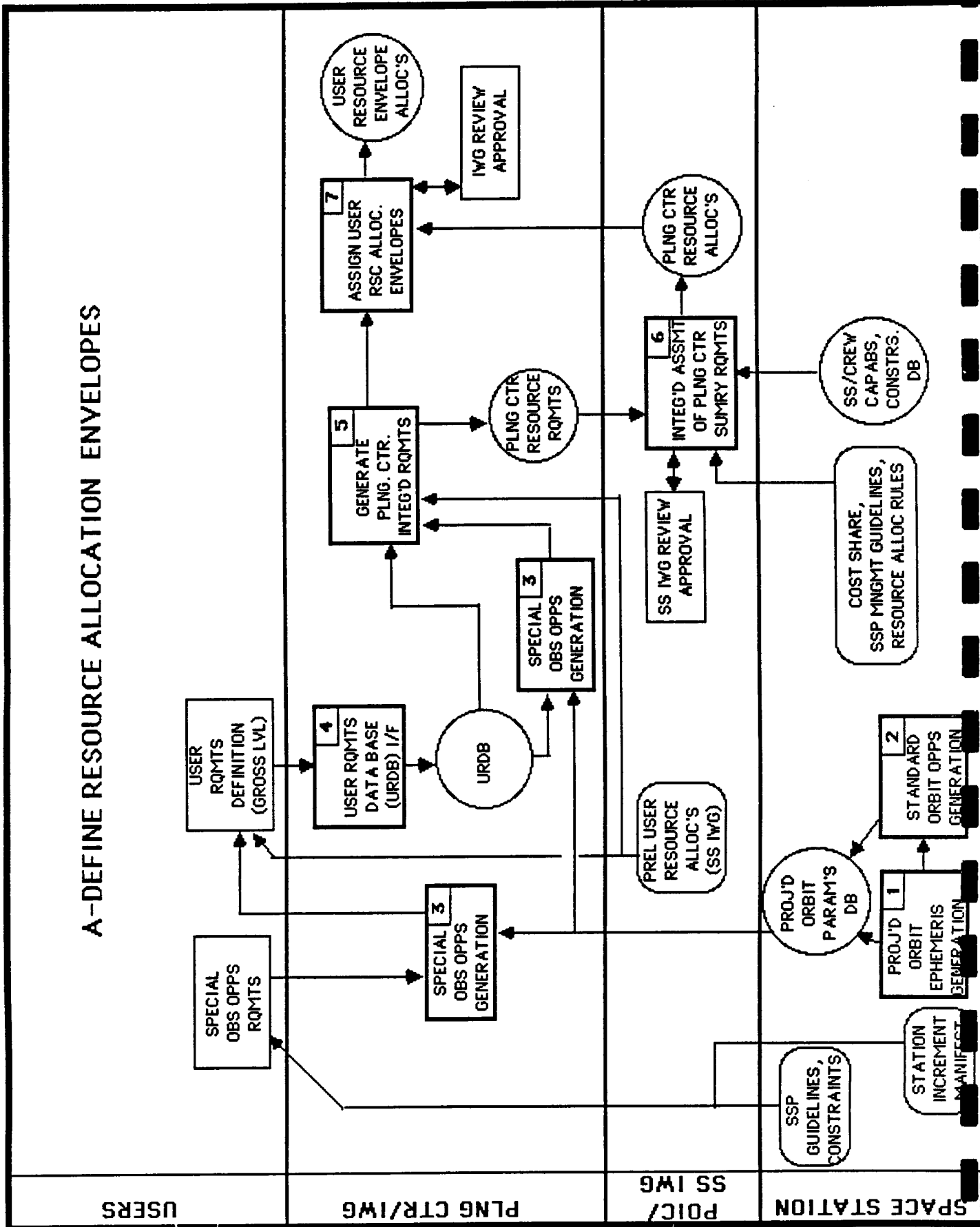
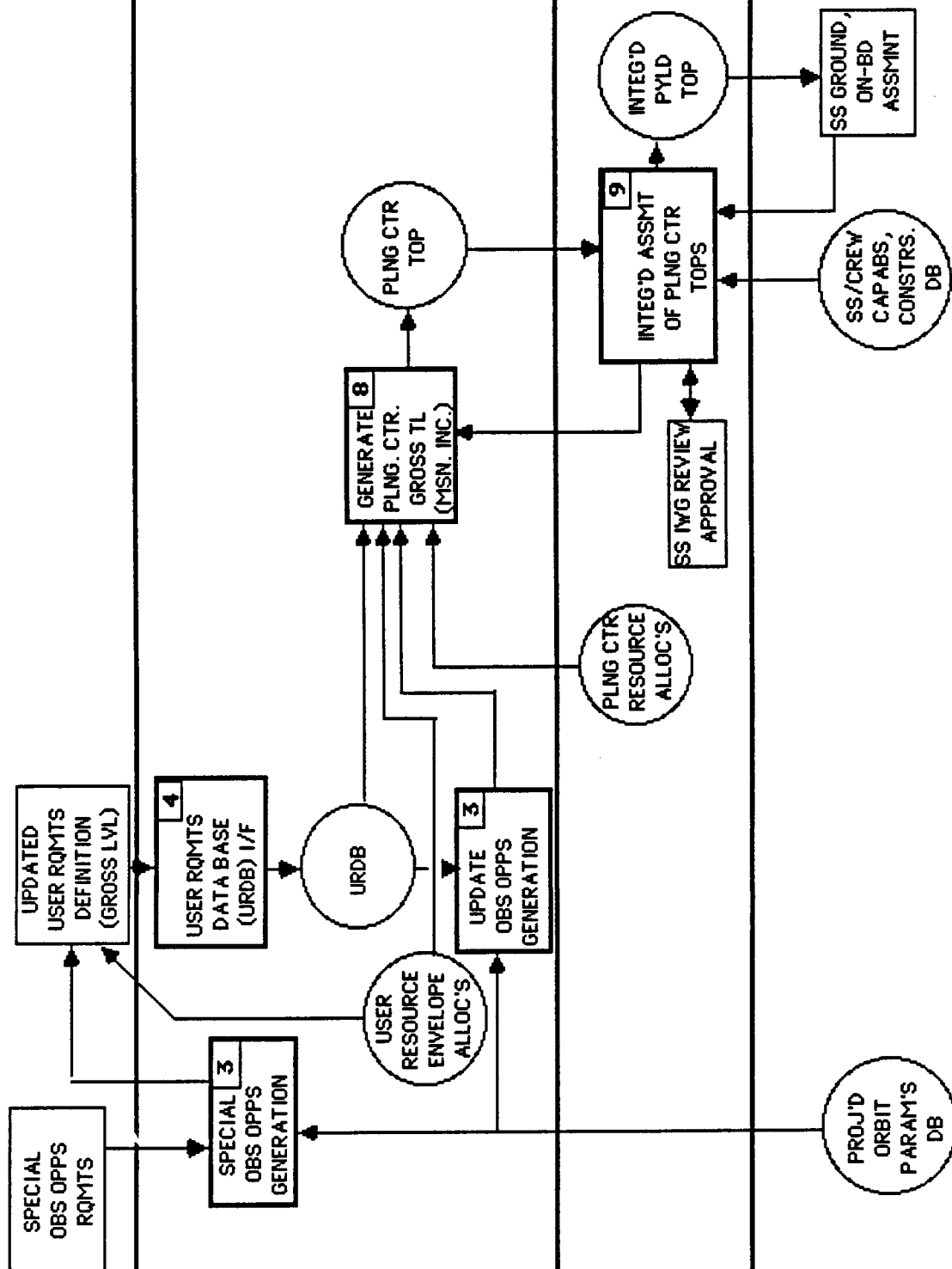


FIGURE 6.2.3-1

A-DEFINE RESOURCE ALLOCATION ENVELOPES

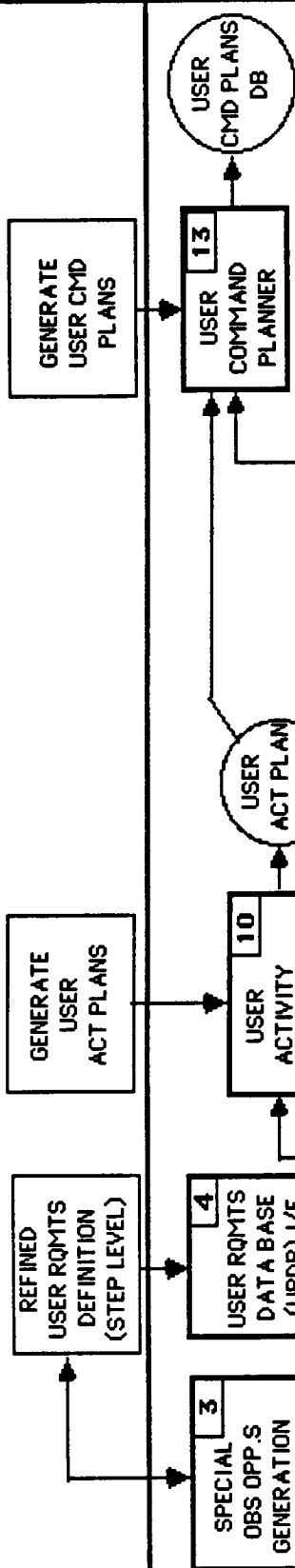


B-GENERATE TACTICAL OPERATIONS PLANS (TOP) (MISSION INCREMENT BASIS)

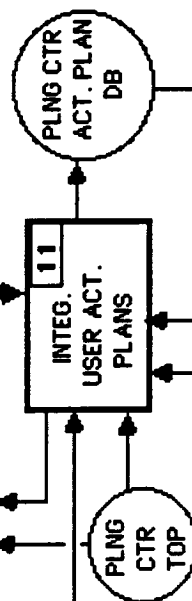


C-GENERATE EXECUTION PLANS (MISSION INCREMENT)

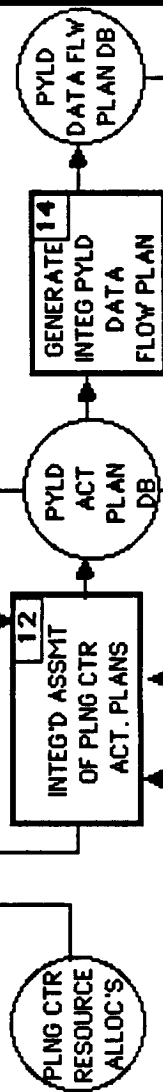
USERS



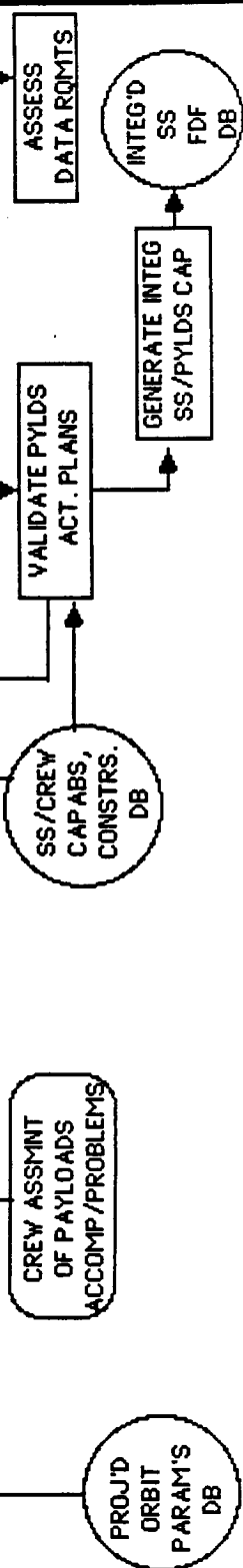
PLNG CTR/IWG

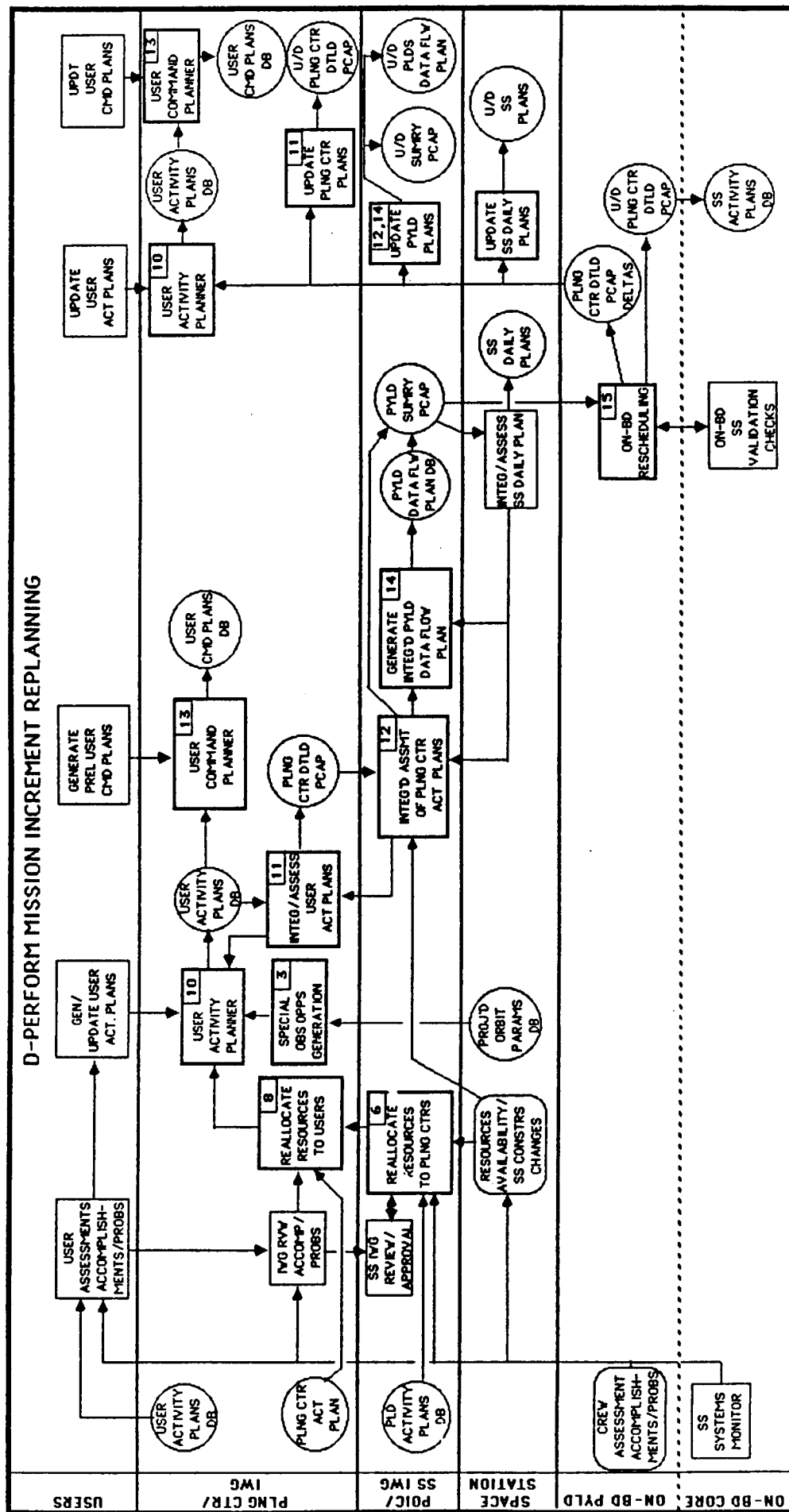


POIC/
SS IWG

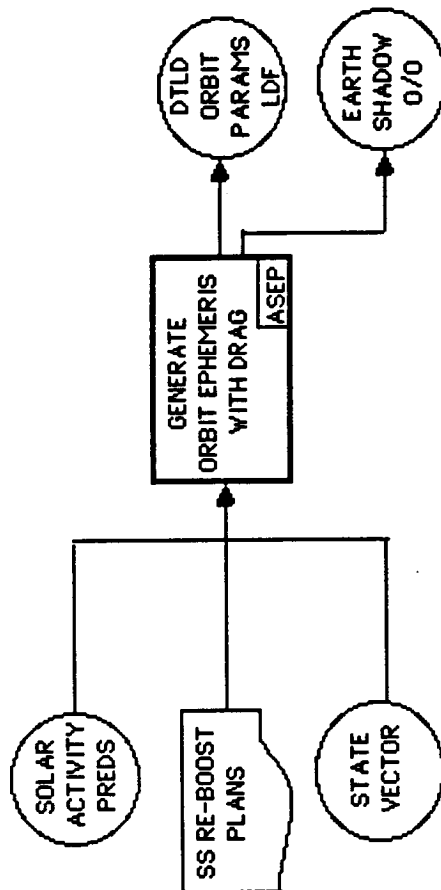


SPACE STATION

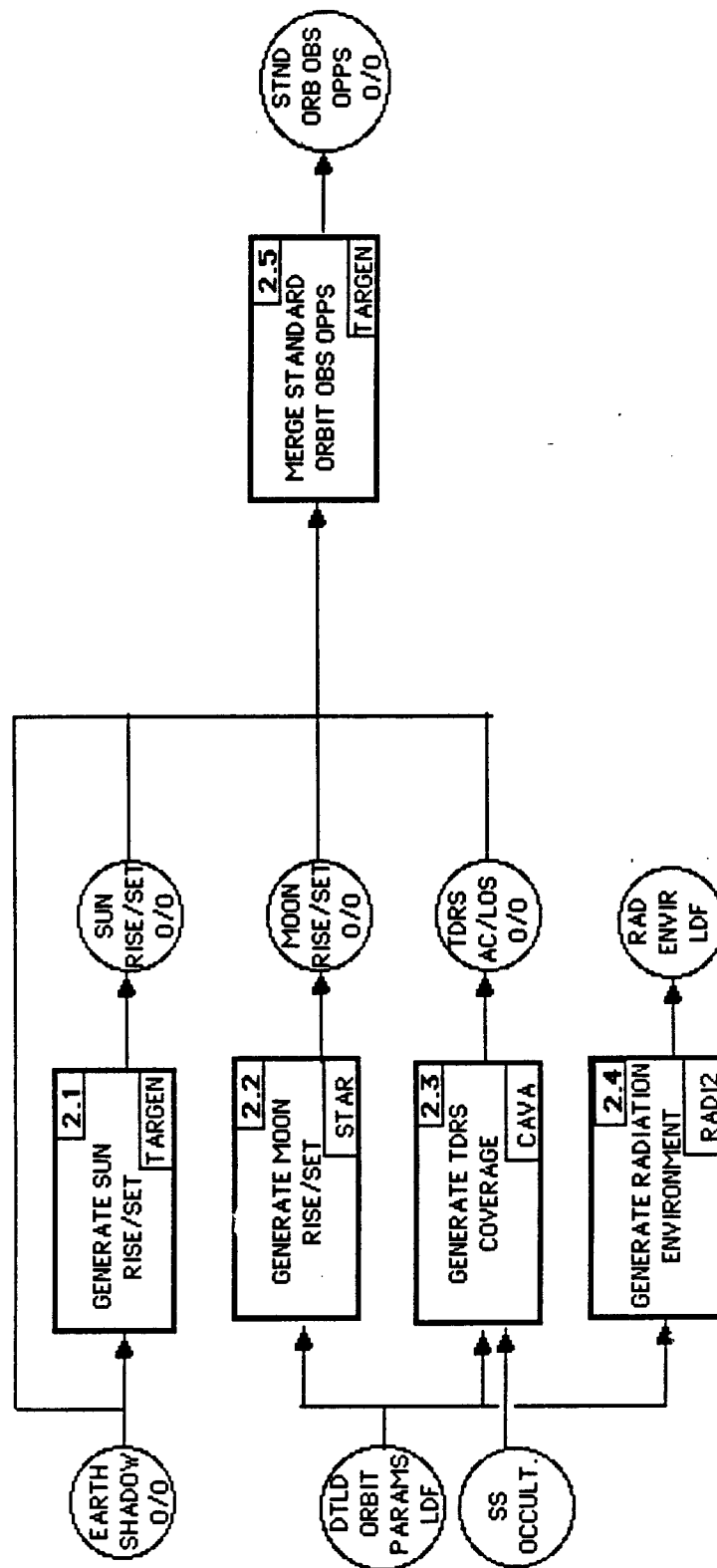




SUBFUNCTION: 1-SS PROJECTED ORBIT EPHEMERIS GENERATION



SUBFUNCTION: 2-STANDARD ORBIT OPPORTUNITIES GENERATION



6.2.3.3 Subfunction 3 - Special Observation Opportunities Generation

This activity involves generation of the observation opportunities that are discipline dependent. It is expanded into task level flow diagrams corresponding to the identified disciplines. Each of these tasks includes the detailed activities required to generate the specific obs opps for each discipline. In practice, this activity will be performed by both SS users and the planning centers. The users will perform analyses to determine when they want to operate and which obs opps they want to utilize. Planning center personnel will access all users obs opps requirements entered in the User Requirements Data Base (URDB) and generate an integrated planning center set of obs opps to serve as input for subsequent mission planning tasks.

6.2.3.4 Subfunction 4 - User Requirements Definition and Data Base (URDB) Interface

This activity includes user interactive input/editing of a data base that contains resources, obs opps, sequencing/concurrency and number of performances/duration requirements as well as operational constraints for each individual experiment entity. An entity can be a single experiment or a group of experiments. Depending on experiment resource profiles, a particular experiment will have from one to many resource envelopes. User friendliness and scheduling complexity as well as resource utilization efficiency will be significant factors in determining the characteristics of the resource profiles.

6.2.3.5 Subfunction 5 - Generate Planning Center Integrated Requirements

User requirements are summarized in a gross scheduling activity based on the URDB entries and the observation opportunities file. No resource checking is performed during scheduling. The output schedule(s) are used to determine the overall planning center resource requirements.

6.2.3.6 Subfunction 6 - Integrated Assessment Of Planning Center Summary Requirements

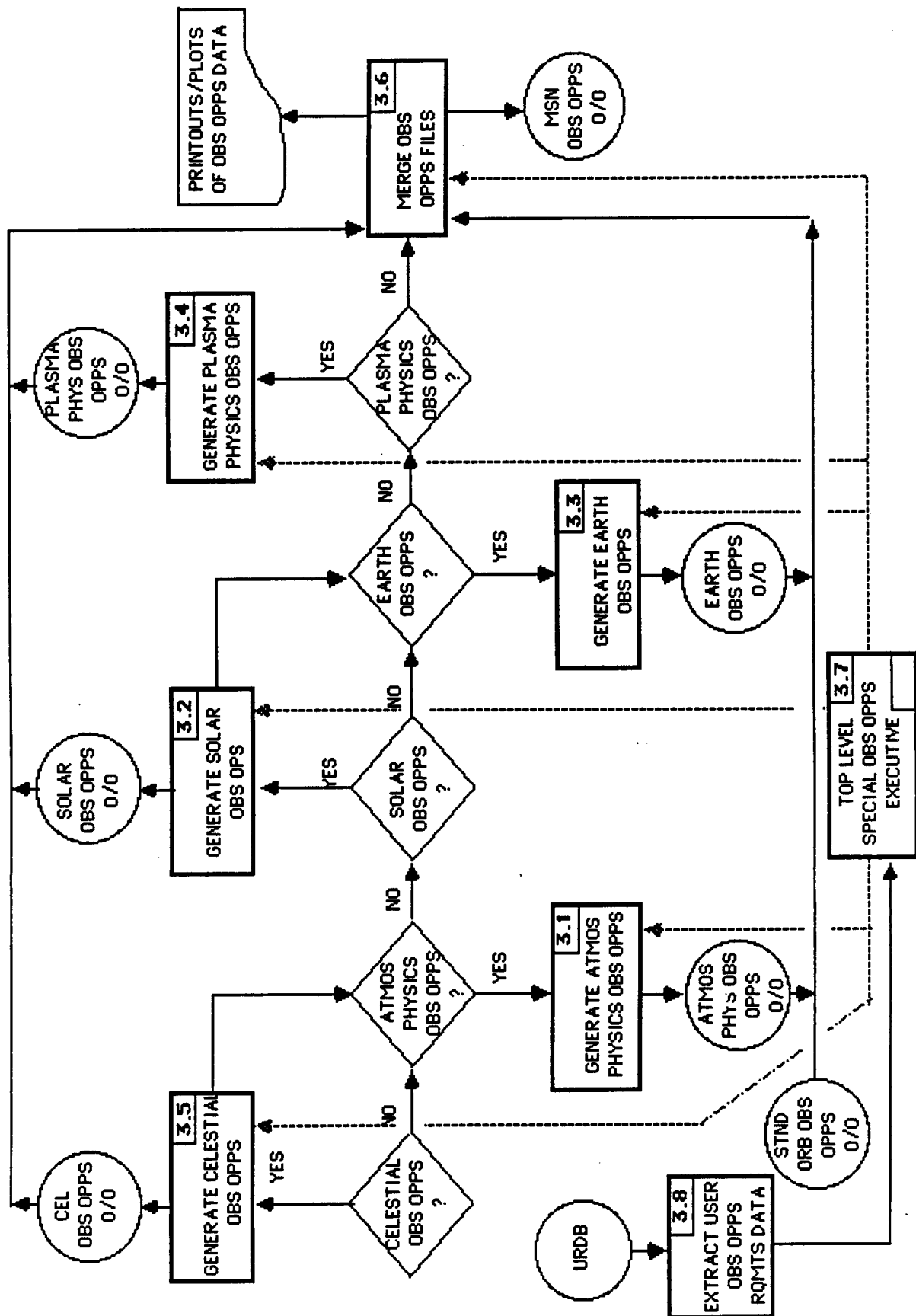
Planning center resource requirements, SSP management guidelines, resource allocation rules, user group guidance and element design constraints are utilized as inputs to assign each planning center a specific set of resource allocations.

6.2.3.7 Subfunction 7 - Assign User Resource Allocation Envelopes

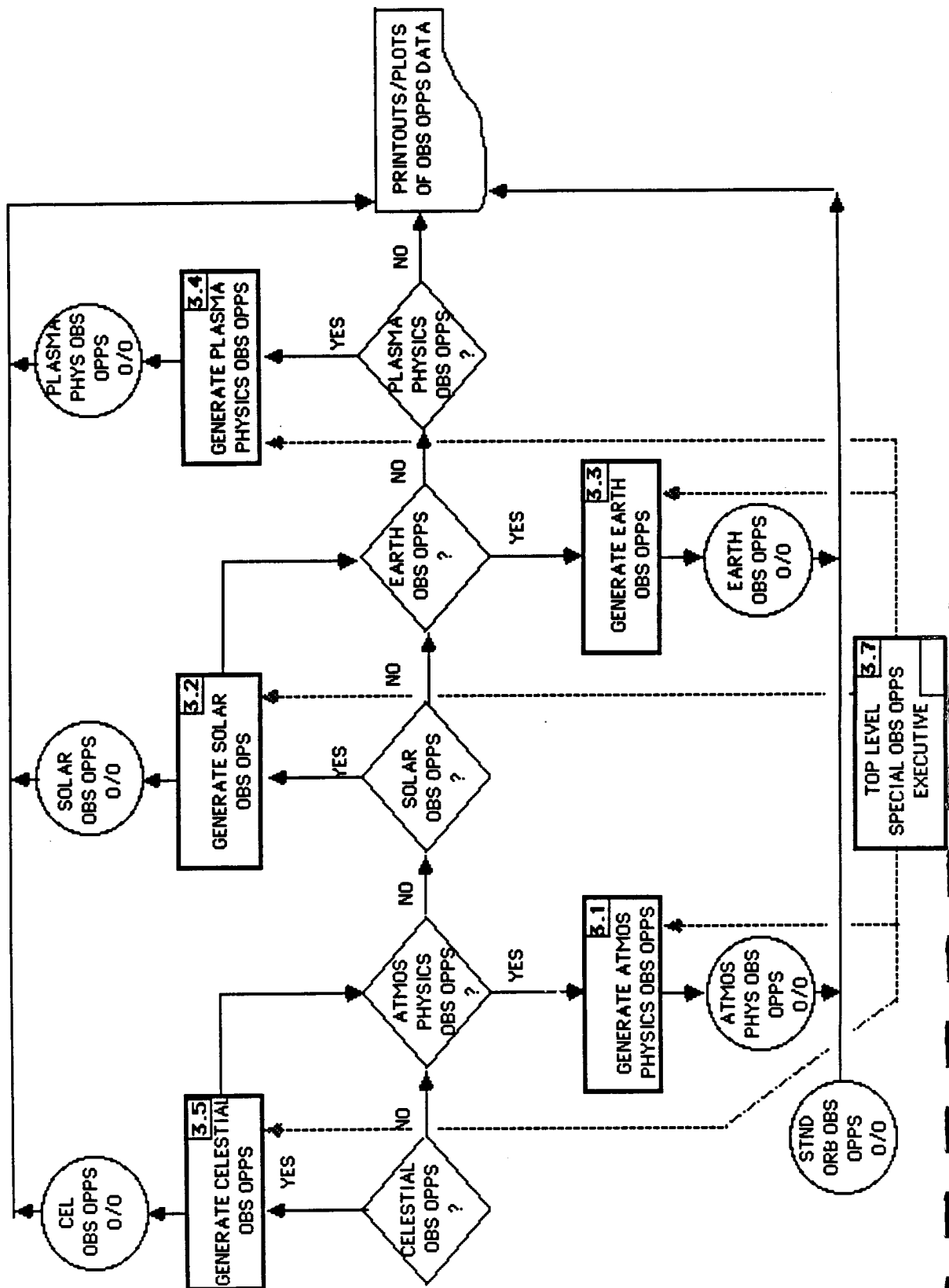
This activity is basically a formal approval process of each users URDB entries. There may be cases where user resource requirements are incompatible with those available to the planning center when considering other manifested users. These cases must be treated separately and will require reduction of user resource requests (redesign, reduced objectives, etc.) or an appeal through channels for increased planning center resource allocations. A thorough compatibility analysis of preliminary user requirements in the strategic level planning/manifesting process should reduce the potential for conflicts at this point.

SUBFUNCTION: 3(PLNG CTR)-SPECIAL OBSERVATION OPPORTUNITIES GENERATION

NOTE: THE MISSION OBSERVATION OPPORTUNITIES GENERATED BY THIS SUBFUNCTION ARE BASED ON PROJECTED SS ORBITAL POSITION AND/OR ENVIRONMENTAL CONDITIONS. USER RQMTS FOR OBSERVATIONS THAT ARE STRICTLY TIME PREFERENCES ARE ACCOMMODATED IN SUBFUNCTION 5,8 TASKS 5.1,8.1.

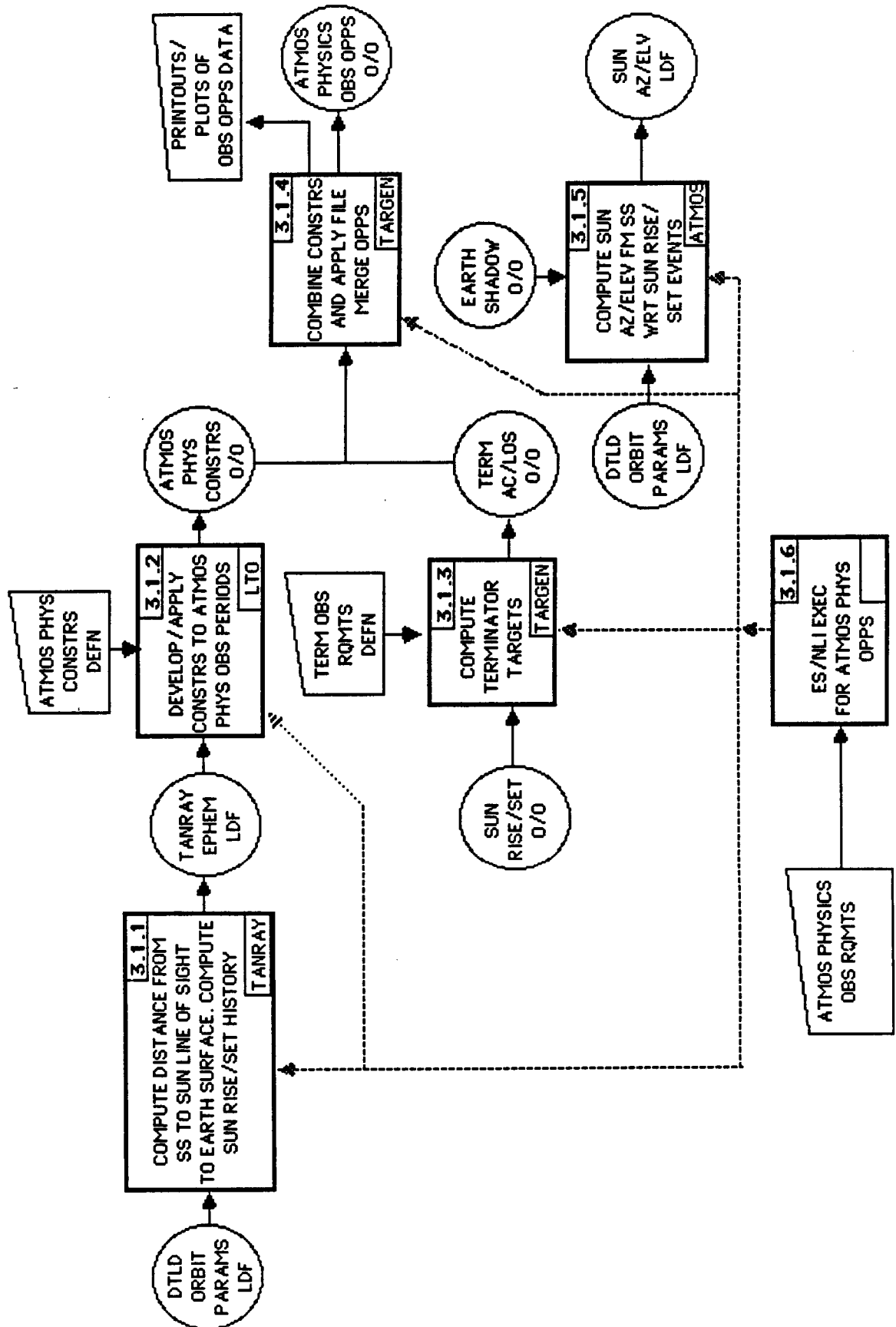


SUBFUNCTION: 3(USER)-SPECIAL OBSERVATION OPPORTUNITIES GENERATION



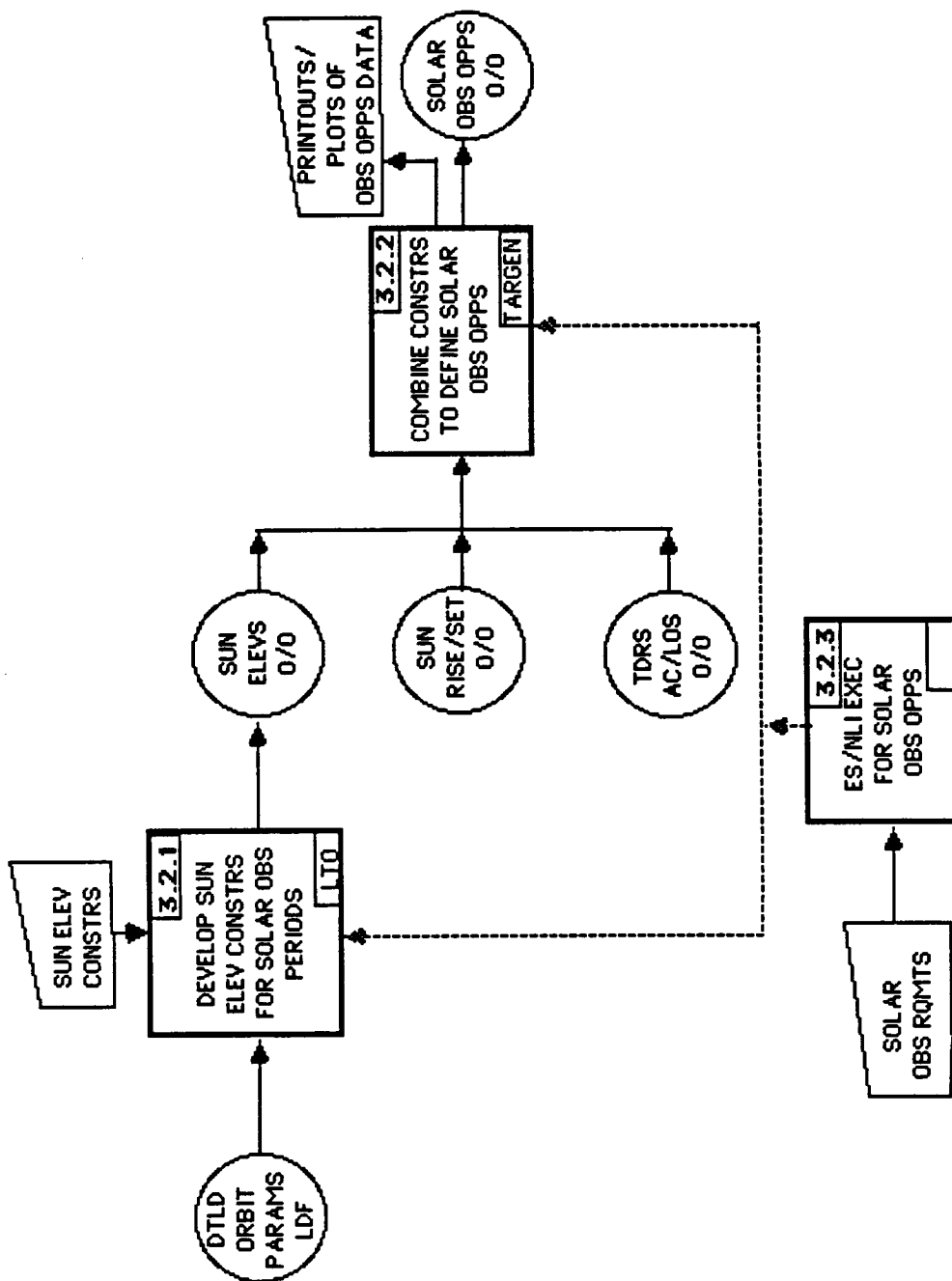
SUBFUNCTION: 3-SPECIAL OBSERVATIONS OPPORTUNITIES GENERATION

TASK: 3.1-GENERATE ATMOSPHERIC PHYSICS OBSERVATION OPPORTUNITIES



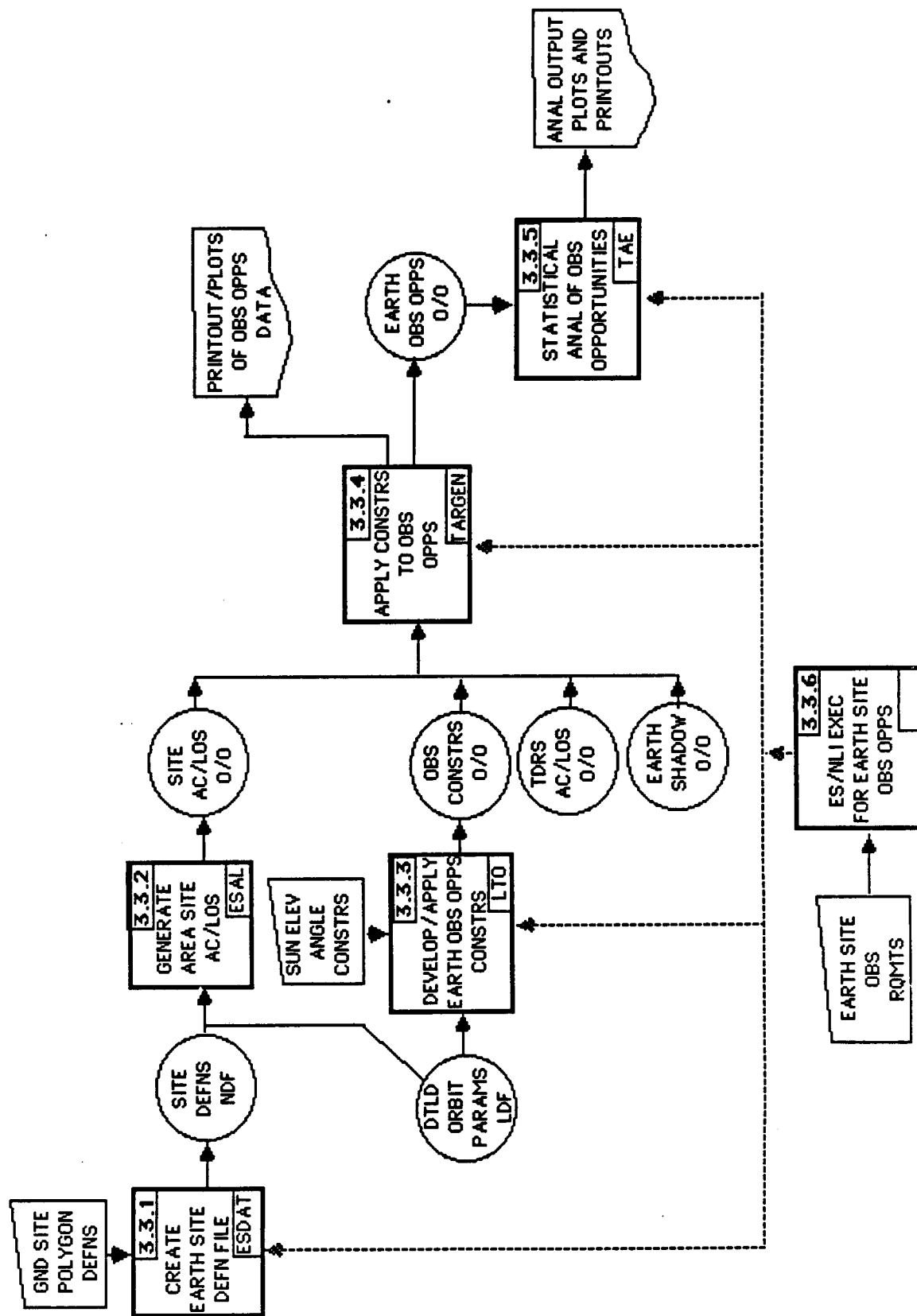
SUBFUNCTION: SPECIAL OBSERVATION OPPORTUNITIES GENERATION

TASK: 3.2-GENERATE SOLAR OBSERVATION OPPORTUNITIES



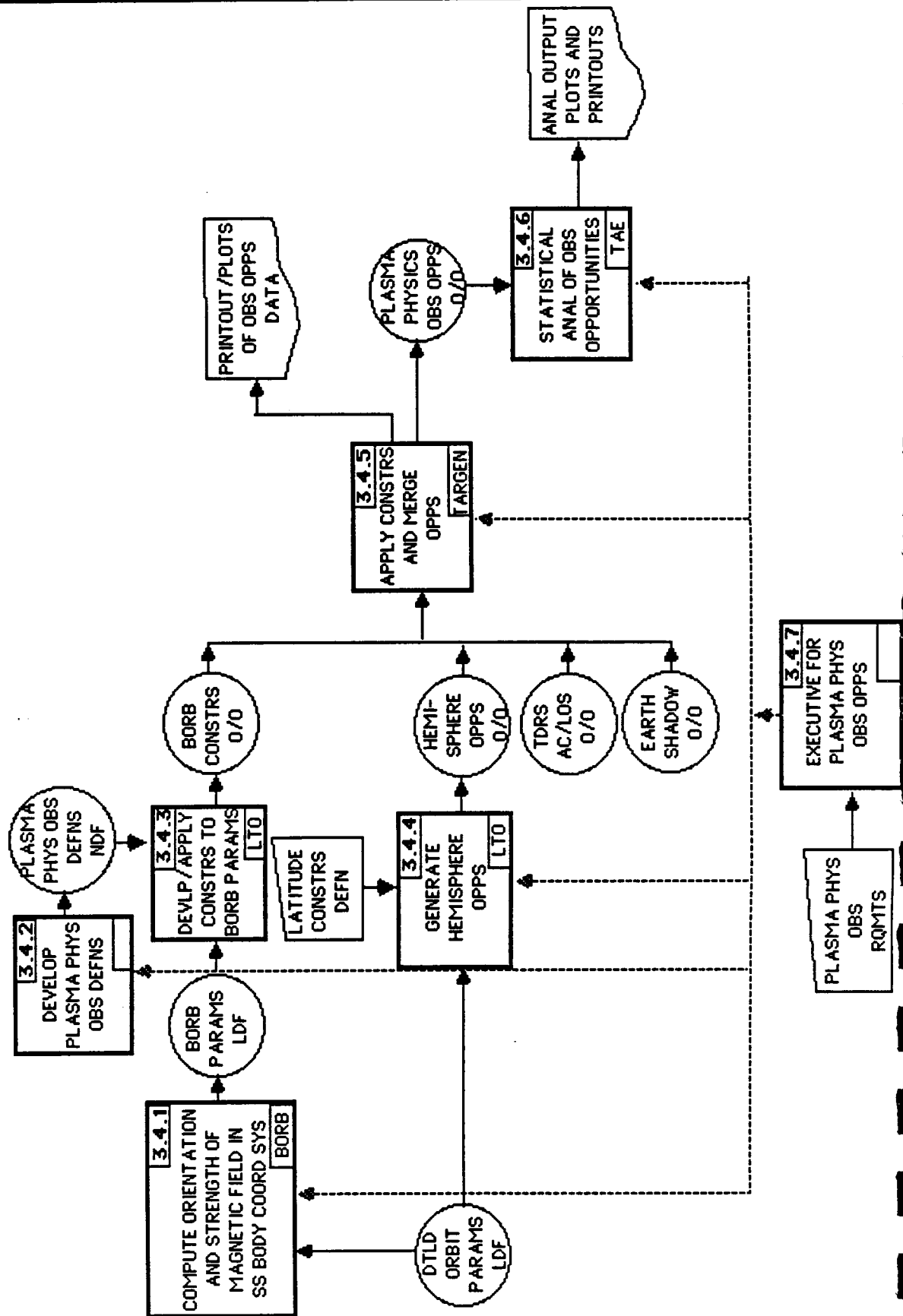
SUBFUNCTION: 3-SPECIAL OBSERVATION OPPORTUNITIES GENERATION

TASK: 3.3-GENERATE EARTH SITE OBSERVATION OPPORTUNITIES



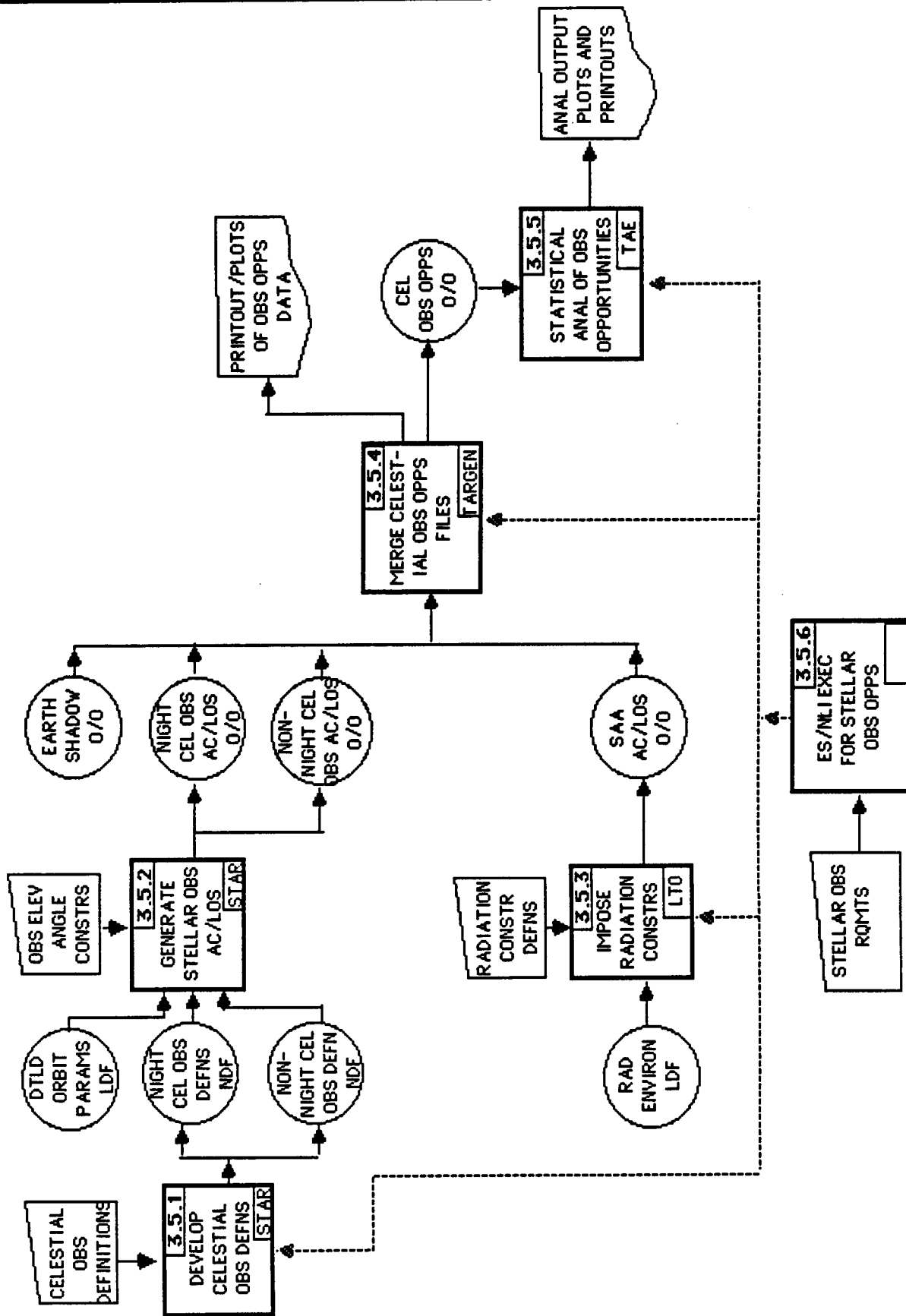
SUBFUNCTION: 3-SPECIAL OBSERVATION OPPORTUNITIES GENERATION

TASK: 3.4-GENERATE PLASMA PHYSICS OBSERVATION OPPORTUNITIES



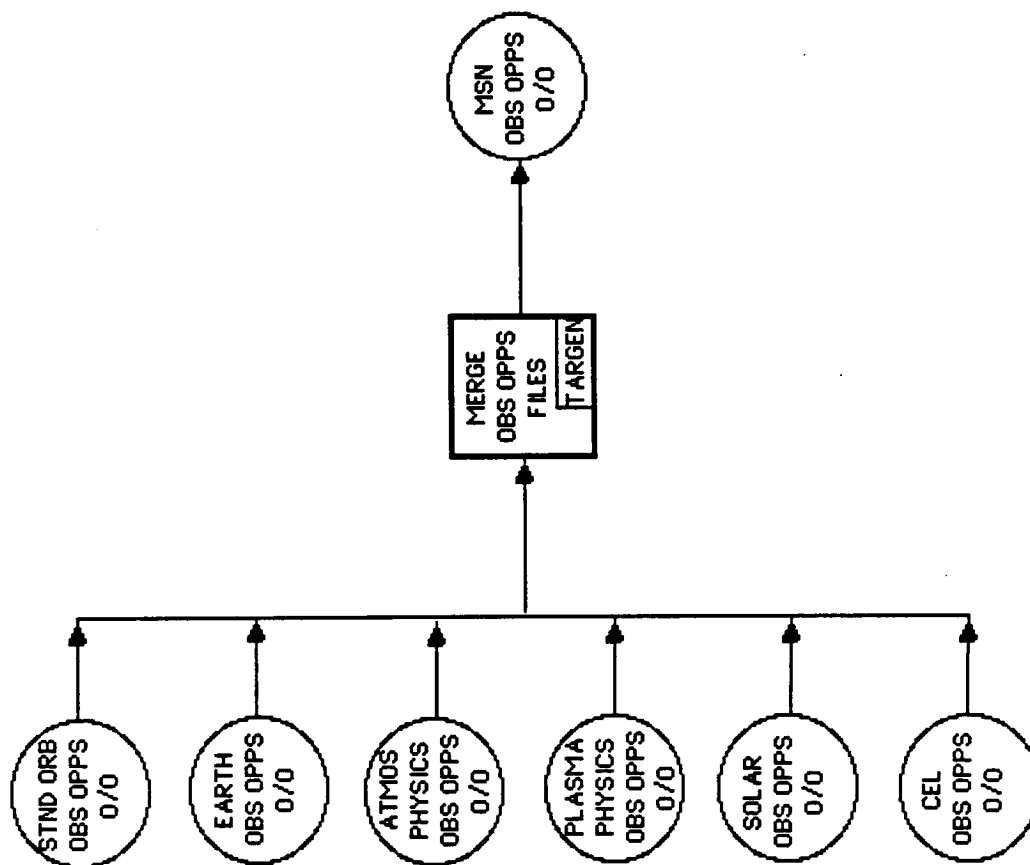
SUBFUNCTION: 3-SPECIAL OBSERVATION OPPORTUNITIES GENERATION

TASK: 3.5-GENERATE CELESTIAL OBSERVATION OPPORTUNITIES



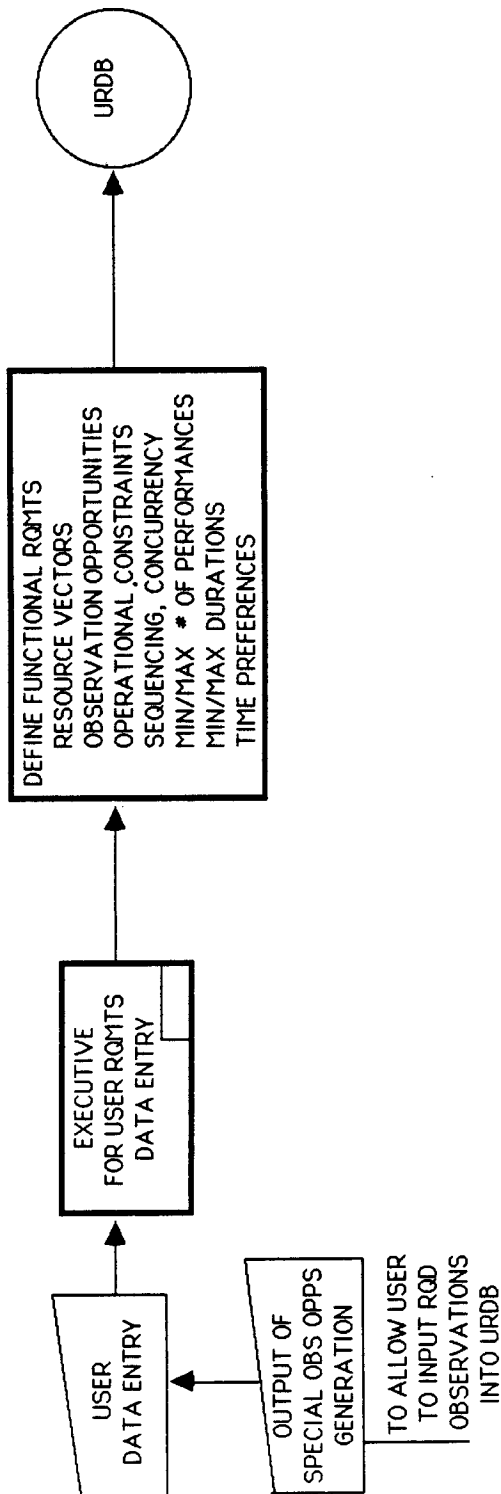
SUBFUNCTION: 3-SPECIAL OBSERVATION OPPORTUNITIES GENERATION

TASK: 3.6-MERGE OBSERVATION OPPORTUNITIES

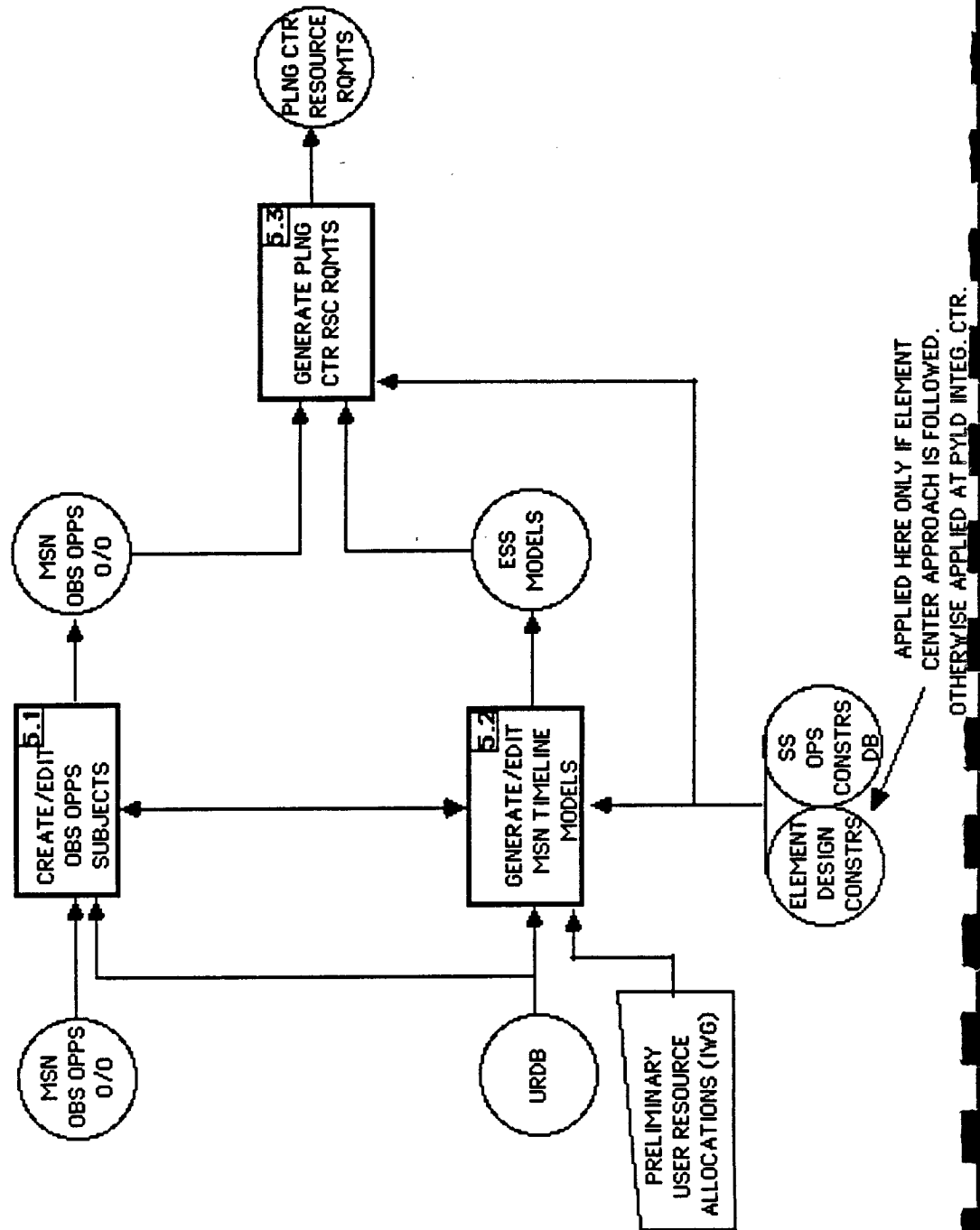


SUBFUNCTION: 4-USER REQUIREMENTS DATA BASE I/F

NOTE: THIS SUBFUNCTION DEPICTS ONLY THOSE USER RQMTS NECESSARY FOR MISSION ON-ORBIT OPERATIONS PLANNING AND SCHEDULING. THE SS OPERATIONS CONCEPT MAY IMPOSE RQMTS FOR A CENTRALIZED DATA BASE INCORPORATING ALL USER INTEGRATION RQMTS. THIS CONCEPT ASSUMES THAT THE URDB I/F SW LEADS THE USER IN BUILDING MSN TL MODELS USEABLE BY THE EXPERIMENT SCHEDULING PROGRAM. THE FIDELITY OF THESE MODELS WILL BE AT THE EXPERIMENT LEVEL UNTIL THE EXECUTION PLNG PHASE AT WHICH TIME THEY WILL BE REFINED TO THE STEP LEVEL.

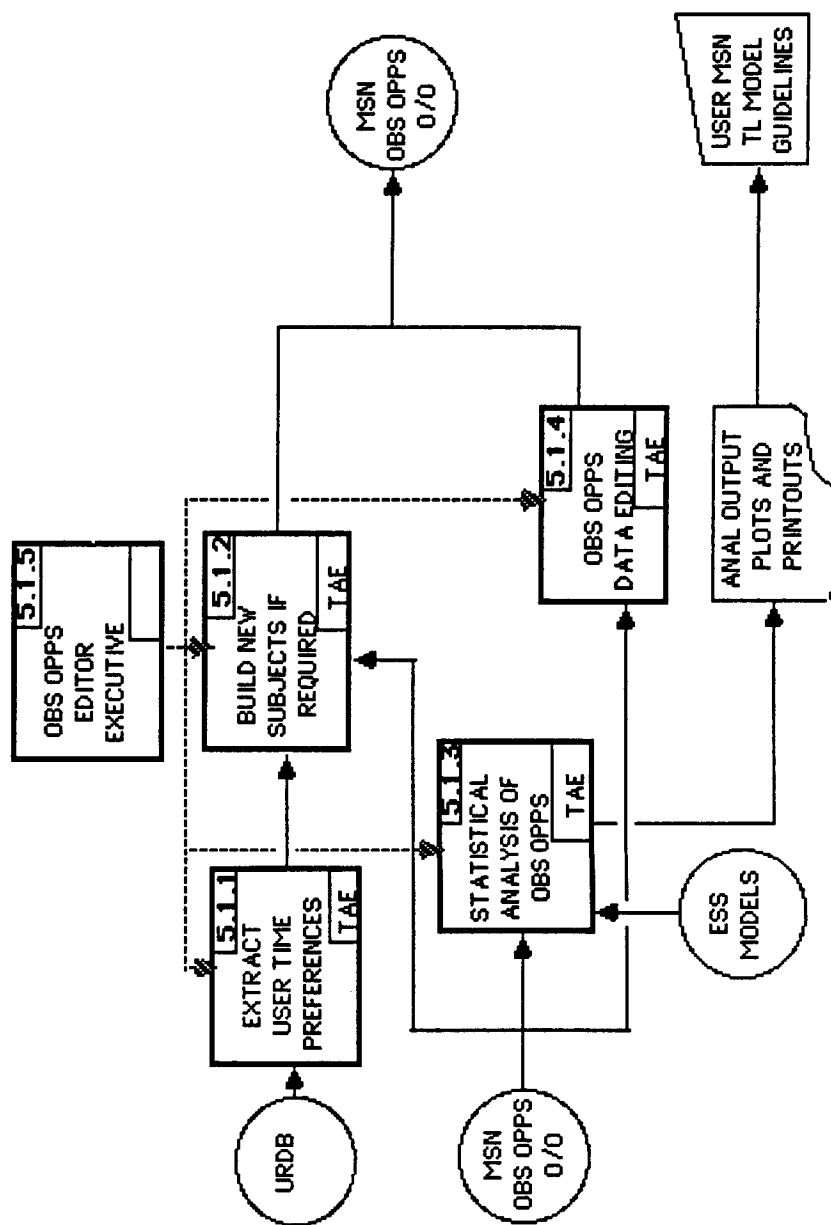


SUBFUNCTION: 5-GENERATE PLANNING CENTER INTEGRATED REQUIREMENTS



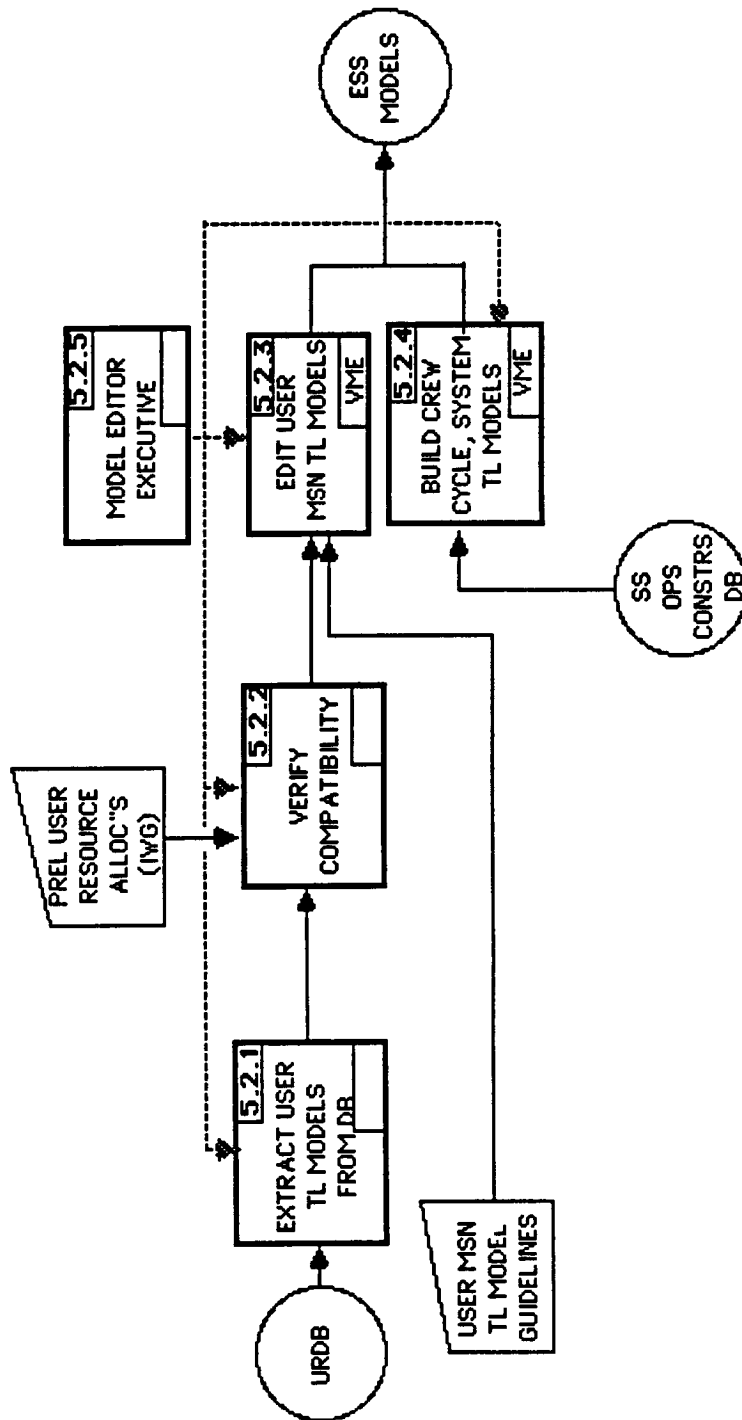
SUBFUNCTION: 5-GENERATE PLANNING CENTER INTEGRATED REQUIREMENTS

TASK: 5.1-CREATE/EDIT OBS OPPS SUBJECTS



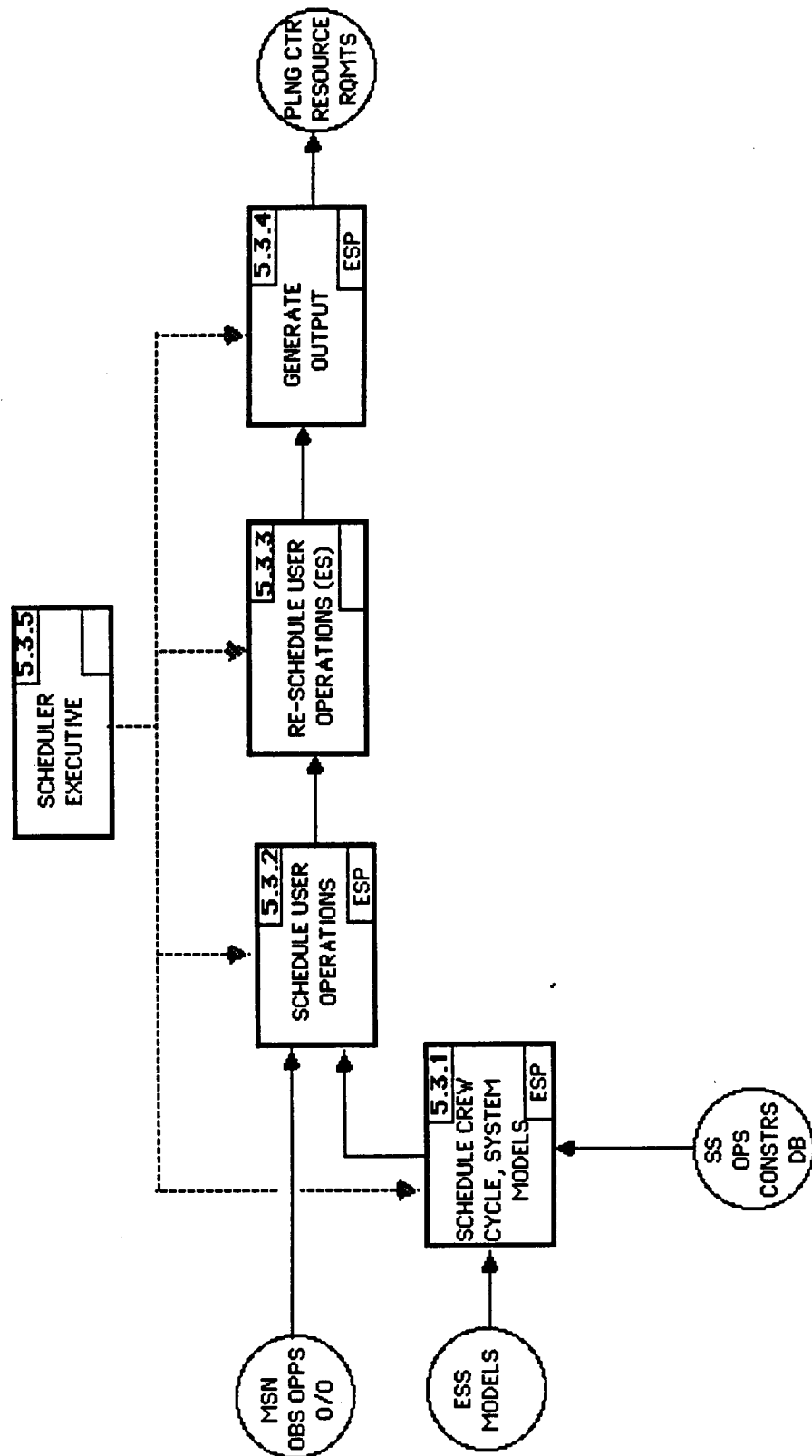
SUBFUNCTION: 5-GENERATE PLANNING CENTER INTEGRATED REQUIREMENTS

TASK: 5.2-GENERATE/EDIT MISSION TIMELINE MODELS

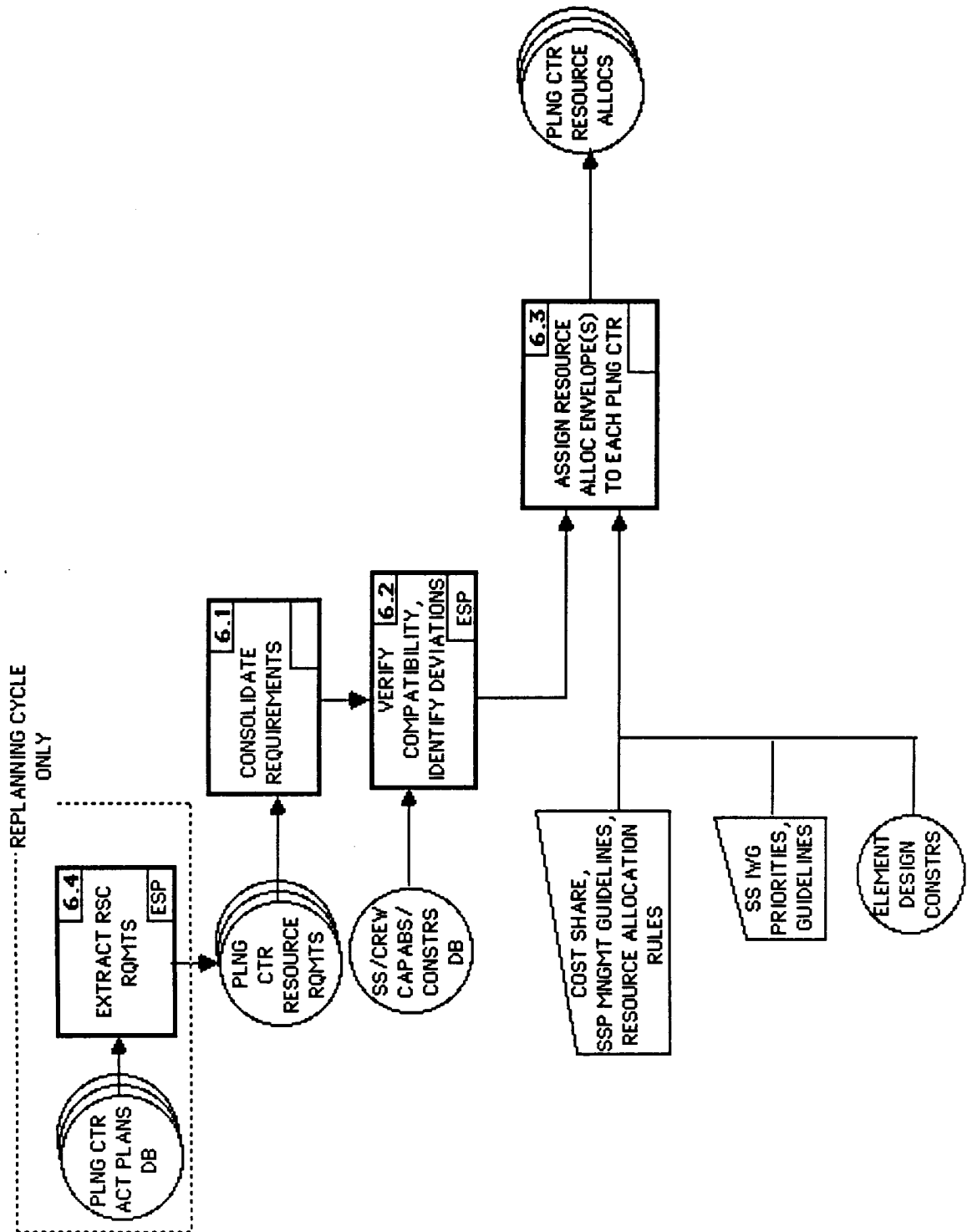


SUBFUNCTION: 5-GENERATE PLANNING CENTER INTEGRATED REQUIREMENTS

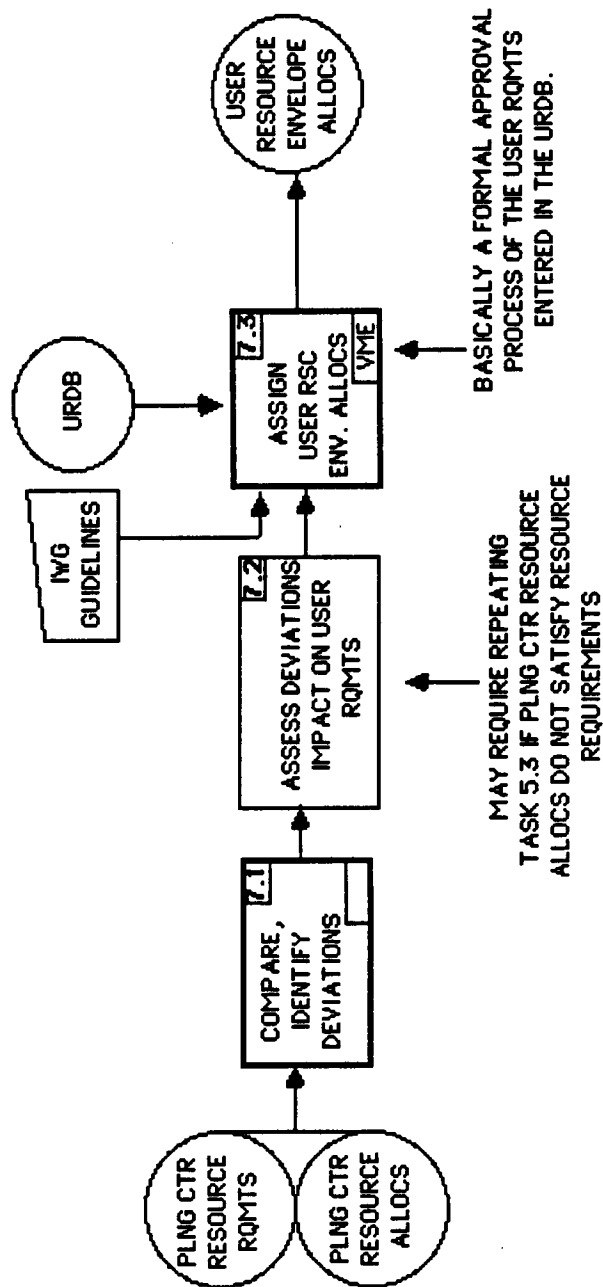
TASK: 5.3-GENERATE PLANNING CENTER RESOURCE REQUIREMENTS



SUBFUNCTION: 6-INTEGRATED ASSESSMENT OF PLANNING CENTER SUMMARY RQMTS



SUBFUNCTION: 7-ASSIGN USER RESOURCE ALLOCATION ENVELOPES



6.2.3.8 Subfunction 8 - Generate Planning Center Gross Timeline

This activity is basically a scheduling function which develops blocks of times within which each experiment may plan operations. This time block allocation approach allows the user flexibility in scheduling his individual operations while minimizing the number of iterations required to obtain a workable timeline. The tolerances included when allocating these time blocks will be a significant factor, essentially trading off user flexibility for schedule efficiency. The subfunction includes editing obs opps subjects; generating/editing mission timeline models; and generating the gross timeline with the appropriate output products to be included in the Tactical Operations Plan (TOP) Data Base.

6.2.3.9 Subfunction 9 - Integrated Assessment of Planning Center TOP's

This subfunction includes the following interrelated tasks: verify that each planning center TOP does not violate the planning center resource allocations; consolidate the planning center TOP's; identify any operational conflicts between users of separate planning centers or between SS systems and users; schedule centralized resources that are not completely handled by resource allocation (crew, data flow etc.); determine what changes are required to eliminate conflicts and satisfy user resource demands; and generate the Integrated Payload TOP.

6.2.3.10 Subfunction 10 - Generate User Activity Plans

This is a detailed activity where each user defines desired experiment operating times within the TOP-allocated time blocks and assigns specific start/stop times to each operating step (mode).

6.2.3.11 Subfunction 11 - Integrate User Activity Plans

This subfunction includes the integration of the activity plans of all users of a particular planning center; verifies compatibility with TOP allocations; consolidates users activity plans; verifies compatibility with other planning center users; re-schedules user operations to eliminate identified resource or operational conflicts; generates the planning center activity plan data base; and, in the daily planning cycle, generates the planning center detailed Payload Crew Activity Plan (PCAP).

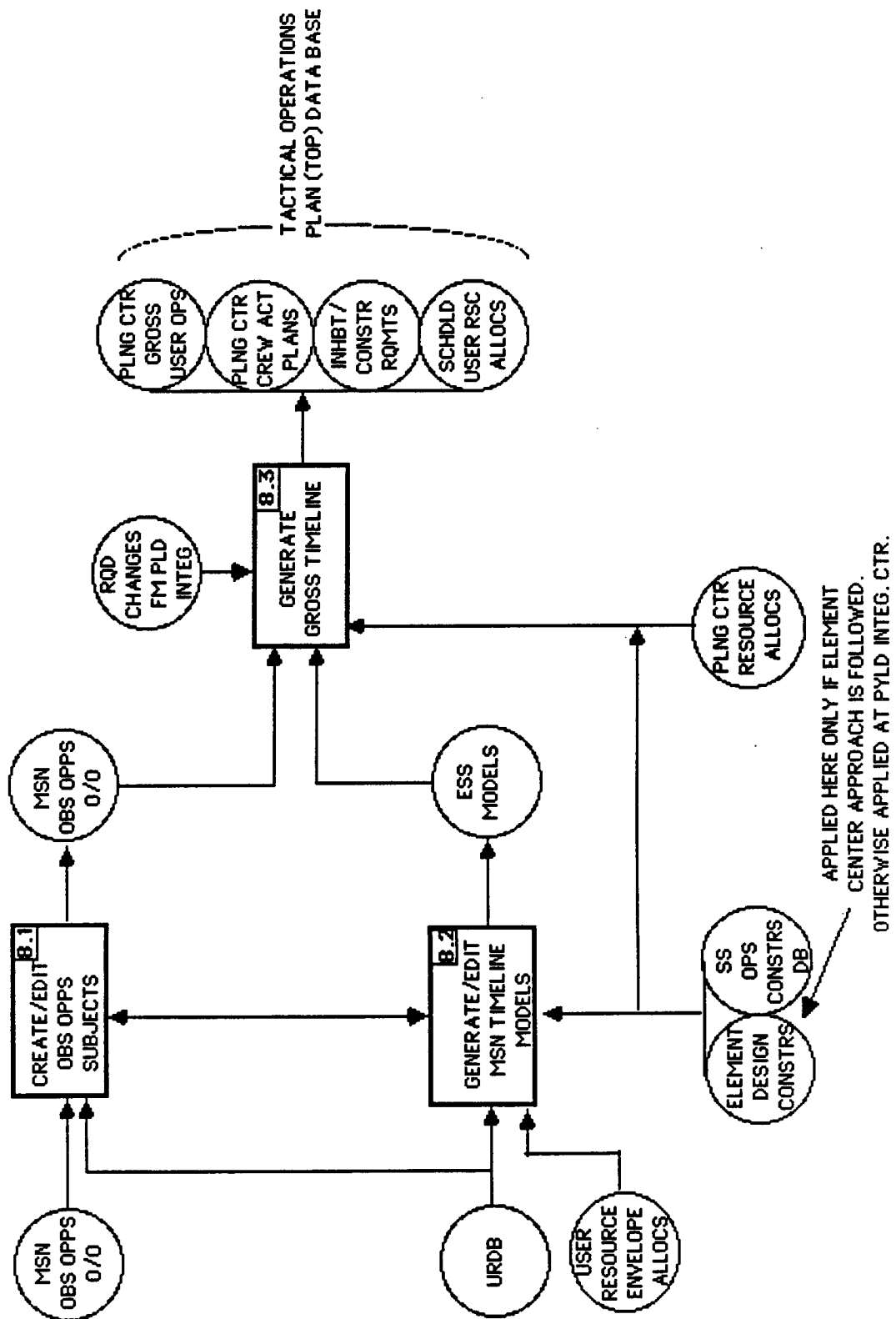
6.2.3.12 Subfunction 12 - Integrated Assessment of Planning Center Activity Plans

This activity is nearly identical to subfunction 11. The difference is that the integration is now at the overall SS payloads level (POIC) instead of the planning center level. The inputs are planning centers' activity plans and the outputs are the payload activity plans data base and payload summary PCAP.

6.2.3.13 Subfunction 13 - Generate User Command Plan

This subfunction includes the user activities of generating a command sequence to transition between steps (modes) of experiment operations and generating time windows during which these commands must occur based on the finalized user activity plans.

SUBFUNCTION: 8-GENERATE PLANNING CENTER GROSS TIMELINE



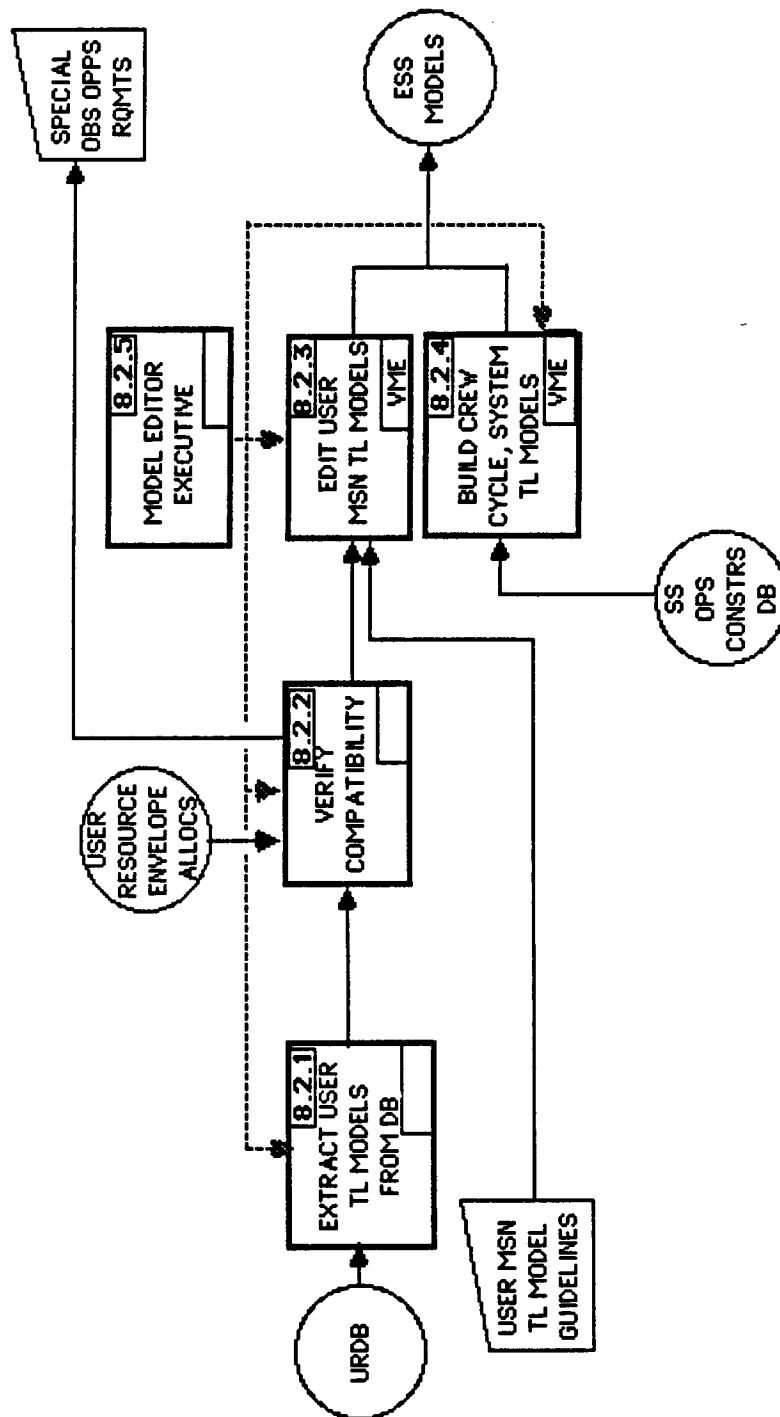
TASK: 8.1 -CREATE/EDIT OBS OPPTS SUBJECTS

[illegible]

SUBFUNCTION: 8-GENERATE PLANNING CENTER GROSS TIMELINE

TASK: 8.2-GENERATE/EDIT MISSION TIMELINE MODELS

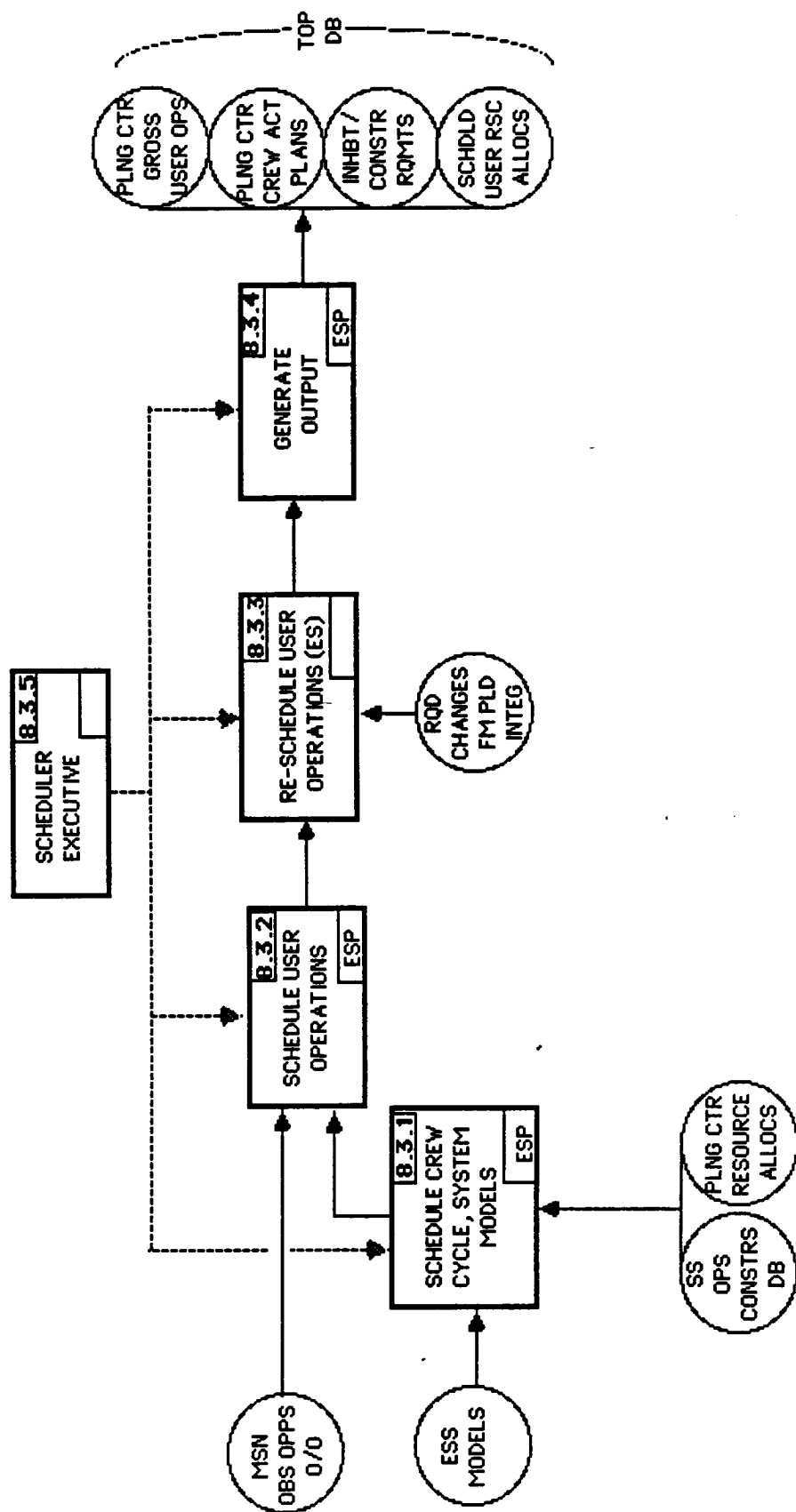
NOTE: SUBTASKS ARE SIMILIAR TO TASK 5.2, AND
ARE REPEATED HERE FOR CLARITY AND TO
IDENTIFY THE UNIQUE INPUTS AND OUTPUTS.



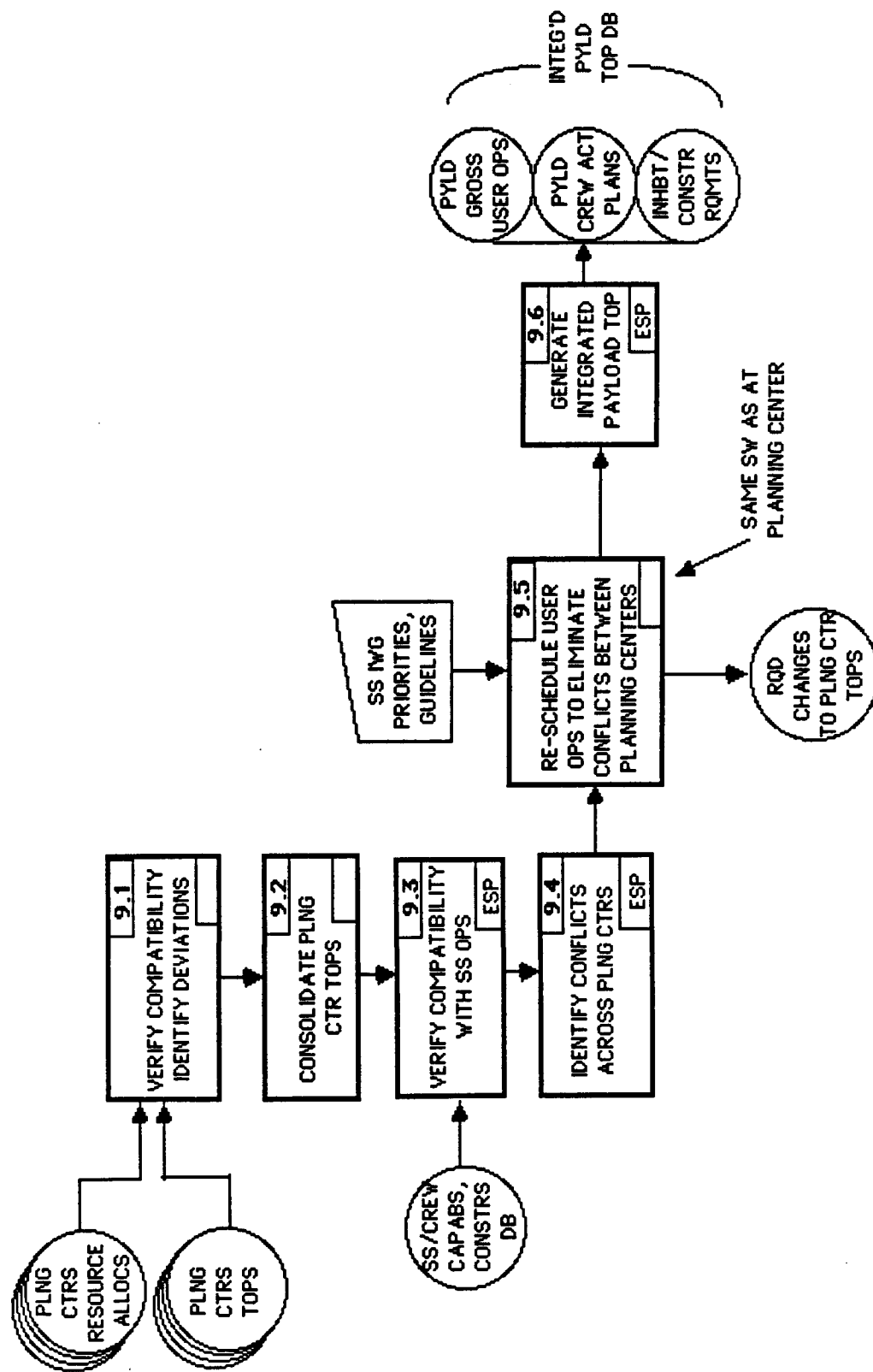
SUBFUNCTION: 8-GENERATE PLANNING CENTER GROSS TIMELINE

TASK: 8.3-GENERATE GROSS TIMELINE

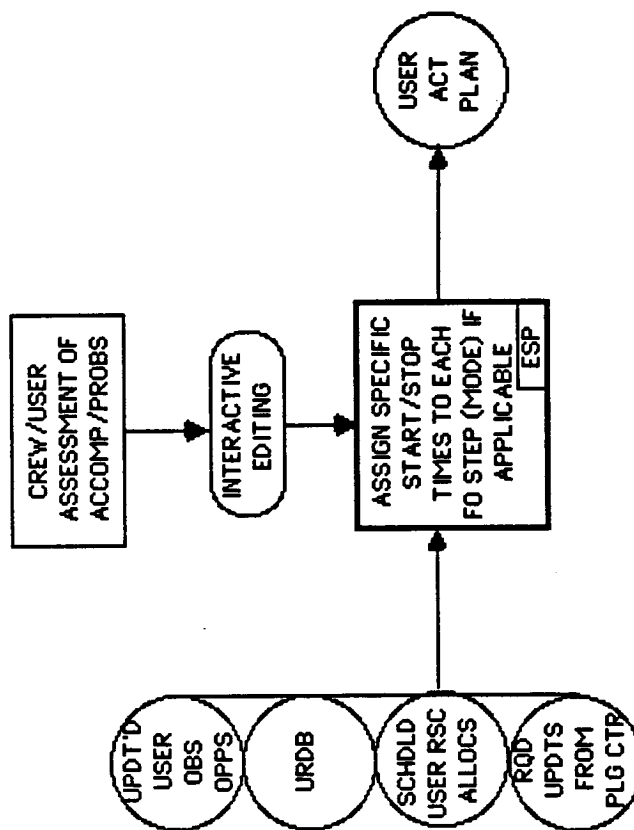
NOTE: SUBTASKS ARE SIMILAR TO TASK 5.3, AND ARE REPEATED HERE FOR CLARITY AND TO IDENTIFY THE UNIQUE INPUTS AND OUTPUTS.



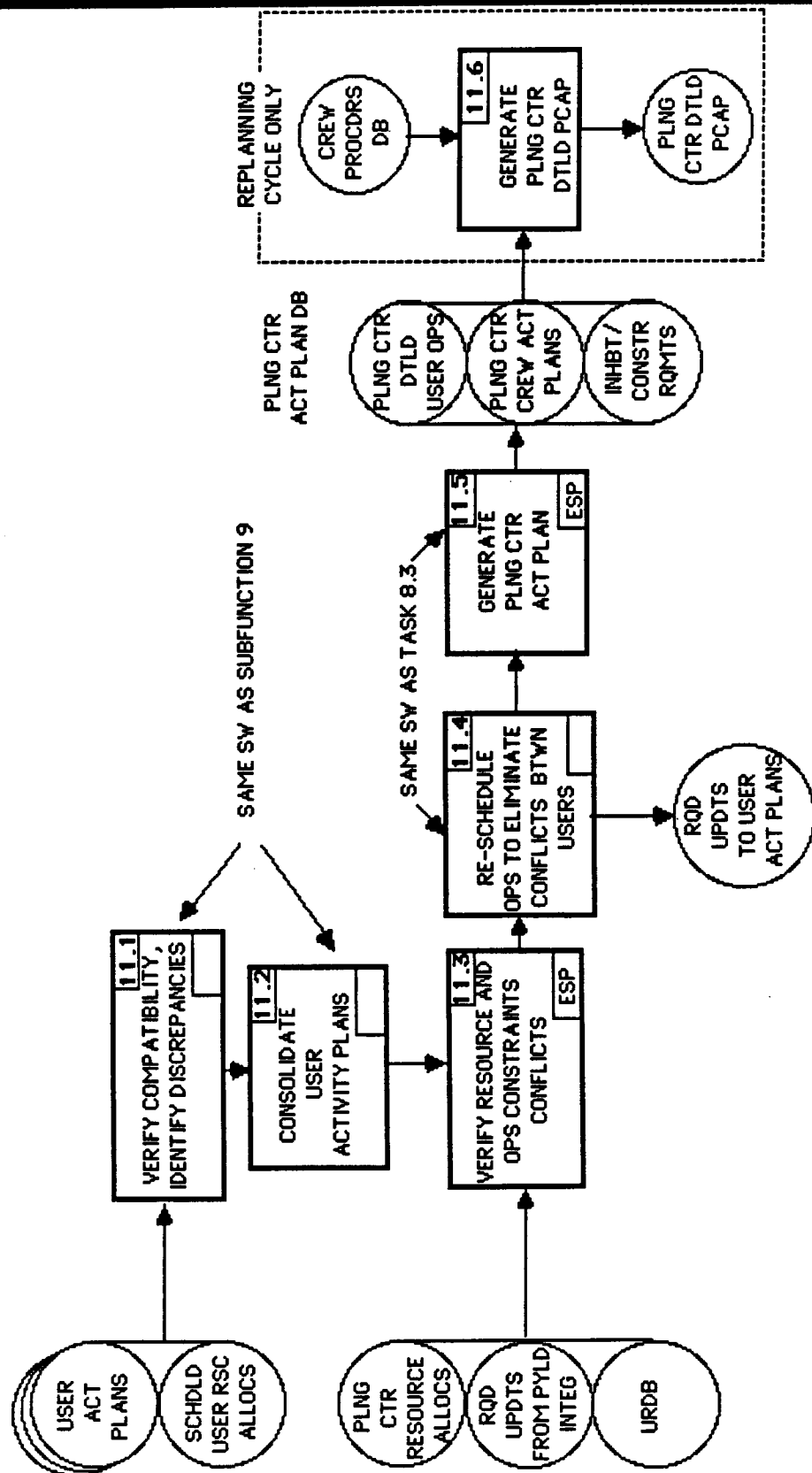
SUBFUNCTION: 9-INTEGRATED ASSESSMENT OF PLNG CTRS TOPS



SUBFUNCTION: 10-GENERATE USER ACTIVITY PLANS

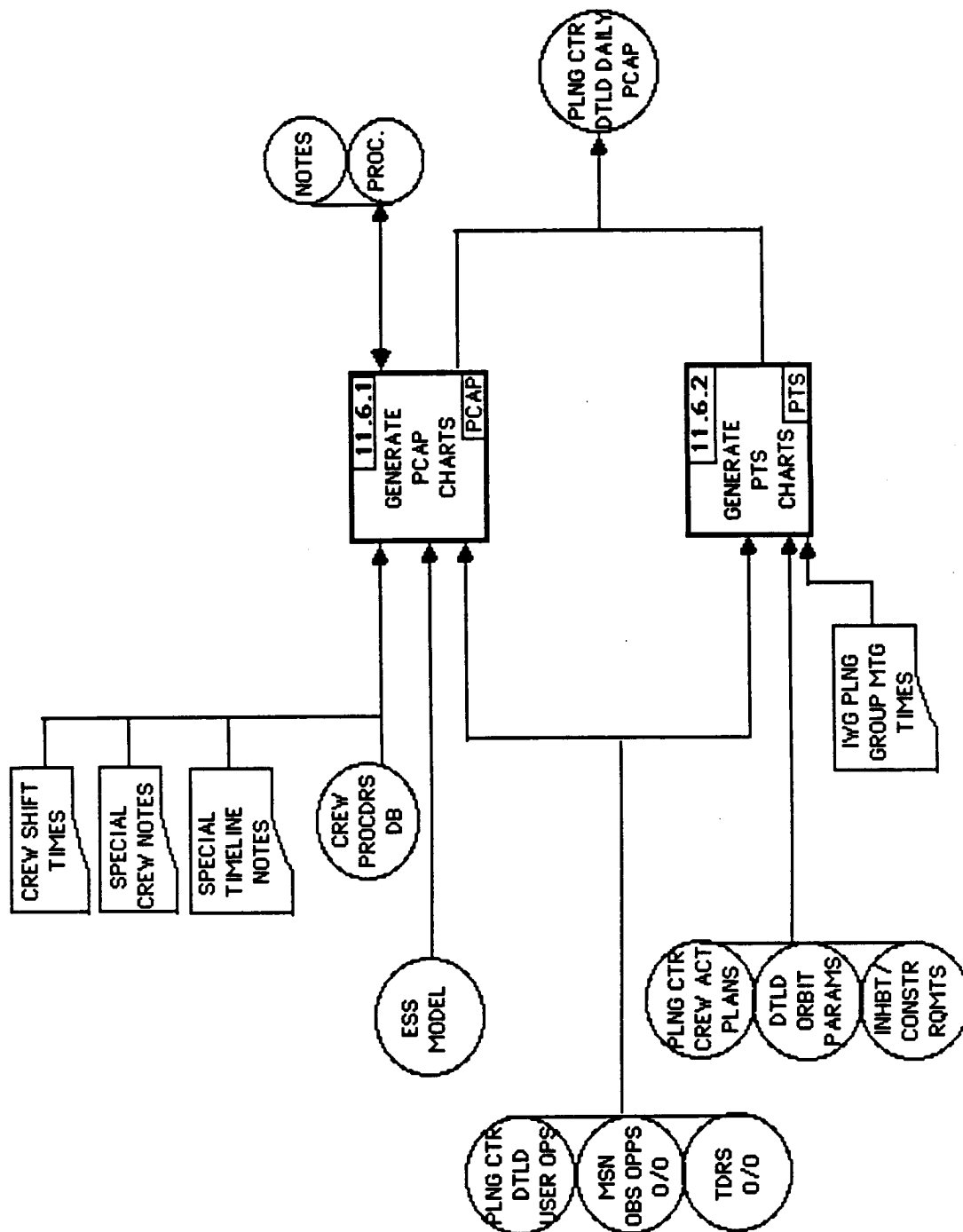


SUBFUNCTION: 11-INTEGRATE USER ACTIVITY PLANS

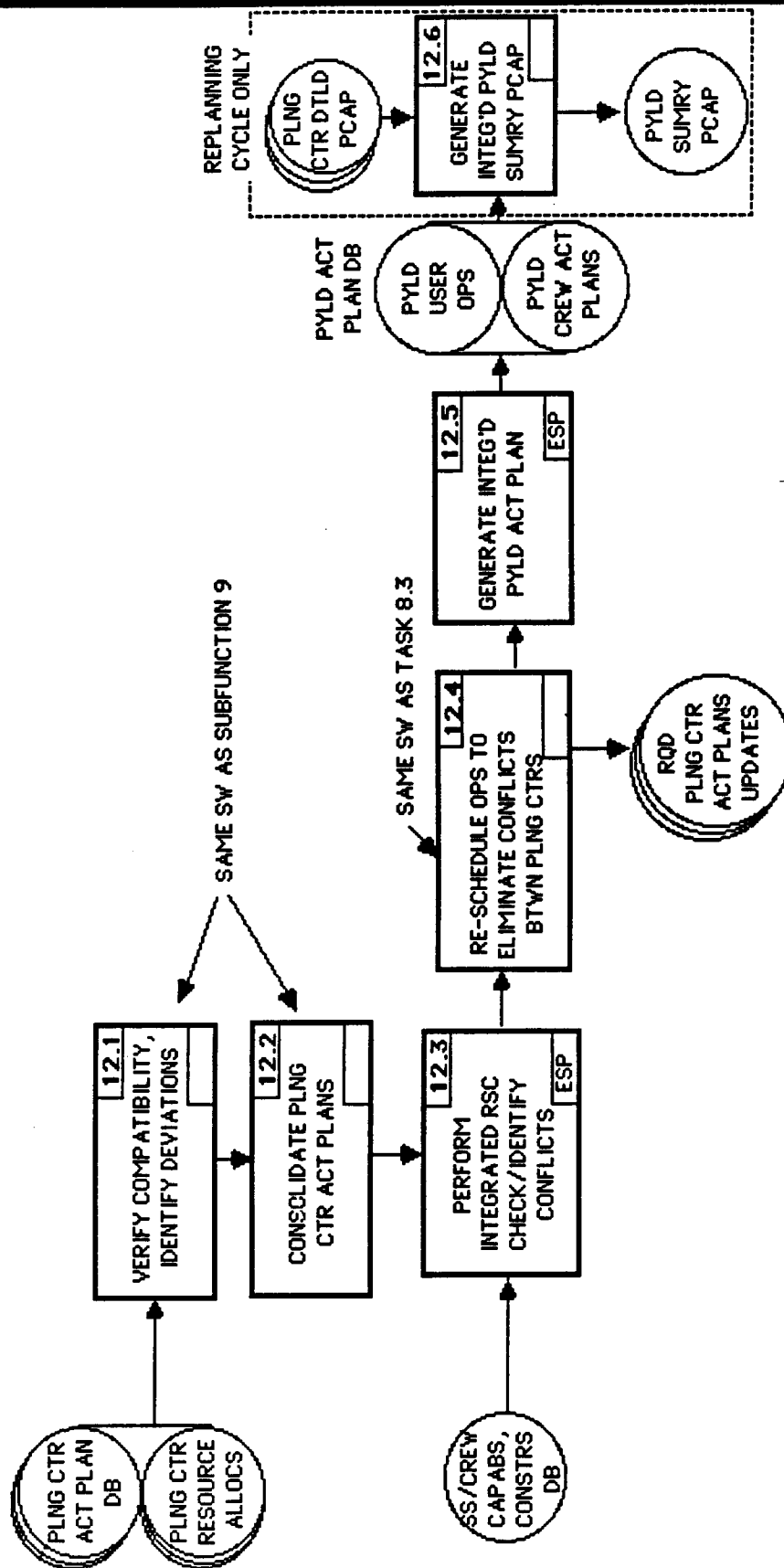


SUBFUNCTION: 11-INTEGRATE USER ACTIVITY PLANS

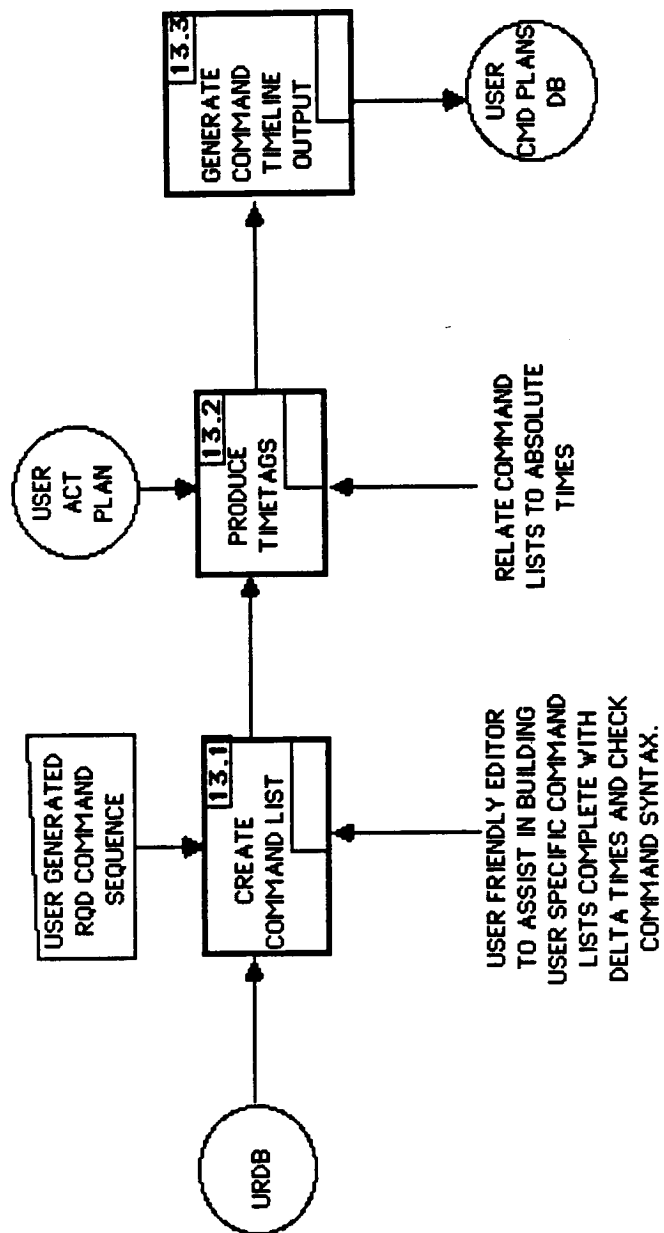
TASK: 11.6-GENERATE PLNG CTR DETAILED PCAP



SUBFUNCTION: 12-INTEGRATED ASSESSMENT OF PLNG CTR ACTIVITY PLANS



SUBFUNCTION: 13-GENERATE USER COMMAND PLANS



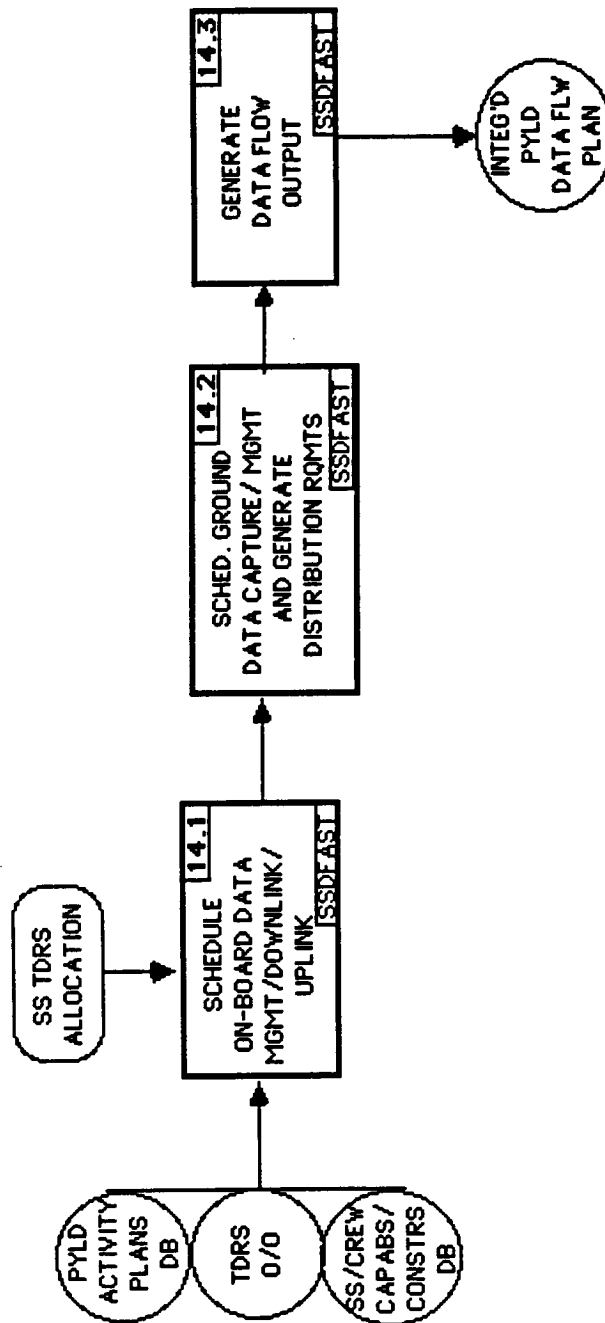
6.2.3.14 Subfunction 14 - Generate Integrated Payload Data Flow Plan

This activity involves scheduling overall payload data flow activities such as on-board data management of downlink/uplink and ground data capture, management and distribution as well as generating the integrated data flow plan.

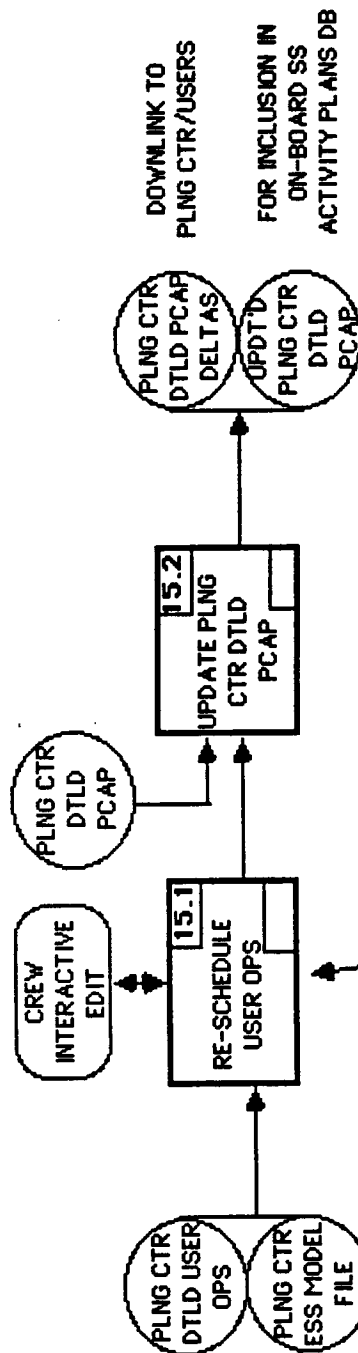
6.2.3.15 Subfunction 15 - Onboard Rescheduling

This subfunction allows for minor re-scheduling on-board by the SS crew. Accompanying activities of updating the PCAP and coordinating the changes with the ground are also required. Whether or not the overall SS operations concepts allow for on-orbit re-scheduling is as yet unresolved. The magnitude of the data requirements and the use of on-orbit crew time may force elimination of this subfunction.

SUBFUNCTION: 14-GENERATE INTEGRATED PAYLOAD DATA FLOW PLAN



SUBFUNCTION: 15-ONBOARD RE-SCHEDULING



SAME SW AS TASK 8.3

Figure 6.3-1a presents a hierarchical structure of the software modules envisioned to be required to implement the SS payload mission planning concept presented in the previous subsection; this structure is oriented toward the SS planning organizations (users, planning centers, POIC) and includes the definition of executive programs to interface with the using organizations and to control the execution of lower level software modules. Figure 6.3-1a also identifies (per the figure legend) the modified SL MIPS, new, and AI-application candidate software programs. (Section 7 presents the rationale for the AI-application candidates.)

Figure 6.3-1b identifies additional software modules required to implement the SS payload mission planning concept. The modules identified are those envisioned to be required to be provided to the on-board crew and the Space Station Systems organization for mission planning and will have to be integrated into the software systems to be developed for the crew and systems organization.

For the purposes of assessing the applicability of AI techniques to the SS MPS in Task 4 of the study, and for generating the Software Development Plan in Task 5, the computer programs identified in Figure 6.3-1 were grouped into software sets - i.e., groups of programs of a similar nature at the same hierarchical level. The software sets are presented in Table 6.3-1. Note that Sets E and F identify Phase I and II versions of the three "System Executives". The distinctions between these versions are explained in the Table 6.3-3 introduced in the following paragraph.

Finally, pages 6-52 through 6-65 of this section present a table (Table 6.3-2) which describes the individual software module requirements to implement the SS MPS concept. The table identifies each required software module by name and whether the module is new or a modified SL MIPS software module. Also provided is a functional description of each module. Finally, each software module is correlated to subfunctions/tasks in the SS mission planning concept functional flow diagrams presented previously.

[illegible]

FIGURE 6.3-1a. SS MPS SW Hierarchy

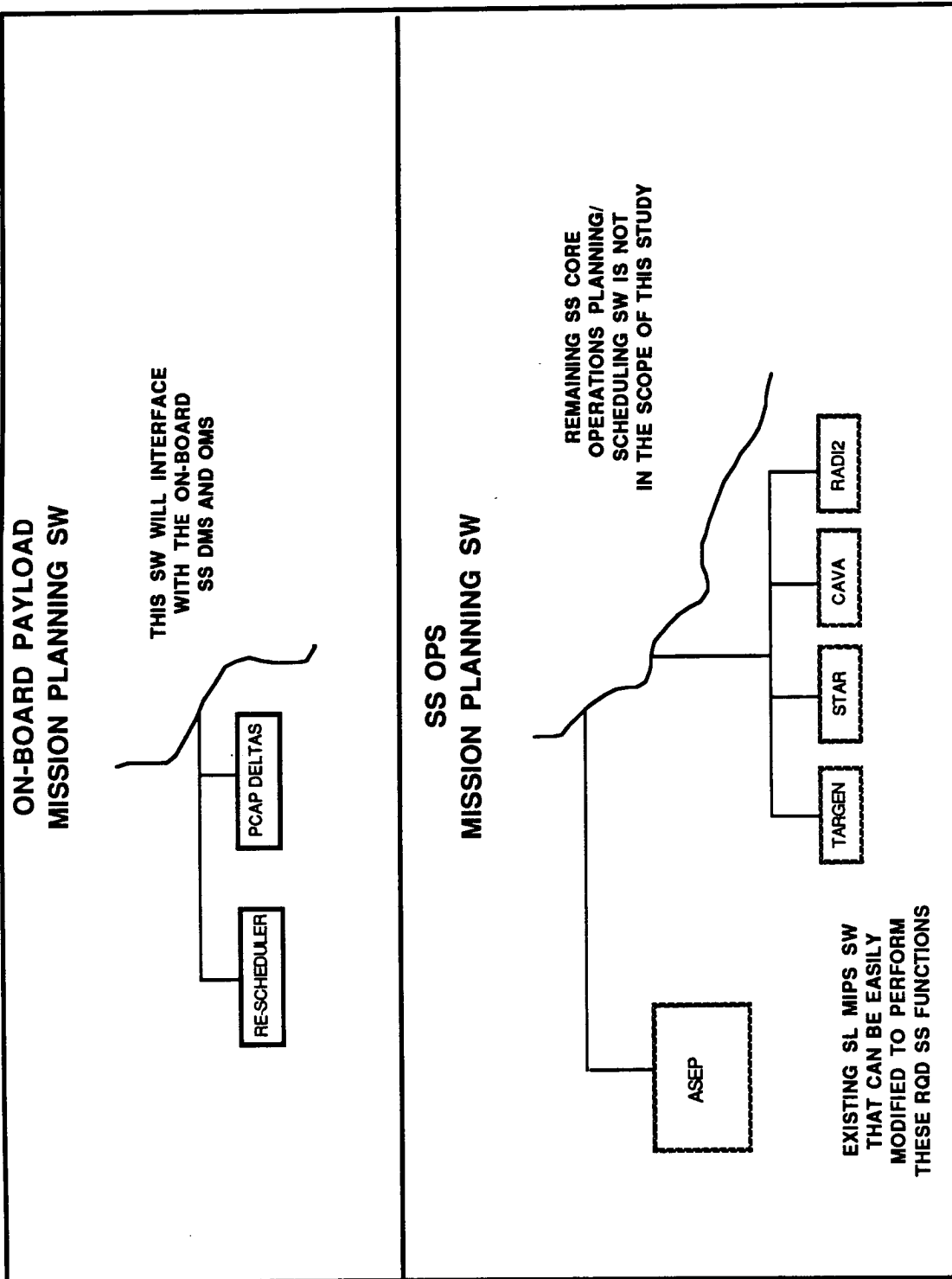


FIGURE 6.3-1b. SS MPS SW HIERARCHY

TABLE 6.3-1. SS MPS SOFTWARE SETS

NEW SOFTWARE

SET A - SPECIAL OBS OPPS EXECUTIVES

TOP LEVEL
ATMOS PHYS
SOLAR
EARTH SITE
PLASMA PHYSICS
CELESTIAL

SET B - URDB I/F

SET C - EDITOR EXECUTIVES

MODEL EDITOR EXEC
OBS OPPS EDITOR EXEC
SCHEDULER EXEC

SET D - RE-SCHEDULER

SET E - SYSTEM EXECUTIVES (PHASE I)

USER MPS EXEC
PLANNING CENTER MPS EXEC
POIC MPS EXEC

SET F - SYSTEM EXECUTIVES (PHASE II)

USER MPS EXEC
PLANNING CENTER MPS EXEC
POIC MPS EXEC

SET G - COMMAND PLANNER

SET H - NEW TIMELINE ANALYSIS MODULES

MDL EXTRACT
MDL COMPARE
TL COMPARE
TL MERGE
PCAP DELTAS
SUMMARY PCAP

SET L - OUTPUT PROCESSOR EXEC

MODIFIED SL MIPS SOFTWARE

SET I - TIMELINE ANALYSIS

ESP
PCAP
PTS
TAE
VME

SET J - ORBIT ANALYSIS

ASEP
ATMOS
BORB
CAVA
ESAL
ESDATA
LTO
RADI2
STAR
TANRAY
TARGEN

SET K - DATA FLOW ANALYSIS

PROFILE
MISSION WINDOWS
ONBOARD RECORDER SCHEDULAR
POSSIBLE FORMATS
FORMAT SCHEDULAR
POSSIBLE POCC CONFIGURATIONS
POCC CONFIGURATION SCHEDULAR
PLAYBACK SCHEDULAR
INTERACTIVE DATA UPDATE SYSTEM
VERIFICATION
COMPARE TDRS
COMPARE MODELS
DATA MANAGEMENT CHECKLIST
DATA SCHEDULE FILE
ANTENNA DISPLAY
IDMS LIBRARY

Table 6.3-2

| SS MPS SOFTWARE REQUIREMENTS SUMMARY | | | PAGE 1 |
|--------------------------------------|--------------------|--|--|
| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK |
| ASEP | MODIFIED | SEE SL MIPS DB. MODIFICATIONS CONSIST OF THE ADDITION OF A ROUTINE TO HANDLE DRAG ANALYTICALLY AND AN OPTION TO WRITE ASCN NODE, DTLD EPHEM AND GND TRCK DATA ON ONE LDF AND EARTH SHADOW ON AN O/O FILE. SHOULD BE SET UP TO PERIODICALLY ACCESS INPUT DATA AND PRODUCE OUTPUT FILES WITH MINIMUM MISSION PLANNERS INTERACTION. INPUT FILES: SOLAR ACTIVITY PROJECTIONS ACTUAL STATE VECTOR OUTPUT FILES: DTLD ORBIT PARAMS LDF EARTH SHADOW O/O | 1 SS PROJECTED ORBIT EPHEMERIS GENERATION |
| TARGEN | MODIFIED | SEE SL MIPS DB. IN THE SS MISSION PLANNING SYSTEM TARGEN WILL BE CALLED AT VARIOUS TIMES TO APPLY CONSTRAINTS AND PERFORM RQD SET THEORY OPERATIONS. MODIFICATIONS RQD WILL CONSIST PRIMARILY OF INTERFACES WITH THE VARIOUS SPECIAL OBSERVATION OPPORTUNITIES EXECUTIVES. INPUT FILES: ANY LIST DIRECTED FILE ANY O/O FILE OUTPUT FILES: AN O/O FILE | 2.1 GENERATE SUN RISE/SET 2.5 MERGE STANDARD ORBIT OBS OPPS 3.1.3 COMPUTE TERMINATOR TARGETS 3.1.4 COMBINE CONSTRS AND APPLY FILE MERGE OPPS 3.2.2 COMBINE CONSTRS TO DEFINE SOLAR OBS OPPS 3.3.4 APPLY CONSTRS TO EARTH SITE OBS OPPS 3.4.5 APPLY CONSTRAINTS AND MERGE OPPS 3.5.4 MERGE CELESTIAL OBS OPPS FILES 3.6 MERGE OBS OPPS FILES |
| STAR | MODIFIED | SEE SL MIPS DB-ADDITIONS/MODIFICATIONS WILL BE TO READ DTLD ORBIT PARAMS FILE INSTEAD OF AN ASCN NODE FILE. PROGRAM WILL ALSO GENERATE SEPARATE NIGHT AND NON-NIGHT CELESTIAL OBS DEFN AND OBS OPPS FILES. CAPABILITY MUST BE PROVIDED TO ENTER DEFNS AND CONSTRAINTS INTERACTIVELY THROUGH THE CELESTIAL OBS OPPS EXECUTIVE. INPUT FILES: DTLD ORBIT PARAMS LDF OUTPUT FILES: MOON RISE/SET O/O CEL OBS DEFNS NDF CEL OBS AC/LOSS O/O | 2.2 GENERATE MOON RISE/SET 3.5.1 DEVELOP CELESTIAL OBS DEFNS 3.5.2 GENERATE STELLAR OBS AC/LOS |

SS MPS SOFTWARE REQUIREMENTS SUMMARY

PAGE 2

| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK |
|----------------------|--------------------|---|--|
| CAVA | MODIFIED | SEE SL MIPS DB-A LARGE PORTION OF SL MIPS CAVA IS DEVOTED TO HANDLING MANEUVER AND ATT TLS WHICH WILL NOT BE RQD FOR SS.THE ROUTINES APPLICABLE TO TRAJECTORY DATA,SENSOR DATA, TARGET DATA, SENSOR TARGET VISIBILITY, OCCULTATION AND FILE MGMT CAN BE USED WITH MODIFICATIONS AND A MODIFIED DRIVER. INPUT FILES: DTLD ORBIT PARAMS LDF OUTPUT FILES: TDRS ACLOS O/O FILE | 2.3 GENERATE TDRS COVERAGE |
| RAD12 | MODIFIED | SEE SL MIPS DB-MUST READ DTLD ORBIT PARAMS FILE INSTEAD OF DTLD EPHEMERIS. INPUT FILES: DTLD ORBIT PARAMS LDF OUTPUT FILES:RAD ENVIR LDF | 2.4 GENERATE RADIATION ENVIR- ONMENT |
| TANRAY | MODIFIED | SEE SL MIPS DB-MUST READ DTLD ORBIT PARAMS FILE INSTEAD OF MANUALLY INPUT STATE VECTOR INPUT FILES:DTLD ORBIT PARAMS LDF OUTPUT FILES: TANRAY EPHEM LDF. | 3.1.1 COMPUTE DISTANCE FROM SS TO SUN LINE OF SITE TO EARTH SURFACE. COMPUTE SUN RISE/ SET HISTORY. |
| LTO | MODIFIED | SEE SL MIPS DB. IN THE SS MISSION PLANNING SYSTEM LTO WILL BE CALLED AT VARIOUS TIMES TO APPLY ACCEPTANCE CONDITIONS TO A LDF AND PRODUCE AN O/O FILE. MODIFICATIONS RQD WILL CONSIST PRIMARILY OF INTERFACES WITH THE VARIOUS SPECIAL OBSERVATION OPPORTUNITIES EXECUTIVES. INPUT FILES: ANY LIST DIRECTED FILE OUTPUT FILES: AN O/O FILE | 3.1.2 DEVELOP/APPLY CONSTRS TO ATMOS PHYS OBS PERIODS 3.2.1 DEVELOP SUN ELEV CONSTRS FOR SOLAR OBS PERIODS 3.3.3 DEVELOP/APPLY EARTH OBS OPPS CONSTRS 3.4.3 DEVL P/APPLY CONSTRS TO BORB PARAMS 3.4.4 GENERATE HEMISPHERE OPPS 3.5.3 IMPOSE RADIATION CONSTRS |
| ATMOS | MODIFIED | SEE SL MIPS DB-WILL READ DTLD ORBIT PARAMS FILE INSTEAD OF ASCN NODE LDF AND EARTH SHADOW O/O FILES. CAPABILITY TO HANDLE VARIABLE ATTITUDES IS NO LONGER NEEDED.MODIFIC- ATIONS RQD WILL CONSIST PRIMARILY OF INTERFACES WITH THE VARIOUS SPECIAL OBSERVATION OPPORTUNITIES EXECUTIVES. INPUT FILES: DTLD ORBIT PARAMS LDF OUTPUT FILES:SUN AZ/ELEV LDF | 3.1.5 COMPUTE SUN AZ/ELEV FM SS WRT SUN RISE/SET EVENTS |

| SS MPS SOFTWARE REQUIREMENTS SUMMARY | | | | PAGE 3 |
|--------------------------------------|--------------------|--|---------------------------------------|--------|
| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK | |
| ATMOS PHYS EXEC | NEW | EXPERT SYSTEM EXECUTIVE THAT AIDS THE USER/MSN PLANNER IN DEFINING ATMOSPHERIC PHYSICS OBSERVATION PQMTS AND CALCULATING OBSERVATION OPPORTUNITIES. SYSTEM MUST PROVIDE A USER FRIENDLY INTERFACE WITH ON-LINE HELP AND EXPLANATION FEATURES. THIS EXECUTIVE SHALL SELECT AND SEQUENCE THE APPLICABLE CALCULATION ROUTINES (TANRAY, LTO, TARGEN, ATMOS). ACTUAL ROUTINE CALLS SHALL BE TRANSPARENT TO THE USER. THE USER SHALL BE PROVIDED WITH A "GENERIC" SET OF INPUT DEFAULT VALUES THAT ARE CONSTANTLY UPDATED BASED ON USER INPUTS TO OTHER FIELDS. THIS PROVIDES A WORKING MODEL OF USER OBS OPPS REQUIREMENTS REGARDLESS OF THE AMOUNT OF USER DEFINITION. THE EXECUTIVE SHALL HAVE THE CAPABILITY TO RECOGNIZE REQUESTS OUTSIDE ITS CURRENT KNOWLEDGE DOMAIN AND REQUEST ASSISTANCE FROM THE USER/OPERATOR. | 3.1.6 EXEC FOR ATMOS PHYS OBS OPPS | |
| ESDAT | MODIFIED | SEE SL MIPS DB-GND SITE DEFNS WILL BE INPUT THROUGH EARTH SITE EXECUTIVE. INPUT FILES: NONE OUTPUT FILES: SITE AC/LOS O/O | 3.3.1 CREATE EARTH SITE DEFN FILE | |
| ESAL | MODIFIED | SEE SL MIPS DB-WILL READ DTLD ORBIT PARAMS LDF INSTEAD OF ASCN NODE. INPUT FILES: SITE DEFNS NDF OUTPUT FILES: SITE AC/LOS O/O | 3.3.2 GENERATE AREA SITE AC/LOS | |

SS MPS SOFTWARE REQUIREMENTS SUMMARY

PAGE 4

| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK |
|-----------------------|--------------------|---|--|
| EARTH SITE EXEC | NEW | EXPERT SYSTEM EXECUTIVE THAT AIDS THE USER/MSN PLANNER IN DEFINING EARTH SITE OBSERVATION ROIMTS AND CALCULATING OBSERVATION OPPORTUNITIES. SYSTEM MUST PROVIDE A USER FRIENDLY INTERFACE WITH ON-LINE HELP AND ALSO INTERFACE WITH THE APPLICABLE CALCULATION ROUTINES (ESDAT, ESAL, LTO, TARGEN, TAE). ACTUAL ROUTINE CALLS SHALL BE AS TRANSPARENT AS POSSIBLE TO THE USER. THIS EXEC CONTAINS FEATURES IDENTICAL TO THE ATMOS PHYS. EXECUTIVE. | 3.3.6 EXEC FOR EARTH SITE OBS OPPS |
| TAE | MODIFIED | SEE SL MIPS DB- STATISTICAL ANALYSIS ROUTINE WILL BE CALLED BY ORBITAL ANALYSIS EXECUTIVE PROGRAMS. OO FILES WILL NOT BE REFORMATED AND BUILDING NEW SUBJECTS AND EDITING DATA WILL BE DONE IN THE O/O FILE FORMAT. MUST ALSO BE MODIFIED TO EXTRACT USER TIME PREFERENCES DIRECTLY FROM THE URDB. INPUT FILES: MSN OBS OPPS O/O OR URDB OUTPUT FILES: MSN OBS OPPS O/O | 3.3.5 STATISTICAL ANAL OF OBS OPPS 3.4.6 STATISTICAL ANAL OF OBS OPPS 3.5.5 STATISTICAL ANAL OF OBS OPPS 5.1.1 EXTRACT USR TIME PREFERENCES 5.1.2 BUILD NEW SUBJECTS IF RQD 5.1.3 STATISTICAL ANALYSIS OF OBS OPPS 5.1.4 OBS OPPS DATA EDITING 8.1.1 EXTRACT USR TIME PREFERENCES 8.1.2 BUILD NEW SUBJECTS IF RQD 8.1.3 STATISTICAL ANALYSIS OF OBS OPPS 8.1.4 OBS OPPS DATA EDITING |
| BORB | MODIFIED | SEE SL MIPS DB-MODIFIED TO READ DTLTD ORBIT PARAMS LDF INSTEAD OF ASCN NODE FILE. CAPABILITY TO HANDLE ATT TL O/O FILE IS NO LONGER REQUIRED. PROGRAM WILL BE DRIVEN BY PLAMSA PHYS EXECUTIVE. INPUT FILES: DTLTD ORBIT PARAMS LDF OUTPUT FILES: BORB PARAMS LDF | 3.4.1 COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN SS BODY COORD SYS 3.4.2 DEVELOP PLASMA PHYSICS OBS DFNS |

SS MPS SOFTWARE REQUIREMENTS SUMMARY

PAGE 5

| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK |
|------------------------|--------------------|--|--|
| PLASMA PHYS EXEC | NEW | EXPERT SYSTEM EXECUTIVE THAT AIDS THE USER/MSN PLANNER IN DEFINING PLASMA PHYS OBSERVATION ROUTES AND CALCULATING OBSERVATION OPPORTUNITIES. SYSTEM MUST PROVIDE A USER FRIENDLY INTERFACE WITH ON-LINE HELP AND ALSO INTERFACE WITH THE APPLICABLE CALCULATION ROUTINES (BORB, LTO, TARGEN, TAE). ACTUAL ROUTINE CALLS SHALL BE AS TRANSPARENT AS POSSIBLE TO THE USER. THIS EXECUTIVE HAS FEATURES IDENTICAL TO THE ATMOS PHYSICS EXECUTIVE. | 3.4.2 DEVELOP PLASMA PHYSICS OBS DEFNS 3.4.7 EXEC FOR PLASMA PHYS OBS OPPS |
| CELEST- IAL EXEC | NEW | EXPERT SYSTEM EXECUTIVE THAT AIDS THE USER/MSN PLANNER IN DEFINING CELESTIAL OBSERVATION ROUTES AND CALCULATING OBSERVATION OPPORTUNITIES. SYSTEM MUST PROVIDE A USER FRIENDLY INTERFACE WITH ON-LINE HELP AND ALSO INTERFACE WITH THE APPLICABLE CALCULATION ROUTINES (STAR, TARGEN, LTO, TAE). ACTUAL ROUTINE CALLS SHALL BE AS TRANSPARENT AS POSSIBLE TO THE USER. THIS EXECUTIVE HAS FEATURES IDENTICAL TO THE ATMOS PHYSICS EXECUTIVE. | 3.5.6 EXEC FOR STELLAR OBS OPPS |
| VME | MODIFIED | SEE SL MIPS DB-INTERFACE MUST BE MODIFIED FOR COMPATIBILITY WITH MODEL EDITOR EXECUTIVE. INPUT FILES: USER MDLS SS OPS CONSTRS DB OUTPUT FILES: ESS MODELS | 5.2.3 EDIT USER MSN TL MODELS 5.2.4 BUILD CREW CYCLE SYSTEM MODELS 8.2.3 EDIT USER MSN TL MODELS 8.2.4 BUILD CREW CYCLE SYSTEM MODELS 7.3 ASSIGN USER RSC ENV ALLOCS |

SS MPS SOFTWARE REQUIREMENTS SUMMARY

PAGE 6

| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK |
|----------------------------|--------------------|---|--|
| OBS OPPS EDITOR EXEC | NEW | EXPERT SYSTEM EXECUTIVE THAT GUIDES THE USER MISSION PLANNER IN BUILDING SPECIAL OBSERVATION OPPORTUNITIES. ALLOWS USER TO INPUT REQUIRED OPERATING TIMES THAT ARE A RESULT OF SOMETHING OTHER THAN AN ORBITAL OPPORTUNITY. MUST INTERFACE WITH TAE ROUTINE THAT BUILDS NEW SUBJECTS. THE EXECUTIVE MUST HAVE THE ABILITY TO CONSTRUCT A VALID OBS OPPS SUBJECT IF REQUIRED, AND EXPLAIN THE PROCESS TO THE MISSION PLANNER. IT SHOULD BE CAPABLE OF ACTIVATING THE STATISTICAL ANALYSIS ROUTINE AND ASSESSING THE RESULTS FOR CONFIDENCE AND RELIABILITY OF THE OUTPUT WINDOW OF OBS OPPS. | 5.1.5 OBS OPPS EDITOR EXEC 8.1.5 OBS OPPS EDITOR EXEC |
| MODEL EDITOR EXEC | NEW | EXPERT SYSTEM TO AID MISSION PLANNERS IN DEVELOPING MISSION TIMELINE MODELS. MUST PROVIDE INTERFACE WITH VME. THE EXECUTIVE SHOULD PERFORM ALL SOFTWARE MODULE SELECTION AND SEQUENCING AND PROVIDE GUIDELINES AND DEFAULT EXAMPLES FOR CONSTRUCTING TIMELINE MODELS. IT SHOULD MONITOR ALL TL MODELS FOR INTERNAL CONSISTENCY AND CONSTRAINTS. AN EXPLANATORY, USER FRIENDLY INTERFACE SHOULD BE PROVIDED. | 5.2.5 MODEL EDITOR EXECUTIVE 8.2.5 MODEL EDITOR EXECUTIVE |
| MDL EXTRACT | NEW | INTERFACE WITH THE USER REQUIREMENTS DATA BASE AND EXTRACT THE APPROPRIATE USER TL MODELS DATA INPUT FILES: URDB OUTPUT FILES: USER MODELS | 5.2.1 EXTRACT USER TL MDLS FROM DB 8.2.1 EXTRACT USER TL MDLS FROM DB |
| MDL COMPARE | NEW | VERIFY COMPATIBILITY OF USER MODELS WITH USER RESOURCE ALLOCATIONS. INPUT FILES: USER MODELS USER RSC ENV ALLOCS OUTPUT FILES: USER MODELS | 5.2.2 VERIFY COMPATIBILITY 8.2.2 VERIFY COMPATIBILITY |

| SS MPS SOFTWARE REQUIREMENTS SUMMARY | | | | PAGE 7 |
|--------------------------------------|--------------------|---|---|--------|
| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK | |
| SCHED- ULER EXEC | NEW | EXPERT SYSTEM EXECUTIVE THAT WILL GUIDE MISSION PLAN- NERS AT BOTH THE PLANNING CENTERS AND PAYLOAD INTEGR- ATION CENTER IN USE OF ALL SCHEDULING SW MODULES. MAJOR MODULES WILL CONSIST OF A REVISED VERSION OF ESP, A NEW EXPERT RE-SCHEDULER AND VARIOUS UTILITY ROUTINES TO ALLOW COMBINING SCHEDULES, COMPARING SCHEDULES, AND GENERATING INTERFACE FILES ETC. MODULES AND UTILITY ROUTINES CURRENTLY IDENTIFIED ARE DISCUSSED INDIVIDUALLY. THE EXECUTIVE SHOULD MONITOR ALL IO FOR CONSISTENCY AND INTERMODULE CONSTRAINTS. IT SHOULD BE CAPABLE OF PERFORMING CONFIDENCE ASSESSMENTS (E.G. 95%) OF ALL THE OUTPUT REQUIREMENTS AND IDENTIFY INPUTS THAT ARE NOT STRONGLY SUPPORTED BY THE CURRENT KNOWLEDGE BASE. | 5.3.5 SCHEDULER EXECUTIVE 8.3.5 SCHEDULER EXECUTIVE | |
| ESP | MODIFIED | SEE SL MIPS DB THE BASIC SCHEDULING PROCESS IS THE SAME AS SL MIPS ESP. THE CORE OF THE SCHEDULER SHOULD REQUIRE ONLY MINOR MODIFICATIONS. ADDITIONS/MODIFICATIONS WILL EXIST PRIMARILY IN THE AREAS OF INTERFACE FILES WITH THE ES RE-SCHEDULER AND THE OVERALL SCHEDULER EXECUTIVE. INPUT FILES: MSN OBS OPPS O/O ESS MODELS RESOURCE ALLOCS FILE (PLNG CTR BASIS OR PLD INTEG CTR BASIS) OUTPUT FILES: PLNG CTR GROSS USER OPS PLNG CTR CREW ACT PLANS INHB/CONSTR RQMTS SCHDLD USER RSC ALLOCS INTEG PLD CNSLDTD SCHDL | 5.3.1 SCHED CREW CYCLE, SYSTEM MODELS 5.3.2 SCHED USER OPERATIONS 5.3.4 GENERATE OUTPUT 6.2 VFY COMPATIBILITY, IDENTIFY DEVIATIONS 6.4 EXTRACT RESOURCE REQUIREMENTS 8.3.1 SCHED CREW CYCLE, SYSTEM MODELS 8.3.2 SCHED USER OPERATIONS 8.3.4 GENERATE OUTPUT 9.3 VERIFY COMPATIBILITY WITH SS OPS 9.4 IDENTIFY CONFLICTS ACROSS PLNG CTRS 9.6 GENERATE INTGD PLD TOP 10 GENERATE USER ACTIVITY PLANS 11.3 VERIFY RSC AND OPS CONSTR S CONFLICTS 11.5 GEN PLNG CTR ACT PLAN 12.3 PERFORM INTEGRATED RSC CHECK/IDENTIFY CONFLICTS 12.5 GENERATE INTEG'D PYLD ACT PLAN | |

| SS MPS SOFTWARE REQUIREMENTS SUMMARY | | | | PAGE 8 |
|--------------------------------------|--------------------|--|---|--------|
| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK | |
| TL COMPARE | NEW | UTILITY ROUTINE THAT WILL COMPARE--AND IDENTIFY DEVIATIONS--A RESOURCE ALLOCATIONS PROFILE WITH A SCHEDULE THAT WAS DEVELOPED TO FIT WITHIN THOSE ALLOCATIONS FOR VERIFICATION PURPOSES. MUST INTERFACE UPWARDS WITH THE SCHEDULER EXECUTIVE INPUT FILES: RESOURCE ALLOCATION FILES ESS TL FILES(USERS OR PLNG CTRS) OUTPUT FILES: VERIFIED ESS TL FILES WITH DEVIATIONS | 7.1 COMPARE, IDENTIFY DEVIATIONS 9.1 VERIFY COMPATIBILITY, IDENTIFY DEVIATIONS 11.1 VERIFY COMPATIBILITY, IDENTIFY DEVIATIONS 12.1 VERIFY COMPATIBILITY, IDENTIFY DEVIATIONS | |
| TL MERGE | NEW | UTILITY ROUTINE THAT WILL CONSOLIDATE VERIFIED SCHEDULES/TIMELINES INTO AN INTEGRATED SCHEDULE OF THE ESS FORMAT. MUST INTERFACE UPWARDS WITH THE SCHEDULER EXECUTIVE INPUT FILES: ESS TL FILES OUTPUT FILES: CONSOLIDATED ESS TL FILES | 6.1 CONSOLIDATE REQUIREMENTS 9.2 CONSOLIDATE PLANNING CTR TOPS 11.2 CONSOLIDATE USER ACT PLANS 12.2 CONSOLIDATE PLNG CTR ACT PLANS | |
| CMD PLANNER | NEW | USER FRIENDLY EDITOR TO GUIDE/ASSIST USERS IN BUILDING USER SPECIFIC COMMAND LISTS COMPLETE WITH DELTA TIMES. ALSO TAKES ACTIVITY PLANS AS INPUT AND TIMETAGS COMMAND LISTS WITH ABSOLUTE TIMES. THE EXECUTIVE SHOULD RELY ON INPUT FROM THE URDB TO ASSEMBLE RECOMMENDED COMMAND SEQUENCES BASED ON CONSTRAINTS AND THE CURRENT EXPERIENCE BASE. IT SHOULD ALSO BE ABLE TO PERFORM REORGANIZATION OF GROUPS OF COMMANDS BASED ON USER INPUTS. INPUT FILES: URDB USER ACTIVITY PLANS DB OUTPUT FILES: USER CMD PLANS DB | 13.1 CREATE COMMAND LIST 13.2 PRODUCE TIMETAGS 13.3 GENERATE CMD TL OUTPUT | |

SS MPS SOFTWARE REQUIREMENTS SUMMARY

PAGE 9

| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK |
|-----------------------|--------------------|---|--|
| RE- SCHED- ULER | NEW | THIS MODULE IS AN EXPERT SYSTEM THAT WILL TAKE A SCHEDULE/TIMELINE GENERATED BY ESP AND ALLOW RE-SCHEDULING OF OPERATIONS TO REFINE THE ORIGINAL SCHEDULE, MAKE CHANGES REQUIRED AS A RESULT OF OVERALL PAYLOAD INTEGRATION OR ALLOW ON-BOARD RE-SCHEDULING BY THE CREW. THIS MODULE WILL READ AN ESS FORMATED TIMELINE, AID THE OPERATOR IN RE-SCHEDULING OPERATIONS AND CREATE AN OUTPUT FILE IN THE ESS FORMAT. PLANET AND MAESTRO ARE POTENTIAL BASELINE MODELS FOR THIS MODULE. SINCE PLANS ARE FOR ONE VERSION OF THIS MODULE TO RESIDE ON-BOARD THE SS RQMT THAT ALL FLIGHT SW BE WRITTEN IN ADA MUST BE CONSIDERED. INPUT FILES: ESS TL FILES ESS MODEL FILES OUTPUT FILES: UPDTD ESS TL FILES | 5.3.3 RE-SCHEDULE USER OPS(ES) 8.3.3 RE-SCHEDULE USER OPS(ES) 9.5 RE-SCHEDULE USER OPS TO ELIMINATE CONFLICTS BETWEEN PLNG CTRS 11.4 RE-SCHEDULE OPS TO ELIMINATE CONFLICTS BETWEEN USERS 12.4 RE-SCHEDULE OPS TO ELIMINATE CONFLICTS BETWEEN PLNG CTRS 15.1 ON-BOARD RE-SCHEDULING |
| PCAP | MODIFIED | SEE SL MIPS DB- SS OPERATIONS MAY IMPOSE NEW RQMTS ON THE LAYOUT OF THE PCAP CHARTS. INTERFACE FILES WILL BE SOMEWHAT DIFFERENT AND HARDCOPY OUTPUT WILL BE OPTIONAL. INPUT FILES: CREW PRCDRS DB ESS MODELS FILE PLNG CTR USER OPS MSN OBS OPPS O/O OUTPUT FILES: PCAP CHARTS FILE NOTES FILE PROCEDURES FILE | 11.6.1 GENERATE PCAP CHARTS |
| PTS | MODIFIED | SEE SL MIPS DB- SS OPERATIONS MAY IMPOSE NEW RQMTS ON THE LAYOUT OF THE PTS CHARTS. INTERFACE FILES WILL BE SOMEWHAT DIFFERENT AND HARDCOPY OUTPUT WILL BE OPTIONAL INPUT FILES: PLNG CTR CREW ACT PLANS DTLD ORBIT PARAMS LDF INHB/CONSTR RQMTS PLNG CTR USER OPS MSN OBS OPPS O/O OUTPUT FILES: PTS CHARTS FILE | 11.6.2 GENERATE PTS CHARTS |

SS MPS SOFTWARE REQUIREMENTS SUMMARY

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| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK |
|-------------------------------------|--------------------|--|--|
| SUMMARY PCAP | NEW | PROGRAM THAT WILL GENERATE A SUMMARY PCAP FROM THE CONSOLIDATED PAYLOAD OPERATIONS SCHEDULE AND THE DETAILED PLANNING CENTER DAILY PCAPS. INPUT FILES:DTLD PLNG CTR DAILY PCAP CNSLDTD PYLD ESS TL FILE OUTPUT FILES: PYLD SUMRY PCAP | 12.6 GENERATE INTEGRATED PAYLOAD SUMMARY PCAP |
| PCAP DELTAS | NEW | ON-BOARD SW THAT WILL, AFTER CREW CHANGES ARE MADE TO THE INDIVIDUAL PLANNING CENTER TIMELINES, MODIFY THE EXISTING ON-BOARD PCAP, SEND THE UPDATED PCAP TO THE ON-BOARD SS ACTIVITY PLANS DATA BASE AND DOWNLINK A DELTAS FILE THAT WILL ALLOW GROUND PERSONNEL TO UPDATE THEIR VERSION OF THE PCAP. INPUT FILES: PLNG CTR DTLD DAILY PCAP ESS TL FILE OUTPUT FILES: UPDSTD PLNG CTR DTLD PCAP PLNG CTR DTLD PCAP DELTAS PYLD SUMRY PCAP | 15.2 UPDATE PLNG CTR DTLD PCAP |
| PLNG CTR MISSION PLNG EXEC | NEW | PHASE I: A HIGH LEVEL EXECUTIVE WHICH PROVIDES MISSION PLANNING PERSONNEL A STANDARDIZED SYSTEM FOR USE OF LOWER LEVEL EXECUTIVES AND CALCULATION ROUTINES AS WELL AS PROVIDING A FILE MANAGEMENT SYSTEM. THE NATURAL LANGUAGE INTERFACE DESCRIBED IN THE USER PLANNING EXECUTIVE WILL BE AVAILABLE WITH A VOCABULARY TAILORED TO THE PLANNING CENTER APPLICATIONS. PHASE II: CAPABILITY WILL BE PROVIDED SUCH THAT LOGIC MAY BE ENCODED IN SUBFUNCTIONS 5, 7, 8 AND 10 OF PLANNING CYCLES A, B, AND C AND USED AS AN ADVISOR IN SUBFUNCTIONS 8 AND 10 OF THE REPLANNING CYCLE D. | |

SS MPS SOFTWARE REQUIREMENTS SUMMARY

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| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK |
|---|--------------------|---|--|
| PYLD OPS INTEG. CTR MSN PLNG EXEC | NEW | <p>PHASE I: A HIGH LEVEL EXECUTIVE WHICH PROVIDES MISSION PLANNING PERSONNEL A STANDARDIZED SYSTEM FOR USE OF LOWER LEVEL EXECUTIVES AND CALCULATION ROUTINES AS WELL AS PROVIDING A FILE MANAGEMENT SYSTEM. SIMILAR TO THE PLNG CTR VERSION.</p> <p>PHASE II: CAPABILITY WILL BE PROVIDED SO THAT LOGIC MAY BE ENCODED DURING SUBFUNCTIONS 6, 9, 11, AND 12 OF PLANNING CYCLES A, B, C AND USED AS AN ADVISOR IN SUBFUNCTIONS 6, 11, AND 12 OF THE REPLANNING CYCLE D.</p> | |
| SPECIAL OBS OPPS EXEC | NEW | <p>A HIGH LEVEL EXECUTIVE THAT INTERFACES UPWARDS TO THE USER AND PLANNING CENTER MISSION PLANNING EXECUTIVES AND DOWNWARD TO THE INDIVIDUAL DISCIPLINE OBSERVATION OPPORTUNITIES EXECUTIVES. THE BASIC FUNCTION IS TO IDENTIFY THE APPLICABLE DISCIPLINE(S) IMPLIED BY THE USER INPUT OBS OPPS DEFINITIONS. THE APPLICABLE DISCIPLINE EXECUTIVES ARE ACTIVATED TO PROVIDE THE DETAILED OBS OPPS DEFINITION. OTHER MODULES ARE POLLED TO IDENTIFY ANY POSSIBLE CONSTRAINTS. THE EXECUTIVE SHALL PROVIDE A USER FRIENDLY INTERFACE WITH BUILT IN TRAINING AND EXPLANATION FEATURES. IT SHALL BE ABLE TO ORGANIZE AND MANIPULATE THE OBS OPPS SETS FOR BEST FIT EVALUATIONS BY THE USER. IT SHALL BE ABLE TO TAG SELECTED SETS FOR LATER RECALL BY THE MISSION PLANNER.</p> | <p>3.7 TOP LEVEL SPECIAL SPECIAL OBS OPPS EXECUTIVE</p> <p>3.8 EXTRACT USER OBS OPPS RQMITS DATA</p> |

SS MPS SOFTWARE REQUIREMENTS SUMMARY

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| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK |
|-----------------------------|--------------------|---|--------------------------------|
| USER MSN PLNG EXEC | NEW | <p>PHASE I: A HIGH LEVEL EXECUTIVE THAT PROVIDES A USER FRIENDLY INTERFACE FOR SS USERS TO THE USER MISSION PLANNING SOFTWARE RESIDENT AT THE PLANNING CENTERS. THE EXECUTIVE WILL PROVIDE CURRENT MISSION INCREMENT INFORMATION SUCH AS DATES/TIMES WHEN USER RQMTS/DETAILED SCHEDULES MUST BE COMPLETED, A GENERAL OVERVIEW OF THE MISSION PLANNING PROCESS AND PROVIDE A HIGH LEVEL GUIDE TO THE USE OF THE APPROPRIATE MISSION PLANNING SW MODULES. THE USER SHALL BE ABLE TO DIALOG WITH THE EXECUTIVE VIA A NATURAL LANGUAGE INTERFACE. THE EXECUTIVE MUST BE ABLE TO MAKE ASSUMPTIONS RELIABLY AND PERFORM CONSISTENCY AND CONSTRAINT CHECKING ON ALL USER INPUT/OUTPUT. IT MUST BE CAPABLE OF RECOGNIZING ITS KNOWLEDGE DOMAIN LIMITATIONS AND REQUESTING USER/OPERATOR ASSISTANCE WHEN REQUIRED.</p> <p>PHASE II: ADD THE CAPABILITY FOR THE EXECUTIVE TO EXTRACT THE REASONING BEHIND THE INPUT DECISIONS SPECIFIED BY THE USER. THESE REASONS WILL BE ENCODED AND MODELED FOR USE IN AN ADVISORY CAPACITY DURING THE MISSION REPLANNING CYCLE.</p> | |

| SS MPS SOFTWARE REQUIREMENTS SUMMARY | | | | PAGE 13 |
|--------------------------------------|--------------------|---|----------------------------------|---------|
| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK | |
| SOLAR EXEC | NEW | <p>EXPERT SYSTEM EXECUTIVE THAT AIDS THE USER/MSN PLANNER IN DEFINING SOLAR OBSERVATION RQMTS AND CALCULATING OBSERVATION OPPORTUNITIES. SYSTEM MUST PROVIDE A USER FRIENDLY INTERFACE WITH ON-LINE HELP AND EXPLANATION FEATURES. THIS EXECUTIVE SHALL SELECT AND SEQUENCE THE APPLICABLE CALCULATION ROUTINES (LTO, TARGEN). ACTUAL ROUTINE CALLS SHALL BE TRANSPARENT TO THE USER. THE USER SHALL BE PROVIDED WITH A "GENERIC" SET OF INPUT DEFAULT VALUES THAT ARE CONSTANTLY UPDATED BASED ON USER INPUT TO OTHER FIELDS. THIS PROVIDES A WORKING MODEL OF USER OBS OPPS REQUIREMENTS REGARDLESS OF THE AMOUNT OF USER DEFINITION. THE EXECUTIVE SHALL HAVE THE CAPABILITY TO RECOGNIZE REQUESTS OUTSIDE ITS CURRENT KNOWLEDGE DOMAIN AND REQUEST ASSISTANCE FROM THE USER/OPERATOR.</p> | 3.2.3 EXEC FOR SOLAR OBS OPPS | |
| OUTPUT PROCESS- OR EXEC | NEW | <p>EXECUTIVE PROGRAM THAT AIDS THE MISSION PLANNER IN THE USE OF THE PTS, PCAP, AND SUMMARY PCAP PROGRAMS. MUST INTERFACE UPWARDS WITH THE PLANNING CENTER AND PAYLOAD OPERATIONS INTEGRATION CENTER SYSTEM EXECUTIVES.</p> | | |

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| SW MODULE NAME | NEW OR MODIFIED | SW MODULE FUNCTIONAL DESCRIPTION | APPLICABLE SUBFUNCTION/TASK |
|----------------------|--------------------|--|-----------------------------------|
| URDB I/F | NEW | <p>EXPERT SYSTEM THAT GUIDES/PROMPTS USERS IN ENTERING FUNCTIONAL REQUIREMENTS INTO THE DATA BASE TO PROVIDE MISSION PLANNERS THE APPROPRIATE INFORMATION FOR PLANNING AND SCHEDULING. THE SYSTEM SHOULD ALLOW INTER-ACTIVE FORM EDITING BY THE USER WITH ON-LINE HELP, DATA ENTRY RULES AND MEANINGFUL DEFAULT VALUES. THE TYPES OF REQUIREMENTS TO BE INCLUDED ARE: RQD RESOURCE VECTORS (POWER, CREW, THERMAL, DATA, ETC.); RQD OBSERVATIONS DEFINITIONS; OPERATIONAL CONSTRAINTS (INHIBITS, ETC.); SEQUENCING, CONCURRENCY RQMTS; AND MIN/MAX # OF PERFORMANCES, DURATIONS. THE DB INTERFACE MUST PROVIDE THE CAPABILITY TO RECOGNIZE REQUESTS/INPUTS OUTSIDE OF ITS KNOWLEDGE DOMAIN AND REQUEST HUMAN EXPERT ASSISTANCE WHEN KNOWLEDGE BASE IS INADEQUATE. THE SYSTEM SHOULD BE ABLE TO INTELLIGENTLY UPGRADE DEFAULT VALUES BASED UPON LATEST INPUT DATA FROM USER. THE SYSTEM SHOULD GENERALIZE LOWER LEVEL DETAILS INTO UPPER LEVEL REQUIREMENTS. ALL REQUIREMENTS SHOULD BE CHECKED FOR CONSTRAINTS IN ALL SIX DISCIPLINES. OUTPUT REQUIREMENTS SHALL HAVE ALL ASSUMPTIONS NOTED AND CONFIDENCE FACTORS (E.G. 90%) ASSIGNED.</p> | 4 USER REQUIREMENTS DATA BASE I/F |

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Section 7

TASK 4 - INVESTIGATE ARTIFICIAL INTELLIGENCE APPLICATIONS

7.1 ACTIVITIES AND ACCOMPLISHMENTS

The objectives of this task were to:

- (1) Define AI techniques that could be applied to SS MPS tasks.
- (2) Identify and evaluate all tasks that could use the AI techniques.
- (3) Recommend a methodology for implementation of the identified AI tasks.

These objectives were accomplished as illustrated in Figure 7.1-1. Two areas of effort contributed to accomplishment of the objectives specified above. The first effort was to conduct a survey of the current AI technology. The second effort was to compile a list of desired criteria for an AI software development program. Both efforts increased the quality and scope of the recommended hardware and software methodology.

7.2 DEFINITION OF ARTIFICIAL INTELLIGENCE

Artificial Intelligence is the emulation of human intelligence and thought processes by computational models. It is the branch of Computer Science concerned with designing intelligent computer systems that exhibit the characteristics associated with intelligence in human behavior - reasoning, understanding language, solving problems, etc.

Expert systems are AI programs that are designed to execute a highly specialized and difficult task with the proficiency of a human expert. They employ domain-specific problem-solving strategies as opposed to broad, general-purpose strategies.

7.3 SURVEY OF AI TECHNOLOGY

A limited survey was conducted of the efforts of various companies and Government agencies to summarize the type of problems that were being solved with AI techniques and the degree of success in their performance. Three areas of the technology were addressed: expert systems, natural language interfaces, and automatic programming. Expert systems were categorized into training and instruction, trending and prediction, design and configuration, information and data interpretation, and planning and scheduling.

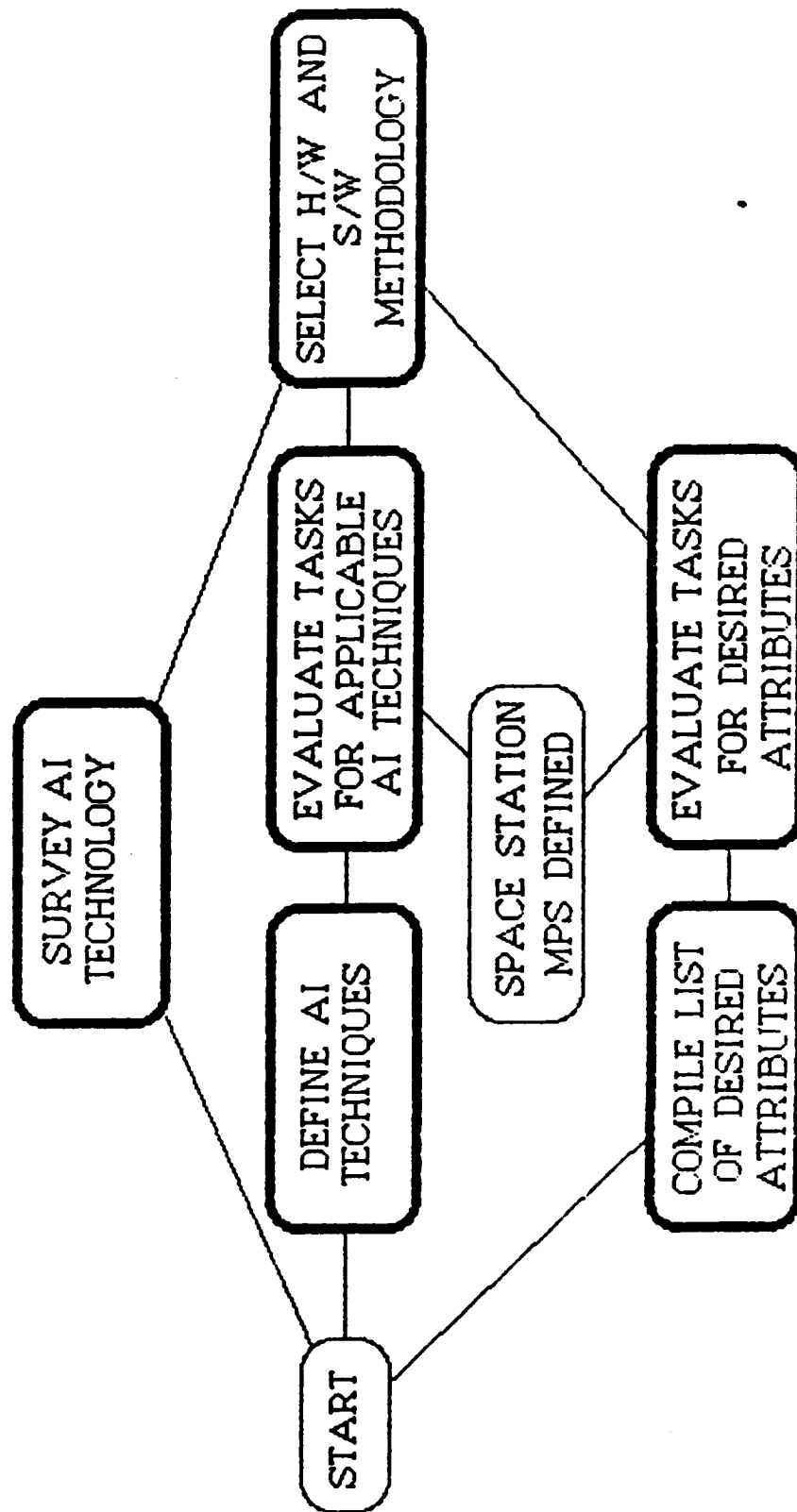
7.3.1 Expert Systems

7.3.1.1 Training and Instruction

The Army Missile Command Research and Development Center successfully deployed an expert system to train operators in the use of an air defense system. It was prototyped with a commercially available expert system tool on an IBM PC and then ported to a larger mainframe. The system performs off-line simulations to improve operator proficiency and provide advisory functions during actual real-time situations.

FIGURE 7.1-1

TASK 4 FLOW CHART



7.3.1.2 Trending and Prediction

The KNOMES system designed by MDAC-Huntington Beach is a hierarchical object-oriented program that performs fault isolation, correction and prediction. It is implemented in ADA on the VAX and has been tested on the SS Data Management System test bed at NASA JSC.

The Systems Autonomy Demonstration Program funded by NASA Ames Research Center has designed several subsystems using autonomous control modules.

Autonomous control is the goal of the NASA Goddard Research Center in designing controllers for the Space Station power distribution system.

The Navigation Subsystem Technical Assistant, designed by Boeing Seattle for the USAF, monitors GNC data and provides recommended actions to the human operator. It was implemented in prototype on an IBM PC using the Personal Consultant shell from Texas Instruments.

The STALEX system designed by NASA JSC performs launch window selection given the many time-dependent factors of orbital mechanics and ground tracking site availabilities.

7.3.1.3 Design and Configuration

The most renowned configuration expert system in the industry is Digital Equipment Corporation's XCON, formerly called R1. It is implemented in OPS5 on the VAX. XCON specifies a detailed computer hardware configuration, including integration and test instructions, from an input of customer requirements.

The KATE system from NASA KSC is currently being prototyped to capture the design knowledge of the existing LES expert system. LES allows the generator to access all electrical schematics of the LOX fueling system of the Shuttle. The KATE system will allow a higher level of user interface to this data base and promote faster electrical redesign. KATE is being implemented on IBM PC/AT's.

The HICLASS system from Hughes Aircraft is tailored to CAD/CAM applications. It was originally coded in FORTRAN and SPL, later converted to PASCAL running on the HP 3000. It has lately been re-coded into C on the Apollo workstations under the Unix operating system.

7.3.1.4 Information and Data Interpretation

The EAGLE system, developed at INCO (a MDAC subsidiary), processes large quantities of numerical and qualitative data to provide advisory information to Air Force operators in the NORAD system. The initial prototype was implemented in LISP using KEE on the Symbolics machine.

The XSEL system in use at DEC is the front end processor for the XCON system mentioned above. It interprets multifformat input data from the customer and outputs standardized configuration requirements.

7.3.1.5 Planning and Scheduling

The KNOMES system from MDAC-HB includes several expert system routines devoted to planning and scheduling for the Space Station.

The MARS system, formerly PLANNET, from MDAC-Kennedy Space Center Division, has been implemented in several prototypes to schedule Shuttle payload integration operations for the entire O&C building. It is implemented in LISP on a Symbolics machine.

The PLAN-IT system from NASA JPL was derived from Voyager mission experiment scheduling. Modules applicable to Spacelab mission planning to are currently being sought. PLAN-IT is implemented in LISP on a Symbolics machine.

The MAESTRO system for Martin Marietta Corporation addresses the problem of experiment scheduling for the Space Station. It is implemented in LISP on a Symbolics machine.

The Space Station Expert System from Lockheed in Houston is a scheduling system prototype to provide on-board advice to operators for reconfiguring resources to meet a hazardous or unexpected event.

7.3.2 Natural Language Interfaces

Intellect, from Artificial Intelligence Corporation, was one of the first NLI's available for information retrieval from existing data bases in finance, businesses and marketing. Lifer is a follow up to the Intellect tool. It facilitates queries to conventional data bases.

Language Craft is a tool available from Technowledge. It is implemented on the Symbolics machine.

Chat-80 is a NLI developed at Stanford and marketed by SRI. It is implemented in Prolog on the Symbolics machine.

Savvy is a tool from a MDAC subsidiary that is implemented on an IBM PC/XT.

7.3.3 Automatic Programming

The ABE system, being developed by Technowledge for DARPA, is a comprehensive attempt to gather an assortment of existing AI tools, languages and techniques, and to develop a system that will allow an operator to assemble expert systems at a high level. The languages contained thus far are: Common LISP, MRS, Knowledge Craft and SI. The logic frameworks include: Blackboards, Data Flow Paths, Intermodule Transaction, and Data Importer. The total system is still in a prototype phase.

7.4 ASSUMPTIONS PRIOR TO CANDIDATE EVALUATION

Experience gained from the early phases of the project allowed several assumptions to be made prior to evaluation of the SS MPS candidates.

7.4.1 ADA Software

It is assumed that all new non-AI mission planning software tasks will be coded in ADA for compatibility with Space Station program requirements.

All AI techniques can be implemented in LISP, PROLOG or ADA. LISP and PROLOG have only a few advantages over ADA, as explained in subsection 7.7 below.

7.4.2 Specialized AI Hardware

If specialized AI hardware is required, assume a Symbolics architecture. LISP and PROLOG are not viable languages unless executed on specialized AI processors. Symbolics is the best processor currently on the market.

The execution of LISP on coprocessor boards installed in conventional computers is not considered; however, their emergence on the market is imminent.

7.4.3 Conventional Hardware

Assume a DEC VAX architecture for all ADA software implementations.

7.4.4 Candidate Evaluation Criteria

The criteria for candidate evaluation are not discrete. They are frequently interrelated.

The criteria are qualitative rather than quantitative. Also, not all criteria are of equal importance.

The evaluation of each software set against the criteria is subjective. The evaluation is highly dependent on definitive information about AI techniques and Space Station operations concepts.

7.5 DESIRED ATTRIBUTES OF MPS TASKS

This list of desired attributes is based upon industry accepted standards for a software development project. Several attributes have been added or modified to tailor them to software projects employing AI techniques.

The desired attributes for candidate MPS tasks are shown in Figure 7.5-1. Each software set received a "+" if the set contained the desired attribute and a "-" if the attribute was missing and could cause potential problems in the implementation of the task.

7.5.1 Task Domain

Domain Knowledge Base is Bounded and Stable

The knowledge base required to accomplish the task must be bounded to have some defined limits; otherwise, the software data base is

FIGURE 7.5-1

ATTRIBUTES OF MPS TASKS

| | TASK GROUP | | | | | | | | | | | |
|------------------------------------|------------|---|---|---|---|---|---|---|---|---|---|---|
| | A | B | C | D | E | F | G | H | I | J | K | L |
| TASK DOMAIN | | | | | | | | | | | | |
| Domain is bounded and stable | + | - | + | + | - | - | + | + | + | + | - | - |
| Domain is specialized and detailed | + | + | + | + | + | - | - | + | + | + | + | + |
| TASK EXPERTISE | | | | | | | | | | | | |
| Expertise to be lost | + | + | + | + | + | + | - | + | + | + | + | + |
| Expertise is scarce | + | + | - | + | + | + | - | + | + | + | + | + |
| Single point expert | + | - | + | + | - | - | - | + | + | + | + | + |
| Expert is dedicated | + | + | + | + | - | - | + | + | + | + | + | + |
| TASK INTERFACES AND METHODS | | | | | | | | | | | | |
| System can monitor real world | + | - | + | + | - | - | - | + | + | + | + | + |
| I/O and methods can be defined | - | - | + | - | + | - | - | + | + | + | + | + |
| Debugging the software | + | + | + | + | - | - | + | + | + | + | + | + |
| ORGANIZATIONAL ISSUES | | | | | | | | | | | | |
| Required Documentation | + | + | + | + | + | + | + | + | + | + | + | + |
| Configuration control | + | + | + | + | + | + | + | + | + | + | + | + |
| System acceptance testing | + | - | + | + | - | - | + | + | + | + | + | + |
| MANAGEMENT ISSUES | | | | | | | | | | | | |
| Realistic schedules and milestones | + | + | + | + | - | - | + | + | + | + | + | + |
| Resource commitment | + | + | + | + | + | + | + | + | + | + | + | + |
| Low initial cost | + | - | + | - | - | - | + | + | + | + | + | + |
| Long term manhour savings | + | + | - | + | + | + | - | + | + | + | + | + |
| PROPOSED USERS OF TASK | | | | | | | | | | | | |
| User acceptance | + | - | + | + | - | - | + | + | + | + | + | + |

SOFTWARE SETS

A - SPECIAL OBS OPPS EXECUTIVES
B - USER REQUIREMENTS DATA BASE INTERFACE
C - EDITOR EXECUTIVES
D - RESCHEDULER
E - SYSTEM EXECUTIVES PHASE I
F - SYSTEM EXECUTIVES PHASE II

G - COMMAND PLANNER
H - NEW TIMELINE SOFTWARE
I - MODIFIED TIMELINE SOFTWARE
J - MODIFIED ORBITAL MECHANICS SOFTWARE
K - MODIFIED DATA FLOW SOFTWARE
L - OUTPUT PROCESSOR EXECUTIVE

never complete and a goal state for project completion is impossible to define. Object oriented programming styles can alleviate this problem somewhat by providing a workable system by declaring objects or modules to deal with input that is outside the current domain. These objects would contain generic methods and generic rules that cover all possible cases. This will allow the system to "soft fail" when confronted with an inquiry outside the task's domain. However, this type of program is likely to be in a state of constant revision.

The knowledge base must also be stable; otherwise, when the system is released, it is already out of date. Frequent mandatory updates to the knowledge base detract from the manpower savings realized from initially automating the task.

Domain Knowledge Base is Specialized and Detailed

Assuming that the task domain is bounded, the ideal domain should consist of specialized knowledge instead of a broad expanse of general knowledge. Specialized knowledge usually lends itself to representation using one or two programming techniques, thus reducing the modeling task complexity.

Detailed knowledge implies that the task contains some expertise (is not a trivial problem), and is therefore worth the effort to code the task.

7.5.2 Task Expertise

Expertise to be Lost

If the expert now performing the task will soon be retiring, advancing, etc., and it will be difficult and expensive to train another expert, then automation may be justified.

Expertise is Scarce

If the expert could be useful in many different locations at the same time, then automation and duplication may be justified.

Single Point Expert

A few people, or preferably one person, must be designated as the domain expert. Multiple experts cause problems such as conflict of information, and organization of segments of knowledge from different experts.

Expert is Dedicated

The expert must be able to suspend his normal duties when needed to assist on the project. This may be difficult since true experts usually have a high demand for their time. Of course, the expert must possess the communications skills to have his knowledge encoded correctly and possess the patience to verify that the system performs correctly. The expert must be interested in the success of the project.

7.5.3

Task Interfaces and Methods

System Can Monitor State of the Real World

All significant communications from the user must be capable of capture by the system. For example, current systems are not capable of capturing facial expressions and voice inflections of a human. The system will contain a model of the real world (within its domain limitations) that it can use to formulate responses. This model must be easily updated by the user (keyboard, voice recognition, etc.). Manual data entry is not a task that humans perform efficiently; preferable interfaces are with automatic stimuli (data stream from other computers, sensors, etc.).

Input, Output and Methods Can be Defined

The expert or pool of experts must be able to define "acceptable" input and output. An expert(s) must be able to define the scope of the task and the methods used to perform the task. The AI capability of rapid prototyping can be cost effective in the early phases of requirements definition to define the methods.

Debugging the Software

During the design phase, unexpected responses by the software are still difficult to detect and isolate. In conventional code, paths of procedural flow may occur that the designer never intended nor had perceived. This is becoming less of a problem for conventional software with improved editors and debuggers. In LISP and especially PROLOG, similar bugs may exist as loops in the knowledge base which cause incorrect assumptions. LISP and PROLOG editors and debuggers are also very powerful and improving. This problem seems to be based on the complexity of the task rather than the choice of software language for implementation.

7.5.4

Organizational Issues

Required Documentation

Automation reduces the amount of documentation required by the user to complete the task, but it necessitates creation of a new set of documentation for maintenance of the new hardware and software. This documentation must describe in detail the implementation of the task on the machine.

Documentation also includes comment lines within the source code. Contrary to rumors about the readability of LISP and PROLOG, these AI languages must contain complete comments to code just as in conventional languages.

Note in Figure 7.5-1 that is an applicable desired attribute for all MPS tasks; therefore, it does not, in effect, serve to identify automated over manual or AI over non-AI tasks.

Configuration Control

Machine hardware and software and the documentation must all be kept in a known state to all users and development and maintenance personnel. Revisions to the hardware and software must be controlled and tracked. A configuration management system is established after early prototyping but prior to Preliminary Design Review. That system continues throughout the life of the project.

This desired attribute is also applicable for all MPS tasks and therefore does not serve to identify automated over manual or AI over non-AI tasks.

System Acceptance Testing

Initial release of the system must be accompanied by testing adequate to provide confidence that the system performs as expected. The test cases used are typically "worst case" or "average" scenarios. If the range of real world problems that the system will encounter is difficult to approximate, then the amount of acceptance testing will be very large in order to obtain a satisfactory level of confidence in the system's performance.

7.5.5 Management Issues

Realistic Schedules and Milestones

A realistic schedule for project completion should contain adequate time for all phases of software development and requirements definition and design. (For example, adequate time may not exist to clearly define the methods to be automated to accomplish MPS tasks previously performed manually.) Significant milestones should be established at the beginning of the project. The level of system performance at these milestones should be well defined to avoid ambiguity about the progress of the system.

The final acceptance milestone should include the explicit definition of "project success".

All milestones serve to verify that the system is developing toward the desired target and to rekindle controlling management's interest and awareness in the project.

Resource Commitment

The necessary resources must be committed, by management, to the project. Budget must be allocated for hardware and software purchases, adequate facilities must be designated, and necessary manpower skills must be committed. (It is assumed that is an applicable desired attribute for all MPS tasks and therefore does not serve to identify automated over manual or AI over non-AI tasks.)

Low Initial Cost

Automation frequently requires a large "up front" investment of capital equipment and man hours. It may take several years of savings from

automation to recover the initial cost. Obviously a low initial cost is preferred to a high one.

Long Term Manhour Savings

The goal of automation is to alleviate humans of the mechanics of performing a task, that they may spend their time in a more cost efficient task. The time required by the user to operate and maintain the system must not exceed the time required to do the task manually for automation to be considered successful.

7.5.6 Proposed Users Of Task

User Acceptance

Unless the system is accepted by the end users it will be ignored and abandoned.

The ideal delivered system should fit into the user's daily routine, impose few new requirements, and demand little or no training in its use or interpretation.

There must be an efficient feedback method from the users to the system designers and maintainers. Ideally the users should maintain the system.

The users must trust the system output. This can be facilitated by heavily involving the users in the design process. The Explanation Capability of AI systems is a good technique to enhance credibility in the eyes of the users.

7.6 ARTIFICIAL INTELLIGENCE TECHNIQUES

An attempt was made to comb through the many books describing AI techniques and pull out the techniques that demonstrate advantages over conventional programming techniques.

The definition of an AI technique versus a conventional technique is subjective and a source of disagreement within the programming community. The boundary between the two is constantly shifting. Many AI techniques were first implemented in LISP or PROLOG and then found their way to conventional implementations in FORTRAN, PASCAL or C. For our definition, AI techniques are most easily implemented in ADA, LISP or PROLOG, while implementations in FORTRAN, etc., are considered to be strictly conventional. Note that ADA holds the middle ground, being a derivative of PASCAL and FORTRAN, but designed to easily implement complex AI techniques.

The following paragraphs describe the AI techniques identified as advantageous over conventional programming techniques. These techniques are listed on Figure 7.6-1. The functions of each software set were evaluated against the list and given a "+" if any of the task functions could be implemented using an AI technique.

FIGURE 7.6-1

AI TECHNIQUES FOR MPS TASKS

| | TASK GROUP | | | | | | | | | | | |
|-------------------------------------|------------|---|---|---|---|---|---|---|---|---|---|---|
| | A | B | C | D | E | F | G | H | I | J | K | L |
| REPRESENTATION OF KNOWLEDGE | | | | | | | | | | | | |
| Production Rules | + | + | + | + | + | + | + | | + | | + | + |
| State space representations | | | | + | | + | | | | | | |
| Frames, Object oriented programming | + | + | + | + | + | + | + | | | | + | + |
| Scripts | + | + | + | + | + | + | + | | | | | |
| Semantic nets | | | | | + | | | | | | | |
| MANIPULATION OF KNOWLEDGE | | | | | | | | | | | | |
| Abstraction | + | + | + | + | + | + | + | | | | | |
| Inheritance | + | + | | + | + | + | | | | | + | |
| Pattern matching | + | + | + | + | + | + | + | | | | | |
| Augmented transition networks | | | | | + | | | | | | | |
| Chaining | | + | | + | + | + | | | | | + | |
| CONTROL STRATEGIES | | | | | | | | | | | | |
| Demons/Methods | + | + | + | + | + | + | + | + | + | + | + | + |
| Blackboards | + | + | + | + | + | + | | | | | | |
| UNCERTAINTY MANAGEMENT | | | | | | | | | | | | |
| Fuzzy logic | + | + | + | | + | + | + | | | | | |
| Dempster shaeffer theory | | + | + | | | + | | | | | | |
| Baysian inference | | | | | | + | | | | | | |
| AUTOMATIC PROGRAMMING | | | | | | | | | | | | |
| Module selection and sequencing | + | | + | | + | | | | | | | |
| Learning capability | | | | | | + | | | | | | |
| EXPLANATION CAPABILITY | | | | | | | | | | | | |
| META KNOWLEDGE | + | + | + | | + | + | + | | | | | |
| NATURAL LANGUAGE INTERFACES | | | | | | | | | | | | |
| DESIGN CAPTURE | | + | | | | + | | | | | | + |
| | | | | | | | | | | | | |
| | | | | | | | | | | | | |

SOFTWARE SETS

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Representation of KnowledgeProduction Rules

Rules are useful for representation of decision trees, i.e., "if/then" statements to be accessed only when the precedent matches the real world model. This technique exists in all the large mainframe AI tools and it is the principal technique used in many PC-based tools.

Applications in the SS MPS include rules for experiment model building, software module selection and sequencing, scheduling, output formatting, logic manipulations and generic rules for processing data outside the current domain.

State Space Representations

Space state representations are useful to represent domains with a large number of input criteria and a large number of acceptable outputs. This technique is implemented in several expert systems used for route finding, chess playing, etc.

Applications in the SS MPS include rescheduling strategies, and mapping of logic solution paths in experiment integration.

Frames/Object Oriented Programming

Object oriented programming is an organization technique that divides a large "master" program into subprograms, similar to FORTRAN subroutines, called modules or objects. The data used by the module is also stored in the module. When dialog is needed between objects they send standardized messages to each other. (See Demons below under Control Strategies.) Objects can be organized in a traditional hierarchical structure but the emphasis is for module autonomy, i.e., controlling functions and decisions are made at the lowest possible level in the hierarchy. This programming technique has proven especially powerful for modeling real world objects and their interaction with other objects. Implementations have been in graphics, animation, and factory floor simulations.

Object oriented programming also facilitates frequent updates to the domain knowledge base since code is localized. This makes modeling feasible for a task that is constantly changing in definition.

Object oriented programming emphasizes the use of calls to common library routines. This results in a substantial reduction of the number of lines of source code required.

This technique has been implemented in languages like SMALLTALK-80, MODULA2, SIMULA-67, CLU, and LISP tools like ART and KEE. Object oriented programming is one of the primary goals of ADA. It can be implemented by data hiding, function hiding, common module libraries, operator overloading and other techniques.

Scripts

Scripts is a set of data patterned after real world scenarios which can be used to provide default values and predict typical responses in a real world situation. Scripts are typically stored in frames (objects) and selected using the pattern matching technique.

Applications in the SS MPS include supplying default values to incomplete user input data, supporting intelligent dialog in a natural language interface, and providing canned planning strategies for default planning by the user.

Semantic Nets

Semantic nets represent objects, actions, or events as nodes and their relationships as interconnecting links. The technique is useful for mapping multidimensional inheritance trees and viewing it from any perspective. The technique is used extensively in Natural Language Interfaces to map text into paraphrases and primitives to be processed by separate routines.

7.6.2 Manipulation of Knowledge

Abstraction

AI languages facilitate the implementation of abstraction by their ability to encode heuristics and logic functions. Abstraction techniques are used to efficiently search through a large set of detailed, possibly incomplete, data to produce sets of possible solutions. Abstraction is a technique which allows the software to create previously undefined configurations from a domain of apparently unrelated facts. In conventional programs all processing paths are predetermined by the programmer which is likely to result in an inefficient search through the possible solution space.

Applications in the SS MPS include: generalization of specific details to a higher level of requirements definition; inference of non-specific English language to specific meaning; and restricting rescheduling in the effort to locate a "best" solution.

Inheritance

Links between objects in object oriented programming allow them to obtain information from their "parent" objects (objects established as "above" them in a hierarchy structure). This significantly reduces the amount of code required at the lower element levels, and assures continuity throughout a branch of related objects.

Applications to the SS MPS include: tailoring dialog in a NLI to a particular person; passing constraints of an experiment down to the step level; filling in meaningful data where scripts don't apply or fall short; and allowing vocabulary modules to inherit meanings from a particular discipline.

Pattern Matching

AI languages have the capability to search through the knowledge base and pick a particular object based upon its content without having to know the object identifier or data type. This eliminates the need for arbitrary naming and data typing necessary in conventional languages.

Applications to SS MPS include activation of rule sets, selection of scripts, and analysis of decisions at state space nodes.

Augmented Transition Networks

When coupled with Semantic Nets this is the most popular technique used in Natural Language interfaces.

Chaining

Forward chaining (data driven search) and backward chaining (goal directed search) are the two established methods for searching through a state space. Forward chaining is typically used for design and configuration problems. Backward chaining is typically used for diagnostic problems.

Applications to the SS MPS include concept formulation, and generation of queries from the knowledge base to the user.

7.6.3 Control Strategies

Demons/Methods

Demons, or methods, are used with object oriented programming to pass messages between objects (causing the receiver to perform an action upon itself). They are programs that wait for a particular condition to occur.

Applications to the SS MPS include consistency checking input data from the real world against previous input data or known conditions, and activation of rule sets to handle queries outside of the current domain.

Blackboards

Blackboards is a technique used widely in "Sensor Fusion", allowing separate routines to post their proposed solutions in a global or restricted data area to be accessed by other routines for constraint monitoring. This technique serves as a "checks and balances" technique to verify that no routine exceeds its authority in making decisions or assumptions. This strategy of executing routines based upon the contents of the blackboard differs from the hierarchical control strategy of conventional programs in which each program has a predefined and limited set of possible activators.

Applications to the SS MPS include constraint checking between large modules of varying conditions, i.e., intermodule, interdepartment, interdisciplinary, and interexperiment.

7.6.4

Uncertainty Management

Fuzzy Logic

Values needed by the system can be inferred from values provided by the user or otherwise presently known by the system. This could be useful in supplementing an incomplete data entry by the user. The technique of Fuzzy Logic can be used to perform reliable assumptions. Fuzzy Logic grades or qualifies statements rather than evaluating them to be strictly true or false. The results of Fuzzy reasoning are not as definite as those derived by strict logic, but they cover a wider range of possibilities.

Applications to the SS MPS include tracking of assumptions, and encoding qualitative rather than quantitative information about an object.

Dempster Shafer

Humans sometimes have the uncanny ability to know the "goodness" of a particular solution. The Dempster Shafer technique allows the system to arrive at an overall numeric value, representing total confidence in the final solution, by summing the confidence factors at each decision node of a State Space, Decision Tree, etc. This can not be easily implemented in conventional languages because of the difficulty in tracing the decision path. AI programs supplement this further by their explanation capability that allows the human to view the logic path that produced the decision. Applications to the SS MPS include the generation of confidence factors for a particular solution.

Bayesian Inference

This is a sophisticated technique that deals in probability computations. In Bayesian Inference, the overall probability that a particular assumption is true is based on a computation of the individual probabilities and the conditional probabilities of each assumption prerequisite. This technique provides more accurate confidence factors than the Dempster Shafer technique, but requires an exponentially greater number of computations.

7.6.5

Automatic Programming

Module Selection and Sequencing

The most widely demonstrated technique of Automatic Programming is module selection and sequencing. This technique selects predefined software modules for problem solution and sequences their execution. Module selection and sequencing is typically aided by heuristic rules of operation.

Learning Capability

The capability of expert systems to learn how to solve problems outside of the current domain is based on the ability of programs to create and execute their own code. This process is similar to conventional programs generating strings of text.

The process is somewhat more straightforward in LISP since, in that language, text and data are treated identically, i.e., newly created code appears identical to all other LISP code. With the use of the EVAL statement in LISP the new code can be forced to execute. Since LISP machines have runtime linking and dynamic memory management, the code can execute immediately after its creation. Conventional systems would have to go through the compile/link cycles.

In theory these processes could be used for systems that "learn" and for "automatic programming". In practice, only a few successful and limited applications have been implemented (VLSI design, image processing, animation). More definition is required in the area of programming conventions. Most early attempts have been met with some rather bizarre computational results when attempting to generate code in real time.

Applications to the SS MPS include the ability to successfully encode changing human logic patterns used in the building of a MPS schedule.

7.6.6 Explanation Capability

The ability of AI languages to form links between elements(nodes) facilitates the display of the solution path. This is difficult to implement in conventional languages without a dedicated trace routine. Explanation of which branches were taken and why help reinforce the user's confidence in the system.

In a LISP machine the explanation path is easily traced back through the memory links performed at runtime during the solution of the problem. ADA systems implement explanation via stack pointers.

Applications to the SS MPS include the ability to explain any computation/assumption to the user to build his confidence in the system output.

7.6.7 Meta Knowledge

Problem solving systems contain the complete set of knowledge for their domain. However, they cannot handle problems outside of their domain and are unaware of this inability. When the program is queried by the user, it searches its data base for a solution. Since the domain fails to produce an affirmative answer, the answer returned is "no" instead of the correct answer of "I don't know". Programming in knowledge of "what a system does not know" significantly increases the size of the software. Definition of meta-knowledge is still a research area of AI. Techniques used include "Metadata" and Data Dictionaries.

Applications to the SS MPS include the capability of any module to recognize its own limitations and request assistance from the next higher module in the hierarchy or from the human expert.

7.6.8 Natural Language Interfaces

Natural language interfaces are one of the most successful and active areas of AI technology. Several commercially available software packages have shown adequate vocabularies in bounded domains.

Applications in the SS MPS include interfacing to the individual users to allow them access to the MPS software system.

7.6.9 Design Capture

Much emphasis is given in AI technology to programming problem solutions on a higher level. Conventional programs classically encode the numeric solution to a problem, logically supported by comments to the code. If the design of the coded module must change to meet new requirements it must be done manually, based on the comments and other supporting documentation. The capability of AI languages to directly encode the logic of the problem solution allows programs to encode the requirements more directly and to be more adaptive to requirements changes.

Applications to the SS MPS include appending logic reasoning to objects representing users, experts and experiment/Space Station hardware.

7.7 METHODOLOGY FOR CANDIDATE IMPLEMENTATION

The methodology for hardware and software host selection is illustrated in Figure 7.7-1. The software sets were evaluated against the attributes described below and given a "+" if they exhibited a need for that attribute. They were given a "-" if they had no need for that attribute.

7.7.1 VAX Vs. Symbolics Architecture

Commercial Support of Hardware

Compared to conventional computer manufacturers, very few companies are involved in the sales and service of LISP machines. For overall reliability, maintainability, proven performance, and acceptance by industry, the VAX is the best alternative. In the SS MPS, if a task is time critical, i.e., if machine downtime must be kept to a minimum, then the VAX is the preferred processor.

However, Symbolics machines are constantly improving and the market share does not appear to be diminishing. The LISP computers will not soon become a dinosaur. But, DEC will likely market a co-processor board for its VAX computers in an effort to gain their share of the LISP computer market.

Real Time Environment

VAX processors are fast enough to support real time environments. Some LISP machines are burdened with the problem of garbage collection which is very detrimental in a real time environment.

FIGURE 7.7-1

AI METHODOLOGY FOR MPS TASKS

| | TASK GROUP | | | | | | | | | | | |
|---------------------------------------|------------|---|---|---|---|---|---|---|---|---|---|---|
| | A | B | C | D | E | F | G | H | I | J | K | L |
| VAX vs. SYMBOLICS | | | | | | | | | | | | |
| Commercial support | + | - | + | + | + | + | + | + | + | + | + | + |
| Real time environment | + | - | + | + | + | + | + | + | + | + | + | + |
| Many users | + | + | - | + | + | + | + | + | + | + | + | + |
| ADA LANGUAGE | | | | | | | | | | | | |
| Standardization | + | + | + | + | + | + | + | + | + | + | + | + |
| Size of source code | + | + | + | + | + | + | + | + | + | + | + | + |
| Capability to implement AI techniques | + | + | + | + | + | + | + | + | + | + | + | + |
| LISP LANGUAGE | | | | | | | | | | | | |
| Rapid prototype environment | - | + | - | + | + | + | - | - | - | - | - | - |
| LISP language advantages | - | + | - | + | - | + | - | - | - | - | - | - |
| Tools available | - | + | - | - | + | + | - | - | - | - | - | - |
| PROLOG LANGUAGE | | | | | | | | | | | | |
| Predicate calculus | - | + | - | + | + | + | - | - | - | - | - | - |
| Parallel processing | - | - | - | + | + | + | - | - | - | - | - | - |
| | | | | | | | | | | | | |
| | | | | | | | | | | | | |
| RECOMMENDED METHODOLOGY | | | | | | | | | | | | |
| Deliver on VAX in ADA (no AI) | | | | | | | | + | + | + | + | + |
| Deliver on VAX in ADA (use AI) | + | + | + | | + | + | + | | | | | |
| Prototype on SYMBOLICS in LISP | | + | | + | + | + | | | | | | |
| Deliver on SYMBOLICS linked to VAX | | | | + | | | | | | | | |
| Implement in Spacelab MIPS | + | + | | + | | | | | | | | |
| | | | | | | | | | | | | |

SOFTWARE SETS

A - SPECIAL OBS OPPS EXECUTIVES

B - USER REQUIREMENTS DATA BASE INTERFACE

C - EDITOR EXECUTIVES

D - RESCHEDULER

E - SYSTEM EXECUTIVES PHASE I

F - SYSTEM EXECUTIVES PHASE II

G - COMMAND PLANNER

H - NEW TIMELINE SOFTWARE

I - MODIFIED TIMELINE SOFTWARE

J - MODIFIED ORBITAL MECHANICS SOFTWARE

K - MODIFIED DATA FLOW SOFTWARE

L - OUTPUT PROCESSOR EXECUTIVE

Also, LISP and PROLOG do not execute efficiently on VAX hardware. Neither does ADA execute efficiently on a LISP machine. Therefore the only two reasonable alternatives for software in a delivery environment is LISP on a LISP machine or ADA on a VAX.

Workstations or Timeshared Terminals

LISP machines are usually dedicated workstations with a higher unit cost than the multi-user environments of VAX hardware. VAX computers are the best alternative for support of more than a few users.

7.7.2 ADA Language

Standardization of Software

ADA (ANSI approved MIL-STD 1815A) has been adopted as the language to be used in all Space Station on-board system applications. Some LISP programmers feel that this requirement will be waived for LISP subroutine calls from an ADA supervisor. The official position is not yet known.

ADA also interfaces more easily with conventional code modules like FORTRAN. Whenever interfacing is required between LISP code and conventional code it is usually performed by a hardware interface between two dedicated processors.

LISP is a powerful language well suited to solving some problems; however, there are some arguments against LISP. Since LISP is a relatively young language several dialects exist. There is as yet no industry wide standard, but DARPA has selected Common LISP. Common LISP contains a small subset of the functions available in Zetalisp (used on the Symbolics mainframes). This reduced set of functions limits the power of the language and increases the amount of code the programmer must generate. Since the language is extensible, programmers could build their own library functions. But then the variations within subroutines written by different programmers is a problem during integration of the larger program. This problem seems to be solved by adoption of a standard, such as Common LISP, but the extensibility feature is then lost.

PROLOG is also a powerful language well suited to solving problems in logic. Japan has selected it as the basis for their Fifth Generation Project, a new breed of computers they hope will replace current processors. However, PROLOG is not widely used or accepted in this country.

Size Of Source Code

LISP code is roughly equivalent to the level of detail found in assembly language for ADA. This translates to more lines for a LISP programmer to generate for an equivalent function in ADA. LISP code also requires a larger dictionary of functions. Commercial tools exist for LISP which raise the level significantly, but processing speed is decreased slightly due to the increased software overhead. Language flexibility is also decreased slightly by the rigid structure of the tool.

7.7.3 LISP Language

7.7.3.1 Rapid Prototyping Environment

Less Time In Edit/Compile/Link/Debug Cycle

In conventional hardware file manipulation must be performed between each cycle. On a LISP machine these utilities are all available under the same monitor, so transfer from one cycle to the next is instantaneous.

User Friendly Editor

Many features are easy to implement on LISP machine editors. These include windows, graphics and syntax check and correction.

Dynamic Linking

LISP elements may be manipulated independent of the values of those elements. Declaration of their value is only required just prior to output of the final solution. In conventional programs all variables must have a declared value. No such limitation exists for a LISP machine which performs dynamic linking at runtime.

Interpreters and Compilers

Incremental compilers (interpreters) are efficient, by industry opinion, on LISP machines. Interpreters are also available for conventional hardware, but are very slow and inefficient. LISP interpreters facilitate debugging during compile.

LISP machines also support traditional compilation of source code files. Benchmark tests indicate that compiling increases execution speed by 4 to 30 times and reduces source code size up to 1/3 to 1/10 of original.

Incremental Execution

This utility allows programmers to debug with the editor as errors are encountered during execution and then continue execution.

Incomplete Input Data

LISP listener environments allow programmers to execute programs and have partial solutions returned which contain the undefined data. Conventional programs will not execute without complete input data.

Dynamic Memory Management

Since memory allocations are performed at run time in LISP machines the programmer does not have to declare these as in conventional languages.

Supports Bottom-Up or Top-Down Design

LISP subroutines can be executed and will return a value. Conventional subroutines require overhead software to call the subroutine. This means software modules can be developed and tested at any time, even if the modules above it in the hierarchy are not yet implemented.

LISP will also support top down design in the same manner as conventional languages.

7.7.3.2 LISP Language Advantages

Function Library

12,000 functions currently exist in Zetalisp, enough for nearly every application by today's standards. In addition, LISP is extensible by the programmer. Extensibility allows the user to add new functions to the existing library without having to change the compiler.

Encode Heuristics

LISP lists and elements easily encompass numbers and variables used by conventional programs such as FORTRAN. Therefore, LISP can represent a wider variety of data types. The ability to encode heuristics is possible in ADA but is slightly more straightforward in LISP.

Ability To Implement Recursive Solutions

LISP functions can call themselves without limit. This is difficult to implement in ADA or any conventional languages. This technique has been implemented to represent infinite series mathematical equations, language syntax, and multidimensional organizational trees.

7.7.3.3 Tools Available

LISP tools such as ART, KEE and Knowledge Craft allow the user to code a large task with a limited knowledge of LISP. User interfaces are extremely friendly and interactive. Tools do tend to be a large overhead which uses up memory and slows processing time. But, all three vendors offer production model shells with a reduced amount of code overhead. Some projects like PLAN-IT were forced to code their own inference engine to get the needed execution speed.

7.7.4 PROLOG Language

Predicate Calculus

Predicate calculus is the most widely accepted mathematical language for modeling of logic based problems and theorems. Many of the problems in the field of AI are heavily logic based, so predicate calculus is the natural choice. LISP and ADA can implement predicate calculus equations but implementation is more straightforward in PROLOG.

Parallel Processing

Parallel Processing has been proposed as one possible solution to speed up large complex software programs. However, the serial method of solution in conventional software does not lend itself easily to division into parallel tasks. PROLOG is exceptionally well suited to this division by allocation of each decision node to a processor. The processor could be dedicated, for a massive computer by today's standards, or allocated from a common pool of available processors. The precedent matching technique of rule firings in a production system could also be divided into parallel tasks.

To date, no practical large scale systems have yet been implemented due to hardware limitations. Current efforts of Japan's Fifth Generation Computer project are focused in the field of building such processors. Techniques for programming in PROLOG also need to be refined to reduce the combinatorial explosion problem in the solution search space.

7.8 RESULTS OF EVALUATION

The evaluation of each SS MPS task against the Desired Attributes criteria produced a list of benefits and concerns for the implementation of each software set. These benefits and concerns are summarized in subsection 7.8.1.

The summation and weighing of all evaluations performed previously, resulted in the task methodology recommended for implementation. This recommendation is shown on the bottom half of Figure 7.7-1 and summarized in subsection 7.8.2.

7.8.1 Benefits/Concerns

Each software set is listed below accompanied by its:

Benefits - Those characteristics which will result in the biggest payoff after the task is automated.

Concerns - Those possible pitfalls that must be avoided during project development and implementation.

Set A - Special Obs Opps Executives

Benefits- Since expertise is scarce and it is expensive to train an expert, automation and replication of this task will result in a big payoff. Most functions are easily implemented which will result in a low initial cost and high user acceptance of output data.

Concerns- There may be a minor difficulty in defining a method for modeling a new user defined target currently outside of the knowledge base.

Set B - User Requirements Data Base Interface

Benefits- Since expertise is scarce and it is expensive to train an expert, automation and replication of this task will result in a big payoff.

Concerns- Problems may be encountered in correctly modeling the experiment as described by the user. Definition of the information extraction technique may be difficult. The user may become easily frustrated if the system fails to dialog intelligently with him. The cost of software requirements definition will be high due to the prototyping phase and the cost of a LISP tool. There is no "generic" or "worst case" acceptance test available so acceptance tests will have to be numerous.

Set C - Editor Executives

Benefits - Automation would significantly speed up this process in all cycles of planning and save significant manpower.

Concerns- None.

Set D - Rescheduler

Benefits - Since expertise is scarce and it is expensive to train an expert, automation of this task will result in a big payoff. Replication of this task will also result in a big payoff since it is needed at seven points in the MPS phases of planning. Interfaces are readily defined since they are all in electronic format. If the software is used only on the ground (planning centers and the Payload Operations Integration Center), and since the task boundary is well defined, it could be executed on a specialized LISP processor and interfaced to the VAX.

Concerns - Several methods currently exist for rescheduling. Deciding on one strategy or set of strategies could be difficult. Experience has shown on several systems that the LISP inference engine must be coded from scratch to obtain acceptable operating speed. This increases the level of effort in prototyping. However, several organizations have already developed working prototypes to address this problem.

Set E - System Executives Phase I

Benefits - Since expertise is scarce and it is expensive to train an expert, automation and replication of this task will result in a big payoff. A Natural Language Interface (NLI) would be a powerful interface tool. Several off-the-shelf commercial tools already exist.

Concerns - Defining and debugging the specialized vocabulary for the NLI will require a large manhour effort. The experts allocated to the task may not be motivated to debug the user interface down to the level of refinement necessary for the user to accept and use the NLI. Since there is no "generic" or "worst case" test for the NLI, acceptance will have to consist of an extensive battery of tests. Several successful NLI's have been developed, but they should still be considered a moderate schedule risk.

Set F - System Executives Phase II

Benefits - Implementation of the Apprentice/Advisor would result in a tremendous manpower savings over the life of the Space Station. It would be the primary step needed for future transfer of the bulk of mission planning to on-board the station.

Concerns - The highly unstable and broad domain may be difficult to model completely and accurately. The expert may be difficult to isolate and motivate to invest the time needed to train the Apprentice. System acceptance tests will have to be extensive and confidence factors tested for Planning Center and Payload Integration Center management to accept the Advisor output.

Set G - Command Planner

Benefits - Information in the SS MPS such as canned typical command timelines may be very useful to the novice user. They impose no restriction on the experienced PI.

Concerns - Since expertise is a plentiful resource for each dedicated PI and the time required to create a command timeline is relatively short, the cost of encoding intelligent software to assist the user may not be cost effective.

Set H - New Timeline Software

Benefits - Tasks to be implemented are straightforward and do not require a large use of AI techniques.

Concerns- none.

Set I - Modified Timeline Software

Benefits - Tasks to be implemented are straightforward and do not require a large use of AI techniques.

Concerns - none.

Set J - Modified Orbital Mechanics Software

Benefits - Tasks to be implemented are straightforward and do not require a large use of AI techniques.

Concerns - none.

Set K - Modified Data Flow Software

Benefits - This task could be more flexible to a changing Space Station configuration environment by using Object Oriented Programming to model the individual hardware elements and the constraints and interactions between the elements and the station.

Concerns - The major portion of the task is already coded (SSDFAST in FORTRAN), therefore recoding into ADA would not be cost effective.

Set L - Output Processor Executive

Benefits - This task could be more flexible to a changing Space Station configuration environment by using Object Oriented Programming and Design Capture to model the requirements for printing/display formats and the constraints on each module of information.

Concerns - The major portion of the task is already coded (PCAP and PTS in FORTRAN), therefore recoding into ADA would not be cost effective.

7.8.2 Methodology Summary

Fourteen tasks were selected as candidates for using AI techniques. Thirteen tasks are recommended to be delivered in ADA on the VAX.

One task is recommended to be delivered on the Symbolics in LISP with a hardware interface to the VAX. At a future date it should be ported to the VAX prior to installation on-board the Space Station.

Machine. Four tasks are recommended for prototyping on the Symbolics

MIPS. Three tasks are recommended for implementation in the Spacelab

7.9 CONCLUSIONS AND RECOMMENDATIONS

7.9.1 AI Technology

AI technology is still very young. The experience base of expert systems performance is small compared to conventional programs. However, the systems in existence do strongly support the many advantages of incorporating this technology into the workplace. AI has proven effective in solving many of the problems where conventional programs fail.

7.9.2 Hardware/Software Architecture

The conclusion to largely use ADA on a VAX is also supported by a study conducted by MDAC-HB for the JSC Space Station Phase B contract.

The largest value of LISP and PROLOG is in the rapid prototyping environment.

7.9.3 Software Tools

Use is recommended during prototyping of an expert system development tool and a natural language development tool.

An in-depth technology survey, with the targeted MPS candidates in mind, should be performed immediately prior to purchase of any off-the-shelf AI tools.

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Section 8

TASK 5 - SOFTWARE DEVELOPMENT PLAN

The objective of this task was to generate a Software (SW) Development Plan for the definition, design and implementation of the SS MPS.

The approach taken to this task consisted of four subtasks. First, assumptions inherent in the generation of the SW Development Plan were identified; these pertained to SW development facilities, computer operating systems, coding languages and standards, required formal reviews, required documentation, etc. The second subtask involved developing a technical description of the project - SW requirements, SW hierarchy, etc., and a detailed description of the activities required to successfully complete the development project. Based on the assumptions of subtask 1 and the descriptions of subtask 2, subtask 3 was performed to generate cost estimates for individual or sets of required SS MPS computer programs in terms of manpower and schedule using the Constructive Cost Model (COCOMO), and integrating these into project level manpower requirements and schedule recommendations. The fourth and final subtask was to document and publish the SW Development Plan.

Inputs to this study task were derived from:

- Task 3 products (SS MPS Functional Flows and SW Requirements Summary)
- Task 4 products (AI recommendations and implementation requirements)
- COCOMO Model
- Existing SW development plans (boilerplates)

The product of this task is the SS MPS SW Development Plan, which constitutes Volume III of the Study final report. In summary, the SW Development Plan documents requirements for a 4841 manmonth effort over a 64 month period to successfully complete the SS MPS software development project.

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Section 9

CONCLUSIONS AND RECOMMENDATIONS

Based on the results of the SS MPS Development Study presented in the previous section, the following conclusions have been drawn:

- 1) A detailed definition of the Spacelab payload mission planning process and SL MIPS software has been derived; this definition (functional flow diagrams and data base) will be of great value for training Spacelab mission planning personnel and for assessing and improving the process.
- 2) A baseline concept for performing SS manned base payload mission planning has been developed; this concept is consistent with current Space Station design/operations concepts and philosophies; however, those concepts and philosophies are the results of Phase B studies and will therefore gain further definition and changes as the Space Station Program progresses.
- 3) SS MPS software requirements have been defined. These software requirements make maximum use of SL MIPS software with modifications, but do include requirements for new software to accommodate the complexity of the SS mission planning concept and to maximize automation of the concept. Also, requirements for new software include candidate programs for the application of AI techniques to capture and make more effective use of mission planning expertise and to involve SS users directly in the mission planning process.
- 4) A SS MPS Software Development Plan has been developed which phases efforts for the development of software to implement the SS mission planning concept. The efforts are phased for the immediate start of development of long-lead-time software programs, but for delayed development of programs with a high dependence on SS design/operations concepts. The development schedule, relative to the current overall Space Station Program schedule, indicates the development effort should begin as soon as possible.
- 5) The estimated manpower requirements to develop the SS MPS are significant; however, the scope of the SS mission planning problem is significant and the process of development is recommended to be highly structured and rigidly controlled. Nonetheless, the software system concept is intended to provide uniform methods of planning payload operations across all equivalent planning levels in order to facilitate the integration of planning, and is intended to maximize the automation of mission planning to minimize long-term mission planning costs.

Based on the conclusions above, the following recommendations are offered:

- 1) Use the definition (functional flows and data base) of the Spacelab payload mission planning process and software to train mission planning personnel and to evaluate and improve the process. As improvements are made, update the flow diagrams and data base.

2) Proceed with implementation of the SS MPS Software Development Plan, including the structured and controlled process for software development.

3) Maintain the SS mission planning concept, software system concept, and Software Development Plan consistent with SS design/operations concepts and program schedules.

4) Use Spacelab mission planning as a test bed for testing prototypes of AI applications.

APPENDIX A

SPACELAB MIPS DATA BASE

The SL MIPS data base was developed in order to provide activity summary data, software description and requirements data, and activity time and skill requirements data. The level of detail of the data base is consistent with the level of detail in the Spacelab mission planning process detailed flow diagrams; that is, entries exist in the data base corresponding to each lowest hierarchical level activity (function, subfunction, task or subtask) identified for every function in the flow diagrams. When assessed in conjunction with the detailed flows, the data base provides a comprehensive definition of the Spacelab payload mission planning process.

The data base consists of eight (8) interrelated tables of data:

- o Activity Summary Data
- o Activity Time and Skill Requirements
- o Software Used by Activity
- o Software Description
- o Software Peripherals Required
- o Activity Input/Outputs
- o Computer Input/Output Summary
- o Manual Input/Output Summary

Table 1 provides the activity summary data which identifies an activity and its position in the hierarchy of activities (function, subfunction, task, subtask), the activity objective, method of accomplishment (manual or automatic), and the need for the activity.

Table 2 provides the activity time and skill requirements data which includes, for each activity, skill type and skill level, manpower requirements and throughput calendar time for each cycle the activity is performed. Time here refers to the total amount of time required to accomplish the activity (data collection and assessment, analysis, computer setup time required, and evaluation of results). The mission planning cycles are, in sequence, preliminary (P), basic (B), update (U), and replanning (R).

Table 3 provides the software identification for activities that are automated, and the required computer setup time for each cycle the activity is performed. Time here is inclusive of time required for file updates/edits, runstream development, and software interaction.

Table 4 provides a description and the resource requirements for each software module. This table is linked to other tables in the data base by software name. Data included are: software function definition, mode of operation, skill requirements, language, lines of code, memory requirements, and estimated CPU time. In this table, where a software name is followed by a number (NAME-1), the number links the software to a particular activity in other tables. A software module may be used to accomplish several different activities; the difference in this table is the required CPU time.

Table 5 identifies the interface peripheral required by a user to exercise a particular software module.

Table 6 provides an input/output summary for all the activities performed during the mission planning process. For each activity the following data is provided: input/output name, I/O form (computer or manual), software module association, the I/O type (input or output), source or destination of I/O, and an indication of which planning cycles utilize the I/O.

Table 7 provides summary data for all computer input/outputs. Data included are: input/output name, file size (maximum, minimum) and a brief description of the data contained in the input/output file. For some input/outputs only one entry is made for file size (minimum). These values are provided as an average (typical) file size.

Table 8 provides summary data for all manual input/outputs. Data included are: input/output name, type (form) of input/output (verbal, written, formal or informal document), name of document (if applicable) in which the input/output is published, and a brief description of the data contained in the input/output.

TABLE 1
ACTIVITY SUMMARY DATA

ACTIVITY SUMMARY DATA

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PAGE

1

| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|-------------------------|---|---|---|---|-------------------|---------|
| PAYLOAD DATA COLLECTION | PAYLOAD DATA COLLECTION | PAYLOAD DATA COLLECTION | PAYLOAD DATA COLLECTION | ONCE THE PAYLOAD COMPLEMENT HAS BEEN SELECTED AND DEFINED, PAYLOAD DATA IS COLLECTED AND PROCESSED AS NECESSARY TO PERFORM REQUIRED MISSION PLANNING ANALYSIS. SOURCES OF DATA INCLUDE PAYLOAD COMPLEMENT DEFINITION, ERD'S AND PI INTERFACE. AFTER PAYLOAD DATA HAS BEEN COLLECTED AND EVALUATED INPUTS/UPDATES ARE MADE TO THE IPRD AND O&IA. | MANUAL | ROUTINE |
| ORBITAL ANALYSIS | ORBIT REQUIREMENTS EVALUATION AND SELECTION | ORBIT REQUIREMENTS EVALUATION AND SELECTION | ORBIT REQUIREMENTS EVALUATION AND SELECTION | SELECT A MISSION ORBIT WHICH BEST MEETS SCIENCE OBJECTIVES AND SYSTEMS REQUIREMENTS. THE TASK INVOLVES SELECTION OF AN ORBIT ALTITUDE AND INCLINATION WHICH MEETS REQUIREMENTS/CONSTRAINTS WHILE SATISFYING PHASING REQUIREMENTS, IF ANY. THIS IS NORMALLY A MANUAL TASK, HOWEVER IF A TRADES ANALYSIS IS REQUIRED SOME SOFTWARE MAY BE USED. THE SUBTASKS AND SOFTWARE MODULES USED FOR TRADES ANALYSIS ARE PERFORMED ELSEWHERE IN THE MISSION PLANNING PROCESS. | MANUAL/AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | LAUNCH WINDOW/LAUNCH H TIME SELECTION | LAUNCH WINDOW/LAUNCH H TIME SELECTION | LAUNCH WINDOW/LAUNCH TIME SELECTION | DETERMINE THE AVAILABLE LAUNCH WINDOW BASED ON PAYLOAD/EXPERIMENT AND STS CONSTRAINTS, AND PICK LAUNCH TIME THAT MAXIMIZES LAUNCH WINDOW | AUTOMATIC/MA NUAL | ROUTINE |

ACTIVITY SUMMARY DATA

DATE 04/08/87

PAGE

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|------------------|---|--|--|---|-----------|---------------------|
| ORBITAL ANALYSIS | STATE VECTOR GENERATION/E PHEMERIS DATA DEVELOPMENT | GENERATE STATE VECTOR | GENERATE STATE VECTOR | PROJECT A STATE VECTOR AT THE TIME OF INSERTION. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | STATE VECTOR GENERATION/E PHEMERIS DATA DEVELOPMENT | CONVERT/STOR E STATE VECTOR | CONVERT/STOR E STATE VECTOR | TAKE AS INPUT A LAUNCH DATE AND TIME, AND A STATE VECTOR AS ACQUIRED FROM JSC (IN ENGLISH UNIT), CONVERT TO APPROPRIATE FORM AND UNITS, AND STORE IN CASE STORAGE FILES FOR USE BY ASEP, NSEP, AND/OR TANRAY. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | STATE VECTOR GENERATION/E PHEMERIS DATA DEVELOPMENT | GENERATE ORBITAL EPHMERIS BY NUMERICAL INTEGRATION | GENERATE ORBITAL EPHMERIS BY NUMERICAL INTEGRATION | GENERATE ORBITAL EPHMERIS DATA BY NUMERICAL INTEGRATION OF THE EQUATIONS OF MOTION. THIS TASK CONSIDERS ATMOSPHERIC DRAG FOR MISSIONS (LOW ORBITS) WHERE DRAG IS A FACTOR IN CALCULATING THE EPHMERIS DATA. | AUTOMATIC | IF DRAG IS A FACTOR |
| ORBITAL ANALYSIS | STATE VECTOR GENERATION/E PHEMERIS DATA DEVELOPMENT | GENERATE REQUIRED EPHMERIS DATA FOR SUBSEQUENT ANALYSIS. | GENERATE REQUIRED EPHMERIS DATA FOR SUBSEQUENT ANALYSIS. | | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIES GENERATION | GENERATE MISSION INDEPENDENT TARGETS | GENERATE EARTH SHADOW ACQ/LOSS | DETERMINE THE TIMES WHEN THE ORBITER ENTERS AND LEAVES THE EARTH'S SHADOW. | AUTOMATIC | ROUTINE |

ACTIVITY SUMMARY DATA

DATE 04/08/87

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|------------------|--------------------------------------|--------------------------------------|------------------------------------|--|-----------|---------|
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE MISSION INDEPENDENT TARGETS | GENERATE SUN RISE/SET | DETERMINE THE TIMES WHEN THE ORBITER ENTERS AND LEAVES SUNLIGHT. THIS IS DONE BY TAKING THE COMPLEMENT OF THE EARTH SHADOW ACQ/LOSS TIMES. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE MISSION INDEPENDENT TARGETS | GENERATE MOON RISE/SET | COMPUTE THE ACQUISITION AND LOSS TIMES OF THE MOON AS SEEN BY THE ORBITER IN ORBIT. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE MISSION INDEPENDENT TARGETS | GENERATE PRELIMINARY TDRS COVERAGE | GENERATE PRELIMINARY TDRS COVERAGE BASED ON BEST ESTIMATE (PRELIMINARY) ATTITUDE TIMELINE. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE MISSION INDEPENDENT TARGETS | GENERATE GROUND STATION COVERAGE | DETERMINE THE TIMES THAT THE ORBITER IS IN COMMUNICATION WITH GROUND TRACKING STATIONS. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE MISSION INDEPENDENT TARGETS | GENERATE RADIATION ENVIRONMENT | DETERMINE THE FLUX OF CHARGED PARTICLES (PROTONS, ELECTRONS AT SOLAR MAXIMUM AND ELECTRONS AT SOLAR MINIMUM) GIVEN A SET OF POINTS (ALTITUDE, LATITUDE, LONGITUDE) IN SPACE AND AN ENERGY THRESHOLD. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE MISSION INDEPENDENT TARGETS | IMPOSE RADIATION CONSTRAINTS | DETERMINE OPERATING PERIODS BASED ON ACCEPTABLE RADIATION LEVELS/CONDITIONS. | AUTOMATIC | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|------------------|-------------------------------------|--------------------------------------|---|---|-----------|-------------|
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIES GENERATION | GENERATE MISSION INDEPENDENT TARGETS | MERGE MISSION INDEPENDENT TARGETS INTO ONE COMMON FILE. THESE TARGETS INCLUDE SUN RISE/SET TIMES, MOON RISE/SET TIMES, PRELIMINARY TDRS COVERAGE, GROUND STATION COVERAGE, AND SOUTH ATLANTIC ANOMALY RADIATION CONSTRAINT TIMES (SAA). | AUTOMATIC | ROUTINE | |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIES GENERATION | GENERATE CELESTIAL TARGETS | DEVELOP CELESTIAL TARGETS (NON-IPS MISSIONS) | DEVELOP THE CELESTIAL TARGET FILE. THE TARGET FILE IS MANUALLY DEVELOPED USING THE STAR PROGRAM. | MANUAL | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIES GENERATION | GENERATE CELESTIAL TARGETS | DEVELOP CELESTIAL TARGETS (IPS MISSIONS) | DEVELOP THE CELESTIAL TARGET FILE. THIS IS DONE USING THE SUBCOD PROGRAM TO DEVELOP A SUBSET OF THE PI COBSERVATION FILE. | AUTOMATIC | IPS MISSION |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIES GENERATION | GENERATE CELESTIAL TARGETS | GENERATE CELESTIAL TARGET(S) ACQ/LOSS | DETERMINE THE ACQUISITION AND LOSS TIMES FOR THE CANDIDATE TARGETS. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIES GENERATION | GENERATE CELESTIAL TARGETS | COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | DETERMINE VIEWING PERIODS BASED ON STELLAR OBSERVATION REQUIREMENTS/CONSTRAINTS AND MERGE RESULTANT FILES INTO ONE COMMON FILE. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIES GENERATION | GENERATE ATMOSPHERIC PHYSICS TARGETS | COMPUTE DISTANCE FROM TANGENT POINT ON THE ORBITER TO SUN LINE-OF-SIGHT TO THE EARTH SURFACE | DETERMINE OPPORTUNITIES TO VIEW THE SUN THROUGH CERTAIN LAYERS OF THE ATMOSPHERE. | AUTOMATIC | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|------------------|--------------------------------------|--------------------------------------|---|--|-----------|-------------------|
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE ATMOSPHERIC PHYSICS TARGETS | DEVELOP/APPLY CONSTRAINTS TO ATMOSPHERIC PHYSICS OBSERVATION PERIODS | DEVELOP ATMOSPHERIC PHYSICS OBSERVATION PERIODS WITHIN ACCEPTABLE CONDITIONS. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE ATMOSPHERIC PHYSICS TARGETS | COMPUTE ORBIT TERMINATOR TARGETS | DETERMINE TARGETS AT, OR NEAR, THE TERMINATOR. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE ATMOSPHERIC PHYSICS TARGETS | COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | DETERMINE OPERATING PERIODS BASED ON ATMOSPHERIC PHYSICS REQUIREMENTS/CONSTRAINTS. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE ATMOSPHERIC PHYSICS TARGETS | COMPUTE SUN AZIMUTH AND ELEVATION FROM ORBITING VEHICLE WITH RESPECT TO SUN | DETERMINE THE SUN AZIMUTH AND ELEVATION AS SEEN FROM AN ORBITING VEHICLE WITH RESPECT TO THE SUN | AUTOMATIC | REALTIME FUNCTION |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE SOLAR TARGETS | DEVELOP CONSTRAINTS FOR SOLAR VIEWING PERIODS | DEVELOP SOLAR VIEWING PERIODS WITHIN ACCEPTABLE CONDITIONS. | AUTOMATIC | IPS MISSION |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE SOLAR TARGETS | COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | DETERMINE VIEWING PERIODS BASED ON SOLAR REQUIREMENTS/CONSTRAINTS. | AUTOMATIC | IPS MISSION |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE EARTH OBSERVATION TARGETS | CREATE EARTH SITE DEFINITION FILE | DEFINE POLYGONAL AREAS OF CANDIDATE GROUND TARGET SITES. | AUTOMATIC | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|------------------|--------------------------------------|------------------------------------|---|--|-----------|---------|
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE EARTH OBSERVATION TARGETS | GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | GENERATE ACQUISITION AND LOSS TIMES OF THE DEFINED EARTH TARGET AREAS. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE EARTH OBSERVATION TARGETS | DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | DEVELOP EARTH OBSERVATION PERIODS WITHIN ACCEPTABLE CONDITIONS. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE EARTH OBSERVATION TARGETS | COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | DETERMINE OPERATING PERIODS BASED ON EARTH OBSERVATION REQUIREMENTS/CONSTRAINTS. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE PLASMA PHYSICS TARGETS | COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | COMPUTE THE ORIENTATION AND STRENGTH OF THE MAGNETIC FIELD RELATIVE TO THE ORBITER POSITION. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE PLASMA PHYSICS TARGETS | DEVELOP/APPLY CONSTRAINTS TO BOHB PARAMETERS | DEVELOP PLASMA PHYSICS OBSERVATION PERIODS WITHIN ACCEPTABLE CONDITIONS. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE PLASMA PHYSICS TARGETS | GENERATE HEMISPHERE OPPORTUNITIES | DEVELOP HEMISPHERE OPPORTUNITIES BASED ON LATITUDE CONSTRAINTS. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE PLASMA PHYSICS TARGETS | COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | DETERMINE OPERATING PERIODS BASED ON PLASMA PHYSICS REQUIREMENTS/CONSTRAINTS. | AUTOMATIC | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|------------------|--------------------------------------|---|--|--|-------------------|---------|
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | MERGE ALL EXPERIMENT TARGET FILES | MERGE ALL EXPERIMENT TARGET FILES | MERGE ALL EXPERIMENT TARGET FILES (OPPORTUNITIES) INTO ONE COMMON FILE. THESE TARGET FILES (OUTPUTS) ARE THE RESULT OF ORBITAL ANALYSIS FOR DIFFERENT SCIENCE DIDCIPLINES. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE MATERIAL AND/OR LIFE SCIENCE TARGETS | GENERATE MATERIAL AND/OR LIFE SCIENCE TARGETS | NOTE: MATERIAL/LIFE SCIENCE TARGET PERIODS ARE DEVELOPED/ASSIGNED BASED ON EXPERIMENT OPERATIONS AND SYSTEMS REQUIREMENTS IN THE "MISSION PROFILE GENERATION" SUBFUNCTION. THE PURPOSE OF THIS TASK IS TO SHOW THAT MATERIAL AND LIFE SCIENCE EXPERIMENTS ARE COMMON TO THE SPACELAB MISSION PLANNING SYSTEM. THESE SCIENCE DISCIPLINES ARE NOT DEPENDENT ON ORBIT GEOMETRY, BUT RATHER MORE SO ON LOW-GRAVITY ATTITUDES, AND TDRS COVERAGE. | N/A | N/A |
| ORBITAL ANALYSIS | EXPERIMENT OPPORTUNITIE S GENERATION | GENERATE CO-ORBITING TARGETS | PERFORM PARAMETRIC ANALYSIS TO DESIGN/DEVELOP CO-ORBITING TRAJECTORIES THAT SATISFY OBJECTIVES AND CONSTRAINTS | PERFORM A PARAMETRIC ANALYSIS CONSIDERING THE BASIC CO-ORBITING REQUIREMENTS (SUCH AS DEPLOYMENT, PROXIMITY OPS, GRAPPLE/CAPTURE) AS WELL AS THE PI REQUIREMENTS/CONSTRAINTS AND STS REQUIREMENTS/CONSTRAINTS. | AUTOMATIC/MA NUAL | ROUTINE |
| ORBITAL ANALYSIS | MISSION PROFILE GENERATION | DEVELOP GROSS MISSION TIMELINE | DEVELOP GROSS MISSION TIMELINE | DEVELOP A GROSS MISSION TIMELINE BY EVALUATING EXPERIMENT OPPORTUNITIES AND TAKING INTO CONSIDERATION MISSION PRIORITIES, DIFFICULTY OF EXPERIMENT SCHEDULING, CREW ACTIVITIES, SYSTEMS REQUIREMENTS, AND MANAGEMENT DIRECTION. | MANUAL | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|------------------|---|---|---|--|-----------|---------------------------|
| ORBITAL ANALYSIS | MISSION PROFILE GENERATION | DEVELOP PRELIMINARY ATTITUDE TIMELINE | DEVELOP PRELIMINARY ATTITUDE TIMELINE | DEVELOP A PRELIMINARY ATTITUDE TIMELINE BASED ON THE ATTITUDE REQUIREMENTS FROM THE "AGREED TO" GROSS MISSION TIMELINE. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | DEDICATED STELLAR OBSERVATION GENERATION | CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | READ TARGET AND TARGET DEFINITION FILES AND REFORMAT/EDIT THIS DATA INTO A NEW FORMAT WHICH THE ASTAR PROGRAM WILL ACCEPT. | AUTOMATIC | DEDICATED STELLAR MISSION |
| ORBITAL ANALYSIS | DEDICATED STELLAR OBSERVATION GENERATION | CREATE RESERVE PERIOD FILE (DS) | CREATE RESERVE PERIOD FILE (DS) | CREATE A FILE CONTAINING RESERVED PERIODS FOR TARGETS THAT ARE DIFFICULT TO VIEW, CREW ACTIVITY, SYSTEM REQUIREMENTS, MOVING TARGETS, ETC. | MANUAL | DEDICATED STELLAR MISSION |
| ORBITAL ANALYSIS | DEDICATED STELLAR OBSERVATION GENERATION | SCHEDULE SCIENCE OBSERVATIONS (DS) | SCHEDULE SCIENCE OBSERVATIONS (DS) | GENERATE VIEWING SCHEDULE FOR DEDICATED ASTRONOMY MISSIONS. GENERATE THIS SCHEDULE MAKING EFFICIENT USE OF AVAILABLE OBSERVATION TIME WHILE MINIMIZING VEHICLE ATTITUDE CHANGES. | AUTOMATIC | DEDICATED STELLAR MISSION |
| ORBITAL ANALYSIS | ATTITUDE/MAN EUVER TIMELINE GENERATION (MULTIDISCIP LINE) | GENERATE ATTITUDE TIMELINE | GENERATE ATTITUDE TIMELINE | GENERATE AN ATTITUDE TIMELINE WHERE THE MANEUVER TIMELINE IS PROVIDED BY THE "MISSION TIMELINE GENERATION" SUBFUNCTION AND THE KEYWORD FILE IS DEVELOPED AS PART OF THIS TASK. | AUTOMATIC | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|------------------|---|--|--|---|-----------|---------------------------|
| ORBITAL ANALYSIS | ATTITUDE/MAN EUVER TIMELINE TO GENERATION (MULTIDISCIP LINE) | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | EDIT CURRENT ATTITUDE DATA TO INCLUDE ADDITIONS OR CHANGES BASED ON STS OR OTHER REQUIREMENTS/CONSTRAINTS (THIS MAY ONLY INVOLVE MINOR UPDATES TO THE PRELIMINARY ATTITUDE TIMELINE). | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | ATTITUDE/MAN EUVER TIMELINE GENERATION (DEDICATED STELLAR) | GENERATE MANEUVER TIMELINE (DS) | GENERATE MANEUVER TIMELINE (DS) | GENERATE A MANEUVER TIMELINE BY DEFINING THE MANEUVERS REQUIRED TO GO BETWEEN TARGET ATTITUDES WITH EIGEN AXIS MANEUVERS. | AUTOMATIC | DEDICATED STELLAR MISSION |
| ORBITAL ANALYSIS | ATTITUDE/MAN EUVER TIMELINE GENERATION (DEDICATED STELLAR) | GENERATE ATTITUDE TIMELINE (DS) | GENERATE ATTITUDE TIMELINE (DS) | GENERATE AN ATTITUDE TIMELINE USING A KEYWORD FILE (TABLES) AND A MANEUVER TIMELINE AS INPUT FROM PROGRAM PAAC. | AUTOMATIC | DEDICATED STELLAR MISSION |
| ORBITAL ANALYSIS | ATTITUDE/MAN EUVER TIMELINE GENERATION (DEDICATED STELLAR) | EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | EDIT CURRENT ATTITUDE DATA TO INCLUDE ADDITIONS OR CHANGES BASED ON STS OR OTHER REQUIREMENTS/CONSTRAINTS (DEVELOPED FOR DEDICATED STELLAR MISSIONS). | AUTOMATIC | DEDICATED STELLAR MISSION |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|------------------|----------------------------------|--------------------------------------|--------------------------------------|--|-----------|-------------------|
| ORBITAL ANALYSIS | ORBITER POINTING DATA GENERATION | GENERATE ORBITER POINTING DATA | GENERATE ORBITER POINTING DATA | GENERATE A TIME HISTORY OF THE ORBITER ATTITUDE AND COMPUTE BODY-REFERENCED POINTING DIRECTIONS FOR VARIOUS ORBIT-RELATED ITEMS (VELOCITY VECTOR, ANGULAR MOMENTUM VECTOR, ETC.), CELESTIAL OBJECTS (SUN, MOON), AND TDRS AS A FUNCTION OF TIME. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | TDRS ACQ/LOSS GENERATION | GENERATE TDRS COVERAGE | GENERATE TDRS COVERAGE | GENERATE TDRS COVERAGE BASED ON THE LATEST ATTITUDE DATA. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | TDRS ACQ/LOSS GENERATION | PROCESS TDRS DATA FOR ENHANCEMENT | PROCESS TDRS DATA FOR ENHANCEMENT | PROCESS TDRS AND GROUND STATION ACQUISITION/LOSS TIME DATA THROUGH UNIONS AND INTERSECTIONS TO ENHANCE USABILITY OF THIS DATA. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | POCC MMU DATA SET GENERATION | GENERATE POCC MMU DATA SET | GENERATE POCC MMU DATA SET | REFORMAT DATA AND GENERATE A FILE (USER DATA SET) FOR USE BY THE ONBOARD EXPERIMENT COMPUTER OPERATING SYSTEM (ECOS) TIMELINE MAINTENANCE SYSTEM. | AUTOMATIC | ROUTINE |
| ORBITAL ANALYSIS | OBJECTIVE LOADS GENERATION | GENERATE CANDIDATE SOLAR GUIDE STARS | GENERATE CANDIDATE SOLAR GUIDE STARS | DETERMINE ACCEPTABLE GUIDE STARS AVAILABLE FOR IPS SOLAR POINTING. | AUTOMATIC | IPS SOLAR MISSION |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|------------------|----------------------------|---|---|---|-----------|---------------------------------|
| ORBITAL ANALYSIS | OBJECTIVE LOADS GENERATION | DEVELOP STRAY LIGHT CONSTRAINTS | DEVELOP STRAY LIGHT CONSTRAINTS | DEVELOP STRAY LIGHT CONSTRAINTS BY PLOTTING THE FIELDS OF VIEW ABOUT GUIDE STARS AND CERTAIN OTHER VECTORS IN THE ORBITER COORDINATE SYSTEM. THIS IS DONE TO DETECT SOURCES OF STRAY LIGHT (REFLECTION OF THE SUN OFF THE ORBITER) THAT WOULD INTERFERE WITH IPS TARGET ACQUISITION AND TRACKING. | AUTOMATIC | IPS SOLAR MISSION |
| ORBITAL ANALYSIS | OBJECTIVE LOADS GENERATION | CHOOSE SOLAR GUIDE STARS | CHOOSE SOLAR GUIDE STARS | FROM CANDIDATE GUIDE STARS, CHOOSE THE SOLAR GUIDE STARS TO BE USED IN GENERATING OBJECTIVE LOADS. SELECTION OF GUIDE STARS ARE BASED PRIMARILY ON STRAY LIGHT CONSIDERATIONS. | MANUAL | IPS SOLAR MISSION |
| ORBITAL ANALYSIS | OBJECTIVE LOADS GENERATION | GENERATE SOLAR OBJECTIVE LOADS | GENERATE SOLAR OBJECTIVE LOADS | GENERATE AN OBJECTIVE LOAD FILE AND AN OBJECTIVE LOAD SUMMARY. THE OBJECTIVE LOADS WILL RESIDE IN THE MMU AND BE USED BY THE INSTRUMENT POINTING SYSTEM (IPS) TO ACQUIRE AND TRACK SOLAR TARGETS OF INTEREST AS A FUNCTION OF ORBITAL POSITION. | AUTOMATIC | IPS SOLAR MISSION |
| ORBITAL ANALYSIS | OBJECTIVE LOADS GENERATION | GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | DETERMINE ACCEPTABLE GUIDE STARS AVAILABLE FOR IPS STELLAR POINTING. | AUTOMATIC | IPS MISSION (DEDICATED STELLAR) |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|------------------|--|--|--|---|-----------|---------------------------------|
| ORBITAL ANALYSIS | OBJECTIVE LOADS GENERATION | FORMAT STELLAR GUIDE STAR CATALOG (DS) | FORMAT STELLAR GUIDE STAR CATALOG (DS) | PROVIDE A GUIDE STAR CATALOG WHICH CONTAINS INFORMATION ON STARS RELATIVE TO THEIR LOCATION IN THE CELESTIAL SPHERE AND RELATIVE TO THEIR LOCATION IN THE IPS STAR TRACKERS. | AUTOMATIC | IPS MISSION (DEDICATED STELLAR) |
| ORBITAL ANALYSIS | OBJECTIVE LOADS GENERATION | SELECT IPS ROLL ANGLES (DS) | SELECT IPS ROLL ANGLES (DS) | DETERMINE THE IPS ROLL ANGLES FOR USE IN STELLAR OBSERVATIONS WHILE MINIMIZING THESE ROLLS BETWEEN OBSERVATIONS. | AUTOMATIC | IPS MISSION (DEDICATED STELLAR) |
| ORBITAL ANALYSIS | OBJECTIVE LOADS GENERATION | GENERATE STELLAR OBJECTIVE LOADS (DS) | GENERATE STELLAR OBJECTIVE LOADS (DS) | GENERATE AN OBJECTIVE LOAD FILE AND AN OBJECTIVE LOAD SUMMARY. THE OBJECTIVE LOADS WILL RESIDE IN THE MMU AND BE USED BY THE INSTRUMENT POINTING SYSTEM (IPS) TO ACQUIRE AND TRACK STELLAR TARGETS OF INTEREST AS A FUNCTION OF ORBITAL POSITION. | AUTOMATIC | IPS MISSION (DEDICATED STELLAR) |
| ORBITAL ANALYSIS | JOINT OPERATIONS TARGET FILE GENERATION(DEDICATED STELLAR) | GENERATE IPS POINTING DATA (DS) | GENERATE IPS POINTING DATA (DS) | REORGANIZE SCIENCE TARGET DATA AND POINTING DATA FOR INPUT INTO THE JOIF PROGRAM AND TO ALLOW THE PI'S TO EDIT DATA TO ADD SEQUENCE NUMBERS TO THE SCIENCE TARGETS. | AUTOMATIC | DEDICATED STELLAR MISSION |
| ORBITAL ANALYSIS | JOINT OPERATIONS TARGET FILE GENERATION(DEDICATED STELLAR) | PI EDITS TO ADD SEQUENCE NUMBER (DS) | PI EDITS TO ADD SEQUENCE NUMBER (DS) | THE PI EDITS THE ID FILE TO ADD A SEQUENCE NUMBER FOR USE BY HIS ONBOARD DEDICATED EXPERIMENT PROCESSOR (DEP) WHICH PERFORMS SPECIFIC OPERATIONS RELATIVE TO A PARTICULAR SEQUENCE NUMBER. | MANUAL | DEDICATED STELLAR MISSION |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|---------------------------|---|--|--|--|------------------|---------------------------|
| ORBITAL ANALYSIS | JOINT OPERATIONS TARGET FILE GENERATION(D EDICATED STELLAR) | GENERATE JOINT OPERATIONS TARGET FILE (DS) | GENERATE JOINT OPERATIONS TARGET FILE (DS) | GENERATE THE JOTF FILE (JOTF DATA SET) TO BE LOADED ON THE MMU. THE FILE CONTAINS DATA FOR EACH OBSERVATION TO BE CALLED UP AND VIEWED BY THE CREW ONBOARD. | AUTOMATIC | DEDICATED STELLAR MISSION |
| MISSION TIMELINE ANALYSIS | CREATE MISSION TIMELINE MODELS | CREATE MISSION TIMELINE MODELS | CREATE MISSION TIMELINE MODELS | CREATE MISSION TIMELINE MODELS AND WRITE THE ESS MODEL FILE. THIS SUBFUNCTION INCLUDES BUILDING EXPERIMENT MODELS, CREW CYCLE MODELS AND SL SYSTEM MODELS AS WELL AS MISSION LEVEL DATA. | MANUAL/AUTOMATIC | ROUTINE |
| MISSION TIMELINE ANALYSIS | GENERATE CREW H/O CYCLE | GENERATE CREW H/O CYCLE | GENERATE CREW H/O CYCLE | USING THE BASIC CREW CYCLE FROM JSC AND THE CREW CYCLE MODELS GENERATED IN THE CREATE MISSION TIMELINE MODELS SUBFUNCTION AS INPUT, GENERATE THE CREW H/O CYCLE. | MANUAL | ROUTINE |
| MISSION TIMELINE ANALYSIS | CREATE ESS TARGET FILE | CREATE ESS TARGET FILE | CREATE ESS TARGET FILE | REFORMAT TARGET, ATTITUDE AND TDRS O/O FILES INTO THE ESS FORMAT. BUILD NEW SUBJECTS/SPECIAL TARGETS IF REQUIRED. EDIT DATA IF REQUIRED. PERFORM STATISTICAL TARGET ANALYSIS TO PREDICT POTENTIAL INCOMPATIBILITIES BETWEEN THE ESS MODELS AND THE ESS TARGETS. THIS SUBTASK IS PERFORMED A MINIMUM OF THREE TIMES PER CYCLE. ONCE FOR INITIAL INPUT TO THE MISSION TL GENERATION SUBTASK AND TWICE MORE DURING ATTITUDE/TDRS ITERATION/UPDATES. | AUTOMATIC | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|---------------------------------|---|---------------------------------|------------------------------|---|------------------------|---------|
| MISSION TIMELINE ANALYSIS | MISSION TIMELINE GENERATION | GENERATE MISSION TIMELINE | GENERATE MISSION TIMELINE | BUILD SELECTION GROUPS, PRIORITIES, GRADING CRITERIA, ETC. AS A BASIS FOR SCHEDULING; SCHEDULE THE CREW CYCLE AND SYSTEM MODELS; SCHEDULE EXPERIMENTS; SCHEDULE PAO TV & PHOTO ACTIVITY; AND GENERATE MISSION TIMELINE OUTPUT FILES & DOCUMENTATION | PRIMARILY AUTOMATIC | ROUTINE |
| MISSION TIMELINE ANALYSIS | MISSION TIMELINE GENERATION | FINALIZE MISSION TIMELINE | FINALIZE MISSION TIMELINE | PERFORM NECESSARY SUBTASKS OF THE GENERATE MISSION TIMELINE TASK, USING THE UPDATES/MODIFICATIONS FROM THE REVIEW CYCLE TO GENERATE A FINAL MISSION TIMELINE. | PRIMARILY MANUAL | ROUTINE |
| MISSION TIMELINE ANALYSIS | PAYLOAD CREW ACTIVITY PLAN DEVELOPMENT | GENERATE PCAP CHARTS | GENERATE PCAP CHARTS | DESIGN THE LAYOUT OF THE CHARTS, DEVELOP PCAP PROCEDURES, DEVELOP NOTES AND GENERATE THE PCAP CHARTS. | AUTOMATIC | ROUTINE |
| MISSION TIMELINE ANALYSIS | PAYLOAD CREW ACTIVITY PLAN DEVELOPMENT | GENERATE PTS CHARTS | GENERATE PTS CHARTS | DESIGN THE LAYOUT OF AND PRODUCE THE PAYLOAD TIMELINE SUMMARY (PTS) CHARTS. | AUTOMATIC | ROUTINE |
| MISSION TIMELINE ANALYSIS | PAYLOAD CREW ACTIVITY PLAN DEVELOPMENT | BUILD PCAP DOCUMENT | BUILD PCAP DOCUMENT | COLLECT THE INFORMATION, ORGANIZE, BIND AND TAB THE PAYLOAD CREW ACTIVITY PLAN (PCAP) DOCUMENTS. | MANUAL | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|--|--|--|--|--|--------|---------|
| FLIGHT DEFINITION DOCUMENT DEVELOPMENT | FLIGHT DEFINITION DOCUMENT DEVELOPMENT | FLIGHT DEFINITION DOCUMENT DEVELOPMENT | FLIGHT DEFINITION DOCUMENT DEVELOPMENT | PREPARE THE FLIGHT DEFINITION DOCUMENT TO PROVIDE A MISSION DESCRIPTION TO BE USED BY SUPPORTING ELEMENTS FOR DESIGN AND OPERATIONS PLANNING FOR THE PAYLOAD. THE FDD IS ALSO USED TO DESCRIBE SCHEDULED OPERATIONS OF THE INDIVIDUAL EXPERIMENTS AND THE SUPPORTING RESOURCE REQUIREMENTS. | MANUAL | ROUTINE |
| FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | PREPARE THE FLIGHT PLANNING ANNEX INPUTS CONTAINING AGREEMENTS BETWEEN THE PYLD AND THE STS ON MATTERS WHICH RELATE TO THE ANALYSIS AND IMPLEMENTATION OF PAYLOAD FLIGHT DESIGN REQUIREMENTS ON THE STS (PART I - ELECTRICAL POWER, ENERGY AND COOLING REQUIREMENTS) (PART II - FLIGHT ACTIVITY PLANNING) (PART III - TRAJECTORY DESIGN). THE ANNEX IS INTENDED TO SUPPLEMENT THE PIP IN PROVIDING ADDITIONAL DETAILS TO FACILITATE MISSION AND FLIGHT OPERATIONS PLANNING. | MANUAL | ROUTINE |
| CREW PROCEDURES DEVELOPMENT | DEVELOP STORAGE BOOK | DEVELOP STORAGE BOOK | DEVELOP STORAGE BOOK | DEVELOP A BOOK CONTAINING A LIST OF ALL EQUIPMENT STORED ONBOARD AND IT'S LOCATION. | MANUAL | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|-----------------------------|------------------------------------|------------------------------------|------------------------------------|---|--------|---------|
| CREW PROCEDURES DEVELOPMENT | DEVELOP TV, PHOTO PROCEDURES | DEVELOP TV, PHOTO PROCEDURES | DEVELOP TV, PHOTO PROCEDURES | GENERATE DETAILED PROCEDURES, BASED ON PI REQUIREMENTS, FOR ALL PAYLOAD TV AND PHOTO OPERATIONS. GENERATE DESIRED SCENES, WHICH CAMERAS TO USE, BEST SET-UP & PERIPHERALS NEEDED (TRIPODS ETC). INCLUDE ACTUAL PHOTOS OF WHAT EACH SCENE IS TO LOOK LIKE. | MANUAL | ROUTINE |
| CREW PROCEDURES DEVELOPMENT | DEVELOP EXPERIMENT CREW PROCEDURES | DEVELOP EXPERIMENT CREW PROCEDURES | DEVELOP EXPERIMENT CREW PROCEDURES | DEVELOP DETAILED CREW PROCEDURES FOR EACH EXPERIMENT. PROCEDURES MUST CONTAIN EXPERIMENT ACTIVATION/DEACTIVATION PROCEDURES, NOMINAL OPERATIONS PROCEDURES, AND MALFUNCTION PROCEDURES. | MANUAL | ROUTINE |
| CREW PROCEDURES DEVELOPMENT | DEVELOP PAYLOAD SYSTEMS HANDBOOK | DEVELOP PAYLOAD SYSTEMS HANDBOOK | DEVELOP PAYLOAD SYSTEMS HANDBOOK | DEVELOP DETAILED CREW PROCEDURES FOR PL COMPLEMENT ACTIVATION/DEACTIVATION, REQUIRED MISSION PECULIAR EQUIPMENT OPERATIONS, AND MALFUNCTION PROCEDURES FOR PAYLOAD COMPLEMENT EQUIPMENT/INTERFACES ANOMALOUS CONDITIONS. | MANUAL | ROUTINE |
| CREW PROCEDURES DEVELOPMENT | DEVELOP CDMS DICTIONARY | DEVELOP CDMS DICTIONARY | DEVELOP CDMS DICTIONARY | DEVELOP A DOCUMENT DESCRIBING ALL PAYLOAD DISPLAYS. | MANUAL | ROUTINE |
| CREW PROCEDURES DEVELOPMENT | BUILD PDF DOCUMENTS | BUILD PDF DOCUMENTS | BUILD PDF DOCUMENTS | PERFORM THE WORD PROCESSING, COPYING, TABBING, ETC., REQUIRED TO BUILD THE PDF DOCUMENTS. | MANUAL | ROUTINE |
| DATA FLOW ANALYSIS | CREATE DATA FLOW MODELS | CREATE DATA FLOW MODELS | CREATE DATA FLOW MODELS | DEVELOP THE REQUIRED DATA FLOW MODELS FROM DATA EXTRACTED FROM THE IPROD AND O&IA DOCUMENTS, AND FROM DIRECT PI INTERFACE WHEN REQUIRED. | MANUAL | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|--------------------|---|--|--|--|-----------|---------|
| DATA FLOW ANALYSIS | GENERATE MISSION DATA REQUIREMENTS PROFILE | GENERATE MISSION DATA REQUIREMENTS PROFILE | GENERATE MISSION DATA REQUIREMENTS PROFILE | CREATE A FILE OF DATA REQUIREMENTS FOR: A) DIGITAL DATA OVER DEDICATED CHANNELS, B) ALL A/V SIGNALS, C) ALL DIRECT ACCESS CHANNEL SIGNALS, D) ALL EXPERIMENT ECIO AND ECIO SUBSETS | AUTOMATIC | ROUTINE |
| DATA FLOW ANALYSIS | SCHEDULE ONBOARD DATA MANAGEMENT AND DOWNLINK | GENERATE MISSION WINDOWS | GENERATE MISSION WINDOWS | DEFINE TDRS COVERAGE AND IDENTIFY OVERLAPS FOR MULTIPLE SATELLITES, WINDOWS OF OPPORTUNITY FOR REALTIME DATA DOWNLINK, WINDOWS OF OPPORTUNITY FOR RECORDER DUMPS, AND REQUIRED HIGH DATA RATE RECORDER (HDRR) RECORD WINDOWS BASED ON TDRS COVERAGE. | AUTOMATIC | ROUTINE |
| DATA FLOW ANALYSIS | SCHEDULE ONBOARD DATA MANAGEMENT AND DOWNLINK | SCHEDULE ONBOARD RECORDER OPERATIONS | SCHEDULE ONBOARD RECORDER OPERATIONS | SCHEDULE REALTIME DOWNLINK OF ANALOG/VIDEO AND DIGITAL DATA, A FILL/DUMP PLAN FOR THE DIGITAL HDRR, A FILL/DUMP PLAN FOR THE ANALOG/VIDEO TAPE RECORDERS, RECORDING ON THE VIDEO CASSETTE RECORDERS (THE USER MUST COMMAND THE RECORDER DUMPS AND CHANGEOUTS). | AUTOMATIC | ROUTINE |
| DATA FLOW ANALYSIS | SCHEDULE ONBOARD DATA MANAGEMENT AND DOWNLINK | GENERATE HRM POSSIBLE FORMATS | GENERATE HRM POSSIBLE FORMATS | IDENTIFY ALL FORMATS POSSIBLE AT ANY GIVEN TIME DURING A MISSION CONSIDERING ACTIVE EXPERIMENTS WITH DIGITAL DATA STREAMS, DIGITAL DUMPS (HDRR AND PAYLOAD RECORDER (PLR)), ANALOG/VIDEO DOWNLINK (REAL TIME AND DUMP), AND THE DOWNLINK OF DACH. | AUTOMATIC | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|--------------------|--|---------------------------------------|---------------------------------------|---|-----------|---------|
| DATA FLOW ANALYSIS | SCHEDULE ONBOARD DATA MANAGEMENT AND DOWNLINK | SCHEDULE HRM FORMATS AND DOWNLINK | SCHEDULE HRM FORMATS AND DOWNLINK | FROM THE HRM FORMATS AVAILABLE DURING A SPACELAB MISSION, SELECT OPTIMUM FORMAT TIMELINE, SCHEDULE KUSP CHANNEL USAGE, SELECT TDRS HANDOVERS, AOS EVALUATION PERIOD - EFFECTIVE AOS TO EFFECTIVE LOS, LOS EVALUATION PERIOD - EFFECTIVE LOS TO EFFECTIVE AOS. | AUTOMATIC | ROUTINE |
| DATA FLOW ANALYSIS | SCHEDULE POCC DATA CAPTURE/MANA GEMENT/DISTR IBUTTON | GENERATE POCC POSSIBLE CONFIGURATIONS | GENERATE POCC POSSIBLE CONFIGURATIONS | IDENTIFY THE LIST OF POSSIBLE REAL TIME POCC CONFIGURATIONS BASED ON THE DATA REQUIREMENTS PROFILE AND THE HDRR DUMP SCHEDULE. | AUTOMATIC | ROUTINE |
| DATA FLOW ANALYSIS | SCHEDULE POCC DATA CAPTURE/MANA GEMENT/DISTR IBUTTON | SCHEDULE POCC CONFIGURATIO NS | SCHEDULE POCC CONFIGURATIONS | GENERATE A SCHEDULE OF CONFIGURATIONS FOR ROUTING REAL-TIME DATA WITHIN THE POCC. | AUTOMATIC | ROUTINE |
| DATA FLOW ANALYSIS | SCHEDULE POCC DATA CAPTURE/MANA GEMENT/DISTR IBUTTON | SCHEDULE RECORDER PLAYBACKS | SCHEDULE RECORDER PLAYBACKS | GENERATE A SCHEDULE OF PLAYBACKS FOR THE HDRR. SCHEDULE TIME PERIODS IN WHICH THE DATA WILL BE PLAYED BACK AND IDENTIFY THE PLAYBACK CONFIGURATIONS. | AUTOMATIC | ROUTINE |
| DATA FLOW ANALYSIS | VERIFICATION OF DATA FLOW SCHEDULES | VERIFY DATA FLOW SCHEDULES | VERIFY DATA FLOW SCHEDULES | INSURE THAT LIMITATIONS (MISSION WINDOWS, HRM FORMATS, POCC CONFIGURATIONS AND TDRS AVAILABILITY) HAVE NOT BEEN VIOLATED BY THE USER WHEN UPDATING/ENHANCING THE DATA FLOW SCHEDULES. | AUTOMATIC | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|----------------------------|--|--------------------------------------|--------------------------------------|--|------------------|---------|
| DATA FLOW ANALYSIS | DATA FLOW AND SYSTEMS CONFIGURATION DOCUMENT DEVELOPMENT | GENERATE DATA FLOW REPORTS | GENERATE DATA FLOW REPORTS | CREATE TABULAR AND PLOT REPORT INFORMATION. THESE REPORTS RANGE FROM SPECIFIC TYPE REPORTS FOR ONE SPECIAL PHASE OF DATA FLOW TO GENERAL REPORTS THAT REFLECT ALL ONBOARD ACTIVITY. THE USER MAY RUN THESE REPORTS FOR A GIVEN TIME SLICE OR FOR THE ENTIRE MISSION. | AUTOMATIC | ROUTINE |
| DATA FLOW ANALYSIS | UPDATE OR ENHANCE EXISTING SCHEDULE | UPDATE OR ENHANCE EXISTING SCHEDULE | UPDATE OR ENHANCE EXISTING SCHEDULE | ENHANCE AN EXISTING SCHEDULE, ADAPT AN OLD SCHEDULE TO A NEW TORS PROFILE, OR INCORPORATE NEW DIGITAL OR ANALOG/VIDEO REQUIREMENTS INTO AN EXISTING SCHEDULE. | AUTOMATIC | ROUTINE |
| MMU LOAD INPUT DEVELOPMENT | CREATE ECOS SUBORDINATE TIMELINES | CREATE SUBORDINATE TIMELINES | CREATE SUBORDINATE TIMELINES | RECEIVE TIMELINE COMMAND INPUTS FROM THE PI'S AND GUIDELINES FROM THE O&I ATIC DOCUMENT AND CREATE SUBORDINATE TIMELINES AS NECESSARY TO ACCOMPLISH EACH PI'S OBJECTIVES (IF COMMANDS ARE VERY FEW THEY ARE INCORPORATED DIRECTLY INTO THE MASTER TIMELINE). | MANUAL/AUTOMATIC | ROUTINE |
| MMU LOAD INPUT DEVELOPMENT | CREATE ECOS SUBORDINATE TIMELINES | CHECK STL SYNTAX | CHECK STL SYNTAX | VERIFY ALL SUBORDINATE TIMELINES ARE LEGAL AND WITHOUT TYPOGRAPHICAL ERROR. | AUTOMATIC | ROUTINE |
| MMU LOAD INPUT DEVELOPMENT | CREATE ECOS SUBORDINATE TIMELINES | DESKTOP STL OPERATIONAL VERIFICATION | DESKTOP STL OPERATIONAL VERIFICATION | VERIFY THAT EACH SUBORDINATE TIMELINE COMPLETES THE OBJECTIVES OF THE P.I. AND DOES NOT CAUSE PERTURBATIONS TO OTHER EXPERIMENTS. | MANUAL | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|----------------------------------|-----------------------------------|---|--|---|-----------|---------|
| MMU LOAD INPUT DEVELOPMENT | CREATE ECOS MASTER TIMELINE | CREATE MASTER INPUT FILES | CREATE MASTER INPUT FILES | ASSEMBLE MISSION TIMELINE INFORMATION AND ARRANGE IN CHRONOLOGICAL ORDER INTO ONE OR SEVERAL MASTER TIMELINES, AS NEEDED. MULTIPLE TIMELINES ARE ACTIVATED SERIALLY DURING A MISSION. | MANUAL | ROUTINE |
| MMU LOAD INPUT DEVELOPMENT | CREATE ECOS MASTER TIMELINE | VERIFY AND COMBINE MASTER INPUT FILES | VERIFY AND COMBINE MASTER INPUT FILES | ASSURE THAT ALL MASTER TIMELINE INPUT FILES ARE LEGAL AND FREE OF TYPOGRAPHICAL ERRORS. MERGE ALL FILES INTO ONE. | AUTOMATIC | ROUTINE |
| MMU LOAD INPUT DEVELOPMENT | CREATE ECOS MASTER TIMELINE | GENERATE MASTER TIMELINE | GENERATE MASTER TIMELINE | COMBINE ALL MASTER INPUT FILES INTO ONE COMPLETE MASTER TIMELINE. | AUTOMATIC | ROUTINE |
| MMU LOAD INPUT DEVELOPMENT | CREATE ECOS MASTER TIMELINE | VERIFY MASTER TIMELINE | VERIFY MASTER TIMELINE | VERIFY ALL COMMANDS AND SUBORDINATE TIMELINE CALLS ARE VALID FOR ENTIRE MISSION. | AUTOMATIC | ROUTINE |
| MMU LOAD INPUT DEVELOPMENT | CREATE ECOS MASTER TIMELINE | GENERATE STL BUFFER UTILIZATION REPORT | GENERATE STL BUFFER UTILIZATION REPORT | SIMULATE THE ACTIVATION AND DE-ACTIVATION OF SUBORDINATE TIMELINES FOR AN ENTIRE MISSION TO VERIFY THAT NO MORE THAN 6 ARE ACTIVE AT ANY ONE TIME. | AUTOMATIC | ROUTINE |
| MMU LOAD INPUT DEVELOPMENT | CREATE ECOS MASTER TIMELINE | DESKTOP MTL OPERATIONAL VERIFICATION | DESKTOP MTL OPERATIONAL VERIFICATION | VERIFY THAT THE MASTER TIMELINE IS FREE OF ERRORS. | MANUAL | ROUTINE |
| MMU LOAD INPUT DEVELOPMENT | BUILD ECOS TIMELINE TAPE | CONVERT TO IBM TAPE FORMAT AND VERIFY | CONVERT TO IBM TAPE FORMAT AND VERIFY | CONVERT ECOS TIMELINE FILE FROM ASCII TO EBCDIC. | AUTOMATIC | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|---|--------------------------|---------------------------------|---------------------------------|---|----------------------|---------|
| MMU LOAD INPUT DEVELOPMENT | BUILD ECOS TIMELINE TAPE | GENERATE ECOS TIMELINE DOCUMENT | GENERATE ECOS TIMELINE DOCUMENT | INCORPORATE THE ECOS TIMELINE PRINTOUT INTO A DOCUMENT. | MANUAL | ROUTINE |
| MMU LOAD INPUT DEVELOPMENT | MMU OPTIMIZATION | CREATE MMU ALLOCATION FILE | CREATE MMU ALLOCATION FILE | CREATE A FILE WHICH SIMULATES THE LOCATION OF ALL FILES ON THE MMU TAPE. | MANUAL/AUTOM ATIC | ROUTINE |
| MMU LOAD INPUT DEVELOPMENT | MMU OPTIMIZATION | EVALUATE MMU TAPE MOVEMENT | EVALUATE MMU TAPE MOVEMENT | FIND THE BEST LOCATION FOR FILES ON THE MMU TAPE TO REDUCE THE AMOUNT OF TAPE MOVEMENT. APPROX. 5 OUTPUTS FROM THIS AND TASK "CREATE MMU ALLOCATION FILE" ARE GENERATED. THESE OUTPUTS ARE COMPARED TO EACH OTHER TO SELECT THE BEST CONFIGURATION. | AUTOMATIC | ROUTINE |
| EXPERIMENT COMMAND PLANNING DEVELOPMENT | CREATE COMMAND LIST | CREATE COMMAND LIST | CREATE COMMAND LIST | GATHER AND COMPILE ALL COMMANDS TO EXPERIMENTS THAT WILL BE PERFORMED DURING THE SUBJECT MISSION. ASSEMBLE ALL COMMANDS IN CHRONOLOGICAL ORDER IN A VAX FILE. | MANUAL/AUTOM ATIC | ROUTINE |
| EXPERIMENT COMMAND PLANNING DEVELOPMENT | CHECK COMMAND SYNTAX | CHECK COMMAND SYNTAX | CHECK COMMAND SYNTAX | VERIFY ALL COMMANDS LISTED ARE LEGAL AND WITHOUT TYPOGRAPHICAL ERRORS. | MANUAL | ROUTINE |
| EXPERIMENT COMMAND PLANNING DEVELOPMENT | PRODUCE COMMAND TIMETAGS | PRODUCE COMMAND TIMETAGS | PRODUCE COMMAND TIMETAGS | APPEND A MISSION ELAPSED TIME TO ALL THE COMMANDS LISTED IN THE INPUT FILE. | AUTOMATIC | ROUTINE |

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| FUNCTION | SUBFUNCTION | TASK | SUBTASK | ACTIVITY OBJECTIVE | METHOD | NEED |
|--|--|--|---|--|----------------------|---------|
| EXPERIMENT COMMAND PLANNING DEVELOPMENT | GENERATE COMMAND TIMELINE | GENERATE COMMAND TIMELINE | GENERATE COMMAND TIMELINE | PRODUCE THE COMMAND TIMELINE PRINTOUT. | AUTOMATIC | ROUTINE |
| EXPERIMENT COMMAND PLANNING DEVELOPMENT | CREATE POC CHECKLIST | CREATE POC CHECKLIST | CREATE POC CHECKLIST | ACCUMULATE ALL THE POC ACTIVITIES ON A LIST IN A VAX FILE. | MANUAL/AUTOM ATIC | ROUTINE |
| EXPERIMENT COMMAND PLANNING DEVELOPMENT | CHECK ACTIVITY SYNTAX | CHECK ACTIVITY SYNTAX | CHECK ACTIVITY SYNTAX | VERIFY ALL ACTIVITIES LISTED ARE CORRECT AND WITHOUT TYPOGRAPHICAL ERRORS. | AUTOMATIC | ROUTINE |
| EXPERIMENT COMMAND PLANNING DEVELOPMENT | PRODUCE ACTIVITY TIMETAGS | PRODUCE ACTIVITY TIMETAGS | PRODUCE ACTIVITY TIMETAGS | TO APPEND A NET TIME TO ALL ACTIVITIES LISTED IN THE INPUT FILE AND TO COMBINE ALL FILES INTO ONE. | MANUAL | ROUTINE |
| EXPERIMENT COMMAND PLANNING DEVELOPMENT | GENERATE POCC CHECKLIST AND COMMAND TIMELINE | GENERATE POCC CHECKLIST AND COMMAND TIMELINE | GENERATE POC CHECKLIST AND COMMAND TIMELINE | MERGE THE INPUT FILES TO PRODUCE A SCHEDULE OF ON-ORBIT ACTIVITIES AND CORRESPONDING POC ACTIVITIES. | AUTOMATIC | ROUTINE |

TABLE 2
ACTIVITY TIME AND SKILL REQUIREMENTS

ACTIVITY TIME AND SKILL REQUIREMENTS

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| ACTIVITY | CYCLE | CALENDAR TIME(DAYS) | SKILL TYPE | SKILL LEVEL (1-NOV, 2-PRO, 3-EXP) | MANPOWER (MAN-HOURS) |
|--|-------|------------------------|--------------|--------------------------------------|-------------------------|
| PAYLOAD DATA COLLECTION | B | 10 | P/L DATA COL | 3 | 80 |
| PAYLOAD DATA COLLECTION | P | 30 | P/L DATA COL | 3 | 160 |
| PAYLOAD DATA COLLECTION | U | 5 | P/L DATA COL | 3 | 40 |
| ORBIT REQUIREMENTS EVALUATION AND SELECTION | B | 2 | ORBIT | 2 | 16 |
| ORBIT REQUIREMENTS EVALUATION AND SELECTION | P | 5 | ORBIT | 3 | 40 |
| ORBIT REQUIREMENTS EVALUATION AND SELECTION | U | 1 | ORBIT | 2 | 8 |
| LAUNCH WINDOW/LAUNCH TIME SELECTION | B | 2 | ORBIT | 3 | 16 |
| LAUNCH WINDOW/LAUNCH TIME SELECTION | P | 5 | ORBIT | 3 | 40 |
| LAUNCH WINDOW/LAUNCH TIME SELECTION | U | 2 | ORBIT | 3 | 16 |
| GENERATE STATE VECTOR | P | 1 | ORBIT | 2 | 4 |
| CONVERT/STORE STATE VECTOR | B | 0 | ORBIT | 1 | 1 |
| CONVERT/STORE STATE VECTOR | P | 0 | ORBIT | 1 | 1 |
| CONVERT/STORE STATE VECTOR | R | 0 | ORBIT | 1 | 1 |
| CONVERT/STORE STATE VECTOR | U | 0 | ORBIT | 1 | 1 |
| GENERATE ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION | B | 1 | ORBIT | 2 | 4 |
| GENERATE ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION | P | 1 | ORBIT | 2 | 4 |
| GENERATE ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION | R | 0 | ORBIT | 2 | 1 |
| GENERATE ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION | U | 1 | ORBIT | 2 | 4 |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | B | 1 | ORBIT | 2 | 4 |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | P | 1 | ORBIT | 2 | 4 |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | R | 0 | ORBIT | 2 | 1 |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | U | 1 | ORBIT | 2 | 4 |
| GENERATE EARTH SHADOW ACQ/LOSS | B | 0 | ORBIT | 2 | 1 |
| GENERATE EARTH SHADOW ACQ/LOSS | P | 0 | ORBIT | 2 | 1 |
| GENERATE EARTH SHADOW ACQ/LOSS | R | 0 | ORBIT | 2 | 1 |
| GENERATE EARTH SHADOW ACQ/LOSS | U | 0 | ORBIT | 2 | 1 |
| GENERATE SUN RISE/SET | B | 0 | ORBIT | 1 | 1 |
| GENERATE SUN RISE/SET | P | 0 | ORBIT | 1 | 1 |
| GENERATE SUN RISE/SET | R | 0 | ORBIT | 1 | 1 |
| GENERATE SUN RISE/SET | U | 0 | ORBIT | 1 | 1 |

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| ACTIVITY | CYCLE | CALENDAR TIME(DAYS) | SKILL TYPE | SKILL LEVEL (1-NOV, 2-PRO, 3-EXP) | MANPOWER (MAN-HOURS) |
|--|-------|------------------------|------------|--------------------------------------|-------------------------|
| GENERATE MOON RISE/SET | B | 0 | ORBIT | 2 | 1 |
| GENERATE MOON RISE/SET | P | 0 | ORBIT | 2 | 1 |
| GENERATE MOON RISE/SET | R | 0 | ORBIT | 2 | 1 |
| GENERATE MOON RISE/SET | U | 0 | ORBIT | 2 | 1 |
| GENERATE PRELIMINARY TDRS COVERAGE | B | 1 | ORBIT | 2 | 1 |
| GENERATE PRELIMINARY TDRS COVERAGE | P | 1 | ORBIT | 2 | 1 |
| GENERATE PRELIMINARY TDRS COVERAGE | U | 1 | ORBIT | 2 | 1 |
| GENERATE GROUND STATION COVERAGE | B | 0 | ORBIT | 2 | 2 |
| GENERATE GROUND STATION COVERAGE | P | 0 | ORBIT | 2 | 2 |
| GENERATE GROUND STATION COVERAGE | R | 0 | ORBIT | 2 | 1 |
| GENERATE GROUND STATION COVERAGE | U | 0 | ORBIT | 2 | 2 |
| GENERATE RADIATION ENVIRONMENT | B | 1 | ORBIT | 2 | 4 |
| GENERATE RADIATION ENVIRONMENT | P | 1 | ORBIT | 2 | 4 |
| GENERATE RADIATION ENVIRONMENT | R | 0 | ORBIT | 2 | 1 |
| GENERATE RADIATION ENVIRONMENT | U | 1 | ORBIT | 2 | 4 |
| IMPOSE RADIATION CONSTRAINTS | B | 0 | ORBIT | 1 | 1 |
| IMPOSE RADIATION CONSTRAINTS | P | 0 | ORBIT | 1 | 1 |
| IMPOSE RADIATION CONSTRAINTS | R | 0 | ORBIT | 1 | 1 |
| IMPOSE RADIATION CONSTRAINTS | U | 0 | ORBIT | 1 | 1 |
| MERGE MISSION INDEPENDENT TARGETS | B | 0 | ORBIT | 1 | 1 |
| MERGE MISSION INDEPENDENT TARGETS | P | 0 | ORBIT | 1 | 1 |
| MERGE MISSION INDEPENDENT TARGETS | R | 0 | ORBIT | 1 | 1 |
| MERGE MISSION INDEPENDENT TARGETS | U | 0 | ORBIT | 1 | 1 |
| DEVELOP CELESTIAL TARGETS (NON-IPS MISSIONS) | B | 0 | ORBIT | 2 | 1 |
| DEVELOP CELESTIAL TARGETS (NON-IPS MISSIONS) | P | 1 | ORBIT | 2 | 4 |
| DEVELOP CELESTIAL TARGETS (NON-IPS MISSIONS) | U | 0 | ORBIT | 2 | 1 |
| DEVELOP CELESTIAL TARGETS (IPS MISSIONS) | B | 1 | ORBIT | 2 | 4 |
| DEVELOP CELESTIAL TARGETS (IPS MISSION) | P | 1 | ORBIT | 2 | 4 |
| DEVELOP CELESTIAL TARGETS (IPS MISSIONS) | U | 1 | ORBIT | 2 | 4 |
| GENERATE CELESTIAL TARGET(S) ACQ/LOSS | B | 1 | ORBIT | 2 | 1 |
| GENERATE CELESTIAL TARGET(S) ACQ/LOSS | P | 1 | ORBIT | 2 | 1 |
| GENERATE CELESTIAL TARGET(S) ACQ/LOSS | R | 0 | ORBIT | 2 | 1 |
| GENERATE CELESTIAL TARGET(S) ACQ/LOSS | U | 1 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | B | 1 | ORBIT | 2 | 4 |

ACTIVITY TIME AND SKILL REQUIREMENTS

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| ACTIVITY | CYCLE | CALENDAR TIME(DAYS) | SKILL TYPE | SKILL LEVEL (1-NOV,2-PRO,3-EXP) | MANPOWER (MAN-HOURS) |
|--|-------|------------------------|------------|------------------------------------|-------------------------|
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | P | 1 | ORBIT | 2 | 4 |
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | R | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | U | 1 | ORBIT | 2 | 4 |
| COMPUTE DISTANCE FROM TANGENT POINT ON THE ORBITER TO SUN LINE-OF-SIGHT TO THE EARTH SURFACE | B | 1 | ORBIT | 2 | 4 |
| COMPUTE DISTANCE FROM TANGENT POINT ON THE ORBITER TO SUN LINE-OF-SIGHT TO THE EARTH SURFACE | P | 1 | ORBIT | 2 | 4 |
| COMPUTE DISTANCE FROM TANGENT POINT ON THE ORBITER TO SUN LINE-OF-SIGHT TO THE EARTH SURFACE | R | 0 | ORBIT | 2 | 1 |
| COMPUTE DISTANCE FROM TANGENT POINT ON THE ORBITER TO SUN LINE-OF-SIGHT TO THE EARTH SURFACE | U | 1 | ORBIT | 2 | 4 |
| DEVELOP/APPLY CONSTRAINTS TO ATMOSPHERIC PHYSICS OBSERVATION PERIODS | B | 0 | ORBIT | 2 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO ATMOSPHERIC PHYSICS OBSERVATION PERIODS | P | 0 | ORBIT | 2 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO ATMOSPHERIC PHYSICS OBSERVATION PERIODS | R | 0 | ORBIT | 2 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO ATMOSPHERIC PHYSICS OBSERVATION PERIODS | U | 0 | ORBIT | 2 | 1 |
| COMPUTE ORBIT TERMINATOR TARGETS | B | 0 | ORBIT | 2 | 1 |
| COMPUTE ORBIT TERMINATOR TARGETS | P | 0 | ORBIT | 2 | 1 |
| COMPUTE ORBIT TERMINATOR TARGETS | R | 0 | ORBIT | 2 | 1 |
| COMPUTE ORBIT TERMINATOR TARGETS | U | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | B | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | P | 0 | ORBIT | 2 | 1 |

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ACTIVITY TIME AND SKILL REQUIREMENTS

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| ACTIVITY | CALENDAR | | SKILL TYPE | SKILL LEVEL (1-NOV, 2-PRO, 3-EXP) | MANPOWER (MAN-HOURS) |
|---|----------|------------|------------|--------------------------------------|-------------------------|
| | CYCLE | TIME(DAYS) | | | |
| COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | R | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | U | 0 | ORBIT | 2 | 1 |
| COMPUTE SUN AZIMUTH AND ELEVATION FROM ORBITING VEHICLE WITH RESPECT TO SUN RISE/SET EVENTS | R | 1 | ORBIT | 2 | 1 |
| DEVELOP CONSTRAINTS FOR SOLAR VIEWING PERIODS | B | 0 | ORBIT | 2 | 1 |
| DEVELOP CONSTRAINTS FOR SOLAR VIEWING PERIODS | P | 0 | ORBIT | 2 | 1 |
| DEVELOP CONSTRAINTS FOR SOLAR VIEWING PERIODS | R | 0 | ORBIT | 2 | 1 |
| DEVELOP CONSTRAINTS FOR SOLAR VIEWING PERIODS | U | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | B | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | P | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | R | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | U | 0 | ORBIT | 2 | 1 |
| CREATE EARTH SITE DEFINITION FILE | B | 0 | ORBIT | 2 | 2 |
| CREATE EARTH SITE DEFINITION FILE | P | 1 | ORBIT | 2 | 4 |
| CREATE EARTH SITE DEFINITION FILE | U | 0 | ORBIT | 2 | 2 |
| GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | B | 0 | ORBIT | 2 | 1 |
| GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | P | 0 | ORBIT | 2 | 1 |
| GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | R | 0 | ORBIT | 2 | 1 |
| GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | U | 0 | ORBIT | 2 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | B | 0 | ORBIT | 2 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | P | 0 | ORBIT | 2 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | R | 0 | ORBIT | 2 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | U | 0 | ORBIT | 2 | 1 |

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| ACTIVITY | CYCLE | CALENDAR TIME(DAYS) | SKILL TYPE | SKILL LEVEL (1-NOV,2-PRO,3-EXP) | MANPOWER (MAN-HOURS) |
|--|-------|------------------------|------------|------------------------------------|-------------------------|
| COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | B | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | P | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | R | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | U | 0 | ORBIT | 2 | 1 |
| COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | B | 1 | ORBIT | 2 | 4 |
| COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | P | 1 | ORBIT | 2 | 4 |
| COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | R | 0 | ORBIT | 2 | 1 |
| COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | U | 1 | ORBIT | 2 | 4 |
| DEVELOP/APPLY CONSTRAINTS TO BORB PARAMETERS | B | 0 | ORBIT | 2 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO BORB PARAMETERS | P | 0 | ORBIT | 2 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO BORB PARAMETERS | R | 0 | ORBIT | 2 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO BORB PARAMETERS | U | 0 | ORBIT | 2 | 1 |
| GENERATE HEMISPHERE OPPORTUNITIES | B | 0 | ORBIT | 1 | 1 |
| GENERATE HEMISPHERE OPPORTUNITIES | P | 0 | ORBIT | 1 | 1 |
| GENERATE HEMISPHERE OPPORTUNITIES | R | 0 | ORBIT | 1 | 1 |
| GENERATE HEMISPHERE OPPORTUNITIES | U | 0 | ORBIT | 1 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | B | 0 | ORBIT | 2 | 2 |
| COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | P | 1 | ORBIT | 2 | 4 |
| COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | R | 0 | ORBIT | 2 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | U | 0 | ORBIT | 2 | 1 |
| MERGE ALL EXPERIMENT TARGET FILES | B | 0 | ORBIT | 1 | 1 |
| MERGE ALL EXPERIMENT TARGET FILES | P | 0 | ORBIT | 1 | 1 |
| MERGE ALL EXPERIMENT TARGET FILES | R | 0 | ORBIT | 1 | 1 |

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| ACTIVITY | CYCLE | CALENDAR TIME(DAYS) | SKILL TYPE | SKILL LEVEL (1-NOV, 2-PRO, 3-EXP) | MANPOWER (MAN-HOURS) |
|---|-------|------------------------|------------|--------------------------------------|-------------------------|
| MERGE ALL EXPERIMENT TARGET FILES | U | 0 | ORBIT | 1 | 1 |
| PERFORM PARAMETRIC ANALYSIS TO DESIGN/DEVELOP | B | 5 | ORBIT | 3 | 40 |
| CO-ORBITING TRAJECTORIES THAT SATISFY OBJECTIVES AND CONSTRAINTS | | | | | |
| PERFORM PARAMETRIC ANALYSIS TO DESIGN/DEVELOP | P | 5 | ORBIT | 3 | 40 |
| CO-ORBITING TRAJECTORIES THAT SATISFY OBJECTIVES AND CONSTRAINTS | | | | | |
| PERFORM PARAMETRIC ANALYSIS TO DESIGN/DEVELOP | U | 5 | ORBIT | 3 | 40 |
| CO-ORBITING TRAJECTORIES THAT SATISFY OBJECTIVES AND CONSTRAINTS | | | | | |
| DEVELOP GROSS MISSION TIMELINE | B | 2 | ORBIT | 3 | 40 |
| DEVELOP GROSS MISSION TIMELINE | P | 2 | ORBIT | 2 | 40 |
| DEVELOP GROSS MISSION TIMELINE | U | 1 | ORBIT | 3 | 8 |
| DEVELOP PRELIMINARY ATTITUDE TIMELINE | B | 1 | ORBIT | 3 | 8 |
| DEVELOP PRELIMINARY ATTITUDE TIMELINE | P | 1 | ORBIT | 3 | 8 |
| DEVELOP PRELIMINARY ATTITUDE TIMELINE | U | 1 | ORBIT | 3 | 4 |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | B | 1 | ORBIT | 3 | 4 |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | P | 1 | ORBIT | 3 | 4 |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | U | 1 | ORBIT | 3 | 4 |
| CREATE RESERVE PERIOD FILE (DS) | B | 10 | ORBIT | 3 | 40 |
| CREATE RESERVE PERIOD FILE (DS) | P | 10 | ORBIT | 3 | 40 |
| CREATE RESERVE PERIOD FILE (DS) | U | 10 | ORBIT | 3 | 20 |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | B | 5 | ORBIT | 3 | 40 |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | P | 5 | ORBIT | 3 | 40 |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | R | 1 | ORBIT | 3 | 8 |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | U | 5 | ORBIT | 3 | 40 |
| GENERATE ATTITUDE TIMELINE | B | 1 | ORBIT | 3 | 8 |
| GENERATE ATTITUDE TIMELINE | P | 1 | ORBIT | 3 | 8 |
| GENERATE ATTITUDE TIMELINE | R | 0 | ORBIT | 3 | 4 |
| GENERATE ATTITUDE TIMELINE | U | 0 | ORBIT | 3 | 8 |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | B | 1 | ORBIT | 3 | 1 |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | P | 1 | ORBIT | 3 | 1 |

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| ACTIVITY | CYCLE | CALENDAR TIME(DAYS) | SKILL TYPE | SKILL LEVEL (1-NOV,2-PRO,3-EXP) | MANPOWER (MAN-HOURS) |
|--|-------|------------------------|------------|------------------------------------|-------------------------|
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS | R | 0 | ORBIT | 3 | 1 |
| OR OTHER REQMTS | | | | | |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS | U | 0 | ORBIT | 3 | 1 |
| OR OTHER REQMTS | | | | | |
| GENERATE MANEUVER TIMELINE (DS) | B | 1 | ORBIT | 3 | 8 |
| GENERATE MANEUVER TIMELINE (DS) | P | 1 | ORBIT | 3 | 8 |
| GENERATE MANEUVER TIMELINE (DS) | R | 0 | ORBIT | 3 | 1 |
| GENERATE MANEUVER TIMELINE (DS) | U | 1 | ORBIT | 3 | 4 |
| GENERATE ATTITUDE TIMELINE (DS) | B | 1 | ORBIT | 3 | 4 |
| GENERATE ATTITUDE TIMELINE (DS) | P | 1 | ORBIT | 3 | 4 |
| GENERATE ATTITUDE TIMELINE (DS) | R | 0 | ORBIT | 3 | 2 |
| GENERATE ATTITUDE TIMELINE (DS) | U | 1 | ORBIT | 3 | 4 |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | B | 0 | ORBIT | 2 | 2 |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | P | 0 | ORBIT | 2 | 2 |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | R | 0 | ORBIT | 2 | 1 |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | U | 0 | ORBIT | 2 | 1 |
| GENERATE ORBITER POINTING DATA | B | 0 | ORBIT | 2 | 1 |
| GENERATE ORBITER POINTING DATA | P | 0 | ORBIT | 2 | 1 |
| GENERATE ORBITER POINTING DATA | R | 0 | ORBIT | 2 | 1 |
| GENERATE ORBITER POINTING DATA | U | 0 | ORBIT | 2 | 1 |
| GENERATE TDRS COVERAGE | B | 1 | ORBIT | 3 | 1 |
| GENERATE TDRS COVERAGE | P | 1 | ORBIT | 3 | 1 |
| GENERATE TDRS COVERAGE | R | 0 | ORBIT | 3 | 1 |
| GENERATE TDRS COVERAGE | U | 1 | ORBIT | 3 | 1 |
| PROCESS TDRS DATA FOR ENHANCEMENT | B | 0 | ORBIT | 3 | 1 |
| PROCESS TDRS DATA FOR ENHANCEMENT | P | 0 | ORBIT | 3 | 1 |
| PROCESS TDRS DATA FOR ENHANCEMENT | R | 0 | ORBIT | 3 | 1 |
| PROCESS TDRS DATA FOR ENHANCEMENT | U | 0 | ORBIT | 3 | 1 |
| GENERATE POCC MMU DATA SET | B | 0 | ORBIT | 2 | 1 |
| GENERATE POCC MMU DATA SET | P | 0 | ORBIT | 2 | 1 |
| GENERATE POCC MMU DATA SET | R | 0 | ORBIT | 2 | 1 |
| GENERATE POCC MMU DATA SET | U | 0 | ORBIT | 2 | 1 |
| GENERATE CANDIDATE SOLAR GUIDE STARS | B | 5 | ORBIT | 3 | 40 |
| GENERATE CANDIDATE SOLAR GUIDE STARS | P | 5 | ORBIT | 3 | 40 |
| GENERATE CANDIDATE SOLAR GUIDE STARS | U | 2 | ORBIT | 3 | 16 |

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| ACTIVITY | CYCLE | CALENDAR TIME(DAYS) | SKILL TYPE | SKILL LEVEL (1-NOV, 2-PRO, 3-EXP) | MANPOWER (MAN-HOURS) |
|---|-------|------------------------|------------|--------------------------------------|-------------------------|
| DEVELOP STRAY LIGHT CONSTRAINTS | B | 5 | ORBIT | 3 | 40 |
| DEVELOP STRAY LIGHT CONSTRAINTS | P | 1 | ORBIT | 3 | 4 |
| DEVELOP STRAY LIGHT CONSTRAINTS | U | 2 | ORBIT | 3 | 16 |
| CHOOSE SOLAR GUIDE STARS | B | 5 | ORBIT | 3 | 40 |
| CHOOSE SOLAR GUIDE STARS | P | 5 | ORBIT | 3 | 40 |
| CHOOSE SOLAR GUIDE STARS | U | 5 | ORBIT | 3 | 40 |
| GENERATE SOLAR OBJECTIVE LOADS | B | 5 | ORBIT | 3 | 40 |
| GENERATE SOLAR OBJECTIVE LOADS | P | 5 | ORBIT | 3 | 40 |
| GENERATE SOLAR OBJECTIVE LOADS | U | 3 | ORBIT | 3 | 20 |
| GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | B | 5 | ORBIT | 3 | 40 |
| GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | P | 5 | ORBIT | 3 | 40 |
| GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | U | 3 | ORBIT | 3 | 20 |
| FORMAT STELLAR GUIDE STAR CATALOG (DS) | B | 0 | ORBIT | 3 | 2 |
| FORMAT STELLAR GUIDE STAR CATALOG (DS) | P | 0 | ORBIT | 3 | 2 |
| FORMAT STELLAR GUIDE STAR CATALOG (DS) | U | 0 | ORBIT | 3 | 2 |
| SELECT IPS ROLL ANGLES (DS) | B | 1 | ORBIT | 3 | 4 |
| SELECT IPS ROLL ANGLES (DS) | P | 1 | ORBIT | 3 | 4 |
| SELECT IPS ROLL ANGLES (DS) | U | 1 | ORBIT | 3 | 4 |
| GENERATE STELLAR OBJECTIVE LOADS (DS) | B | 5 | ORBIT | 3 | 40 |
| GENERATE STELLAR OBJECTIVE LOADS (DS) | P | 5 | ORBIT | 3 | 40 |
| GENERATE STELLAR OBJECTIVE LOADS (DS) | U | 3 | ORBIT | 3 | 20 |
| GENERATE IPS POINTING DATA (DS) | B | 0 | ORBIT | 3 | 2 |
| GENERATE IPS POINTING DATA (DS) | P | 0 | ORBIT | 3 | 2 |
| GENERATE IPS POINTING DATA (DS) | R | 0 | ORBIT | 3 | 1 |
| GENERATE IPS POINTING DATA (DS) | U | 0 | ORBIT | 3 | 2 |
| PI EDITS TO ADD SEQUENCE NUMBER (DS) | B | 5 | ORBIT | 2 | 8 |
| PI EDITS TO ADD SEQUENCE NUMBER (DS) | P | 5 | ORBIT | 2 | 8 |
| PI EDITS TO ADD SEQUENCE NUMBER (DS) | U | 5 | ORBIT | 2 | 8 |
| GENERATE JOINT OPERATIONS TARGET FILE (DS) | B | 1 | ORBIT | 3 | 4 |
| GENERATE JOINT OPERATIONS TARGET FILE (DS) | P | 1 | ORBIT | 3 | 4 |
| GENERATE JOINT OPERATIONS TARGET FILE (DS) | R | 0 | ORBIT | 3 | 1 |
| GENERATE JOINT OPERATIONS TARGET FILE (DS) | U | 1 | ORBIT | 3 | 4 |
| CREATE MISSION TIMELINE MODELS | B | 21 | TIMELINE | 3 | 240 |
| CREATE MISSION TIMELINE MODELS | P | 21 | TIMELINE | 3 | 240 |
| CREATE MISSION TIMELINE MODELS | U | 14 | TIMELINE | 3 | 160 |

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| ACTIVITY | CYCLE | CALENDAR TIME(DAYS) | SKILL TYPE | SKILL LEVEL (1-NOV,2-PRO,3-EXP) | MANPOWER (MAN-HOURS) |
|---|-------|------------------------|--------------|------------------------------------|-------------------------|
| GENERATE CREW H/O CYCLE | B | 3 | TIMELINE | 2 | 24 |
| GENERATE CREW H/O CYCLE | P | 3 | TIMELINE | 2 | 24 |
| GENERATE CREW H/O CYCLE | R | 0 | TIMELINE | 2 | 2 |
| GENERATE CREW H/O CYCLE | U | 3 | TIMELINE | 2 | 24 |
| CREATE ESS TARGET FILE | B | 2 | TIMELINE | 2 | 12 |
| CREATE ESS TARGET FILE | P | 3 | TIMELINE | 2 | 20 |
| CREATE ESS TARGET FILE | R | 0 | TIMELINE | 2 | 1 |
| CREATE ESS TARGET FILE | U | 2 | TIMELINE | 2 | 12 |
| GENERATE MISSION TIMELINE | B | 25 | TIMELINE | 2 | 600 |
| GENERATE MISSION TIMELINE | P | 20 | TIMELINE | 2 | 480 |
| GENERATE MISSION TIMELINE | R | 1 | TIMELINE | 3 | 6 |
| GENERATE MISSION TIMELINE | U | 20 | TIMELINE | 2 | 640 |
| FINALIZE MISSION TIMELINE | B | 5 | TIMELINE | 2 | 120 |
| FINALIZE MISSION TIMELINE | P | 5 | TIMELINE | 2 | 120 |
| FINALIZE MISSION TIMELINE | R | 0 | TIMELINE | 3 | 2 |
| FINALIZE MISSION TIMELINE | U | 5 | TIMELINE | 2 | 160 |
| GENERATE PCAP CHARTS | B | 25 | TIMELINE | 3 | 400 |
| GENERATE PCAP CHARTS | R | 0 | TIMELINE | 3 | 2 |
| GENERATE PCAP CHARTS | U | 30 | TIMELINE | 3 | 720 |
| GENERATE PTS CHARTS | B | 5 | TIMELINE | 2 | 40 |
| GENERATE PTS CHARTS | P | 5 | TIMELINE | 2 | 40 |
| GENERATE PTS CHARTS | R | 0 | TIMELINE | 2 | 1 |
| GENERATE PTS CHARTS | U | 5 | TIMELINE | 2 | 40 |
| BUILD PCAP DOCUMENT | B | 5 | TIMELINE | 2 | 80 |
| BUILD PCAP DOCUMENT | R | 0 | TIMELINE | 2 | 1 |
| BUILD PCAP DOCUMENT | U | 5 | TIMELINE | 3 | 120 |
| FLIGHT DEFINITION DOCUMENT DEVELOPMENT | B | 14 | FDD DEVELOP | 2 | 168 |
| FLIGHT DEFINITION DOCUMENT DEVELOPMENT | P | 14 | FDD DEVELOP | 2 | 168 |
| FLIGHT DEFINITION DOCUMENT DEVELOPMENT | U | 14 | FDD DEVELOP | 2 | 168 |
| FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | B | 14 | FPA DEVELOP | 2 | 168 |
| FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | P | 14 | FPA DEVELOP | 2 | 168 |
| FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | U | 14 | FPA DEVELOP | 2 | 168 |
| DEVELOP STOMAGE BOOK | B | 210 | CREW PRO DEV | 2 | 360 |
| DEVELOP STOMAGE BOOK | P | 90 | CREW PRO DEV | 2 | 155 |
| DEVELOP STOMAGE BOOK | U | 120 | CREW PRO DEV | 2 | 206 |

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| ACTIVITY | CALENDAR | SKILL TYPE | SKILL LEVEL (1-NOV,2-PRO,3-EXP) | MANPOWER (MAN-HOURS) |
|--|------------------|--------------|------------------------------------|-------------------------|
| | CYCLE TIME(DAYS) | | | |
| DEVELOP TV, PHOTO PROCEDURES | B 210 | CREW PRO DEV | 2 | 1200 |
| DEVELOP TV, PHOTO PROCEDURES | P 90 | CREW PRO DEV | 2 | 514 |
| DEVELOP TV, PHOTO PROCEDURES | U 120 | CREW PRO DEV | 2 | 686 |
| DEVELOP EXPERIMENT CREW PROCEDURES | B 210 | CREW PRO DEV | 2 | 3600 |
| DEVELOP EXPERIMENT CREW PROCEDURES | P 90 | CREW PRO DEV | 2 | 1543 |
| DEVELOP EXPERIMENT CREW PROCEDURES | U 120 | CREW PRO DEV | 2 | 686 |
| DEVELOP PAYLOAD SYSTEMS HANDBOOK | B 210 | CREW PRO DEV | 2 | 720 |
| DEVELOP PAYLOAD SYSTEMS HANDBOOK | P 90 | CREW PRO DEV | 2 | 309 |
| DEVELOP PAYLOAD SYSTEMS HANDBOOK | U 120 | CREW PRO DEV | 2 | 411 |
| DEVELOP CDMS DICTIONARY | B 210 | CREW PRO DEV | 2 | 240 |
| DEVELOP CDMS DICTIONARY | P 90 | CREW PRO DEV | 2 | 103 |
| DEVELOP CDMS DICTIONARY | U 120 | CREW PRO DEV | 2 | 137 |
| BUILD PDFD DOCUMENTS | B 21 | CREW PRO DEV | 2 | 240 |
| BUILD PDFD DOCUMENTS | P 21 | CREW PRO DEV | 2 | 240 |
| BUILD PDFD DOCUMENTS | U 21 | CREW PRO DEV | 2 | 240 |
| CREATE DATA FLOW MODELS | B 10 | DATA | 3 | 40 |
| CREATE DATA FLOW MODELS | P 90 | DATA | 3 | 100 |
| CREATE DATA FLOW MODELS | U 2 | DATA | 3 | 10 |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | B 10 | DATA | 3 | 40 |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | P 90 | DATA | 3 | 100 |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | R 1 | DATA | 3 | 1 |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | U 3 | DATA | 3 | 20 |
| GENERATE MISSION WINDOWS | B 3 | DATA | 3 | 10 |
| GENERATE MISSION WINDOWS | P 10 | DATA | 3 | 50 |
| GENERATE MISSION WINDOWS | R 1 | DATA | 3 | 1 |
| GENERATE MISSION WINDOWS | U 1 | DATA | 3 | 5 |
| SCHEDULE ONBOARD RECORDER OPERATIONS | B 1 | DATA | 2 | 5 |
| SCHEDULE ONBOARD RECORDER OPERATIONS | P 3 | DATA | 2 | 20 |
| SCHEDULE ONBOARD RECORDER OPERATIONS | R 1 | DATA | 2 | 1 |
| SCHEDULE ONBOARD RECORDER OPERATIONS | U 1 | DATA | 2 | 2 |
| GENERATE HRM POSSIBLE FORMATS | B 2 | DATA | 2 | 10 |
| GENERATE HRM POSSIBLE FORMATS | P 1 | DATA | 2 | 5 |
| GENERATE HRM POSSIBLE FORMATS | U 1 | DATA | 2 | 2 |
| SCHEDULE HRM FORMATS AND DOWNLINK | B 1 | DATA | 2 | 5 |
| SCHEDULE HRM FORMATS AND DOWNLINK | P 3 | DATA | 2 | 20 |

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ACTIVITY TIME AND SKILL REQUIREMENTS

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| ACTIVITY | CYCLE | CALENDAR TIME(DAYS) | SKILL TYPE | SKILL LEVEL (1-NOV,2-PRO,3-EXP) | MANPOWER (MAN-HOURS) |
|--|-------|------------------------|------------|------------------------------------|-------------------------|
| SCHEDULE HRM FORMATS AND DOWNLINK | R | 1 | DATA | 2 | 1 |
| SCHEDULE HRM FORMATS AND DOWNLINK | U | 1 | DATA | 2 | 2 |
| GENERATE POCC POSSIBLE CONFIGURATIONS | B | 1 | DATA | 2 | 5 |
| GENERATE POCC POSSIBLE CONFIGURATIONS | P | 1 | DATA | 2 | 5 |
| GENERATE POCC POSSIBLE CONFIGURATIONS | U | 1 | DATA | 2 | 1 |
| SCHEDULE POCC CONFIGURATIONS | B | 1 | DATA | 2 | 2 |
| SCHEDULE POCC CONFIGURATIONS | P | 2 | DATA | 2 | 10 |
| SCHEDULE POCC CONFIGURATIONS | R | 1 | DATA | 2 | 1 |
| SCHEDULE POCC CONFIGURATIONS | U | 1 | DATA | 2 | 1 |
| SCHEDULE RECORDER PLAYBACKS | B | 1 | DATA | 3 | 5 |
| SCHEDULE RECORDER PLAYBACKS | P | 3 | DATA | 3 | 20 |
| SCHEDULE RECORDER PLAYBACKS | R | 1 | DATA | 3 | 1 |
| SCHEDULE RECORDER PLAYBACKS | U | 1 | DATA | 3 | 2 |
| VERIFY DATA FLOW SCHEDULES | B | 3 | DATA | 3 | 20 |
| VERIFY DATA FLOW SCHEDULES | R | 1 | DATA | 3 | 1 |
| VERIFY DATA FLOW SCHEDULES | U | 3 | DATA | 3 | 20 |
| GENERATE DATA FLOW REPORTS | B | 5 | DATA | 3 | 20 |
| GENERATE DATA FLOW REPORTS | P | 3 | DATA | 3 | 10 |
| GENERATE DATA FLOW REPORTS | R | 1 | DATA | 3 | 1 |
| GENERATE DATA FLOW REPORTS | U | 1 | DATA | 3 | 5 |
| UPDATE OR ENHANCE EXISTING SCHEDULES | B | 5 | DATA | 3 | 40 |
| UPDATE OR ENHANCE EXISTING SCHEDULES | R | 1 | DATA | 3 | 1 |
| UPDATE OR ENHANCE EXISTING SCHEDULES | U | 2 | DATA | 3 | 10 |
| CREATE SUBORDINATE TIMELINES | B | 60 | MMU | 2 | 320 |
| CHECK STL SYNTAX | B | 5 | MMU | 2 | 40 |
| DESKTOP STL OPERATIONAL VERIFICATION | B | 5 | MMU | 2 | 60 |
| CREATE MASTER INPUT FILES | B | 30 | MMU | 2 | 100 |
| VERIFY AND COMBINE MASTER INPUT FILES | B | 1 | MMU | 2 | 40 |
| GENERATE MASTER TIMELINE | B | 1 | MMU | 2 | 4 |
| VERIFY MASTER TIMELINE | B | 1 | MMU | 2 | 8 |
| GENERATE STL BUFFER UTILIZATION REPORT | B | 7 | MMU | 2 | 40 |
| DESKTOP MTL OPERATIONAL VERIFICATION | B | 7 | MMU | 2 | 120 |
| CONVERT TO IBM TAPE FORMAT AND VERIFY | B | 2 | MMU | 2 | 6 |
| GENERATE ECOS TIMELINE DOCUMENT | B | 4 | MMU | 2 | 8 |
| CREATE MMU ALLOCATION FILE | B | 3 | MMU | 2 | 24 |

ACTIVITY TIME AND SKILL REQUIREMENTS

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| ACTIVITY | CYCLE | CALENDAR TIME(DAYS) | SKILL TYPE | SKILL LEVEL (1-NOV,2-PRO,3-EXP) | MANPOWER (MAN-HOURS) |
|--|-------|------------------------|--------------|------------------------------------|-------------------------|
| EVALUATE MMU TAPE MOVEMENT | B | 3 | MMU | 3 | 24 |
| CREATE COMMAND LIST | B | 15 | EXP CMD PLNG | 3 | 160 |
| CREATE COMMAND LIST | R | 1 | EXP CMD PLNG | 3 | 8 |
| CREATE COMMAND LIST | U | 5 | EXP CMD PLNG | 3 | 32 |
| CHECK COMMAND SYNTAX | B | 1 | EXP CMD PLNG | 2 | 4 |
| CHECK COMMAND SYNTAX | R | 1 | EXP CMD PLNG | 2 | 8 |
| CHECK COMMAND SYNTAX | U | 1 | EXP CMD PLNG | 2 | 4 |
| PRODUCE COMMAND TIMETAGS | B | 1 | EXP CMD PLNG | 2 | 4 |
| PRODUCE COMMAND TIMETAGS | R | 1 | EXP CMD PLNG | 2 | 8 |
| PRODUCE COMMAND TIMETAGS | U | 1 | EXP CMD PLNG | 2 | 4 |
| GENERATE COMMAND TIMELINE | B | 3 | EXP CMD PLNG | 2 | 8 |
| GENERATE COMMAND TIMELINE | R | 1 | EXP CMD PLNG | 2 | 8 |
| GENERATE COMMAND TIMELINE | U | 3 | EXP CMD PLNG | 2 | 8 |
| CREATE POCC CHECKLIST | B | 10 | EXP CMD PLNG | 3 | 160 |
| CREATE POCC CHECKLIST | R | 1 | EXP CMD PLNG | 3 | 8 |
| CREATE POCC CHECKLIST | U | 5 | EXP CMD PLNG | 3 | 32 |
| CHECK ACTIVITY SYNTAX | B | 1 | EXP CMD PLNG | 2 | 4 |
| CHECK ACTIVITY SYNTAX | R | 1 | EXP CMD PLNG | 2 | 8 |
| CHECK ACTIVITY SYNTAX | U | 1 | EXP CMD PLNG | 2 | 4 |
| PRODUCE ACTIVITY TIMETAGS | B | 1 | EXP CMD PLNG | 2 | 4 |
| PRODUCE ACTIVITY TIMETAGS | R | 1 | EXP CMD PLNG | 2 | 8 |
| PRODUCE ACTIVITY TIMETAGS | U | 1 | EXP CMD PLNG | 2 | 4 |
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | B | 3 | EXP CMD PLNG | 2 | 8 |
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | R | 1 | EXP CMD PLNG | 2 | 8 |
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | U | 3 | EXP CMD PLNG | 2 | 8 |

TABLE 3
SOFTWARE USED BY ACTIVITY

SOFTWARE USED BY ACTIVITY

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| ACTIVITY NAME | SOFTWARE NAME | TIME REQUIRED FOR USE BY PLANNING CYCLE (HRS) | PRELIMINARY | BASIC | UPDATE | REPLANNING |
|---|---------------|---|-------------|-------|--------|------------|
| LAUNCH WINDOW/LAUNCH TIME SELECTION | LWAP | 1 | 1 | 1 | 1 | 0 |
| GENERATE STATE VECTOR | LANTIM | 1 | 0 | 0 | 0 | 0 |
| CONVERT/STORE STATE VECTOR | VECTOR | 1 | 1 | 1 | 1 | 1 |
| GENERATE ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION | NSEP | 2 | 2 | 2 | 2 | 1 |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | ASEP-1 | 1 | 1 | 1 | 1 | 1 |
| GENERATE EARTH SHADOW ACQ/LOSS | ASEP-2 | 1 | 1 | 1 | 1 | 1 |
| GENERATE SUN RISE/SET | TARGEN | 1 | 1 | 1 | 1 | 1 |
| GENERATE MOON RISE/SET | STAR | 1 | 1 | 1 | 1 | 1 |
| GENERATE PRELIMINARY TDRS COVERAGE | CAVA/CAVINP | 1 | 1 | 1 | 1 | 0 |
| GENERATE GROUND STATION COVERAGE | SALES | 1 | 1 | 1 | 1 | 1 |
| GENERATE RADIATION ENVIRONMENT | RAD12 | 1 | 1 | 1 | 1 | 1 |
| IMPOSE RADIATION CONSTRAINTS | LTO-1 | 1 | 1 | 1 | 1 | 1 |
| MERGE MISSION INDEPENDENT TARGETS | TARGEN | 1 | 1 | 1 | 1 | 1 |
| DEVELOP CELESTIAL TARGETS (NON-IPS MISSIONS) | STAR-1 | 1 | 1 | 1 | 1 | 0 |
| DEVELOP CELESTIAL TARGETS (IPS MISSIONS) | SUBCOO | 1 | 1 | 1 | 1 | 0 |

| ACTIVITY NAME | SOFTWARE NAME | PRELIMINARY | BASIC | UPDATE | REPLANNING |
|--|---------------|-------------|-------|--------|------------|
| GENERATE CELESTIAL TARGET(S) ACQ/LOSS | STAR-2 | 1 | 1 | 1 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | TARGEN | 1 | 1 | 1 | 1 |
| COMPUTE DISTANCE FROM TANGENT POINT ON THE ORBITER TO SUN LINE-OF-SIGHT TO THE EARTH SURFACE | TANRAY | 1 | 1 | 1 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO ATMOSPHERIC PHYSICS OBSERVATION PERIODS | LTO-4 | 1 | 1 | 1 | 1 |
| COMPUTE ORBIT TERMINATOR TARGETS | TARGEN | 1 | 1 | 1 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | TARGEN | 1 | 1 | 1 | 1 |
| COMPUTE SUN AZIMUTH AND ELEVATION FROM ORBITING VEHICLE WITH RESPECT TO SUN RISE/SET EVENTS | ATMOS | 0 | 0 | 0 | 1 |
| DEVELOP CONSTRAINTS FOR SOLAR VIEWING PERIODS | LTO-1 | 1 | 1 | 1 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | TARGEN | 1 | 1 | 1 | 1 |
| CREATE EARTH SITE DEFINITION FILE | ESDAT | 2 | 2 | 2 | 0 |
| GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | ESAL | 1 | 1 | 1 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | LTO-3 | 1 | 1 | 1 | 1 |

| ACTIVITY NAME | SOFTWARE NAME | TIME REQUIRED FOR USE BY PLANNING CYCLE (HRS) | | |
|--|---------------|---|-------|-------------------|
| | | PRELIMINARY | BASIC | UPDATE REPLANNING |
| COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | TARGEN | 1 | 1 | 1 |
| COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | BORB | 1 | 1 | 1 |
| DEVELOP/APPLY CONSTRAINTS TO BORB PARAMETERS | LTO-2 | 1 | 1 | 1 |
| GENERATE HEMISPHERE OPPORTUNITIES | LTO-1 | 1 | 1 | 1 |
| COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | TARGEN | 4 | 4 | 4 |
| MERGE ALL EXPERIMENT TARGET FILES | TARGEN | 1 | 1 | 1 |
| PERFORM PARAMETRIC ANALYSIS TO DESIGN/DEVELOP CO-ORBITING TRAJECTORIES THAT SATISFY OBJECTIVES AND CONSTRAINTS | RELMO | 10 | 10 | 10 |
| DEVELOP PRELIMINARY ATTITUDE TIMELINE | CAVA/CAVINP | 1 | 1 | 0 |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | READP1 | 2 | 2 | 2 |
| CREATE RESERVE PERIOD FILE (DS) | EDT | 8 | 8 | 4 |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | ASTAR | 80 | 80 | 80 |
| GENERATE ATTITUDE TIMELINE | CAVA/KEYGEN | 8 | 8 | 8 |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | CAVA/CAVINP | 1 | 1 | 1 |

| ACTIVITY NAME | SOFTWARE NAME | TIME REQUIRED FOR USE BY PLANNING CYCLE (HRS) | | |
|--|---------------|---|-------|-------------------|
| | | PRELIMINARY | BASIC | UPDATE REPLANNING |
| GENERATE MANEUVER TIMELINE (DS) | PAAC | 4 | 4 | 1 |
| GENERATE ATTITUDE TIMELINE (DS) | CAVA/KEYGEN | 4 | 4 | 1 |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | CAVA/CAVINP | 1 | 1 | 0 |
| GENERATE ORBITER POINTING DATA | PROCAM | 1 | 1 | 1 |
| GENERATE TDRS COVERAGE | CAVA/CAVINP | 1 | 1 | 1 |
| PROCESS TDRS DATA FOR ENHANCEMENT | TARGEN | 1 | 1 | 1 |
| GENERATE POCB MMU DATA SET | PHDSG | 1 | 1 | 1 |
| GENERATE CANDIDATE SOLAR GUIDE STARS | SCATGEN | 40 | 40 | 0 |
| DEVELOP STRAY LIGHT CONSTRAINTS | ASTRO | 0 | 8 | 0 |
| GENERATE SOLAR OBJECTIVE LOADS | SCATGEN | 40 | 40 | 0 |
| GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | GSOLP-1 | 40 | 40 | 0 |
| FORMAT STELLAR GUIDE STAR CATALOG (DS) | GSOLP-2 | 2 | 2 | 0 |
| SELECT IPS ROLL ANGLES (DS) | GIMBAL | 4 | 4 | 0 |
| GENERATE STELLAR OBJECTIVE LOADS (DS) | GSOLP-2 | 40 | 40 | 0 |
| GENERATE IPS POINTING DATA (DS) | IPOL | 2 | 2 | 2 |
| PI EDITS TO ADD SEQUENCE NUMBER (DS) | EDT | 2 | 2 | 0 |

| ACTIVITY NAME | SOFTWARE NAME | TIME REQUIRED FOR USE BY PLANNING CYCLE (HRS) | | |
|---|---------------|---|-------|------------|
| | | PRELIMINARY | BASIC | UPDATE |
| | | | | REPLANNING |
| GENERATE JOINT OPERATIONS TARGET FILE (DS) | JOTF | 4 | 4 | 4 |
| CREATE MISSION TIMELINE MODELS | VME | 108 | 108 | 108 |
| CREATE ESS TARGET FILE | TAE | 10 | 10 | 10 |
| GENERATE MISSION TIMELINE | ESP | 128 | 160 | 128 |
| FINALIZE MISSION TIMELINE | ESP | 0 | 32 | 32 |
| GENERATE PCAP CHARTS | PCAP | 0 | 160 | 192 |
| GENERATE PTS CHARTS | PTS | 24 | 24 | 24 |
| CREATE DATA FLOW MODELS | EDT | 15 | 5 | 1 |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | DF/MORPG | 50 | 20 | 10 |
| GENERATE MISSION WINDOWS | DF/MMG | 40 | 8 | 4 |
| SCHEDULE ONBOARD RECORDER OPERATIONS | DF/ORS | 16 | 4 | 1 |
| GENERATE HRM POSSIBLE FORMATS | DF/HPFG | 4 | 8 | 2 |
| SCHEDULE HRM FORMATS AND DOWNLINK | DF/HFS | 16 | 4 | 1 |
| GENERATE POCC POSSIBLE CONFIGURATIONS | DF/PPCG | 4 | 4 | 1 |
| SCHEDULE POCC CONFIGURATIONS | DF/PCS | 8 | 1 | 1 |
| SCHEDULE RECORDER PLAYBACKS | DF/PBS | 16 | 4 | 1 |

| ACTIVITY NAME | SOFTWARE NAME | TIME REQUIRED FOR USE BY PLANNING CYCLE (HRS) | | | |
|--|---------------|---|-------|--------|------------|
| | | PRELIMINARY | BASIC | UPDATE | REPLANNING |
| VERIFY DATA FLOW SCHEDULES | DF/DWM | 0 | 16 | 16 | 1 |
| GENERATE DATA FLOW REPORTS | DF/DFRG | 5 | 10 | 3 | 1 |
| UPDATE OR ENHANCE EXISTING SCHEDULES | IDUS | 0 | 40 | 10 | 1 |
| CREATE SUBORDINATE TIMELINES | EDT | 0 | 320 | 0 | 0 |
| CHECK STL SYNTAX | VERSTL | 0 | 40 | 0 | 0 |
| CREATE MASTER INPUT FILES | EDT | 0 | 160 | 0 | 0 |
| VERIFY AND COMBINE MASTER INPUT FILES | VERMI | 0 | 40 | 0 | 0 |
| GENERATE MASTER TIMELINE | GENMTL | 0 | 8 | 0 | 0 |
| VERIFY MASTER TIMELINE | VERMTL | 0 | 8 | 0 | 0 |
| GENERATE STL BUFFER UTILIZATION REPORT | STLBUF | 0 | 40 | 0 | 0 |
| CONVERT TO IBM TAPE FORMAT AND VERIFY | DEL.COM | 0 | 16 | 0 | 0 |
| CREATE MMU ALLOCATION FILE | EDT | 0 | 24 | 0 | 0 |
| EVALUATE MMU TAPE MOVEMENT | MMJALL | 0 | 8 | 0 | 0 |
| CREATE COMMAND LIST | EDT | 0 | 120 | 40 | 4 |
| CHECK COMMAND SYNTAX | CHECK | 0 | 8 | 8 | 4 |
| PRODUCE COMMAND TIMETAGS | MET | 0 | 8 | 8 | 4 |
| GENERATE COMMAND TIMELINE | CMDATG | 0 | 24 | 24 | 8 |

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| ACTIVITY NAME | SOFTWARE NAME | PRELIMINARY | BASIC | UPDATE | REPLANNING |
|---|---------------|-------------|-------|--------|------------|
| CREATE POCC CHECKLIST | EDT | 0 | 80 | 40 | 4 |
| CHECK ACTIVITY SYNTAX | CHECK | 0 | 8 | 8 | 4 |
| PRODUCE ACTIVITY TIMETAGS | MET | 0 | 8 | 8 | 4 |
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | CG | 0 | 24 | 24 | 8 |

TABLE 4
SOFTWARE DESCRIPTION

SOFTWARE DESCRIPTION

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PAGE 1

| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS | | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|--|-------------|--------------|------------|----------|-------|-------------------|------------------|
| | | | TYPE | LEVEL | | | | |
| ASEP-1 | (ANALYTIC SATELLITE EPHEMERIS PROGRAM) THIS PROGRAM COMPUTES (ANALYTICALLY) AN EARTH SATELLITE EPHEMERIS. THE PROGRAM CONSISTS OF SEVERAL COMPUTATIONAL OPTIONS: 1) A DETAILED TRAJECTORY CONSISTING OF A TIME HISTORY (ANY DELTA TIME) OF 110 VARIABLES, 2) A GROUND TRACK TRAJECTORY CONSISTING OF A TIME HISTORY OF THE LATITUDE AND LONGITUDE OF THE SATELLITE SUB-POINT AND 10 OTHER VARIABLES, 3) EARTH SHADOW ACQUISITION AND LOSS COMPUTATION, 4) A LAUNCH TIME PARAMETERIZATION WHERE MULTIPLE DETAILED TRAJECTORIES ARE GENERATED FOR A USER SPECIFIED RANGE OF LAUNCH TIMES AND, 5) THE CAPABILITY TO DRIVE THOSE ABOVE OPTIONS WITH AN EPHEMERIS FILE GENERATED BY NSEP. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAM | 14191 | 1474000 | 110 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS | | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|--|-------------|--------------|------------|----------|-------|-------------------|------------------|
| | | | TYPE | LEVEL | | | | |
| ASEP-2 | (ANALYTIC SATELLITE EPHEMERIS PROGRAM) THIS PROGRAM COMPUTES (ANALYTICALLY) AN EARTH SATELLITE EPHEMERIS. THE PROGRAM CONSISTS OF SEVERAL COMPUTATIONAL OPTIONS: 1) A DETAILED TRAJECTORY CONSISTING OF A TIME HISTORY (ANY DELTA TIME) OF 110 VARIABLES, 2) A GROUND TRACK TRAJECTORY CONSISTING OF A TIME HISTORY OF THE LATITUDE AND LONGITUDE OF THE SATELLITE SUB-POINT AND 10 OTHER VARIABLES, 3) EARTH SHADOW ACQUISITION AND LOSS COMPUTATION, 4) A LAUNCH TIME PARAMETERIZATION WHERE MULTIPLE DETAILED TRAJECTORIES ARE GENERATED FOR A USER SPECIFIED RANGE OF LAUNCH TIMES AND, 5) THE CAPABILITY TO DRIVE THOSE ABOVE OPTIONS WITH AN EPHEMERIS FILE GENERATED BY NSEP. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN | 14191 | 1474000 | 20 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|----------------------|------------|----------|-------|-------------------|------------------|
| ASTAR | (ASTRONOMY SCHEDULER) THIS PROGRAM IS DESIGNED TO PRODUCE A VIEWING SCHEDULE FOR SPACE SHUTTLE BASED ASTRONOMY MISSIONS. IT EMPLOYS AN AUTOMATED SCHEDULING ALGORITHM TO GENERATE OBSERVATION SEQUENCES THAT MAKE EFFICIENT USE OF AVAILABLE OBSERVATORY TIME. FACTORS INCLUDE PRIORITY, WINDOWS, SLEW TIMES, VIEWING CONSTRAINTS, AND RESERVED PERIODS. OUTPUT INCLUDES VIEWING SCHEDULE, STATISTICS, GRAPHS AND FILES FOR INPUT TO THE CREW TIMELINE AND ORBITER ATTITUDE PROGRAMS. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN | 11000 | 5431000 | 15 |
| ASTRO | THIS PROGRAM IS DESIGNED TO PROVIDE THE CAPABILITIES FOR ENTERING ASTRONOMY DATA WITH SPECIFIED FIELDS OF VIEWS (FOV), VIEWING AVAILABILITY ON THE CELESTIAL SPHERE, AND PLOTTING THE DATA. THE CELESTIAL SPHERE IS REPRESENTED BY A RECTANGULAR GRID OF RIGHT ASCENSION(S) AND DECLINATION(S). THE SUN, MOON, PLANETS AND STELLAR TARGETS MAY BE VIEWED GRAPHICALLY IN VARIOUS PLANES AND FOV'S. THE INPUT IS VIA DATA FILES OR A CROSS-HAIR EDITOR. OUTPUT IS VIA A DATA FILE AND THE TERMINAL. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN | 5410 | 1232 | 60 |

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| SW NAME | SW FUNCTION | MODE OF USE | | | SKILL REQMTS | | | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|-------|------|--------------|------------|--------|----------|-------|-------------------|------------------|
| | | INTERACTIVE | ORBIT | TYPE | LEVEL | PROFICIENT | FORTAN | | | | |
| ATHOS | (ATHOS EXPERIMENT PROGRAM) THIS PROGRAM DOES CALCULATIONS OF THE AZIMUTH AND ELEVATION OF THE SUN AS SEEN FROM THE VEHICLE AT USER SPECIFIED TIMES WITH RESPECT TO SUN RISE AND SET EVENTS. THE PROGRAM REQUIRES A NODE FILE AND A SPECIFICATION OF THE ATTITUDE AS INPUT. OUTPUT IS A TABLE OF SUN AZIMUTHS AND ELEVATIONS OR A LIST-DIRECTED FILE CONTAINING THAT INFORMATION. | INTERACTIVE | ORBIT | | | PROFICIENT | FORTAN | | 1650 | 1525000 | 10 |
| BORB | (B-VECTOR IN THE ORBITER COORDINATE SYSTEM) THIS PROGRAM COMPUTES THE STRENGTH OF THE GEOMAGNETIC FIELD AND THE DIRECTIONS OF THIS FIELD AT THE CURRENT POSITION OF THE ORBITER IN SEVERAL COORDINATE SYSTEMS INCLUDING THE ORBITER BODY SYSTEM. THE PROGRAM CONTAINS THE OPTION TO USE EITHER THE 1975 OR 1980 INTERNATIONAL GEOMAGNETIC REFERENCE FIELD (IGRF) MODEL. BORB OUTPUTS 54 PARAMETERS ON A LIST DIRECTED FILE INCLUDING THE SPHERICAL POLAR COMPONENTS OF THE FIELD, THE TOTAL MAGNITUDE OF THE FIELD AND THE MCILWAIN PARAMETERS. BORB IS A PART OF CAVA AND DOES NOT RUN STANDALONE. THE VEHICLE'S EMPHEMERIS AND ATTITUDE TIMELINE MUST BE LOADED IN VIA THE OTHER OPTIONS IN CAVA BEFORE RUNNING BORB. | INTERACTIVE | ORBIT | | | PROFICIENT | FORTAN | | 2100 | 1263000 | 45 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|----------------------|--|-------------|----------------------|------------|----------|-------|-------------------|------------------|
| CAVA/CAVNP /KEYGN | (COMBINED ATTITUDE AND VISIBILITY ANALYSIS) THIS PROGRAM COMBINES SEVERAL INTERRELATED FUNCTIONS FOR ATTITUDE AND VISIBILITY ANALYSIS. IT COMPUTES AN ATTITUDE TIMELINE USING KEYWORD TABLE DATA AND A MANEUVER TIMELINE (KEYGN); GENERATES A SPACECRAFT OCCULTATION IMAGE ON A CELESTIAL SPHERE AS SEEN FROM A SPECIFIED ANTENNA OR SENSOR COORDINATE SYSTEM ABOARD A SPACECRAFT (ASOC); GENERATES ACQUISITION AND LOSS TIMES OF CELESTIAL OR EARTH-FIXED TARGETS BY SENSORS ABOARD A SPACECRAFT IN EARTH ORBIT (CAVINP); COMPUTES GEOMAGNETIC FIELD PARAMETERS IN THE ORBITER COORDINATE SYSTEM (BOB); AND DISPLAYS CALCULATED ATTITUDE DATA VIA A TEKTRONIX 4014 TERMINAL OR ALTERNATE FILE (ATTD0C). | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN | 20072 | 1539000 | 25 |
| CG | COMBINES ALL INPUT FILES TO PRODUCE A PRINTOUT OF ALL "ON ORBIT" ACTIVITIES AND ALL CORRESPONDING POCC ACTIVITIES. | INTERACTIVE | EXP CHD PLNG | PROFICIENT | FORTRAN | 2600 | 45824 | 180 |
| CHECK | VERIFIES THAT THE COMMANDS LISTED IN THE INPUT FILE ARE LEGAL COMMANDS TO THAT EXPERIMENT AND WITHOUT TYPOGRAPHICAL ERROR. | INTERACTIVE | EXP CHD PLNG | PROFICIENT | FORTRAN | 2460 | 13824 | 45 |
| CHDATG | PRODUCES THE COMMAND TIMELINE PRINTOUT FROM THE INPUT FILE. | INTERACTIVE | EXP CHD PLNG | PROFICIENT | FORTRAN | 2046 | 26368 | 120 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL RIGHTS | | | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|-----------------|---|-------------|--------------|------------|--------|-------|----------------|---------------|
| | | | TYPE | LEVEL | DCL | | | |
| DEL.COM | CONVERT FILE FORMAT FROM ASCII TO EBCDIC | INTERACTIVE | MMU | PROFICIENT | DCL | 338 | 14848 | 150 |
| DFAST/DPRG | THIS DFAST MODULE CONSISTS OF A SET OF EXECUTABLES THAT CREATE TABULAR OR PLOT REPORT INFORMATION. THESE EXECUTABLES INCLUDE DATA MANAGEMENT CHECKLIST, MISSION ACTIVITY PLOTS, PLAYBACK REPORTS, ANTENNA DISPLAYS, ETC.. | INTERACTIVE | DATA | EXPERT | FORTAN | 42383 | 921600 | 50 |
| DFAST/DWM | THIS DFAST MODULE IS USED TO VERIFY DATA FLOW SCHEDULES. | INTERACTIVE | DATA | EXPERT | FORTAN | 10838 | 665600 | 3 |
| DFAST/HFS | THIS DFAST MODULE SELECTS THE FORMATS THAT WILL BE IN OPERATION DURING A SPACELAB MISSION. | INTERACTIVE | DATA | PROFICIENT | FORTAN | 7908 | 358400 | 1 |
| DFAST/HPFG | THIS DFAST MODULE IDENTIFIES ALL FORMATS POSSIBLE AT ANY GIVEN TIME DURING A MISSION. | INTERACTIVE | DATA | PROFICIENT | FORTAN | 2630 | 204800 | 1 |
| DFAST/HDRP G | THIS DFAST MODULE DEVELOPS A FILE TO BE USED IN THE DATA FLOW ANALYSIS TO DEFINE EXPERIMENT AND SYSTEM REQUIREMENTS AND CAPABILITIES. | INTERACTIVE | DATA | EXPERT | FORTAN | 5031 | 563200 | 1 |
| DFAST/MMG | THIS DFAST MODULE DEFINES TDRS COVERAGE, REALTIME DOWNLINK, HDRR DUMP WINDOWS. | INTERACTIVE | DATA | EXPERT | FORTAN | 16608 | 819200 | 1 |
| DFAST/ORS | THIS DFAST MODULE SCHEDULES ONBOARD RECORDER AND REALTIME DOWNLINKS. | INTERACTIVE | DATA | EXPERT | FORTAN | 15587 | 819200 | 1 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|------------|---|-------------|----------------------|------------|-------------------|-------|-------------------|------------------|
| DFAST/PBS | THIS DFAST MODULE SCHEDULES PLAYBACKS FOR THE HDRR DUMPS. | INTERACTIVE | DATA | PROFICIENT | FORTAN | 16498 | 307200 | 1 |
| DFAST/PCS | THIS DFAST MODULE DEFINES ROUTING OF REALTIME DATA IN THE POC. | INTERACTIVE | DATA | PROFICIENT | FORTAN | 5555 | 307200 | 1 |
| DFAST/PPCG | THIS DFAST MODULE ESTABLISHES A LIST OF POSSIBLE REALTIME POC CONFIGURATION. | INTERACTIVE | DATA | PROFICIENT | FORTAN | 9396 | 358400 | 1 |
| EDT | THE VAX EDITOR ALLOWS RANDOM ACCESS TO A LIST OF COMMANDS/ACTIVITIES IN COMPUTER MEMORY. EACH ITEM IS ADDED, DELETED OR REVISED, ONE AT A TIME BY THE OPERATOR. | INTERACTIVE | ALL | NOVICE | VAX OPS SYSTEM | 0 | 0 | 0 |
| ESAL | (EARTH SITE ACQUISITION AND LOSS PROGRAM) THIS PROGRAM ANALYTICALLY COMPUTES THE TIMES AN ORBITING VEHICLE'S GROUND TRACK WILL ENTER AND EXIT POLYGONAL GROUND SITES. | INTERACTIVE | ORBIT | PROFICIENT | FORTAN | 1850 | 1286000 | 30 |
| ESDAT | (EARTH SITE DATA) THIS PROGRAM ALLOWS THE USER TO DEFINE POLYGONAL GROUND SITES FOR USE IN THE ESAL PROGRAM. | INTERACTIVE | ORBIT | PROFICIENT | FORTAN | 700 | 1220000 | 45 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|----------------------|------------|----------|-------|-------------------|------------------|
| ESP | THE EXPERIMENT SCHEDULING PROGRAM (ESP) GENERATES, MODIFIES, VERIFIES, & DOCUMENTS EXPERIMENT T/L'S. THE USER MAY SELECT EITHER AUTOMATIC SCHEDULING OR MANUAL KEY IN OF ACTIVITIES. IN EITHER CASE THE PROGRAM VERIFIES THAT CONSTRAINTS ARE NOT VIOLATED. WHEN EDITING ACTIVITIES MANUALLY, THE USER MAY ELECT TO ACCEPT WARNINGS THAT CERTAIN CONSTRAINTS HAVE BEEN VIOLATED & PRODUCE A T/L WITH VIOLATIONS. THE PROGRAM ALSO HAS AN OPTION TO VERIFY THAT CHANGES TO THE MODELS, MISSION CONSTRAINTS, AND/OR TARGET/ATTITUDE FILES HAVE NOT INVALIDATED A PREVIOUS EXPERIMENT TIMELINE. | INTERACTIVE | TIMELINE | PROFICIENT | FORTAN | 90000 | 3800 | 60 |
| GENMTL | ASSEMBLES ALL MASTER TIMELINE EVENTS INTO CHRONOLOGICAL ORDER. | INTERACTIVE | NNU | PROFICIENT | FORTAN | 2528 | 57856 | 120 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS | | | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|--------------|------------|-------|-------------------|------------------|
| | | | TYPE | LEVEL | ORBIT | | |
| GINBAL | (GINBAL ANGLE PROGRAM) THIS PROGRAM ALLOWS THE USER TO DETERMINE THE CELESTIAL ROLLS TO USE IN IPS STELLAR OBSERVATIONS WHILE MINIMIZING THOSE ROLLS BETWEEN OBSERVATIONS. THIS IS DONE VIA AN ITERATIVE PROCESS IN WHICH THE USER CAN SPECIFY THE IPS GIMBAL ROLL IN AN INPUT FILE. GIMBAL OUTPUTS A LIST-DIRECTED FILE LISTING NET, SCIENCE AND GUIDE STAR POSITIONS, AND GIMBAL AND CELESTIAL ANGLES FOR EACH OBSERVATION. | INTERACTIVE | ORBIT | EXPERT | 1100 | 1216000 | 60 |
| GSOLP-1 | THE GUIDE STAR OBJECTIVE LOAD PROGRAM (GSOLP) HAS BEEN DEVELOPED EXCLUSIVELY TO SELECT GUIDE STARS FOR USE BY THE INERTIAL POINTING SYSTEM (IPS) ON ASTRO MISSIONS. THE PROGRAM HAS 5 MODULES TO PROVIDE: 1) SELECTION OF GUIDE STARS, 2) TABULATION AND PLOTTING OF SELECTED DATA, 3) OPTIMIZATION OF IPS GIMBAL ANGLES FOR OBJECTIVE LOADS, 4) GENERATION OF OBJECTIVE LOADS IN ASCII FORMAT, AND 5) GENERATION OF OBJECTIVE LOADS TO MMU FORMAT FOR REAL TIME UPLINK. | INTERACTIVE | ORBIT | PROFICIENT | 17460 | 3000 | 960 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS | | | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|--------------|------------|----------|-------------------|------------------|
| | | | TYPE | LEVEL | LANGUAGE | | |
| GSOLP-2 | THE GUIDE STAR OBJECTIVE LOAD PROGRAM (GSOLP) HAS BEEN DEVELOPED EXCLUSIVELY TO SELECT GUIDE STARS FOR USE BY THE INERTIAL POINTING SYSTEM (IPS) ON ASTRO MISSIONS. THE PROGRAM HAS 5 MODULES TO PROVIDE: 1) SELECTION OF GUIDE STARS, 2) TABULATION AND PLOTTING OF SELECTED DATA, 3) OPTIMIZATION OF IPS GIMBAL ANGLES FOR OBJECTIVE LOADS, 4) GENERATION OF OBJECTIVE LOADS IN ASCII FORMAT, AND 5) GENERATION OF OBJECTIVE LOADS TO MMU FORMAT FOR REAL TIME UPLINK. | INTERACTIVE | ORBIT | PROFICIENT | 17460 | 3000 | 60 |
| IDUS | INTERACTIVE DATA UPDATE SYSTEM FOR MODIFYING DATA FLOW SCHEDULES | INTERACTIVE | DATA | EXPERT | 31790 | 1177600 | 1 |
| IPOL | (IPS POINTING LISTS PROGRAM) THIS PROGRAM OUTPUTS TWO FILES WHICH ARE LISTS DESCRIBING THE POINTINGS DONE BY THE IPS ON ASTRO MISSIONS. ONE IS AN ASCII FILE WHICH LISTS ALL THE ID'S OF SCHEDULED TARGETS IN ID ORDER. THIS FILE IS FOR THE PI TO ADD SEQUENCE NUMBERS TO AND THEN IT FEEDS INTO JOTF. THE SECOND OUTPUT FILE IS A LIST-DIRECTED FILE GIVING DATA ON EACH POINTING IN TIME ORDER. THIS FILE IS READ BY PCAP AND JOTF. | INTERACTIVE | ORBIT | PROFICIENT | 2200 | 2265000 | 30 |

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| SW NAME | SW FUNCTION | MODE OF USE | | | SKILL RIGHTS | | | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|-------|------|--------------|----------|------------|-------|-------------------|------------------|
| | | INTERACTIVE | ORBIT | TYPE | LEVEL | LANGUAGE | PROFICIENT | | | |
| JOTF | (JOINT OPERATIONS TARGET FILE PROGRAM) JOTF GENERATES THE JOINT OPERATIONS TARGET FILE WHICH IS AN MMJ FORMAT FILE THAT IS UPLINKED TO THE ON-BOARD COMPUTER. THE JOTF FILE CONTAINS DATA FOR EACH OBSERVATION AND THAT DATA IS DISPLAYED ON BOARD BY CREW MEMBERS WHEN CALLED UP BY TARGET ID. THE DATA INCLUDES TARGET ID, NAME, RA, DEC, OBJECTIVE LOAD ROLL, START AND STOP TIMES, SEQUENCE NUMBERS, ETC. INPUTS REQUIRED BY JOTF ARE AN IPOL FILE, A SEQUENCE FILE, AN ASTRON SCHEULE, AND A COO FILE. | INTERACTIVE | ORBIT | | PROFICIENT | ORBIT | | 1300 | 1184000 | 45 |
| LANTIM | GENERATE AN INSERTAION VECTOR FROM ANY OF 4 SETS OF INPUT DATA. | INTERACTIVE | ORBIT | | PROFICIENT | FORTRAN | | 2548 | 1228000 | 10 |
| LTO-1 | (LIST DIRECTED TO ON/OFF FILE). THE LTO PROGRAM TAKES AS INPUT A USER SPECIFIED LIST-DIRECTED FILE AND ACCEPTANCE CONDITIONS TO APPLY TO THAT FILE. LTO READS THE INPUT FILE SCANNING FOR PERIODS OF TIME DURING WHICH THE SPECIFIED CONDITIONS ARE SATISFIED. THE OUTPUT IS AN ON/OFF SUBJECT FILE CONTAINING THE TIMES AT WHICH THE SPECIFIED CONDITIONS ARE SATISFIED. THIS OUTPUT SUBJECT MAY BE WRITTEN ON A NEW ON/OFF FILE OR ADDED TO A EXISTING ONE. | INTERACTIVE | ORBIT | | PROFICIENT | FORTRAN | | 351 | 1139000 | 15 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL RIGHTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|----------------------|------------|----------|-------|-------------------|------------------|
| LTO-2 | (LIST-DIRECTED TO ON/OFF FILE). THE LTO PROGRAM TAKES AS INPUT A USER SPECIFIED LIST-DIRECTED FILE AND ACCEPTANCE CONDITIONS TO APPLY TO THAT FILE. LTO READS THE INPUT FILE SCANNING FOR PERIODS OF TIME DURING WHICH THE SPECIFIED CONDITIONS ARE SATISFIED. THE OUTPUT IS AN ON/OFF SUBJECT CONTAINING THE TIMES AT WHICH THE SPECIFIED CONDITIONS ARE SATISFIED. THIS OUTPUT SUBJECT MAY BE WRITTEN ON A NEW ON/OFF FILE OR ADDED TO AN EXISTING ONE. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN | 351 | 1139000 | 150 |
| LTO-3 | (LIST DIRECTED TO ON/OFF FILE). THE LTO PROGRAM TAKES AS INPUT A USER SPECIFIED LIST-DIRECTED FILE AND ACCEPTANCE CONDITIONS TO APPLY TO THAT FILE. LTO READS THE INPUT FILE SCANNING FOR PERIODS OF TIME DURING WHICH THE SPECIFIED CONDITIONS ARE SATISFIED. THE OUTPUT IS AN ON/OFF SUBJECT FILE CONTAINING THE TIMES AT WHICH THE SPECIFIED CONDITIONS ARE SATISFIED. THIS OUTPUT SUBJECT MAY BE WRITTEN ON A NEW ON/OFF FILE OR ADDED TO A EXISTING ONE. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN | 351 | 1139000 | 30 |

SOFTWARE DESCRIPTION

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|----------------------|------------|----------|-------|-------------------|------------------|
| LTO-4 | (LIST DIRECTED TO ON/OFF FILE). THE LTO PROGRAM TAKES AS INPUT A USER SPECIFIED LIST-DIRECTED FILE AND ACCEPTANCE CONDITIONS TO APPLY TO THAT FILE. LTO READS THE INPUT FILE SCANNING FOR PERIODS OF TIME DURING WHICH THE SPECIFIED CONDITIONS ARE SATISFIED. THE OUTPUT IS AN ON/OFF SUBJECT FILE CONTAINING THE TIMES AT WHICH THE SPECIFIED CONDITIONS ARE SATISFIED. THIS OUTPUT SUBJECT MAY BE WRITTEN ON A NEW ON/OFF FILE OR ADDED TO A EXISTING ONE. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN | 351 | 1139000 | 60 |
| LWAP | TO COMPUTE AND/OR DISPLAY LAUNCH WINDOW OPEN/CLOSE DATA BASED UPON LAUNCH, LANDING, AOA, TAL, RTLS, SHADOW ENTRY/EXIT, AND BELTA ANGLE CONSTRAINT DATA. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN | 3900 | 2995000 | 12 |
| MET | APPENDS MET TIMETAGS TO ALL ITEMS IN THE INPUT FILES(S). GENERATES MULTIPLE ON/OFF TIMES FOR ACTIVITIES. CREATES SEPERATE SECTION FOR EACH .INP FILE THEN MERGES THEM TOGETHER. | INTERACTIVE | EXP CMD PLNG | PROFICIENT | FORTRAN | 2046 | 19712 | 30 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS | | | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|--|-------------|--------------|--------|-------|-------------------|------------------|
| | | | TYPE | LEVEL | LINES | | |
| MMUALL | (MMU ALLOCATION) CREATES A MODEL OF THE LOCATIONS OF THE MMU FILES ON THE MMU MAGNETIC TAPE, USING THE ADDRESSES AND FILENAMES SPECIFIED BY THE INPUT. THE MODEL IS THEN USED TO SUM THE MOVEMENT BETWEEN TAPE LOCATIONS FOR THE ENTIRE MISSION. THE OUTPUT IS A SINGLE NUMBER REPRESENTING THE TOTAL DISTANCE TRAVELED (AND TOTAL FILE ACCESS WAIT TIME). | INTERACTIVE | MMU | EXPERT | 689 | 1000 | 30 |

| | | | | | | | | |
|------|--|-------------|-------|------------|------------|------|---------|----|
| NSEP | (NUMERICAL SATELLITE EPHEMERIS PROGRAM). THIS PROGRAM GENERATES AN ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION OF THE COMELL EQUATIONS OF MOTION. IT CONSIDERS THE SPACECRAFT'S ATTITUDE IN DETERMINING A COEFFICIENT OF DRAG AND THEN INTEGRATES THE EFFECTS OF ATMOSPHERIC DRAG USING JACCHIA'S MODEL OF THE ATMOSPHERE. INPUT INTO NSEP CONSISTS OF LAUNCH DATE AND TIME, AN INSERTION VECTOR, A CAVA INPUT FILE FROM WHICH ATTITUDE DATA IS READ, SOLAR ACTIVITY TABLES, VEHICLE MASS AND VEHICLE REFERENCE AREA. A COEFFICIENT OF DRAG TABLE MAY BE USED INSTEAD OF THE ATTITUDE DATA. OUTPUT IS IN THE FORM OF NSEP EPHEMERIS FILE. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN 77 | 3991 | 1211000 | 30 |
|------|--|-------------|-------|------------|------------|------|---------|----|

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|--|-------------|----------------------|------------|----------|-------|-------------------|------------------|
| PAAC | (PROGRAM FOR ASTRONOMY ATTITUDE CALCULATIONS) THIS PROGRAM READS IN AN ON/OFF FILE AS WRITTEN BY ASTAR AND OUTPUTS A MANEUVER TIMELINE BY GOING BETWEEN TARGETS WITH EIGEN AXIS MANEUVERS. PAAC OUTPUTS AN ON/OFF MANEUVER TIMELINE AND A KEYWORD FILE SUITABLE FOR READING BY CAVA. | INTERACTIVE | ORBIT | EXPERT | FORTRAN | 740 | 1392000 | 30 |
| PCAP | BUILD, DISPLAY & PRINT PAYLOAD CREW ACTIVITY PLAN (PCAP) CHARTS WHICH DESCRIBE THE HOUR-BY-HOUR ACTIVITIES OF THE PAYLOAD CREW FOR SPACE SHUTTLE MISSIONS. THE MAIN PRODUCT OF THE PROGRAM IS DETAILED 1 HOUR CHARTS, PRODUCED ON A HIGH RESOLUTION LASER PRINTER, WHICH BECOME PART OF THE PAYLOAD FLIGHT DATA FILE (PDF). THE PROGRAM ALLOWS THE USER TO DEFINE THE LAYOUT OF THE PLAN, TO SELECT THE DATA SOURCES, AND TO EDIT THE CREW PROCEDURES & NOTES. | INTERACTIVE | TIMELINE | PROFICIENT | FORTRAN | 27000 | 2300 | 120 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQTS | | | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|-------------|------------|------------|---------|----------------|---------------|
| | | | TYPE | LEVEL | PROFICIENT | | | |
| PHDSG | (POCC MMU DATA SET) GENERATOR. THIS PROGRAM GENERATES THE POCC MMU (MASS MEMORY UNIT) DATA SETS. THIS IS USED BY THE COMMAND SHELL FILLER PROGRAM TO GENERATE USER DATA SETS FOR EVENTUAL UPLINKING. DATA FROM AN ASEP NODE FILE, AND ON/OFF FILES CONTAINING SUN RISE/SET, MOON RISE/SET AND TDRS COVERAGE ARE READ IN. THE USER MUST SPECIFY THE START TIME OF THE DATA SET. A VERIFICATION REPORT WILL ALSO BE GENERATED IF DESIRED. | INTERACTIVE | ORBIT | PROFICIENT | 1200 | 1228000 | 20 | |

| | | | | | | | |
|--------|--|-------------|-------|------------|------|---------|----|
| PROCAM | (PROPELLANT CONSUMED BY ATTITUDE MANEUVERS) PROCAM CALCULATES THE AMOUNT OF RCS PROPELLANT CONSUMED BY ATTITUDE MANEUVERS DURING A MISSION. IT ALSO, AND MORE IMPORTANTLY, CALCULATES ATTITUDE RELATED INFORMATION. PROCAM READS IN AN ASEP NODE FILE AND A CAVA ATTITUDE TIMELINE FILE AND OUTPUTS A LIST DIRECTED FILE AND A CAVA ATTITUDE TIMELINE FILE AND OUTPUTS A LIST DIRECTED FILE. THE OUTPUT FILE CONTAINS RCS USAGE DATA ALONG WITH ATTITUDE RELATED DATA SUCH AS THE ATTITUDE AND ATTITUDE RATES IN 3 REFERENCE FRAMES; AZIMUTH AND ELEVATION OF THE RADIUS, VELOCITY, ANGULAR MOMENTUM, LUNAR AND SOLAR VECTORS; AND THE INERTIAL TO BODY TRANSFORMATION MATRIX. | INTERACTIVE | ORBIT | PROFICIENT | 4200 | 1328000 | 30 |
|--------|--|-------------|-------|------------|------|---------|----|

SOFTWARE DESCRIPTION

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQTS | | | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|-------------|------------|------------|----------|---------|-------------------|------------------|
| | | | TYPE | LEVEL | PROFICIENT | | | | |
| PTS | PRODUCE PAYLOAD TIMELINE SUMMARY (PTS) CHARTS WHICH SUMMARIZE THE PAYLOAD OPERATIONS OF A SHUTTLE MISSION IN 6-HOUR INCREMENTS. | INTERACTIVE | TIMELINE | PROFICIENT | FORTRAM | 6200 | 1400 | 120 | |
| RAD12 | (RADIATION 11 PROGRAM) THE RAD12 PROGRAM READS IN AN ASEP LIST-DIRECTED FILE (GROUND TRACK OR DETAILED) AND CALCULATES THE FLUX OF CHARGED PARTICLES AT EACH POINT ON THE FILE (LATITUDE, LONGITUDE AND ALTITUDE) ABOVE A SPECIFIED ENERGY THRESHOLD. THE FLUX CALCULATED IS FOR PROTONS, ELECTRONS AT SOLAR MAXIMUM AND ELECTRONS AT SOLAR MINIMUM. IT IS DETERMINED BY USING SOLAR MIN AND MAX ELECTRON MODELS AND SMOOTHER KLUGU-LENGHART PROTONS CONTAINED IN THE PROGRAM. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAM | 8060 | 1252000 | 30 | |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|--|-------------|----------------------|------------|----------|-------|-------------------|------------------|
| READPI | (READ PI FILE) READPI IS DESIGNED TO CREATE AN INPUT FILE FOR ASTAR BY READING A PI PARAMETER (COO) FILE PLUS THE APPROPRIATE TARGET ACQ/LOSS, SHADOW, SAA AND TIMEGOAL FILES, AND STRUCTURES THIS DATA IN A FORM THAT ASTAR CAN INTERPRET. IT ALSO PROVIDES THE CAPABILITY TO EDIT INDIVIDUAL PARAMETERS IN THE COO FILES AND PERFORMS A NUMBER OF QUALITY CHECKS ON THE INPUT DATA FILES. AN ANGULAR DISTANCE MATRIX FOR ALL COMBINATIONS OF TARGETS IS THEN COMPUTED, AND, IF THE OPTION IS SELECTED, A SUN AVOIDANCE ROUTINE DETERMINES A "FORBIDDEN PAIR" LIST TO CURRENT ASTAR FROM SCHEDULING A MANEUVER PATH TOO CLOSE TO THE SUN. | INTERACTIVE | ORBIT | PROFICIENT | FORTAN | 2900 | 3468000 | 30 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|--|-------------|----------------------|------------|----------|-------|-------------------|------------------|
| RELMO | RELMO IS DESIGNED TO GENERATE A TIMELINE HISTORY OF POSITIONS AND VELOCITIES FOR A SUBSATELLITE RELATIVE TO A SPACECRAFT BY SETTING UP A COORDINATE SYSTEM CENTERED IN ONE OF THE BODIES, AND USING THE DIFFERENTIAL EQUATIONS OF MOTIONS FOR THE OTHER BODY TO OBTAIN AN APPROXIMATE SOLUTION. THE PROGRAM OFFERS THE USER FOUR TARGETING OPTIONS AND PERFORMS THE CALCULATIONS FOR THE SELECTED OPTION. THE CAPABILITY EXISTS FOR DISPLAYING THIS TRAJECTORY DATA ON THE SCREEN, PRINTING THE DATA AND/OR STORING THE DATA ON A FILE. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN | 2700 | 1241000 | 60 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|--|-------------|----------------------|------------|----------|-------|-------------------|------------------|
| SALES | (SATELLITE ACQUISITION AND LOSS OVER EARTH SITE) THIS PROGRAM COMPUTES THE TIMES THAT AN EARTH SATELLITE ACQUIRES AND LOSES AN EARTH SITE. ACQUISITION/LOSS IS DEFINED BY AN ELEVATION ANGLE CONSTRAINT. FOR PURPOSES OF DETERMINING THE TIMES THAT A SATELLITE IS IN COMMUNICATION WITH A TRACKING STATION, TERRAIN EFFECTS CAN BE SIMULATED. TABULAR DATA AND PLOTS PROVIDE SUMMARY INFORMATION SUCH AS THE EARTH-FIXED AZIMUTH AND ELEVATION OF THE VEHICLE, THE TIME AND DURATION OF THE PASS, MAXIMUM ELEVATION ANGLE REACHED. SALES READS AN ASEP EPHEMERIS FILE TO GET THE ORBITAL DATA NEEDED. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN | 2886 | 1512000 | 45 |

| | | | | | | | | |
|---------|--|-------------|-------|------------|------------|------|---------|----|
| SCATGEN | THE STAR CATALOG GENERATOR (SCATGEN) PROGRAM IS USED TO SELECT IPS GUIDE STARS AND GENERATE IPS OBJECTIVE LOADS FOR SOLAR OBSERVATIONS. A CATALOG OF CANDIDATE GUIDE STARS ARE SELECTED BY APPLYING CONSTRAINTS TO VARIOUS VARIABLES, AND MAKING COMPUTATIONS FROM VARIOUS VARIABLES ON THE MSFC SKYMAP CATALOG. ANY COMPUTATION CAN BE FILTERED BY APPLYING A CONSTRAINT TO ITS OUTPUT. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN 77 | 4000 | 3733000 | 30 |
|---------|--|-------------|-------|------------|------------|------|---------|----|

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| SV NAME | SW FUNCTION | MODE OF USE | SKILL RECHTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|--|-------------|----------------------|------------|------------|-------|-------------------|------------------|
| STAR-1 | (STELLAR TARGET ACQUISITION AND LOSS ROUTINE) THIS PROGRAM ANALYTICALLY COMPUTES THE ACQUISITION AND LOSS TIMES OF CELESTIAL OBJECTS FROM ORBIT. ALSO CALCULATED ARE THE LOCATION IN THE ORBIT PLANE OF THE ACQUISITION AND LOSS, OBSERVATION TIME PER ORBIT, OBSERVATION TIME IN DARKNESS AND THE ANGLE BETWEEN THE TARGET AND THE SUN OR MOON. INPUT NEEDED ARE AN ASEP NODE FILE, A NAME-DIRECTED FILE CONTAINING A LIST OF TARGETS, AND A USER SPECIFIED ORBITAL ELEVATION ANGLE. OUTPUT IS IN THE FORM OF TABULATION AND/OR AN ON/OFF FILE. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAM 77 | 2054 | 1308000 | 15 |
| STAR-2 | (STELLAR TARGET ACQUISITION AND LOSS ROUTINE) THIS PROGRAM ANALYTICALLY COMPUTES THE ACQUISITION AND LOSS TIMES OF CELESTIAL OBJECTS FROM ORBIT. ALSO CALCULATED ARE THE LOCATION IN THE ORBIT PLANE OF THE ACQUISITION AND LOSS, OBSERVATION TIME PER ORBIT, OBSERVATION TIME IN DARKNESS AND THE ANGLE BETWEEN THE TARGET AND THE SUN OR MOON. INPUT NEEDED ARE AN ASEP NODE FILE, A NAME-DIRECTED FILE CONTAINING A LIST OF TARGETS, AND A USER SPECIFIED ORBITAL ELEVATION ANGLE. OUTPUT IS IN THE FORM OF TABULATION AND/OR AN ON/OFF FILE. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAM 77 | 2054 | 1308000 | 60 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQTS | | | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|-------------|------------|-------|----------------|---------------|
| | | | TYPE | LEVEL | LINES | | |
| STLBUF | GENERATE SUBORDINATE TIMELINE BUFFER UTILIZATION REPORT | AUTOMATIC | MMU | PROFICIENT | 3648 | 48640 | 600 |
| SUBCOO | SUBCOO IS A PROGRAM USED TO CREATE NEW FILES THAT ARE A SUBSET OF THE PI COOBSERVATION FILE. IT GENERATES SUBSETS ACCORDING TO USER CONTROLLED CONSTRAINTS BY INTERSECTIONS, UNIONS AND INDIVIDUAL PARAMETERS. IT CAN WRITE AN ASCII FILE LIKE THE COOBSERVATION FILE OR A NAME-DIRECTED FILE FOR OTHER SOFTWARE. | INTERACTIVE | ORBIT | PROFICIENT | 1500 | 1261000 | 30 |
| TAE | THE TARGET/ATTITUDE EDITOR (TAE) PROGRAM PERFORMS DATA BASE OPERATIONS ON ESS TARGET/ATTITUDE FILES. THE OPERATIONS AVAILABLE INCLUDE EDITING, DELETING AND ADDING ACQUISITION PERIODS; DELETING SUBJECTS; CREATING UNIONS, / INTERSECTIONS & COMPLEMENTS OF SUBJECTS; COPYING SUBJECTS FROM OTHER FILES AND DISPLAYING & PRINTING DATA. THE PROGRAM CAN READ O/O FILES & CONVERT THEM TO THE ESS TARGET/ATTITUDE FORMAT. WHEN RE-FORMATTING, ACQUISITION PERIODS CAN BE MAINTAINED EXACTLY OR EXPANDED TO THE NEXT WHOLE MINUTE ON EACH END. | INTERACTIVE | TIMELINE | PROFICIENT | 10000 | 3100 | 30 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS | | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|--|-------------|--------------|------------|----------|-------|----------------|---------------|
| | | | TYPE | LEVEL | | | | |
| TANRAY | (TANGENT RAY PROGRAM) TANRAY COMPUTES A DETAILED TIME HISTORY OF SUN RISES AND SUN SETS AS SEEN ALONG A LINE OF SIGHT (RAY) FROM THE ORBITER TO THE SUN. THAT LINE WILL HAVE A POINT ON IT WHICH IS CLOSEST TO THE SURFACE OF THE EARTH, REFERED TO AS THE TANGENT POINT. THE USER INPUTS A VECTOR FOR THE ORBIT WHICH TANRAY PROPOGATES INTERNALLY, A PAIR OF START AND STOP TIMES, A DELTA TIME FOR THE CALCULATIONS, AND A MINIMUM AND MAXIMUM TANGENT RAY HEIGHT. OUTPUTS FROM TANRAY ARE A LIST-DIRECTED FILE OF 63 VARIABLES, A TABULATION OF A SUB-SET OF THOSE VARIABLES TO THE SCREEN, AND/OR ON/OFF FILE OF SUN RISE/SET TIME HISTORY. | INTERACTIVE | ORBIT | PROFICIENT | FORTRAN | 1625 | 1387000 | 60 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|---|-------------|----------------------|------------|-----------|-------|-------------------|------------------|
| TARGEN | (TARGET GENERATOR PROGRAM). THIS PROGRAM IS USED TO GENERATE EXPERIMENT OPERATION OPPORTUNITIES FROM VARIOUS FILES AND FILE TYPES WHICH ARE USED IN ORBITAL ANALYSIS AND MISSION PLANNING. THESE TARGETS OF OPPORTUNITY ARE GENERATED IN TARGEN FROM TARGET DEFINITIONS THE USER CREATES AND MAY STORE. THESE TARGET DEFINITIONS SPECIFY CONSTRAINTS TO BE APPLIED TO FILES AND HOW THESE FILES ARE THEN COMBINED USING ELEMENTARY SET THEORY. THE INPUT FILES MAY BE LIST-DIRECTED OR ON/OF, AND THE OUTPUT FILES ARE ON/OFF TYPE. TARGEN ALSO HAS OPTIONS TO TABULATE OR DRAW BAR CHARTS OF DATA ON AN ON/OFF FILE. | INTERACTIVE | ORBIT | PROFICIENT | FORTAN | 10556 | 1613000 | 15 |
| VECTOR | THIS PROGRAM TAKES AS INPUT A LAUNCH DATE AND TIME AND A STATE VECTOR EXACTLY AS ACQUIRED FROM JSC (IN ENGLISH UNITS). IT THEN CONVERTS THAT VECTOR TO METRIC UNITS AND DISPLAYS THE EQUIVALENT IN MEAN CLASSICAL ELEMENTS FOR THE USER TO CHECK ITS VALIDITY. THE PROGRAM THEN HAS THE ABILITY TO STORE THE VECTOR IN ITS APPROPRIATE FORM AND UNITS IN THE CASE STORAGE FILES OF ASEP, NSEP AND/OR TANRAY. THIS ALLOWS THE USER TO KEY IN THE VECTOR ONLY ONCE FOR ALL THREE PROGRAMS. | INTERACTIVE | ORBIT | PROFICIENT | FORTAN 77 | 1469 | 1378000 | 10 |

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| SW NAME | SW FUNCTION | MODE OF USE | SKILL REQMTS TYPE | LEVEL | LANGUAGE | LINES | MEMORY (BYTES) | RUNTIME (MIN) |
|---------|--|-------------|----------------------|------------|----------|-------|-------------------|------------------|
| VERMI | TO VERIFY THAT THE INPUT .MI FILE IS LEGAL AND FREE OF ALL TYPOGRAPHIC ERROS. | INTERACTIVE | MMU | PROFICIENT | FORTRAN | 826 | 49664 | 5 |
| VERMTL | VERIFY SYNTAX OF ALL MASTER TIMELINES | INTERACTIVE | MMU | PROFICIENT | FORTRAN | 1500 | 91136 | 60 |
| VERSTL | TO VERIFY THE SYNTAX OF SUBORDINATE TIMELINES. | INTERACTIVE | MMU | PROFICIENT | FORTRAN | 1631 | 60416 | 1 |
| VME | THE VT100 MODEL EDITOR PROGRAM(VME) IS A DATA BASE MANAGEMENT TOOL USED TO CREATE, MODIFY AND COPY ESS EXPERIMENT MODEL FILES. VME MAKES EXTENSIVE USE OF FORM EDITING TECHNIQUES TO EDIT THE DATA. IT USES THE SPECIAL FEATURES OF THE VT100 SUCH AS DIRECT CURSOR POSITIONING, SCROLLING AND VIDEO RENDITIONS TO PRESENT DATA AND SOLICIT INPUTS. | INTERACTIVE | TIMELINE | PROFICIENT | FORTRAN | 40000 | 1900 | 60 |

TABLE 5
SOFTWARE PERIPHERALS REQUIRED

| SOFTWARE NAME | PERIPHERAL REQUIRED |
|---------------|---------------------|
| ASEP | TEKTRONIX 4014 |
| ASTAR | ANY TERMINAL |
| ASTRO | ANY TERMINAL |
| ATNOS | ANY TERMINAL |
| BORB | TEKTRONIX 4014 |
| CAVA/CAVINP | TEKTRONIX 4014 |
| CAVA/KEYGEN | TEKTRONIX 4014 |
| CG | VAX TERMINAL |
| CHECK | VAX TERMINAL |
| CHDATG | VAX TERMINAL |
| DEL.CON | VAX TERMINAL |
| DF/DFRG | TEKTRONIX 4014 |
| DF/DVM | VAX TERMINAL |
| DF/HFS | VAX TERMINAL |
| DF/HPFG | VAX TERMINAL |
| DF/MDRPG | VAX TERMINAL |
| DF/MWG | VAX TERMINAL |
| DF/ORS | VAX TERMINAL |
| DF/PBS | VAX TERMINAL |
| DF/PCS | VAX TERMINAL |
| DF/PPCG | VAX TERMINAL |
| EDT | VAX TERMINAL |
| ESAL | TEKTRONIX 4014 |
| ESDAT | TEKTRONIX 4014 |
| ESP | VT241,100,TEKTR. |
| GENMTL | VAX TERMINAL |
| GIMBAL | ANY TERMINAL |
| GSOLP | ANY TERMINAL |
| IDUS | VAX TERMINAL |
| IPOI | ANY TERMINAL |
| JOTF | ANY TERMINAL |
| LANTIM | ANY TERMINAL |
| LTO | ANY TERMINAL |
| LWAP | ANY TERMINAL |
| MET | VAX TERMINAL |
| MUALL | VAX TERMINAL |

| SOFTWARE NAME | PERIPHERAL REQUIRED |
|---------------|---------------------|
| NSEP | ANY TERMINAL |
| PAAC | ANY TERMINAL |
| PCAP | VT241 |
| PMDSG | ANY TERMINAL |
| PROCAM | ANY TERMINAL |
| PTS | VT100, TEKTR. |
| RAD12 | ANY TERMINAL |
| READPI | ANY TERMINAL |
| RELMO | TEKTRONIX 4014 |
| SALES | TEKTRONIX 4014 |
| SCATGEN | ANY TERMINAL |
| STAR | ANY TERMINAL |
| STLBUF | VAX TERMINAL |
| SUBCOO | ANY TERMINAL |
| TAE | VT241, 100, TEKTR. |
| TANRAY | TEKTRONIX 4014 |
| TARGEN | ANY TERMINAL |
| VECTOR | ANY TERMINAL |
| VERMI | VAX TERMINAL |
| VERMTL | VAX TERMINAL |
| VERSTL | VAX TERMINAL |
| VME | VT241, 100, TEKTR. |

TABLE 6
ACTIVITY INPUT/OUTPUTS

ACTIVITY INPUT/OUTPUTS

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | |
|---|-----------------------------------|----------|-------------------|------|---|----------------------------|-------|------------|
| | | | | | | PREL | BASIC | UPDT RPLNG |
| PAYLOAD DATA COLLECTION | PI INTERFACE | MANUAL | NONE | I | PI DISCUSSIONS | Y | Y | N |
| PAYLOAD DATA COLLECTION | ERD'S | MANUAL | NONE | I | PI DEVELOPED (EXPERIMENT REQUIREMENTS) | Y | Y | N |
| PAYLOAD DATA COLLECTION | PAYLOAD COMPLEMENT DEFINITION | MANUAL | NONE | I | HQORS | Y | Y | N |
| PAYLOAD DATA COLLECTION | INPUTS/UPDATES TO O&IA | MANUAL | NONE | O | CREATE DATA FLOW MODEL(S) | Y | Y | N |
| PAYLOAD DATA COLLECTION | INPUTS/UPDATES TO IPRD | MANUAL | NONE | O | PI'S (FOR ASSESSMENT OF ALLOCATIONS) | Y | Y | N |
| PAYLOAD DATA COLLECTION | INPUTS/UPDATES TO O&IA | MANUAL | NONE | O | ORBIT REQUIREMENTS EVALUATION AND SELECTION | Y | Y | N |
| PAYLOAD DATA COLLECTION | INPUTS/UPDATES TO O&IA | MANUAL | NONE | O | CREATE MISSION TIMELINE MODEL(S) | Y | Y | N |
| ORBIT REQUIREMENTS EVALUATION AND SELECTION | PI INTERFACE | MANUAL | NONE | I | PI DISCUSSIONS | Y | Y | N |
| ORBIT REQUIREMENTS EVALUATION AND SELECTION | ERD'S | MANUAL | NONE | I | PI DEVELOPED (EXPERIMENT REQUIREMENTS) | Y | N | N |
| ORBIT REQUIREMENTS EVALUATION AND SELECTION | STS CAPABILITIES DOCUMENTATION | MANUAL | NONE | I | STS CTR DEVELOPED | N | Y | N |

ACTIVITY INPUT/OUTPUTS

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING PREL BASIC UPDT RPLNG |
|---|-----------------------------------|----------|-------------------|------|---|---|
| ORBIT REQUIREMENTS EVALUATION AND SELECTION | O&IA | MANUAL | NONE | I | PL CTR DEVELOPED | N Y Y N |
| ORBIT REQUIREMENTS EVALUATION AND SELECTION | ORBIT DEFINITION PARAMETERS | MANUAL | NONE | O | LAUNCH WINDOW/LAUNCH TIME SELECTION | Y Y Y N |
| ORBIT REQUIREMENTS EVALUATION AND SELECTION | ORBIT DEFINITION PARAMETERS | MANUAL | NONE | O | GENERATE STATE VECTOR | Y Y Y N |
| LAUNCH WINDOW/LAUNCH TIME SELECTION | PL/EXP CONSTRAINTS | MANUAL | LWAP | I | DOCUMENTATION (ERD'S,O&IA'S), PI INTERFACE | Y Y Y N |
| LAUNCH WINDOW/LAUNCH TIME SELECTION | STS CONSTRAINTS | MANUAL | LWAP | I | STS CAPABILITIES DOCUMENTATION | Y Y Y N |
| LAUNCH WINDOW/LAUNCH TIME SELECTION | ORBIT DEFINITION PARAMETERS | MANUAL | LWAP | I | ORBIT REQUIREMENTS EVALUATION AND SELECTION | Y Y Y N |
| LAUNCH WINDOW/LAUNCH TIME SELECTION | LAUNCH WINDOW/LAUNCH TIME DATA | COMPUTER | LWAP | O | GENERATE STATE VECTOR | Y Y Y N |
| LAUNCH WINDOW/LAUNCH TIME SELECTION | LAUNCH WINDOW/LAUNCH TIME DATA | COMPUTER | LWAP | O | GENERATE CANDIDATE SOLAR GUIDE STARS | Y Y Y N |
| GENERATE STATE VECTOR | ORBIT DEFINITION PARAMETERS | MANUAL | LANTIM | I | ORBIT REQUIREMENTS EVALUATION AND SELECTION | Y N N N |

ACTIVITY INPUT/OUTPUTS

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. | | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | | |
|----------------------------|--------------------------------|----------|-----------|--|------|--|----------------------------|-------|------|-------|
| | | | WITH | | | | PREL | BASIC | UPDT | RPLNG |
| GENERATE STATE VECTOR | LAUNCH WINDOW/LAUNCH TIME DATA | MANUAL | LANTIM | | I | LAUNCH WINDOW/LAUNCH TIME SELECTION | Y | N | N | N |
| GENERATE STATE VECTOR | STATE VECTOR PRINTOUT | COMPUTER | LANTIM | | O | COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | Y | N | N | N |
| GENERATE STATE VECTOR | STATE VECTOR PRINTOUT | COMPUTER | LANTIM | | O | GENERATE ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION | Y | N | N | N |
| GENERATE STATE VECTOR | STATE VECTOR PRINTOUT | COMPUTER | LANTIM | | O | COMPUTE DISTANCE FROM TANGENT POINT ON THE ORBITER TO SUN LINE-OF-SIGHT TO EARTH SURFACE | Y | N | N | N |
| GENERATE STATE VECTOR | STATE VECTOR PRINTOUT | COMPUTER | LANTIM | | O | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | N | N | N |
| CONVERT/STORE STATE VECTOR | STATE VECTOR ELECTRONIC | COMPUTER | VECTOR | | I | JSC | N | N | N | Y |
| CONVERT/STORE STATE VECTOR | STATE VECTOR FROM JSC | MANUAL | VECTOR | | I | JSC | Y | Y | Y | Y |
| CONVERT/STORE STATE VECTOR | CASE STO | COMPUTER | VECTOR | | O | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y | Y | Y |

ACTIVITY INPUT/OUTPUTS

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|--|--------------------------|----------|-------------------|------|---|------|--|
| CONVERT/STORE STATE VECTOR | CASE STO | COMPUTER | VECTOR | O | GENERATE ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION | Y | Y Y Y Y |
| CONVERT/STORE STATE VECTOR | CASE STO | COMPUTER | VECTOR | O | GENERATE EARTH SHADOW ACQ/LOSS | Y | Y Y Y Y |
| GENERATE ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION | CASE STO | COMPUTER | NSEP | I | CONVERT/STORE STATE VECTOR | Y | Y Y Y Y |
| GENERATE ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION | STATE VECTOR PRINTOUT | MANUAL | NSEP | I | GENERATE STATE VECTOR | Y | Y Y Y Y |
| GENERATE ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION | NSEP EPHEM | COMPUTER | NSEP | O | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y Y Y Y |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | NSEP EPHEM | COMPUTER | ASEP | I | GENERATE ORBITAL EPHEMERIS BY NUMERICAL INTEGRATION | Y | Y Y Y Y |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | CASE STO | COMPUTER | ASEP | I | CONVERT/STORE STATE VECTOR | Y | Y Y Y Y |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | STATE VECTOR PRINTOUT | MANUAL | ASEP | I | GENERATE STATE VECTOR | Y | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | BASIC | UPDT | RPLNG | CYCLES INPUT/OUTPUT DURING |
|---|-------------------|----------|-------------------|------|---|------|-------|------|-------|----------------------------|
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | GND TRK | COMPUTER | ASEP | 0 | DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | Y | Y | Y | Y | Y |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | DETAIL EPHEM | COMPUTER | ASEP | 0 | GENERATE RADIATION ENVIRONMENT | Y | Y | Y | Y | Y |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | ASCN NODE | COMPUTER | ASEP | 0 | COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | Y | Y | Y | Y | Y |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | GND TRK | COMPUTER | ASEP | 0 | GENERATE PTS CHARTS | Y | Y | Y | Y | Y |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | ASCN NODE | COMPUTER | ASEP | 0 | GENERATE POCC MMU DATA SET | Y | Y | Y | Y | Y |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | DETAIL EPHEM | COMPUTER | ASEP | 0 | GENERATE GROUND STATION COVERAGE | Y | Y | Y | Y | Y |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | ASCN NODE | COMPUTER | ASEP | 0 | GENERATE PRELIMINARY TDRS COVERAGE | Y | Y | Y | Y | N |
| GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | ASCN NODE | COMPUTER | ASEP | 0 | GENERATE TDRS COVERAGE | Y | Y | Y | Y | Y |

ACTIVITY INPUT/OUTPUTS

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | BASIC | UPDT | RPLNG | CYCLES INPUT/OUTPUT DURING |
|--|-------------------|----------|-------------------|------|---|------|-------|------|-------|----------------------------|
| GENERATE REQUIRED EPIHEMERIS DATA FOR OUTPUT | DETAIL EPHM | COMPUTER | ASEP | 0 | GENERATE ORBITER POINTING DATA | Y | Y | Y | Y | |
| GENERATE REQUIRED EPIHEMERIS DATA FOR OUTPUT | GND TRK | COMPUTER | ASEP | 0 | DEVELOP CONSTRAINTS FOR SOLAR VIEWING PERIODS | Y | Y | Y | Y | |
| GENERATE REQUIRED EPIHEMERIS DATA FOR OUTPUT | GND TRK | COMPUTER | ASEP | 0 | GENERATE HEMISPHERE OPPORTUNITIES | Y | Y | Y | Y | |
| GENERATE REQUIRED EPIHEMERIS DATA FOR OUTPUT | DETAIL EPHM | COMPUTER | ASEP | 0 | DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | Y | Y | Y | Y | |
| GENERATE REQUIRED EPIHEMERIS DATA FOR OUTPUT | ASCN NODE | COMPUTER | ASEP | 0 | COMPUTE SUN AZIMUTH AND ELEVATION FROM ORBITING VEHICLE WITH RESPECT TO SUN RISE/SET EVENTS | N | N | N | Y | |
| GENERATE REQUIRED EPIHEMERIS DATA FOR OUTPUT | ASCN NODE | COMPUTER | ASEP | 0 | GENERATE MOON RISE/SET | Y | Y | Y | Y | |
| GENERATE REQUIRED EPIHEMERIS DATA FOR OUTPUT | ASCN NODE | COMPUTER | ASEP | 0 | GENERATE CELESTIAL TARGET(S) ACQ/LOSS | Y | Y | Y | Y | |
| GENERATE REQUIRED EPIHEMERIS DATA FOR OUTPUT | ASCN NODE | COMPUTER | ASEP | 0 | GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | Y | Y | Y | Y | |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|-----------------------------------|-------------------|----------|-------------------|------|---|------|--|
| GENERATE EARTH SHADOW ACQ/LOSS | CASE STO | COMPUTER | ASEP | I | CONVERT/STORE STATE VECTOR | Y | Y Y Y |
| GENERATE EARTH SHADOW ACQ/LOSS | EARTH SHADOW | COMPUTER | ASEP | O | GENERATE SUN RISE/SET | Y | Y Y Y |
| GENERATE EARTH SHADOW ACQ/LOSS | EARTH SHADOW | COMPUTER | ASEP | O | CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | Y | Y Y Y |
| GENERATE EARTH SHADOW ACQ/LOSS | EARTH SHADOW | COMPUTER | ASEP | O | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y | Y Y Y |
| GENERATE EARTH SHADOW ACQ/LOSS | EARTH SHADOW | COMPUTER | ASEP | O | COMPUTE SUN AZIMUTH AND ELEVATION FROM ORBITING VEHICLE WITH RESPECT TO SUN RISE/SET EVENTS | N | N N Y |
| GENERATE EARTH SHADOW ACQ/LOSS | EARTH SHADOW | COMPUTER | ASEP | O | COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | Y | Y Y Y |
| GENERATE SUN RISE/SET | EARTH SHADOW | COMPUTER | TARGEN | I | GENERATE EARTH SHADOW ACQ/LOSS | Y | Y Y Y |
| GENERATE SUN RISE/SET | SUN RISE/SET | COMPUTER | TARGEN | O | COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | Y | Y Y Y |
| GENERATE SUN RISE/SET | SUN RISE/SET | COMPUTER | TARGEN | O | MERGE MISSION INDEPENDENT TARGETS | Y | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. | | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | | |
|---------------------------------------|-------------------|----------|-----------|--------|------|--|----------------------------|------------------|---|---|
| | | | WITH | TARGET | | | PREL | BASIC UPDT RPLNG | | |
| GENERATE SUN RISE/SET | SUN RISE/SET | COMPUTER | | TARGET | O | COMPUTE ORBIT TERMINATOR TARGETS | Y | Y | Y | Y |
| GENERATE MOON RISE/SET | ASCN NODE | COMPUTER | | STAR | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y | Y | Y |
| GENERATE MOON RISE/SET | MOON RISE/SET | COMPUTER | | STAR | O | MERGE MISSION INDEPENDENT TARGETS | Y | Y | Y | Y |
| GENERATE PRELIMINARY TDRS COVERAGE | PREL ATT TL | COMPUTER | | CAVINP | I | DEVELOP PRELIMINARY ATTITUDE TIMELINE | Y | Y | Y | N |
| GENERATE PRELIMINARY TDRS COVERAGE | ASCN NODE | COMPUTER | | CAVINP | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y | Y | N |
| GENERATE PRELIMINARY TDRS COVERAGE | ORBITER OCCULT | COMPUTER | | CAVINP | I | JSC | Y | N | N | N |
| GENERATE PRELIMINARY TDRS COVERAGE | PREL TDRS AC/LOS | COMPUTER | | CAVINP | O | COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | Y | Y | Y | N |
| GENERATE PRELIMINARY TDRS COVERAGE | PREL TDRS AC/LOS | COMPUTER | | CAVINP | O | MERGE MISSION INDEPENDENT TARGETS | Y | Y | Y | N |
| GENERATE PRELIMINARY TDRS COVERAGE | PREL TDRS AC/LOS | COMPUTER | | CAVINP | O | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | Y | Y | Y | N |
| GENERATE GROUND STATION COVERAGE | DETAIL EPHEM | COMPUTER | | SALES | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y | Y | Y |

ACTIVITY INPUT/OUTPUTS

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING PREL BASIC UPDT RPLNG |
|-------------------------------------|--------------------------|----------|-------------------|------|---|---|
| GENERATE GROUND STATION COVERAGE | GND STA AC/LOS | COMPUTER | SALES | 0 | PROCESS TDRS DATA FOR ENHANCEMENT | Y Y Y Y |
| GENERATE GROUND STATION COVERAGE | GND STA AC/LOS | COMPUTER | SALES | 0 | MERGE MISSION INDEPENDENT TARGETS | Y Y Y Y |
| GENERATE RADIATION ENVIRONMENT | DETAIL EPHEM | COMPUTER | RAD12 | 1 | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y Y Y Y |
| GENERATE RADIATION ENVIRONMENT | RAD ENVIR | COMPUTER | RAD12 | 0 | IMPOSE RADIATION CONSTRAINTS | Y Y Y Y |
| IMPOSE RADIATION CONSTRAINTS | RAD ENVIR | COMPUTER | LTO | 1 | GENERATE RADIATION ENVIRONMENT | Y Y Y Y |
| IMPOSE RADIATION CONSTRAINTS | RADIATION CONSTRAINTS | MANUAL | LTO | 1 | PI DEVELOPED (RADIATION CONSTRAINTS) | Y Y Y Y |
| IMPOSE RADIATION CONSTRAINTS | SAA AC/LOS | COMPUTER | LTO | 0 | CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | Y Y Y N |
| IMPOSE RADIATION CONSTRAINTS | SAA AC/LOS | COMPUTER | LTO | 0 | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y Y Y Y |
| IMPOSE RADIATION CONSTRAINTS | SAA AC/LOS | COMPUTER | LTO | 0 | COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | Y Y Y Y |
| IMPOSE RADIATION CONSTRAINTS | SAA AC/LOS | COMPUTER | LTO | 0 | MERGE MISSION INDEPENDENT TARGETS | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|--|------------------------------|----------|-------------------|------|--|------|--|
| MERGE MISSION INDEPENDENT TARGETS | GND STA AC/LOS | COMPUTER | TARGEN | I | GENERATE GROUND STATION COVERAGE | Y | Y Y Y Y |
| MERGE MISSION INDEPENDENT TARGETS | MOON RISE/SET | COMPUTER | TARGEN | I | GENERATE MOON RISE/SET | Y | Y Y Y Y |
| MERGE MISSION INDEPENDENT TARGETS | SUN RISE/SET | COMPUTER | TARGEN | I | GENERATE SUN RISE/SET | Y | Y Y Y Y |
| MERGE MISSION INDEPENDENT TARGETS | PREL TDRS AC/LOS | COMPUTER | TARGEN | I | GENERATE PRELIMINARY TDRS COVERAGE | Y | Y Y Y N |
| MERGE MISSION INDEPENDENT TARGETS | SAA AC/LOS | COMPUTER | TARGEN | I | IMPOSE RADIATION CONSTRAINTS | Y | Y Y Y Y |
| MERGE MISSION INDEPENDENT TARGETS | PRINTOUTS OF MSN IND DATA | COMPUTER | TARGEN | O | DEVELOP GROSS MISSION TIMELINE | Y | Y Y Y Y |
| MERGE MISSION INDEPENDENT TARGETS | MSN IND TARGETS | COMPUTER | TARGEN | O | MERGE ALL EXPERIMENT TARGET FILES | Y | Y Y Y Y |
| MERGE MISSION INDEPENDENT TARGETS | MSN IND TARGETS | COMPUTER | TARGEN | O | GENERATE POCB MHU DATA SET | Y | Y Y Y Y |
| DEVELOP CELESTIAL TARGETS (NON-IPS MISSIONS) | CELESTIAL TARGETS | MANUAL | STAR | I | PI GENERATED (CELESTIAL TARGET REQMTS) | Y | Y Y Y N |
| DEVELOP CELESTIAL TARGETS (NON-IPS MISSIONS) | CEL TAR | COMPUTER | STAR | O | GENERATE CELESTIAL TARGET(S) ACQ/LOSS | Y | Y Y Y N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | BASIC | UPDT | RPLNG |
|--|------------------------------------|----------|-------------------|------|---|------|-------|------|-------|
| DEVELOP CELESTIAL TARGETS (IPS MISSIONS) | COO | COMPUTER | SUBCOO | I | PI DEVELOPED (CELESTIAL TARGETS OBSERVATION REQMTS) | Y | Y | Y | N |
| DEVELOP CELESTIAL TARGETS (IPS MISSIONS) | CEL TAR | COMPUTER | SUBCOO | O | GENERATE CELESTIAL TARGET(S) ACQ/LOSS | Y | Y | Y | N |
| DEVELOP CELESTIAL TARGETS (IPS MISSIONS) | SUBCOO | COMPUTER | SUBCOO | O | CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | Y | Y | Y | N |
| DEVELOP CELESTIAL TARGETS (IPS MISSIONS) | SUBCOO | COMPUTER | SUBCOO | O | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y | Y | Y | N |
| GENERATE CELESTIAL TARGET(S) ACQ/LOSS | ASCN NODE | COMPUTER | STAR | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y | Y | Y |
| GENERATE CELESTIAL TARGET(S) ACQ/LOSS | CEL TAR | COMPUTER | STAR | I | DEVELOP CELESTIAL TARGETS (NON-IPS MISSION) | Y | Y | Y | N |
| GENERATE CELESTIAL TARGET(S) ACQ/LOSS | CEL TAR | COMPUTER | STAR | I | DEVELOP CELESTIAL TARGETS (IPS MISSION) | Y | Y | Y | N |
| GENERATE CELESTIAL TARGET(S) ACQ/LOSS | CEL TARGET(S) ELEV ANGLE CONSTS | MANUAL | STAR | I | PI GENERATED (CELESTIAL TARGET(S) ELEVATION ANGLE CONSTRAINTS) | Y | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---|--------------------------------|----------|-------------------|------|---|------|--|
| GENERATE CELESTIAL TARGET(S) ACQ/LOSS | CEL TAR AC/LOS | COMPUTER | STAR | 0 | COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | CEL TAR AC/LOS | COMPUTER | TARGET | I | GENERATE CELESTIAL TARGET(S) ACQ/LOSS | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | EARTH SHADOW | COMPUTER | TARGET | I | GENERATE EARTH SHADOW ACQ/LOSS | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | SAA AC/LOS | COMPUTER | TARGET | I | IMPOSE RADIATION CONSTRAINTS | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | CEL TARGETS | COMPUTER | TARGET | 0 | MERGE ALL EXPERIMENT TARGET FILES | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | CEL TARGETS | COMPUTER | TARGET | 0 | CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | PRINTOUTS OF CELESTIAL DATA | COMPUTER | TARGET | 0 | DEVELOP GROSS MISSION TIMELINE | Y | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING PREL BASIC UPDT RPLNG |
|--|--------------------------|----------|-------------------|------|--|---|
| COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | CEL TARGETS | COMPUTER | TARGET | O | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y Y Y Y |
| COMPUTE DISTANCE FROM TANGENT POINT ON THE ORBITER TO SUN LINE-OF-SIGHT TO THE EARTH SURFACE | STATE VECTOR PRINTOUT | MANUAL | TANRAY | I | GENERATE STATE VECTOR | Y Y Y N |
| COMPUTE DISTANCE FROM TANGENT POINT ON THE ORBITER TO SUN LINE-OF-SIGHT TO THE EARTH SURFACE | TANRAY EPHEM | COMPUTER | TANRAY | O | DEVELOP/APPLY CONSTRAINTS TO ATMOSPHERIC PHYSICS OBSERVATION PERIODS | Y Y Y Y |
| DEVELOP/APPLY CONSTRAINTS TO ATMOSPHERIC PHYSICS OBSERVATION PERIODS | TANRAY EPHEM | COMPUTER | LTO | I | COMPUTE DISTANCE FROM TANGENT POINT ON THE ORBITER TO SUN LINE-OF-SIGHT TO EARTH SURFACE | Y Y Y Y |
| DEVELOP/APPLY CONSTRAINTS TO ATMOSPHERIC PHYSICS OBSERVATION PERIODS | AT PHY CONSTS | COMPUTER | LTO | O | COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | Y Y Y Y |
| COMPUTE ORBIT TERMINATOR TARGETS | SUN RISE/SET | COMPUTER | TARGET | I | GENERATE SUN RISE/SET | Y Y Y Y |
| COMPUTE ORBIT TERMINATOR TARGETS | P1 ORBIT TERM REQMTS | MANUAL | TARGET | I | P1 GENERATED (ORBIT TERMINATOR REQMTS) | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES PREL | INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---|------------------------------------|----------|-------------------|------|---|----------------|---|
| COMPUTE ORBIT TERMINATOR TARGETS | TERM AC/LOS | COMPUTER | TARGEN | O | COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | TERM AC/LOS | COMPUTER | TARGEN | I | COMPUTE ORBIT TERMINATOR TARGETS | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | AT PHY CONSTS | COMPUTER | TARGEN | I | DEVELOP/APPLY CONSTRAINTS TO ATMOSPHERIC PHYSICS OBSERVATION PERIODS | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | PRINTOUTS OF ATMOS PHYSICS DATA | COMPUTER | TARGEN | O | DEVELOP GROSS MISSION TIMELINE | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | AT PHY TARGETS | COMPUTER | TARGEN | O | MERGE ALL EXPERIMENT TARGET FILES | Y | Y Y Y Y |
| COMPUTE SUN AZIMUTH AND ELEVATION FROM ORBITING VEHICLE WITH RESPECT TO SUN RISE/SET EVENTS | ASCN NODE | COMPUTER | ATMOS | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | N | N N N Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---|-----------------------------------|----------|-------------------|------|--|------|---|
| COMPUTE SUN AZIMUTH AND ELEVATION FROM ORBITING VEHICLE WITH RESPECT TO SUN RISE/SET EVENTS | EARTH SHADOW | COMPUTER | ATMOS | I | GENERATE EARTH SHADOW ACQ/LOSS | N | N N Y |
| COMPUTE SUN AZIMUTH AND ELEVATION FROM ORBITING VEHICLE WITH RESPECT TO SUN RISE/SET EVENTS | ATMOS DESIRED VEHICLE ATTITUDE | MANUAL | ATMOS | I | PI (ATMOS EXP.) GENERATED (DESIRED VEHICLE ATTITUDE) | N | N N Y |
| COMPUTE SUN AZIMUTH AND ELEVATION FROM ORBITING VEHICLE WITH RESPECT TO SUN RISE/SET EVENTS | SUN AZ/ELV | COMPUTER | ATMOS | O | TO MMUM FOR REALTIME UPDATE TO THE EXP. DEP | N | N N Y |
| DEVELOP CONSTRAINTS FOR SOLAR VIEWING PERIODS | GND TRK | COMPUTER | LTO | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y Y Y |
| DEVELOP CONSTRAINTS FOR SOLAR VIEWING PERIODS | SUN ELEV CONSTRAINTS (SOLAR) | MANUAL | LTO | I | PI GENERATED (SOLAR VIEWING SUN ELEVATION ANGLE CONSTRAINTS) | Y | Y Y Y |
| DEVELOP CONSTRAINTS FOR SOLAR VIEWING PERIODS | SOLAR CONSTS | COMPUTER | LTO | O | COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | Y | Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | SUN RISE/SET | COMPUTER | TARGET | I | GENERATE SUN RISE/SET | Y | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING PREL BASIC UPDT RPLNG |
|--|----------------------------|----------|-------------------|------|---|---|
| COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | PREL TDRS AC/LOS | COMPUTER | TARGEN | I | GENERATE PRELIMINARY TDRS COVERAGE | Y Y Y N |
| COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | SOLAR CONSTS | COMPUTER | TARGEN | I | DEVELOP CONSTRAINTS FOR SOLAR VIEWING PERIODS | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | SOLAR TARGETS | COMPUTER | TARGEN | O | MERGE ALL EXPERIMENT TARGET FILES | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE SOLAR DATA TARGETS | PRINTOUTS OF SOLAR DATA | COMPUTER | TARGEN | O | DEVELOP GROSS MISSION TIMELINE | Y Y Y Y |
| CREATE EARTH SITE DEFINITION FILE | GROUND SITE POLYGONS | MANUAL | ESDAT | I | P1 DEVELOPED (DESIRED GROUND OBSERVATION AREAS) | Y Y Y N |
| CREATE EARTH SITE DEFINITION FILE | SITE DEF. | COMPUTER | ESDAT | O | GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | Y Y Y N |
| GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | SITE DEF. | COMPUTER | ESAL | I | CREATE EARTH SITE DEFINITION FILE | Y Y Y Y |
| GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | ASCN NODE | COMPUTER | ESAL | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | BASIC UPDT | RPLNG |
|--|-----------------------------------|----------|-------------------|------|---|------|------------|-------|
| GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | SITE AC/LOS | COMPUTER | ESAL | 0 | COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | Y | Y | Y |
| DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | DETAIL EPHEM | COMPUTER | LTO | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y | Y |
| DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | GND TRK | COMPUTER | LTO | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y | Y |
| DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | EARTH OBSERV SUN ELEV CONSTS | MANUAL | LTO | I | PI GENERATED (EARTH OBSERVATION SUN ELEVATION CONSTRAINTS) | Y | Y | N |
| DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | OBS CONSTS | COMPUTER | LTO | 0 | COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | Y | Y | Y |
| COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | SITE AC/LOS | COMPUTER | TARGEN | I | GENERATE ACQ/LOSS OF GROUND SITE TARGET AREAS | Y | Y | Y |
| COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | OBS CONSTS | COMPUTER | TARGEN | I | DEVELOP/APPLY CONSTRAINTS TO EARTH OBSERVATION TARGETS | Y | Y | Y |
| COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | PRINTOUTS OF EARTH OBSERV DATA | COMPUTER | TARGEN | 0 | DEVELOP GROSS MISSION TIMELINE | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---|-------------------------------------|----------|-------------------|------|--|------|--|
| COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | EARTH TARGETS | COMPUTER | TARGET | O | MERGE ALL EXPERIMENT TARGET FILES | Y | Y Y Y Y |
| COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | ASCN NODE | COMPUTER | BORB | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y Y Y Y |
| COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | ATT TL | COMPUTER | BORB | I | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | Y | Y Y Y Y |
| COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | PREL ATT TL | COMPUTER | BORB | I | DEVELOP PRELIMINARY ATTITUDE TIMELINE | Y | Y Y Y N |
| COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | STATE VECTOR PRINTOUT | MANUAL | BORB | I | GENERATE STATE VECTOR | Y | Y Y Y Y |
| COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | PL PHYSICS DESIRED VEH. ATTITUDE | MANUAL | BORB | I | P1 SUPPLIED DESIRED VEH. ATTITUDE/EL25 DEVELOPED | Y | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | | |
|---|----------------------|----------|-------------------|------|---|----------------------------|-------|------------|---|
| | | | | | | PREL | BASIC | UPDT RPLNG | |
| COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | BORB PAR. | COMPUTER | BORB | 0 | DEVELOP/APPLY CONSTRAINTS TO BORB PARAMETERS | Y | Y | Y | Y |
| DEVELOP/APPLY CONSTRAINTS TO BORB PARAMETERS | BORB PAR. | COMPUTER | LTO | 1 | COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | Y | Y | Y | Y |
| DEVELOP/APPLY CONSTRAINTS TO BORB PARAMETERS | BORB CONSTRAINTS | MANUAL | LTO | 1 | PI GENERATED (BORB CONSTRAINTS) | Y | Y | Y | Y |
| DEVELOP/APPLY CONSTRAINTS TO BORB PARAMETERS | BORB CONSTS | COMPUTER | LTO | 0 | COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | Y | Y | Y | Y |
| GENERATE HEMISPHERE OPPORTUNITIES | GND TRK | COMPUTER | LTO | 1 | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y | Y | Y |
| GENERATE HEMISPHERE OPPORTUNITIES | LATITUDE CONSTRAINTS | MANUAL | LTO | 1 | PI DEVELOPED (HEMISPHERE LATITUDE CONSTRAINTS) | Y | Y | Y | Y |
| GENERATE HEMISPHERE OPPORTUNITIES | HEMSPR CONSTS | COMPUTER | LTO | 0 | COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | Y | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---|-------------------------------------|----------|-------------------|------|---|------|--|
| COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | HEMSPR CONSTS | COMPUTER | TARGET | I | GENERATE HEMISPHERE OPPORTUNITIES | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | BORB CONSTS | COMPUTER | TARGET | I | DEVELOP/APPLY CONSTRAINTS TO BORB PARAMETERS | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | PRINTOUTS OF PLASMA PHYSICS DATA | COMPUTER | TARGET | O | DEVELOP GROSS MISSION TIMELINE | Y | Y Y Y Y |
| COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | PL PHY TARGETS | COMPUTER | TARGET | O | MERGE ALL EXPERIMENT TARGET FILES | Y | Y Y Y Y |
| MERGE ALL EXPERIMENT TARGET FILES | AT PHY TARGETS | COMPUTER | TARGET | I | COMBINE CONSTRAINTS TO DETERMINE ATMOSPHERIC PHYSICS TARGETS | Y | Y Y Y Y |
| MERGE ALL EXPERIMENT TARGET FILES | EARTH TARGETS | COMPUTER | TARGET | I | COMBINE CONSTRAINTS TO DETERMINE EARTH OBSERVATION TARGETS | Y | Y Y Y Y |
| MERGE ALL EXPERIMENT TARGET FILES | SOLAR TARGETS | COMPUTER | TARGET | I | COMBINE CONSTRAINTS TO DETERMINE SOLAR TARGETS | Y | Y Y Y Y |
| MERGE ALL EXPERIMENT TARGET FILES | PL PHY TARGETS | COMPUTER | TARGET | I | COMBINE CONSTRAINTS TO DETERMINE PLASMA PHYSICS TARGETS | Y | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|--|-----------------------------|----------|-------------------|------|---|------|--|
| MERGE ALL EXPERIMENT TARGET FILES | MSN IND TARGETS | COMPUTER | TARGEN | I | MERGE MISSION INDEPENDENT TARGETS | Y | Y Y Y |
| MERGE ALL EXPERIMENT TARGET FILES | CEL TARGETS | COMPUTER | TARGEN | I | COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | Y | Y Y Y |
| MERGE ALL EXPERIMENT TARGET FILES | MSN TARGETS | COMPUTER | TARGEN | O | CREATE ESS TARGET FILE | Y | Y Y Y |
| MERGE ALL EXPERIMENT TARGET FILES | MSN TARGETS | COMPUTER | TARGEN | O | GENERATE PCAP CHARTS | Y | Y Y Y |
| MERGE ALL EXPERIMENT TARGET FILES | MSN TARGETS | COMPUTER | TARGEN | O | GENERATE PTS CHARTS | Y | Y Y Y |
| PERFORM PARAMETRIC ANALYSIS TO DESIGN/DEVELOP CO-ORBITING TRAJECTORIES THAT SATISFY OBJECTIVES AND CONSTRAINTS | BASIC CO-ORBITING REQMTS | MANUAL | RELMO | I | ORBIT ANALYSIS DEVELOPED (EL25) | Y | Y Y N |
| PERFORM PARAMETRIC ANALYSIS TO DESIGN/DEVELOP CO-ORBITING TRAJECTORIES THAT SATISFY OBJECTIVES AND CONSTRAINTS | STS REQMTS/CONSTRAINTS | MANUAL | RELMO | I | STS CTR DEVELOPED (STS REQMTS/CONSTRAINTS) | Y | Y Y N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. | | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | | |
|--|---------------------------------|----------|-----------|-------|------|---------------------------------------|----------------------------|-------|------|-------|
| | | | WITH | RELMO | | | PREL | BASIC | UPDT | RPLNG |
| PERFORM PARAMETRIC ANALYSIS TO DESIGN/DEVELOP CO-ORBITING TRAJECTORIES THAT SATISFY OBJECTIVES AND CONSTRAINTS | PI REQMTS/CONSTRAINTS | MANUAL | RELMO | | I | PI DEVELOPED (CO-ORBITING REQMTS) | Y | Y | Y | N |
| PERFORM PARAMETRIC ANALYSIS TO DESIGN/DEVELOP CO-ORBITING TRAJECTORIES THAT SATISFY OBJECTIVES AND CONSTRAINTS | INPUTS TO MISSION PLANNING | MANUAL | RELMO | | O | DEVELOP GROSS MISSION TIMELINE | Y | Y | Y | N |
| DEVELOP GROSS MISSION TIMELINE | MISSION PROFILE CONSIDERATIONS | MANUAL | NONE | | I | MSN MANAGEMENT, IWG, PI'S, ANALYSTS | Y | Y | Y | N |
| DEVELOP GROSS MISSION TIMELINE | MSN EXP OPPORTUNITIES DATA | MANUAL | NONE | | I | EXPERIMENT OPPORTUNITIES GENERATION | Y | Y | Y | N |
| DEVELOP GROSS MISSION TIMELINE | MGMT AGREEMENT ON GROSS MSN T/L | MANUAL | NONE | | O | DEVELOP PRELIMINARY ATTITUDE TIMELINE | Y | Y | Y | N |
| DEVELOP GROSS MISSION TIMELINE | MGMT AGREEMENT ON GROSS MSN T/L | MANUAL | NONE | | O | CREATE RESERVE PERIOD FILE (DS) | Y | Y | Y | N |
| DEVELOP GROSS MISSION TIMELINE | MGMT AGREEMENT ON GROSS MSN T/L | MANUAL | NONE | | O | GENERATE MISSION TIMELINE | Y | Y | Y | N |
| DEVELOP PRELIMINARY ATTITUDE TIMELINE | MGMT AGREEMENT ON GROSS MSN T/L | MANUAL | CAVIMP | | I | DEVELOP GROSS MISSION TIMELINE | Y | Y | Y | N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | |
|---|-------------------|----------|-------------------|------|---|----------------------------|-------|------------|
| | | | | | | PREL | BASIC | UPDT RPLNG |
| DEVELOP PRELIMINARY ATTITUDE TIMELINE | PREL ATT TL | COMPUTER | CAVINP | 0 | GENERATE PRELIMINARY TDRS COVERAGE | Y | Y | Y N |
| DEVELOP PRELIMINARY ATTITUDE TIMELINE | PREL ATT TL | COMPUTER | CAVINP | 0 | COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | Y | Y | Y N |
| DEVELOP PRELIMINARY ATTITUDE TIMELINE | PREL ATT TL | COMPUTER | CAVINP | 0 | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | Y | Y | Y N |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | CEL TARGETS | COMPUTER | READPI | I | COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | Y | Y | Y N |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | TIME GOAL | COMPUTER | READPI | I | PI DEVELOPED (DESIRED TARGET OBSERVATION TIMES) | Y | Y | Y N |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | SUBCOO | COMPUTER | READPI | I | DEVELOP CELESTIAL TARGETS (IPS MISSION) | Y | Y | Y N |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | EARTH SHADOW | COMPUTER | READPI | I | GENERATE EARTH SHADOW ACQ/LOSS | Y | Y | Y N |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | SAA AC/LOS | COMPUTER | READPI | I | IMPOSE RADIATION CONSTRAINTS | Y | Y | Y N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | |
|---|-----------------------------------|----------|-------------------|------|---|----------------------------|------------------|
| | | | | | | PREL | BASIC UPDT RPLNG |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | COO | COMPUTER | READPI | I | P1 DEVELOPED (CELESTIAL TARGETS OBSERVATION REQMTS) | Y | Y Y N |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | ANG'LR DIST'CE | COMPUTER | READPI | O | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y | Y Y N |
| CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | READPI | COMPUTER | READPI | O | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y | Y Y N |
| CREATE RESERVE PERIOD FILE (DS) | RESERVE PERIODS CONSIDERATIONS | MANUAL | EDT | I | DEVELOP GROSS MISSION TIMELINE | Y | Y Y N |
| CREATE RESERVE PERIOD FILE (DS) | RESERVE PERIODS | COMPUTER | EDT | O | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y | Y Y N |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | SUBCOO | COMPUTER | ASTAR | I | DEVELOP CELESTIAL TARGETS (IPS MISSION) | Y | Y Y N |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | CEL TARGETS | COMPUTER | ASTAR | I | COMBINE CONSTRAINTS TO DETERMINE CELESTIAL TARGETS AND MERGE FILES | Y | Y Y Y |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | RESERVE PERIODS | COMPUTER | ASTAR | I | CREATE RESERVE PERIOD FILE (DS) | Y | Y Y N |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | READPI | COMPUTER | ASTAR | I | CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | Y | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---------------------------------------|-------------------|----------|-------------------|------|---|------|--|
| SCHEDULE SCIENCE OBSERVATIONS (DS) | ANG'LR DIST'CE | COMPUTER | ASTAR | I | CREATE COMMON FILE FOR ASTAR PROGRAM (DS) | Y | Y Y Y |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | EARTH SHADOW | COMPUTER | ASTAR | I | GENERATE EARTH SHADOW ACQ/LOSS | Y | Y Y Y |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | SAA AC/LOS | COMPUTER | ASTAR | I | IMPOSE RADIATION CONSTRAINTS | Y | Y Y Y |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | COO | COMPUTER | ASTAR | I | P1 DEVELOPED (CELESTIAL TARGETS OBSERVATION REQMTS) | Y | Y Y N |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | SCIENCE SCHED'LE | COMPUTER | ASTAR | O | GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | Y | Y Y N |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | SCIENCE SCHED'LE | COMPUTER | ASTAR | O | GENERATE IPS POINTING DATA (DS) | Y | Y Y Y |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | SCIENCE SCHED'LE | COMPUTER | ASTAR | O | GENERATE JOINT OPERATIONS TARGET FILE (DS) | Y | Y Y Y |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | SCIENCE SCHED'LE | COMPUTER | ASTAR | O | CREATE ESS TARGET FILE | Y | Y Y Y |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | SCIENCE SCHED'LE | COMPUTER | ASTAR | O | GENERATE MANEUVER TIMELINE (DS) | Y | Y Y Y |
| SCHEDULE SCIENCE OBSERVATIONS (DS) | SCIENCE SCHED'LE | COMPUTER | ASTAR | O | SELECT IPS ROLL ANGLES (DS) | Y | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING PREL BASIC UPDT RPLNG |
|---|-------------------|----------|-------------------|------|---|---|
| GENERATE ATTITUDE TIMELINE | KEYW'RD | COMPUTER | KEYGEN | I | DEVELOPED AS PART OF THIS SUBTASK BY ORBIT ANALYSIS ENGINEERS | Y Y Y Y |
| GENERATE ATTITUDE TIMELINE | MVR TL | COMPUTER | KEYGEN | I | GENERATE MISSION TIMELINE | Y Y Y Y |
| GENERATE ATTITUDE TIMELINE | ATT TL (NDF) | COMPUTER | KEYGEN | O | GENERATE ORBITER POINTING DATA | Y Y Y Y |
| GENERATE ATTITUDE TIMELINE | ATT TL | COMPUTER | KEYGEN | O | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | Y Y Y Y |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | PREL ATT TL | COMPUTER | CAVINP | I | DEVELOP PRELIMINARY ATTITUDE TIMELINE | Y Y Y N |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | ATT TL | COMPUTER | CAVINP | I | GENERATE ATTITUDE TIMELINE | Y Y Y Y |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | ATTITUDE UPDATES | MANUAL | CAVINP | I | FINALIZE MISSION TIMELINE | Y Y Y Y |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | ATTITUDE UPDATES | MANUAL | CAVINP | I | GENERATE MISSION TIMELINE | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. | | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | | |
|---|-------------------|----------|-----------|--------|------|---|----------------------------|-------|------|-------|
| | | | WITH | CAVINP | | | PREL | BASIC | UPDT | RPLNG |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQNTS | ATT TL | COMPUTER | | CAVINP | 0 | GENERATE POC CHECKLIST AND COMMAND TIMELINE | N | Y | Y | Y |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQNTS | ATT TL | COMPUTER | | CAVINP | 0 | GENERATE PTS CHARTS | Y | Y | Y | Y |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQNTS | ATT TL | COMPUTER | | CAVINP | 0 | GENERATE TDRS COVERAGE | Y | Y | Y | Y |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQNTS | ATT TL | COMPUTER | | CAVINP | 0 | COMPUTE ORIENTATION AND STRENGTH OF MAGNETIC FIELD IN THE ORBITER COORDINATE SYSTEM | Y | Y | Y | Y |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQNTS | ATT TL | COMPUTER | | CAVINP | 0 | GENERATE PCAP CHARTS | Y | Y | Y | Y |
| EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQNTS | ATT TL | COMPUTER | | CAVINP | 0 | CREATE ESS TARGET FILE | Y | Y | Y | Y |
| GENERATE MANEUVER TIMELINE (DS) | SCIENCE SCHED 'LE | COMPUTER | | PAAC | 1 | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES PREL | INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|--|-------------------|----------|-------------------|------|--|----------------|---|
| GENERATE MANEUVER TIMELINE (DS) | MVR TL | COMPUTER | PAAC | O | GENERATE ATTITUDE TIMELINE (DS) | Y | Y Y Y |
| GENERATE MANEUVER TIMELINE (DS) | KEYW*RD | COMPUTER | PAAC | O | GENERATE ATTITUDE TIMELINE (DS) | Y | Y Y Y |
| GENERATE ATTITUDE TIMELINE (DS) | MVR TL | COMPUTER | KEYGEN | I | GENERATE MANEUVER TIMELINE (DS) | Y | Y Y Y |
| GENERATE ATTITUDE TIMELINE (DS) | KEYW*RD | COMPUTER | KEYGEN | I | GENERATE MANEUVER TIMELINE (DS) | Y | Y Y Y |
| GENERATE ATTITUDE TIMELINE (DS) | ATT TL | COMPUTER | KEYGEN | O | EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | Y | Y Y Y |
| GENERATE ATTITUDE TIMELINE (DS) | ATT TL (NDF) | COMPUTER | KEYGEN | O | GENERATE ORBITER POINTING DATA | Y | Y Y Y |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | ATT TL | COMPUTER | CAVINP | I | GENERATE ATTITUDE TIMELINE (DS) | Y | Y Y Y |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | ATTITUDE UPDATES | MANUAL | CAVINP | I | GENERATE MISSION TIMELINE | Y | Y Y Y |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | ATTITUDE UPDATES | MANUAL | CAVINP | I | FINALIZE MISSION TIMELINE | Y | Y Y Y |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | ATT TL | COMPUTER | CAVINP | O | GENERATE PTS CHARTS | Y | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | BASIC | UPDT | RPLNG | CYCLES INPUT/OUTPUT DURING |
|--|-------------------|----------|-------------------|------|--|------|-------|------|-------|----------------------------|
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | ATT TL | COMPUTER | CAVINP | 0 | GENERATE TDRS COVERAGE | Y | Y | Y | Y | Y |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | ATT TL | COMPUTER | CAVINP | 0 | GENERATE POCC CHECKLIST AND COMMAND TIMELINE | N | Y | Y | Y | Y |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | ATT TL | COMPUTER | CAVINP | 0 | CREATE ESS TARGET FILE | Y | Y | Y | Y | Y |
| EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | ATT TL | COMPUTER | CAVINP | 0 | GENERATE PCAP CHARTS | Y | Y | Y | Y | Y |
| GENERATE ORBITER POINTING DATA | ATT TL (NDF) | COMPUTER | PROCAM | 1 | GENERATE ATTITUDE TIMELINE (DS) | Y | Y | Y | Y | Y |
| GENERATE ORBITER POINTING DATA | ATT TL (NDF) | COMPUTER | PROCAM | 1 | GENERATE ATTITUDE TIMELINE | Y | Y | Y | Y | Y |
| GENERATE ORBITER POINTING DATA | DETAIL EPHEM | COMPUTER | PROCAM | 1 | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y | Y | Y | Y |
| GENERATE ORBITER POINTING DATA | PROCAM | COMPUTER | PROCAM | 0 | GENERATE PTS CHARTS | Y | Y | Y | Y | Y |
| GENERATE ORBITER POINTING DATA | PROCAM | COMPUTER | PROCAM | 0 | DEVELOP STRAY LIGHT CONSTRAINTS | Y | Y | Y | Y | Y |
| GENERATE ORBITER POINTING DATA | PROCAM | COMPUTER | PROCAM | 0 | GENERATE IPS POINTING DATA (DS) | Y | Y | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|--------------------------------------|-------------------|----------|-------------------|------|--|------|--|
| GENERATE ORBITER POINTING DATA | PROCAM | COMPUTER | PROCAM | O | SELECT IPS ROLL ANGLES (DS) | Y | Y Y Y Y |
| GENERATE TDRS COVERAGE | ORBITER OCCULT | COMPUTER | CAVINP | I | JSC | Y | Y Y N |
| GENERATE TDRS COVERAGE | ASCN NODE | COMPUTER | CAVINP | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y Y Y Y |
| GENERATE TDRS COVERAGE | ATT TL | COMPUTER | CAVINP | I | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | Y | Y Y Y Y |
| GENERATE TDRS COVERAGE | ATT TL | COMPUTER | CAVINP | I | EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | Y | Y Y Y Y |
| GENERATE TDRS COVERAGE | TDRS AC/LOS | COMPUTER | CAVINP | O | GENERATE PCAP CHARTS | N | Y Y Y Y |
| GENERATE TDRS COVERAGE | TDRS AC/LOS | COMPUTER | CAVINP | O | CREATE ESS TARGET FILE | Y | Y Y Y Y |
| GENERATE TDRS COVERAGE | TDRS AC/LOS | COMPUTER | CAVINP | O | GENERATE PTS CHARTS | Y | Y Y Y Y |
| GENERATE TDRS COVERAGE | TDRS AC/LOS | COMPUTER | CAVINP | O | PROCESS TDRS DATA FOR ENHANCEMENT | Y | Y Y Y Y |
| PROCESS TDRS DATA FOR ENHANCEMENT | TDRS AC/LOS | COMPUTER | TARGEN | I | GENERATE TDRS COVERAGE | Y | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES PREL | INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|--------------------------------------|------------------------------|----------|-------------------|------|---|----------------|---|
| PROCESS TDRS DATA FOR ENHANCEMENT | GND STA AC/LOS | COMPUTER | TARGET | I | GENERATE GROUND STATION COVERAGE | Y | Y Y Y |
| PROCESS TDRS DATA FOR ENHANCEMENT | COMM AC/LOS | COMPUTER | TARGET | O | GENERATE MISSION WINDOWS | Y | Y Y Y |
| PROCESS TDRS DATA FOR ENHANCEMENT | COMM AC/LOS | COMPUTER | TARGET | O | UPDATE OR ENHANCE EXISTING SCHEDULE | Y | Y Y Y |
| PROCESS TDRS DATA FOR ENHANCEMENT | COMM AC/LOS | COMPUTER | TARGET | O | GENERATE DATA FLOW REPORTS | Y | Y Y Y |
| PROCESS TDRS DATA FOR ENHANCEMENT | COMM AC/LOS | COMPUTER | TARGET | O | VERIFY DATA FLOW SCHEDULES | Y | Y Y Y |
| PROCESS TDRS DATA FOR ENHANCEMENT | COMM AC/LOS | COMPUTER | TARGET | O | GENERATE POCC MMU DATA SET | Y | Y Y Y |
| GENERATE POCC MMU DATA SET | COMM AC/LOS | COMPUTER | PMSG | I | PROCESS TDRS DATA FOR ENHANCEMENT | Y | Y Y Y |
| GENERATE POCC MMU DATA SET | ASCN NODE | COMPUTER | PMSG | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y | Y Y Y |
| GENERATE POCC MMU DATA SET | MSN IND TARGETS | COMPUTER | PMSG | I | MERGE MISSION INDEPENDENT TARGETS | Y | Y Y Y |
| GENERATE POCC MMU DATA SET | PRINTOUTS OF MMU DATA SET | COMPUTER | PMSG | O | MANUAL VERIFICATION OF MMU DATA SET | Y | Y Y Y |
| GENERATE POCC MMU DATA SET | MMU DAT SET | COMPUTER | PMSG | O | MMU LOAD INPUT DEVELOPMENT | Y | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES PREL | INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---|-----------------------------------|----------|-------------------|------|---|----------------|---|
| GENERATE CANDIDATE SOLAR GUIDE STARS | SKYMAP DATBASE | COMPUTER | SCATGEN | I | MSFC DEVELOPED | Y | Y Y N |
| GENERATE CANDIDATE SOLAR GUIDE STARS | LAUNCH WINDOW/LAUNCH TIME DATA | MANUAL | SCATGEN | I | LAUNCH WINDOW/LAUNCH TIME SELECTION | Y | Y Y N |
| GENERATE CANDIDATE SOLAR GUIDE STARS | CAND GSTAR | COMPUTER | SCATGEN | O | GENERATE SOLAR OBJECTIVE LOADS | Y | Y Y N |
| GENERATE CANDIDATE SOLAR GUIDE STARS | PRINTOUTS OF CAND GUIDE STARS | COMPUTER | SCATGEN | O | DEVELOP STRAY LIGHT CONSTRAINTS | Y | Y Y N |
| DEVELOP STRAY LIGHT CONSTRAINTS | PROCAM | COMPUTER | ASTRO | I | GENERATE ORBITER POINTING DATA | Y | Y Y N |
| DEVELOP STRAY LIGHT CONSTRAINTS | PRINTOUTS OF CAND GUIDE STARS | MANUAL | ASTRO | I | GENERATE CANDIDATE SOLAR GUIDE STARS | Y | Y Y N |
| DEVELOP STRAY LIGHT CONSTRAINTS | PRINTOUTS OF STRAY LIGHT DATA | COMPUTER | ASTRO | O | CHOOSE SOLAR GUIDE STARS | Y | Y Y N |
| CHOOSE SOLAR GUIDE STARS | PRINTOUTS OF STRAY LIGHT DATA | MANUAL | NONE | I | DEVELOP STRAY LIGHT CONSTRAINTS | Y | Y Y N |
| CHOOSE SOLAR GUIDE STARS | GUIDE STAR CHOICES | MANUAL | NONE | O | GENERATE SOLAR OBJECTIVE LOADS | Y | Y Y N |
| GENERATE SOLAR OBJECTIVE LOADS | CAND GSTAR | COMPUTER | SCATGEN | I | GENERATE CANDIDATE SOLAR GUIDE STARS | Y | Y Y N |
| GENERATE SOLAR OBJECTIVE LOADS | GUIDE STAR CHOICES | MANUAL | SCATGEN | I | CHOOSE SOLAR GUIDE STARS | Y | Y Y N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING PREL BASIC UPDT RPLNG |
|---|---------------------------|----------|-------------------|------|---|---|
| GENERATE SOLAR OBJECTIVE LOADS | SOLAR OBJ LOAD SUMMARY | COMPUTER | SCATGEN | 0 | USED FOR SOLAR OBJECTIVE LOAD VERIFICATION AND GENERAL INFORMATION | Y Y Y N |
| GENERATE SOLAR OBJECTIVE LOADS | OBJ LOAD | COMPUTER | SCATGEN | 0 | MMU LOAD INPUT DEVELOPMENT | Y Y Y N |
| GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | COO | COMPUTER | GSOLP | I | PI DEVELOPED (CELESTIAL TARGETS OBSERVATION REQMTS) | Y Y Y N |
| GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | SCIENCE SCHED'LE | COMPUTER | GSOLP | I | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y Y Y N |
| GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | SKYMAP DATBASE | COMPUTER | GSOLP | I | MSFC DEVELOPED | Y Y Y N |
| GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | CAND GSTAR | COMPUTER | GSOLP | 0 | FORMAT STELLAR GUIDE STAR CATALOG (DS) | Y Y Y N |
| FORMAT STELLAR GUIDE STAR CATALOG (DS) | SKYMAP DATBASE | COMPUTER | GSOLP | I | MSFC DEVELOPED | Y Y Y N |
| FORMAT STELLAR GUIDE STAR CATALOG (DS) | CAND GSTAR | COMPUTER | GSOLP | I | GENERATE CANDIDATE STELLAR GUIDE STARS (DS) | Y Y Y N |
| FORMAT STELLAR GUIDE STAR CATALOG (DS) | GSTAR CAT | COMPUTER | GSOLP | 0 | GENERATE STELLAR OBJECTIVE LOADS (DS) | Y Y Y N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. | | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | | |
|---------------------------------------|--------------------------|----------|-----------|--------|------|--|----------------------------|------------|-------|---|
| | | | WITH | GIMBAL | | | PREL | BASIC UPDT | RPLNG | |
| SELECT IPS ROLL ANGLES (DS) | SCIENCE SCHED'LE | COMPUTER | | GIMBAL | I | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y | Y | Y | N |
| SELECT IPS ROLL ANGLES (DS) | COO | COMPUTER | | GIMBAL | I | P1 DEVELOPED (CELESTIAL TARGETS OBSERVATION REQMTS) | Y | Y | Y | N |
| SELECT IPS ROLL ANGLES (DS) | PROCAM | COMPUTER | | GIMBAL | I | GENERATE ORBITER POINTING DATA | Y | Y | Y | Y |
| SELECT IPS ROLL ANGLES (DS) | IPS ROLL ANGLES | COMPUTER | | GIMBAL | O | GENERATE STELLAR OBJECTIVE LOADS (DS) | Y | Y | Y | N |
| GENERATE STELLAR OBJECTIVE LOADS (DS) | IPS ROLL ANGLES | COMPUTER | | GSOLP | I | SELECT IPS ROLL ANGLES (DS) | Y | Y | Y | N |
| GENERATE STELLAR OBJECTIVE LOADS (DS) | GSTAR CAT | COMPUTER | | GSOLP | I | FORMAT STELLAR GUIDE STAR CATALOG (DS) | Y | Y | Y | N |
| GENERATE STELLAR OBJECTIVE LOADS (DS) | OBJ LOAD | COMPUTER | | GSOLP | O | MMU LOAD INPUT DEVELOPMENT | Y | Y | Y | N |
| GENERATE STELLAR OBJECTIVE LOADS (DS) | STELLAR OBJ LOAD SUMMARY | COMPUTER | | GSOLP | O | USED FOR STELLAR OBJECTIVE LOAD VERIFICATION AND GENERAL INFORMATION | Y | Y | Y | N |
| GENERATE STELLAR OBJECTIVE LOADS (DS) | OBJ LOAD | COMPUTER | | GSOLP | O | GENERATE IPS POINTING DATA (DS) | Y | Y | Y | N |
| GENERATE IPS POINTING DATA (DS) | PROCAM | COMPUTER | | IPOL | I | GENERATE ORBITER POINTING DATA | Y | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | BASIC | UPDT | RPLNG | CYCLES INPUT/OUTPUT DURING |
|--|-------------------|----------|-------------------|------|---|------|-------|------|-------|----------------------------|
| GENERATE IPS POINTING DATA (DS) | COO | COMPUTER | IPOL | I | PI DEVELOPED (CELESTIAL TARGETS OBSERVATION REQMTS) | Y | Y | Y | N | |
| GENERATE IPS POINTING DATA (DS) | OBJ LOAD | COMPUTER | IPOL | I | GENERATE STELLAR OBJECTIVE LOADS (DS) | Y | Y | Y | N | |
| GENERATE IPS POINTING DATA (DS) | SCIENCE SCHED'LE | COMPUTER | IPOL | I | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y | Y | Y | Y | |
| GENERATE IPS POINTING DATA (DS) | ID LIST | COMPUTER | IPOL | O | PI EDITS TO ADD SEQUENCE NUMBER (DS) | Y | Y | Y | N | |
| GENERATE IPS POINTING DATA (DS) | IPOL | COMPUTER | IPOL | O | GENERATE PCAP CHARTS | Y | Y | Y | Y | |
| GENERATE IPS POINTING DATA (DS) | IPOL | COMPUTER | IPOL | O | GENERATE JOINT OPERATIONS TARGET FILE (DS) | Y | Y | Y | Y | |
| PI EDITS TO ADD SEQUENCE NUMBER (DS) | ID LIST | COMPUTER | EDT | I | GENERATE IPS POINTING DATA (DS) | Y | Y | Y | N | |
| PI EDITS TO ADD SEQUENCE NUMBER (DS) | ID/SEQ LIST | COMPUTER | EDT | O | GENERATE JOINT OPERATIONS TARGET FILE (DS) | Y | Y | Y | N | |
| GENERATE JOINT OPERATIONS TARGET FILE (DS) | COO | COMPUTER | JOTF | I | PI DEVELOPED (CELESTIAL TARGETS OBSERVATION REQMTS) | Y | Y | Y | N | |
| GENERATE JOINT OPERATIONS TARGET FILE (DS) | IPOL | COMPUTER | JOTF | I | GENERATE IPS POINTING DATA (DS) | Y | Y | Y | Y | |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | BASIC | UPDT | RPLNG | CYCLES INPUT/OUTPUT DURING |
|--|-----------------------------------|----------|-------------------|------|---|------|-------|------|-------|----------------------------|
| GENERATE JOINT OPERATIONS TARGET FILE (DS) | SCIENCE SCHED'LE | COMPUTER | JOTF | I | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y | Y | Y | Y | Y |
| GENERATE JOINT OPERATIONS TARGET FILE (DS) | ID/SEQ LIST | COMPUTER | JOTF | I | PI EDITS TO ADD SEQUENCE NUMBER (DS) | Y | Y | Y | Y | N |
| GENERATE JOINT OPERATIONS TARGET FILE (DS) | JOTF | COMPUTER | JOTF | O | MMU LOAD INPUT DEVELOPMENT | Y | Y | Y | Y | Y |
| GENERATE JOINT OPERATIONS TARGET FILE (DS) | JOTF VERIFICATION REPORT | COMPUTER | JOTF | O | MANUAL VERIFICATION OF JOTF FILE | Y | Y | Y | Y | Y |
| CREATE MISSION TIMELINE MODELS | STS CAPABILITIES DOCUMENTATION | MANUAL | VME | I | STS CAPABILITIES DEFINITION (JSC) | Y | Y | Y | Y | N |
| CREATE MISSION TIMELINE MODELS | MISSION CONFIGURATION | MANUAL | VME | I | PAYLOAD COMPLEMENT DEFINITION (JSC) | Y | Y | Y | Y | N |
| CREATE MISSION TIMELINE MODELS | ERD'S | MANUAL | VME | I | PI'S EXPERIMENT REQMTS INPUTS | Y | N | N | N | N |
| CREATE MISSION TIMELINE MODELS | O&IA | MANUAL | VME | I | PAYLOAD DATA COLLECTION | N | Y | Y | Y | N |
| CREATE MISSION TIMELINE MODELS | SPAH & SL SYSTEM DOCUMENTATION | MANUAL | VME | I | SPACELAB CONFIGURATION DEFINITION | Y | Y | Y | Y | N |
| CREATE MISSION TIMELINE MODELS | BASIC CREW CYCLE (JSC) | MANUAL | VME | I | CREW SYSTEMS CONSTRAINTS (JSC) | Y | Y | Y | Y | N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | |
|-----------------------------------|---------------------------|----------|-------------------|------|------------------------------------|----------------------------|-------|------------|
| | | | | | | PREL | BASIC | UPDT RPLNG |
| CREATE MISSION TIMELINE MODELS | ESS MODEL | COMPUTER | VME | 0 | GENERATE PCAP CHARTS | Y | Y | N |
| CREATE MISSION TIMELINE MODELS | ESS MODEL | COMPUTER | VME | 0 | CREATE ESS TARGET FILE | Y | Y | N |
| CREATE MISSION TIMELINE MODELS | ESS MODEL | COMPUTER | VME | 0 | DATA FLOW | Y | Y | N |
| CREATE MISSION TIMELINE MODELS | ESS MODEL | COMPUTER | VME | 0 | GENERATE MISSION TIMELINE | Y | Y | N |
| CREATE MISSION TIMELINE MODELS | ESS MODELS PRINTOUT | COMPUTER | VME | 0 | GENERATE CREW H/O CYCLE | Y | Y | N |
| GENERATE CREW H/O CYCLE | BASIC CREW CYCLE (JSC) | MANUAL | NONE | 1 | CREW SYSTEMS CONSTRAINTS (JSC) | Y | Y | Y |
| GENERATE CREW H/O CYCLE | ESS MODELS PRINTOUT | MANUAL | NONE | 1 | CREATE MISSION TIMELINE MODELS | Y | Y | Y |
| GENERATE CREW H/O CYCLE | CREW H/O CYCLE | MANUAL | NONE | 0 | CREATE RESERVE PERIOD FILE (DS) | Y | Y | Y |
| GENERATE CREW H/O CYCLE | CREW H/O CYCLE | MANUAL | NONE | 0 | GENERATE MISSION TIMELINE | Y | Y | Y |
| GENERATE CREW H/O CYCLE | CREW H/O CYCLE | MANUAL | NONE | 0 | DEVELOP GROSS MISSION TIMELINE | Y | Y | Y |
| CREATE ESS TARGET FILE | ESS MODEL | COMPUTER | TAE | 1 | CREATE MISSION TIMELINE MODELS | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | BASIC | UPDT | RPLNG |
|---------------------------|-----------------------------|----------|-------------------|------|--|------|-------|------|-------|
| CREATE ESS TARGET FILE | TDRS AC/LOS | COMPUTER | TAE | I | GENERATE TDRS COVERAGE | Y | Y | Y | Y |
| CREATE ESS TARGET FILE | ATT TL | COMPUTER | TAE | I | EDIT TO INCORPORATE STS OR OTHER REGMTS (DS) | Y | Y | Y | Y |
| CREATE ESS TARGET FILE | ATT TL | COMPUTER | TAE | I | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REGMTS | Y | Y | Y | Y |
| CREATE ESS TARGET FILE | SCIENCE SCHED'LE | COMPUTER | TAE | I | SCHEDULE SCIENCE OBSERVATIONS (DS) | Y | Y | Y | Y |
| CREATE ESS TARGET FILE | MSN TARGETS | COMPUTER | TAE | I | MERGE ALL EXPERIMENT TARGET FILES | Y | Y | Y | Y |
| CREATE ESS TARGET FILE | SPECIAL TARGETS | MANUAL | TAE | I | PI'S, MSN T/L ENGR'S, MGMT DIRECTION | Y | Y | Y | Y |
| CREATE ESS TARGET FILE | ESS TARGET PRINTOUTS | COMPUTER | TAE | O | CREATE MASTER INPUT FILES | N | Y | N | N |
| CREATE ESS TARGET FILE | ESS TARGET | COMPUTER | TAE | O | CREATE MASTER TIMELINE | Y | Y | Y | Y |
| CREATE ESS TARGET FILE | TARGET ANALYSIS PRINTOUT | COMPUTER | TAE | O | MISSION T/L ENG'S | Y | Y | Y | Y |
| CREATE ESS TARGET FILE | ESS TARGET | COMPUTER | TAE | O | GENERATE POCC CHECKLIST AND COMMAND TIMELINE | Y | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|------------------------------|------------------------------------|----------|-------------------|------|--|------|--|
| CREATE ESS TARGET FILE | ESS TARGET | COMPUTER | TAE | O | GENERATE MISSION TIMELINE | Y | Y Y Y |
| GENERATE MISSION TIMELINE | ESS MODEL | COMPUTER | ESP | I | CREATE MISSION TIMELINE MODELS | Y | Y Y Y |
| GENERATE MISSION TIMELINE | ESS TARGET | COMPUTER | ESP | I | CREATE ESS TARGET FILE | Y | Y Y Y |
| GENERATE MISSION TIMELINE | PAO REQMTS | MANUAL | ESP | I | PAO REQUIREMENTS INPUT | N | Y Y Y |
| GENERATE MISSION TIMELINE | MGMT AGREEMENT ON GROSS MSN T/L | MANUAL | ESP | I | DEVELOP GROSS MISSION TIMELINE | Y | Y Y Y |
| GENERATE MISSION TIMELINE | CREW H/O CYCLE | MANUAL | ESP | I | GENERATE CREW H/O CYCLE | Y | Y Y Y |
| GENERATE MISSION TIMELINE | CREW ACT | COMPUTER | ESP | O | GENERATE PTS CHARTS | Y | Y Y Y |
| GENERATE MISSION TIMELINE | EXP T/L | COMPUTER | ESP | O | CREATE MASTER TIMELINE | N | Y Y Y |
| GENERATE MISSION TIMELINE | MVR TL | COMPUTER | ESP | O | GENERATE ATTITUDE TIMELINE | Y | Y Y Y |
| GENERATE MISSION TIMELINE | EXP T/L | COMPUTER | ESP | O | GENERATE POCC CHECKLIST AND COMMAND TIMELINE | N | Y Y Y |
| GENERATE MISSION TIMELINE | EXP | COMPUTER | ESP | O | GENERATE PTS CHARTS | Y | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. | | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | | |
|------------------------------|------------------------------------|----------|-----------|-----|------|--|----------------------------|-------|------|-------|
| | | | WITH | ESP | | | PREL | BASIC | UPDT | RPLNG |
| GENERATE MISSION TIMELINE | INHBT | COMPUTER | ESP | 0 | 0 | GENERATE PTS CHARTS | Y | Y | Y | Y |
| GENERATE MISSION TIMELINE | JSC CAP | COMPUTER | ESP | 0 | 0 | TO JSC FOR REVIEW | N | Y | Y | Y |
| GENERATE MISSION TIMELINE | MSN T/L TABLES, PRINTOUTS | COMPUTER | ESP | 0 | 0 | PI'S, ENGR'S, MGMT FOR REVIEW | Y | Y | Y | Y |
| GENERATE MISSION TIMELINE | EXP T/L | COMPUTER | ESP | 0 | 0 | GENERATE MISSION DATA REQUIREMENTS PROFILE | N | Y | Y | Y |
| GENERATE MISSION TIMELINE | EXP T/L | COMPUTER | ESP | 0 | 0 | GENERATE PCAP CHARTS | N | Y | Y | Y |
| GENERATE MISSION TIMELINE | ATTITUDE UPDATES | MANUAL | ESP | 0 | 0 | EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | Y | Y | Y | Y |
| GENERATE MISSION TIMELINE | ATTITUDE UPDATES | MANUAL | ESP | 0 | 0 | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | Y | Y | Y | Y |
| FINALIZE MISSION TIMELINE | ESS TARGET | COMPUTER | ESP | I | I | CREATE ESS TARGET FILE | N | Y | Y | Y |
| FINALIZE MISSION TIMELINE | ESS MODEL | COMPUTER | ESP | I | I | CREATE MISSION TIMELINE MODELS | N | Y | Y | Y |
| FINALIZE MISSION TIMELINE | REVIEW CYCLE UPDATES | MANUAL | ESP | I | I | DIVISION MANAGEMENT REVIEW | Y | Y | Y | Y |
| FINALIZE MISSION TIMELINE | MGMT AGREEMENT ON GROSS MSN T/L | MANUAL | ESP | I | I | DEVELOP GROSS MISSION TIMELINE | Y | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|------------------------------|------------------------------|----------|-------------------|------|--|------|---|
| FINALIZE MISSION TIMELINE | REVIEW CYCLE UPDATES | MANUAL | ESP | I | JSC REVIEW | N | Y Y Y |
| FINALIZE MISSION TIMELINE | REVIEW CYCLE UPDATES | MANUAL | ESP | I | ATTITUDE/TDRS ITERATION | Y | Y Y Y |
| FINALIZE MISSION TIMELINE | CREW H/O CYCLE | MANUAL | ESP | I | GENERATE CREW H/O CYCLE | Y | Y Y Y |
| FINALIZE MISSION TIMELINE | PAO REQMTS | MANUAL | ESP | I | PAO REQUIREMENTS INPUT | N | Y Y Y |
| FINALIZE MISSION TIMELINE | REVIEW CYCLE UPDATES | MANUAL | ESP | I | STL/MTL REVIEW UPDATES | N | Y Y Y |
| FINALIZE MISSION TIMELINE | EXP T/L PRINTOUTS | COMPUTER | ESP | O | CREATE MASTER INPUT FILES | N | Y N N |
| FINALIZE MISSION TIMELINE | MVR TL | COMPUTER | ESP | O | GENERATE ATTITUDE TIMELINE | Y | Y Y Y |
| FINALIZE MISSION TIMELINE | EXP T/L | COMPUTER | ESP | O | GENERATE PCAP CHARTS | N | Y Y Y |
| FINALIZE MISSION TIMELINE | EXP T/L | COMPUTER | ESP | O | VERIFICATION OF STL'S/MLT'S | N | Y Y Y |
| FINALIZE MISSION TIMELINE | EXP T/L | COMPUTER | ESP | O | DATA FLOW ANALYSIS FOR VERIFICATION | N | Y Y Y |
| FINALIZE MISSION TIMELINE | MSN T/L TABLES, PRINTOUTS | COMPUTER | ESP | O | PI'S, ENGR'S, MGMT FOR APPROVAL | Y | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES PREL | INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---|-------------------|----------|-------------------|------|--|----------------|---|
| FINALIZE MISSION TIMELINE | JSC CAP | COMPUTER | ESP | 0 | TO JSC FOR INCORPORATION INTO THE CAP | N | Y Y Y |
| FINALIZE MISSION TIMELINE | CREW ACT | COMPUTER | ESP | 0 | GENERATE PTS CHARTS | N | Y Y Y |
| FINALIZE MISSION TIMELINE | EXP | COMPUTER | ESP | 0 | GENERATE PTS CHARTS | N | Y Y Y |
| FINALIZE MISSION TIMELINE | INHBT | COMPUTER | ESP | 0 | GENERATE PTS CHARTS | N | Y Y Y |
| FINALIZE MISSION TIMELINE | ATTITUDE UPDATES | MANUAL | ESP | 0 | EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | Y | Y Y Y |
| FINALIZE MISSION TIMELINE | ATTITUDE UPDATES | MANUAL | ESP | 0 | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | Y | Y Y Y |
| GENERATE PCAP CHARTS PROC. | | COMPUTER | PCAP | I | DATABASE | N | Y Y Y |
| GENERATE PCAP CHARTS ESS MODEL | | COMPUTER | PCAP | I | CREATE MISSION TIMELINE MODELS | N | Y Y Y |
| GENERATE PCAP CHARTS TDRS AC/LOS | | COMPUTER | PCAP | I | GENERATE TDRS COVERAGE | N | Y Y Y |
| GENERATE PCAP CHARTS DATA FLOW SCHED | | COMPUTER | PCAP | I | GENERATE DATA FLOW REPORTS | N | Y Y Y |
| GENERATE PCAP CHARTS IPOL | | COMPUTER | PCAP | I | GENERATE IPS POINTING DATA (DS) | N | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | |
|----------------------|-----------------------------|----------|-------------------|------|--|----------------------------|-------|------------|
| | | | | | | PREL | BASIC | UPDT RPLNG |
| GENERATE PCAP CHARTS | MSN TARGETS | COMPUTER | PCAP | I | MERGE ALL EXPERIMENT TARGET FILES | N | Y | Y |
| GENERATE PCAP CHARTS | EXP T/L | COMPUTER | PCAP | I | GENERATE MISSION TIMELINE | N | Y | Y |
| GENERATE PCAP CHARTS | NOTES | COMPUTER | PCAP | I | DATABASE | N | Y | Y |
| GENERATE PCAP CHARTS | ATT TL | COMPUTER | PCAP | I | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | N | Y | Y |
| GENERATE PCAP CHARTS | ATT TL | COMPUTER | PCAP | I | EDIT TO INCORPORATE STS OR OTHER REQMTS | N | Y | Y |
| GENERATE PCAP CHARTS | SPECIAL TIMELINE NOTES | MANUAL | PCAP | I | MISSION T/L ENGINEERS INPUTS | N | Y | Y |
| GENERATE PCAP CHARTS | CREW PROCEDURES DOCUMENT | MANUAL | PCAP | I | DEVELOP EXPERIMENT CREW PROCEDURES | N | Y | Y |
| GENERATE PCAP CHARTS | SPECIAL CREW NOTES | MANUAL | PCAP | I | CREW REQMTS FOR PCAP | N | Y | Y |
| GENERATE PCAP CHARTS | SHIFT TIMES | MANUAL | PCAP | I | MISSION T/L ENGINEERS INPUTS | N | Y | Y |
| GENERATE PCAP CHARTS | PCAP CHARTS | COMPUTER | PCAP | O | BUILD PCAP DOCUMENT | N | Y | Y |
| GENERATE PTS CHARTS | DATA FLOW SCHED | COMPUTER | PTS | I | GENERATE DATA FLOW REPORTS | Y | Y | Y |
| GENERATE PTS CHARTS | PROCAM | COMPUTER | PTS | I | GENERATE ORBITER POINTING DATA | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING PREL BASIC UPDT RPLNG |
|---------------------|-------------------|----------|-------------------|------|--|---|
| GENERATE PTS CHARTS | ATT TL | COMPUTER | PTS | I | EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | Y Y Y Y |
| GENERATE PTS CHARTS | CREW ACT | COMPUTER | PTS | I | GENERATE MISSION TIMELINE | Y Y Y Y |
| GENERATE PTS CHARTS | GND TRK | COMPUTER | PTS | I | GENERATE REQUIRED EPHEMERIS DATA FOR OUTPUT | Y Y Y Y |
| GENERATE PTS CHARTS | MSN TARGETS | COMPUTER | PTS | I | MERGE ALL EXPERIMENT TARGET FILES | Y Y Y Y |
| GENERATE PTS CHARTS | EXP | COMPUTER | PTS | I | GENERATE MISSION TIMELINE | Y Y Y Y |
| GENERATE PTS CHARTS | ATT TL | COMPUTER | PTS | I | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | Y Y Y Y |
| GENERATE PTS CHARTS | INHBT | COMPUTER | PTS | I | MISSION TIMELINE GENERATION | Y Y Y Y |
| GENERATE PTS CHARTS | TDRS AC/LOS | COMPUTER | PTS | I | GENERATE TDRS COVERAGE | Y Y Y Y |
| GENERATE PTS CHARTS | SOPG TIMES | MANUAL | PTS | I | MISSION T/L ENGINEER INPUT | Y Y Y Y |
| GENERATE PTS CHARTS | PTS CHARTS | COMPUTER | PTS | O | FLIGHT DEFINITION DOCUMENT DEVELOPMENT | Y Y Y Y |
| GENERATE PTS CHARTS | PTS CHARTS | COMPUTER | PTS | O | BUILD PCAP DOCUMENT | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | |
|---|------------------------------------|----------|-------------------|------|-------------------------------|----------------------------|-------|------------|
| | | | | | | PREL | BASIC | UPDT RPLNG |
| BUILD PCAP DOCUMENT | PTS CHARTS | MANUAL | NONE | I | GENERATE PTS CHARTS | N | Y | Y |
| BUILD PCAP DOCUMENT | PCAP CHARTS | MANUAL | NONE | I | GENERATE PCAP CHARTS | N | Y | Y |
| BUILD PCAP DOCUMENT | PCAP DOCUMENT | MANUAL | NONE | O | PAYLOAD FLIGHT DATA FILE | N | Y | Y |
| FLIGHT DEFINITION DOCUMENT DEVELOPMENT | DATA FLOW ANALYSIS FDD INPUTS | MANUAL | NONE | I | DATA FLOW ANALYSIS TASKS | Y | Y | N |
| FLIGHT DEFINITION DOCUMENT DEVELOPMENT | ORBIT ANALYSIS FDD INPUTS | MANUAL | NONE | I | ORBIT ANALYSIS TASKS | Y | Y | N |
| FLIGHT DEFINITION DOCUMENT DEVELOPMENT | MISSION T/L ANALYSIS FDD INPUTS | MANUAL | NONE | I | MISSION T/L ANALYSIS TASKS | Y | Y | N |
| FLIGHT DEFINITION DOCUMENT DEVELOPMENT | FLIGHT DEFINITION DOCUMENT | MANUAL | NONE | O | PUBLICATION | Y | Y | N |
| FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | COOLING LOAD RQMTS | MANUAL | NONE | I | THERMAL ANALYSIS (MSFC) | N | Y | N |
| FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | HORR DUMP TIMES | MANUAL | NONE | I | DATA FLOW ANALYSIS TASKS | N | Y | N |
| FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | MISSION T/L ANALYSIS FPA INPUTS | MANUAL | NONE | I | MISSION T/L ANALYSIS TASKS | N | Y | N |
| FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | ORBIT ANALYSIS FPA INPUTS | MANUAL | NONE | I | ORBIT ANALYSIS TASKS | N | Y | N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---|----------------------------------|----------|-------------------|------|--|------|--|
| FLIGHT PLANNING ANNEX INPUT DEVELOPMENT | FPA INPUTS | MANUAL | NONE | O | FLIGHT PLANNING ANNEX INPUTS TO JSC | N | Y N N N |
| DEVELOP STORAGE BOOK | OTHER PI INPUTS | MANUAL | NONE | I | PI REQMTS INPUT | Y | Y Y N N |
| DEVELOP STORAGE BOOK | ERD'S, FO'S AND/OR O&IA DOC | MANUAL | NONE | I | PAYLOAD DATA COLLECTION | Y | Y Y N N |
| DEVELOP STORAGE BOOK | MSN T/L TABLES, PRINTOUTS | MANUAL | NONE | I | GENERATE MISSION TIMELINE | Y | Y Y N N |
| DEVELOP STORAGE BOOK | STORAGE BOOK | MANUAL | NONE | O | BUILD PFDF DOCUMENTS | Y | Y Y N N |
| DEVELOP TV, PHOTO PROCEDURES | TV, PHOTO SYSTEM CAPABILITIES | MANUAL | NONE | I | SPA, ICD'S, ETC. | Y | Y Y N N |
| DEVELOP TV, PHOTO PROCEDURES | MSN T/L TABLES, PRINTOUTS | MANUAL | NONE | I | GENERATE MISSION TIMELINE | Y | Y Y N N |
| DEVELOP TV, PHOTO PROCEDURES | ERD'S, FO'S AND/OR O&IA DOC | MANUAL | NONE | I | PAYLOAD DATA COLLECTION | Y | Y Y N N |
| DEVELOP TV, PHOTO PROCEDURES | OTHER PI INPUTS | MANUAL | NONE | I | PI REQMTS INPUT | Y | Y Y N N |
| DEVELOP TV, PHOTO PROCEDURES | TV, PHOTO OPS HANDBOOK | MANUAL | NONE | O | BUILD PFDF DOCUMENTS | Y | Y Y N N |
| DEVELOP EXPERIMENT CREW PROCEDURES | MSN T/L TABLES, PRINTOUTS | MANUAL | NONE | I | GENERATE MISSION TIMELINE | Y | Y Y N N |
| DEVELOP EXPERIMENT CREW PROCEDURES | OTHER PI INPUTS | MANUAL | NONE | I | PI REQMTS INPUT | Y | Y Y N N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---------------------------------------|--------------------------------|----------|-------------------|------|------------------------------|------|--|
| DEVELOP EXPERIMENT CREW PROCEDURES | ERD'S, FO'S AND/OR O&IA DOC | MANUAL | NONE | I | PAYLOAD DATA COLLECTION | Y | N N N |
| DEVELOP EXPERIMENT CREW PROCEDURES | EXP CREW PROCEDURES DOC | MANUAL | NONE | O | BUILD PDF DOCUMENTS | Y | Y Y N |
| DEVELOP PAYLOAD SYSTEMS HANDBOOK | MSN T/L TABLES, PRINTOUTS | MANUAL | NONE | I | GENERATE MISSION TIMELINE | Y | Y Y N |
| DEVELOP PAYLOAD SYSTEMS HANDBOOK | ERD'S, FO'S AND/OR O&IA DOC | MANUAL | NONE | I | PAYLOAD DATA COLLECTION | Y | Y Y N |
| DEVELOP PAYLOAD SYSTEMS HANDBOOK | OTHER PI INPUTS | MANUAL | NONE | I | PI REQMTS INPUT | Y | Y Y N |
| DEVELOP PAYLOAD SYSTEMS HANDBOOK | MPE OPS REQMTS | MANUAL | NONE | I | SPAH, ICD'S, ECT. | Y | Y Y N |
| DEVELOP PAYLOAD SYSTEMS HANDBOOK | SL/PL INTERFACE DEFINITION | MANUAL | NONE | I | SPAH, ICD'S, ECT. | Y | Y Y N |
| DEVELOP PAYLOAD SYSTEMS HANDBOOK | PAYLOAD SYSTEMS HANDBOOK | MANUAL | NONE | O | BUILD PDF DOCUMENT | Y | Y Y N |
| DEVELOP COMS DICTIONARY | MSN T/L TABLES, PRINTOUTS | MANUAL | NONE | I | GENERATE MISSION TIMELINE | Y | Y Y N |
| DEVELOP COMS DICTIONARY | OTHER PI INPUTS | MANUAL | NONE | I | PI REQMTS INPUT | Y | Y Y N |
| DEVELOP COMS DICTIONARY | ERD'S, FO'S AND/OR O&IA DOC | MANUAL | NONE | I | PAYLOAD DATA COLLECTION | Y | Y Y N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|----------------------------|--------------------------------|----------|-------------------|------|---|------|--|
| DEVELOP CDMS DICTIONARY | CDMS SYSTEM DEFINITION | MANUAL | NONE | I | DESIGN DOCUMENTATION | Y | Y Y N |
| DEVELOP CDMS DICTIONARY | CDMS DICTIONARY | MANUAL | NONE | O | BUILD PFDF DOCUMENTS | N | N Y N |
| BUILD PFDF DOCUMENTS | CDMS DICTIONARY | MANUAL | NONE | I | DEVELOP CDMS DICTIONARY | Y | Y Y N |
| BUILD PFDF DOCUMENTS | TV, PHOTO OPS HANDBOOK | MANUAL | NONE | I | DEVELOP TV, PHOTO PROCEDURES | Y | Y Y N |
| BUILD PFDF DOCUMENTS | STORAGE BOOK | MANUAL | NONE | I | DEVELOP STORAGE BOOK | Y | Y Y N |
| BUILD PFDF DOCUMENTS | EXP CREW PROCEDURES DOC | MANUAL | NONE | I | DEVELOP EXPERIMENT CREW PROCEDURES | Y | Y Y N |
| BUILD PFDF DOCUMENTS | PAYLOAD SYSTEMS HANDBOOK | MANUAL | NONE | I | DEVELOP PAYLOAD SYSTEMS HANDBOOK | Y | Y Y N |
| BUILD PFDF DOCUMENTS | PFDF DOCUMENTS | MANUAL | NONE | O | CREW, MSN SUPPORT PERSONNEL | Y | Y Y N |
| CREATE DATA FLOW MODELS | PI INTERFACE | MANUAL | EDT | I | PI INTERVIEWS, DISCUSSIONS | Y | Y N N |
| CREATE DATA FLOW MODELS | ERD'S, FO'S AND/OR O&IA DOC | MANUAL | EDT | I | PAYLOAD DATA COLLECTION | Y | Y Y N |
| CREATE DATA FLOW MODELS | DATA FLOW MODELS | COMPUTER | EDT | O | GENERATE MISSION DATA REQUIREMENTS PROFILE | Y | Y Y N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | |
|--|------------------------------|----------|-------------------|------|--|----------------------------|-------|------------|
| | | | | | | PREL | BASIC | UPDT RPLNG |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | EXP T/L | COMPUTER | DF/MORPG | I | GENERATE MISSION TIMELINE | Y | Y | Y |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | HRM FORMATS DEFN | COMPUTER | DF/MORPG | I | HRM FORMATS DEFINITIONS | Y | Y | N |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | POCC CONFIG DEFN | COMPUTER | DF/MORPG | I | POCC CONFIGURATION DEFINITION | Y | Y | N |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | DATA FLOW INPUT VARIABLES | COMPUTER | DF/MORPG | I | MISSION SUPPORT ACTIVITIES INPUT VARIABLES | Y | Y | N |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | DATA FLOW MODELS | COMPUTER | DF/MORPG | I | CREATE DATA FLOW MODELS | Y | Y | N |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | DATA FLOW REQMTS | COMPUTER | DF/MORPG | I | DATA FLOW REQUIREMENTS INPUTS | Y | Y | N |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | PROFILE | COMPUTER | DF/MORPG | O | UPDATE OR ENHANCE EXISTING SCHEDULES | Y | Y | Y |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | PROFILE | COMPUTER | DF/MORPG | O | GENERATE POCC POSSIBLE CONFIGURATIONS | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES PREL | INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|--|------------------------------|----------|-------------------|------|---|----------------|---|
| GENERATE MISSION DATA REQUIREMENTS PROFILE | PROFILE | COMPUTER | DF/MORPG | O | SCHEDULE HRM FORMATS AND DOWNLINK | Y | Y Y Y |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | PROFILE | COMPUTER | DF/MORPG | O | GENERATE DATA FLOW REPORTS | Y | Y Y Y |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | PROFILE | COMPUTER | DF/MORPG | O | SCHEDULE RECORDER PLAYBACKS | Y | Y Y Y |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | MORPG ERROR | COMPUTER | DF/MORPG | O | DATA FLOW ANALYSIS ENGINEERS FOR EVALUATION | Y | Y Y Y |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | PROFILE | COMPUTER | DF/MORPG | O | GENERATE HRM POSSIBLE FORMATS | Y | Y Y Y |
| GENERATE MISSION DATA REQUIREMENTS PROFILE | PROFILE | COMPUTER | DF/MORPG | O | GENERATE MISSION WINDOWS | Y | Y Y Y |
| GENERATE MISSION WINDOWS | DATA FLOW INPUT VARIABLES | COMPUTER | DF/MWG | I | MISSION SUPPORT ACTIVITIES INPUT VARIABLES | Y | Y Y Y |
| GENERATE MISSION WINDOWS | HRM FORMATS DEFN | COMPUTER | DF/MWG | I | HRM FORMATS DEFINITION | Y | Y Y Y |
| GENERATE MISSION WINDOWS | COMM AC/LOS | COMPUTER | DF/MWG | I | PROCESS TDRS DATA FOR ENHANCEMENT | Y | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | |
|--------------------------------------|---------------------------|----------|-------------------|------|--|----------------------------|------------------|
| | | | | | | PREL | BASIC UPDT RPLNG |
| GENERATE MISSION WINDOWS | PROFILE | COMPUTER | DF/MNG | I | GENERATE MISSION DATA REQUIREMENTS PROFILE | Y | Y Y Y |
| GENERATE MISSION WINDOWS | DFA SCHEDULE CMDS | COMPUTER | DF/MNG | I | MANUAL FEEDBACK TO UPDATE SCHEDULES BASED ON EVALUATION OF ERROR FILES | Y | Y Y Y |
| GENERATE MISSION WINDOWS | MISSION WINDOWS | COMPUTER | DF/MNG | O | SCHEDULE ONBOARD RECORDER OPERATIONS | Y | Y Y Y |
| GENERATE MISSION WINDOWS | MISSION WINDOWS | COMPUTER | DF/MNG | O | VERIFY DATA FLOW SCHEDULES | Y | Y Y Y |
| GENERATE MISSION WINDOWS | MNG ERROR | COMPUTER | DF/MNG | O | DATA FLOW ANALYSIS ENGINEERS FOR EVALUATION | Y | Y Y Y |
| SCHEDULE ONBOARD RECORDER OPERATIONS | MISSION WINDOWS | COMPUTER | DF/ORS | I | GENERATE MISSION WINDOWS | Y | Y Y Y |
| SCHEDULE ONBOARD RECORDER OPERATIONS | DFA SCHEDULE CMDS | COMPUTER | DF/ORS | I | MANUAL FEEDBACK TO UPDATE SCHEDULES BASED ON EVALUATION OF ERROR FILES | Y | Y Y Y |
| SCHEDULE ONBOARD RECORDER OPERATIONS | DATA FLOW INPUT VARIABLES | COMPUTER | DF/ORS | I | MISSION SUPPORT ACTIVITIES INPUT VARIABLES | Y | Y Y Y |
| SCHEDULE ONBOARD RECORDER OPERATIONS | MASTER FILE | COMPUTER | DF/ORS | O | SCHEDULE HRM FORMATS AND DOWNLINK | Y | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING PREL BASIC UPDT RPLNG |
|---|------------------------------|----------|-------------------|------|---|---|
| SCHEDULE ONBOARD RECORDER OPERATIONS | ORS ERROR | COMPUTER | DF/ORS | 0 | DATA FLOW ANALYSIS ENGINEERS FOR EVALUATION | Y Y Y Y |
| SCHEDULE ONBOARD RECORDER OPERATIONS | MASTER FILE | COMPUTER | DF/ORS | 0 | GENERATE HRM POSSIBLE FORMATS | Y Y Y Y |
| GENERATE HRM POSSIBLE FORMATS | DFA SCHEDULAR CMNDS | COMPUTER | DF/HPFG | I | MANUAL FEEDBACK TO UPDATE SCHEDULES BASED ON EVALUATION OF ERROR FILES | Y Y Y Y |
| GENERATE HRM POSSIBLE FORMATS | PROFILE | COMPUTER | DF/HPFG | I | GENERATE MISSION DATA REQUIREMENTS PROFILE | Y Y Y Y |
| GENERATE HRM POSSIBLE FORMATS | HRM FORMATS DEFN | COMPUTER | DF/HPFG | I | HRM FORMATS DEFINITION | Y Y Y Y |
| GENERATE HRM POSSIBLE FORMATS | DATA FLOW INPUT VARIABLES | COMPUTER | DF/HPFG | I | MISSION SUPPORT ACTIVITIES INPUT VARIABLES | Y Y Y Y |
| GENERATE HRM POSSIBLE FORMATS | MASTER FILE | COMPUTER | DF/HPFG | I | SCHEDULE ONBOARD RECORDER OPERATIONS | Y Y Y Y |
| GENERATE HRM POSSIBLE FORMATS | HRM POSSIBLE FORMATS | COMPUTER | DF/HPFG | 0 | SCHEDULE HRM FORMATS AND DOWNLINK | Y Y Y Y |
| GENERATE HRM POSSIBLE FORMATS | HRM POSSIBLE FORMATS | COMPUTER | DF/HPFG | 0 | GENERATE DATA FLOW REPORTS | Y Y Y Y |
| GENERATE HRM POSSIBLE FORMATS | HRM POSSIBLE FORMATS | COMPUTER | DF/HPFG | 0 | VERIFY DATA FLOW SCHEDULES | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---|------------------------------|----------|-------------------|------|--|------|--|
| SCHEDULE HRM FORMATS AND DOWNLINK | MASTER FILE | COMPUTER | DF/HFS | I | SCHEDULE ONBOARD RECORDER OPERATIONS | Y | Y Y Y |
| SCHEDULE HRM FORMATS AND DOWNLINK | DATA FLOW INPUT VARIABLES | COMPUTER | DF/HFS | I | MISSION SUPPORT ACTIVITIES INPUT VARIABLES | Y | Y Y Y |
| SCHEDULE HRM FORMATS AND DOWNLINK | DFA SCHEDULAR CMNDS | COMPUTER | DF/HFS | I | MANUAL FEEDBACK TO UPDATE SCHEDULES BASED ON EVALUATION OF ERROR FILE | Y | Y Y Y |
| SCHEDULE HRM FORMATS AND DOWNLINK | HRM FORMATS DEFN | COMPUTER | DF/HFS | I | HRM FORMATS DEFINITION | Y | Y Y Y |
| SCHEDULE HRM FORMATS AND DOWNLINK | HRM POSSIBLE FORMATS | COMPUTER | DF/HFS | I | GENERATE HRM POSSIBLE FORMATS | Y | Y Y Y |
| SCHEDULE HRM FORMATS AND DOWNLINK | PROFILE | COMPUTER | DF/HFS | I | GENERATE MISSION DATA REQUIREMENTS PROFILE | Y | Y Y Y |
| SCHEDULE HRM FORMATS AND DOWNLINK | HFS ERROR | COMPUTER | DF/HFS | 0 | DATA FLOW ANALYSIS ENGINEERS FOR EVALUATION | Y | Y Y Y |
| SCHEDULE HRM FORMATS AND DOWNLINK | MASTER FILE | COMPUTER | DF/HFS | 0 | GENERATE POCC POSSIBLE CONFIGURATIONS | Y | Y Y Y |
| GENERATE POCC POSSIBLE CONFIGURATIONS | MASTER FILE | COMPUTER | DF/PPCG | I | SCHEDULE HRM FORMATS AND DOWNLINK | Y | Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING PREL BASIC UPDT RPLNG |
|---|------------------------------|----------|-------------------|------|---|---|
| GENERATE POCC POSSIBLE CONFIGURATIONS | DATA FLOW INPUT VARIABLES | COMPUTER | DF/PPCG | I | MISSION SUPPORT ACTIVITIES INPUT VARIABLES | Y Y Y Y |
| GENERATE POCC POSSIBLE CONFIGURATIONS | PROFILE | COMPUTER | DF/PPCG | I | GENERATE MISSION DATA REQUIREMENTS PROFILE | Y Y Y Y |
| GENERATE POCC POSSIBLE CONFIGURATIONS | POCC CONFIG DEFN | COMPUTER | DF/PPCG | I | POCC CONFIGURATION DEFINITION | Y Y Y Y |
| GENERATE POCC POSSIBLE CONFIGURATIONS | DFA SCHEDULAR CMNDS | COMPUTER | DF/PPCG | I | MANUAL FEEDBACK TO UPDATE SCHEDULES BASED ON EVALUATION OF ERROR FILES | Y Y Y Y |
| GENERATE POCC POSSIBLE CONFIGURATIONS | POCC POSSIBLE CONFIG | COMPUTER | DF/PPCG | O | SCHEDULE POCC CONFIGURATIONS | Y Y Y Y |
| GENERATE POCC POSSIBLE CONFIGURATIONS | POCC POSSIBLE CONFIG | COMPUTER | DF/PPCG | O | SCHEDULE RECORDER PLAYBACKS | Y Y Y Y |
| SCHEDULE POCC CONFIGURATIONS | POCC POSSIBLE CONFIG | COMPUTER | DF/PCS | I | GENERATE POCC POSSIBLE CONFIGURATIONS | Y Y Y Y |
| SCHEDULE POCC CONFIGURATIONS | POCC CONFIG DEFN | COMPUTER | DF/PCS | I | POCC CONFIGURATION DEFINITION | Y Y Y Y |
| SCHEDULE POCC CONFIGURATIONS | MASTER FILE | COMPUTER | DF/PCS | I | SCHEDULE HRM FORMATS AND DOWNLINK | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|---------------------------------|------------------------------|----------|-------------------|------|---|------|--|
| SCHEDULE POCC CONFIGURATIONS | HRM FORMATS DEFN | COMPUTER | DF/PCS | I | HRM FORMATS DEFINITION | Y | Y Y Y Y |
| SCHEDULE POCC CONFIGURATIONS | DATA FLOW INPUT VARIABLES | COMPUTER | DF/PCS | I | MISSION SUPPORT ACTIVITIES INPUT VARIABLES | Y | Y Y Y Y |
| SCHEDULE POCC CONFIGURATIONS | MASTER FILE | COMPUTER | DF/PCS | O | SCHEDULE RECORDER PLAYBACKS | Y | Y Y Y Y |
| SCHEDULE RECORDER PLAYBACKS | PROFILE | COMPUTER | DF/PBS | I | GENERATE MISSION DATA REQUIREMENTS PROFILE | Y | Y Y Y Y |
| SCHEDULE RECORDER PLAYBACKS | MASTER FILE | COMPUTER | DF/PBS | I | SCHEDULE POCC CONFIGURATIONS | Y | Y Y Y Y |
| SCHEDULE RECORDER PLAYBACKS | DATA FLOW INPUT VARIABLES | COMPUTER | DF/PBS | I | MISSION SUPPORT ACTIVITIES INPUT VARIABLES | Y | Y Y Y Y |
| SCHEDULE RECORDER PLAYBACKS | POCC POSSIBLE CONFIG | COMPUTER | DF/PBS | I | GENERATE POCC POSSIBLE CONFIGURATIONS | Y | Y Y Y Y |
| SCHEDULE RECORDER PLAYBACKS | DFA SCHEDULAR CHNDS | COMPUTER | DF/PBS | I | MANUAL FEEDBACK TO UPDATE SCHEDULES BASED ON EVALUATION OF ERROR FILES | Y | Y Y Y Y |
| SCHEDULE RECORDER PLAYBACKS | PBS ERROR | COMPUTER | DF/PBS | O | DATA FLOW ANALYSIS ENGINEERS FOR EVALUATION | Y | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | BASIC | UPDT | RPLNG |
|--------------------------------|----------------------|----------|-------------------|------|---|------|-------|------|-------|
| SCHEDULE RECORDER PLAYBACKS | MASTER FILE | COMPUTER | DF/PBS | 0 | GENERATE DATA FLOW REPORTS | Y | Y | Y | Y |
| SCHEDULE RECORDER PLAYBACKS | MASTER FILE | COMPUTER | DF/PBS | 0 | VERIFY DATA FLOW SCHEDULES | Y | Y | Y | Y |
| SCHEDULE RECORDER PLAYBACKS | POCC POSSIBLE CONFIG | COMPUTER | DF/PBS | 0 | VERIFY DATA FLOW SCHEDULES | Y | Y | Y | Y |
| SCHEDULE RECORDER PLAYBACKS | MASTER FILE | COMPUTER | DF/PBS | 0 | UPDATE OR ENHANCE EXISTING SCHEDULES | Y | Y | Y | Y |
| VERIFY DATA FLOW SCHEDULES | COMM AC/LOS | COMPUTER | DF/DVM | I | PROCESS TDRS DATA FOR ENHANCEMENT | Y | Y | Y | Y |
| VERIFY DATA FLOW SCHEDULES | MASTER FILE | COMPUTER | DF/DVM | I | SCHEDULE RECORDER PLAYBACKS | Y | Y | Y | Y |
| VERIFY DATA FLOW SCHEDULES | MISSION WINDOWS | COMPUTER | DF/DVM | I | GENERATE MISSION WINDOWS | Y | Y | Y | Y |
| VERIFY DATA FLOW SCHEDULES | HRM POSSIBLE FORMATS | COMPUTER | DF/DVM | I | GENERATE HRM POSSIBLE FORMATS | Y | Y | Y | Y |
| VERIFY DATA FLOW SCHEDULES | POCC POSSIBLE CONFIG | COMPUTER | DF/DVM | I | SCHEDULE RECORDER PLAYBACKS | Y | Y | Y | Y |
| VERIFY DATA FLOW SCHEDULES | HRM FORMATS DEFN | COMPUTER | DF/DVM | I | HRM FORMATS DEFINITION | Y | Y | Y | Y |
| VERIFY DATA FLOW SCHEDULES | SPECIFIC CHECKS | MANUAL | DF/DVM | I | DATA FLOW ANALYSIS ENGINEERS COMMAND CHECK OF SPECIFIC FILES | Y | Y | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING PREL BASIC UPDT RPLNG |
|-------------------------------|---------------------------------|----------|-------------------|------|--|---|
| VERIFY DATA FLOW SCHEDULES | VERIFY ERROR | COMPUTER | DF/DVM | 0 | DATA FLOW ANALYSIS ENGINEERS FOR EVALUATION | Y Y Y Y |
| GENERATE DATA FLOW REPORTS | DATA MANGMT CHECKLIST (PREV) | COMPUTER | DF/DFRG | 1 | GENERATE DATA FLOW REPORTS (PREVIOUS CYCLE) | Y Y Y Y |
| GENERATE DATA FLOW REPORTS | DATA FLOW INPUT VARIABLES | COMPUTER | DF/DFRG | 1 | MISSION SUPPORT ACTIVITIES INPUT VARIABLES | Y Y Y Y |
| GENERATE DATA FLOW REPORTS | PROFILE | COMPUTER | DF/DFRG | 1 | GENERATE MISSION DATA REQUIREMENTS PROFILE | Y Y Y Y |
| GENERATE DATA FLOW REPORTS | MASTER FILE | COMPUTER | DF/DFRG | 1 | SCHEDULE RECORDER PLAYBACKS | Y Y Y Y |
| GENERATE DATA FLOW REPORTS | EXPERIMENT HEADER | COMPUTER | DF/DFRG | 1 | MANUAL INPUT OF DEDICATED CHANNELS AND EXPERIMENT ID'S | Y Y Y Y |
| GENERATE DATA FLOW REPORTS | HRM POSSIBLE FORMATS | COMPUTER | DF/DFRG | 1 | GENERATE HRM POSSIBLE FORMATS | Y Y Y Y |
| GENERATE DATA FLOW REPORTS | COMM AC/LOS | COMPUTER | DF/DFRG | 1 | PROCESS TDRS DATA FOR ENHANCEMENT | Y Y Y Y |
| GENERATE DATA FLOW REPORTS | TABULAR REPORTS | COMPUTER | DF/DFRG | 0 | MISSION SUPPORT PERSONNEL | Y Y Y Y |
| GENERATE DATA FLOW REPORTS | DATA FLOW SCHED | COMPUTER | DF/DFRG | 0 | GENERATE PTS CHARTS | Y Y Y Y |

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|---|---------------------------------|----------|-------------------|------|---|---|
| GENERATE DATA FLOW REPORTS | COMPARED DATA MGMT CHECKLIST | COMPUTER | DF/DFRG | O | DATA FLOW ANALYSTS | Y Y Y Y |
| GENERATE DATA FLOW REPORTS | DATA MANGMT CHECKLIST | COMPUTER | DF/DFRG | O | POCC SUPPORT | N N Y Y |
| GENERATE DATA FLOW REPORTS | DATA FLOW SCHED | COMPUTER | DF/DFRG | O | GENERATE PCAP CHARTS | N Y Y Y |
| GENERATE DATA FLOW REPORTS | DATA FLOW SCHED | COMPUTER | DF/DFRG | O | GENERATE POCC CHECKLIST AND COMMAND TIMELINES | N Y Y Y |
| GENERATE DATA FLOW REPORTS | DATA FLOW SCHED | COMPUTER | DF/DFRG | O | GENERATE COMMAND TIMELINE | N Y Y Y |
| UPDATE OR ENHANCE EXISTING SCHEDULES | PROFILE | COMPUTER | IDUS | I | GENERATE MISSION DATA REQUIREMENTS PROFILE | Y Y Y Y |
| UPDATE OR ENHANCE EXISTING SCHEDULES | DATA FLOW INPUT VARIABLES | COMPUTER | IDUS | I | MISSION SUPPORT ACTIVITIES INPUT VARIABLES | Y Y Y Y |
| UPDATE OR ENHANCE EXISTING SCHEDULES | HRM FORMATS DEFN | COMPUTER | IDUS | I | HRM FORMATS DEFINITION | Y Y Y Y |
| UPDATE OR ENHANCE EXISTING SCHEDULES | COMM AC/LOS | COMPUTER | IDUS | I | PROCESS TDRS DATA FOR ENHANCEMENT | Y Y Y Y |
| UPDATE OR ENHANCE EXISTING SCHEDULES | MASTER FILE | COMPUTER | IDUS | I | SCHEDULE RECORDER PLAYBACKS | Y Y Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | |
|--|-------------------|----------|-------------------|------|---|----------------------------|-------|------------|
| | | | | | | PREL | BASIC | UPDT RPLNG |
| UPDATE OR ENHANCE EXISTING SCHEDULES | MODIFICATIONS | MANUAL | IDUS | I | MODIFICATIONS TO SCHEDULES BASED ON USER EXPERIENCE | Y | Y | Y |
| UPDATE OR ENHANCE EXISTING SCHEDULES | MASTER FILE | COMPUTER | IDUS | O | VERIFY DATA FLOW SCHEDULES | Y | Y | Y |
| UPDATE OR ENHANCE EXISTING SCHEDULES | MASTER FILE | COMPUTER | IDUS | O | GENERATE DATA FLOW REPORTS | Y | Y | Y |
| CREATE SUBORDINATE TIMELINES | O&IA | MANUAL | EDT | I | PAYLOAD DATA COLLECTION | N | Y | N |
| CREATE SUBORDINATE TIMELINES | INPUTS FROM PI'S | MANUAL | EDT | I | PI REQMTS INPUT | N | Y | N |
| CREATE SUBORDINATE TIMELINES | ITL | COMPUTER | EDT | O | CHECK STL SYNTAX | N | Y | N |
| CHECK STL SYNTAX | ITL | COMPUTER | VERSTL | I | CREATE SUBORDINATE TIMELINES | N | Y | N |
| CHECK STL SYNTAX | SUB TL | COMPUTER | VERSTL | O | GENERATE STL BUFFER UTILIZATION REPORT | N | Y | N |
| CHECK STL SYNTAX | STL PRINTOUTS | COMPUTER | VERSTL | O | DESKTOP STL OPERATIONAL VERIFICATION | N | Y | N |
| DESKTOP STL OPERATIONAL VERIFICATION | STL PRINTOUTS | MANUAL | NONE | I | CHECK STL SYNTAX | N | Y | N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. | | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | | |
|--|----------------------|----------|-----------|------|------|--|----------------------------|-------|------|-------|
| | | | WITH | NONE | | | PREL | BASIC | UPDT | RPLNG |
| DESKTOP STL OPERATIONAL VERIFICATION | UPDATES TO STL'S | MANUAL | | NONE | O | CREATE SUBORDINATE TIMELINES | N | Y | N | N |
| CREATE MASTER INPUT FILES | ESS TARGET PRINTOUTS | MANUAL | EDT | | I | CREATE ESS TARGET FILE | N | Y | N | N |
| CREATE MASTER INPUT FILES | O&IA | MANUAL | EDT | | I | PAYLOAD DATA COLLECTION | N | Y | N | N |
| CREATE MASTER INPUT FILES | EXP T/L PRINTOUTS | MANUAL | EDT | | I | FINALIZE MISSION TIMELINE | N | Y | N | N |
| CREATE MASTER INPUT FILES | INPUTS FROM PI'S | MANUAL | EDT | | I | PI REQMTS INPUT | N | Y | N | N |
| CREATE MASTER INPUT FILES | MI | COMPUTER | EDT | | O | VERIFY MASTER INPUT FILES | N | Y | N | N |
| VERIFY AND COMBINE MASTER INPUT FILES | MI | COMPUTER | VERMI | | I | CREATE MASTER INPUT FILES | N | Y | N | N |
| VERIFY AND COMBINE MASTER INPUT FILES | MI | COMPUTER | VERMI | | O | GENERATE MASTER TIMELINE | N | Y | N | N |
| GENERATE MASTER TIMELINE | EXP T/L | COMPUTER | GENMTL | | I | FINALIZE MISSION TIMELINE | N | Y | N | N |
| GENERATE MASTER TIMELINE | MI | COMPUTER | GENMTL | | I | VERIFY AND COMBINE MASTER INPUT FILES | N | Y | N | N |
| GENERATE MASTER TIMELINE | ESS TARGET | COMPUTER | GENMTL | | I | CREATE ESS TARGET FILE | N | Y | N | N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | |
|--|----------------------------------|----------|-------------------|------|--|----------------------------|-------|------------|
| | | | | | | PREL | BASIC | UPDT RPLNG |
| GENERATE MASTER TIMELINE | MTL | COMPUTER | GENMTL | O | VERIFY MASTER TIMELINE | N | Y | N N |
| VERIFY MASTER TIMELINE | MTL | COMPUTER | VERMTL | I | GENERATE MASTER TIMELINE | N | Y | N N |
| VERIFY MASTER TIMELINE | MTL | COMPUTER | VERMTL | O | GENERATE STL BUFFER UTILIZATION REPORT | N | Y | N N |
| GENERATE STL BUFFER UTILIZATION REPORT | SUB TL | COMPUTER | STLBUF | I | DESKTOP STL OPERATIONAL VERIFICATION | N | Y | N N |
| GENERATE STL BUFFER UTILIZATION REPORT | MTL | COMPUTER | STLBUF | I | VERIFY MASTER TIMELINE | N | Y | N N |
| GENERATE STL BUFFER UTILIZATION REPORT | ECOS TL PRINTOUTS | COMPUTER | STLBUF | O | DESKTOP MTL OPERATIONAL VERIFICATION | N | Y | N N |
| GENERATE STL BUFFER UTILIZATION REPORT | ECOS TL | COMPUTER | STLBUF | O | CONVERT TO IBM TAPE FORMAT AND VERIFY | N | Y | N N |
| GENERATE STL BUFFER UTILIZATION REPORT | ECOS TL | COMPUTER | STLBUF | O | CREATE MMU ALLOCATION FILE | N | Y | N N |
| DESKTOP MTL OPERATIONAL VERIFICATION | ECOS TL PRINTOUTS | MANUAL | NONE | I | GENERATE STL BUFFER UTILIZATION REPORT | N | Y | N N |
| DESKTOP MTL OPERATIONAL VERIFICATION | UPDATES TO MASTER INPUT FILES | MANUAL | NONE | O | CREATE MASTER INPUT FILES | N | Y | N N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | BASIC | UPDT | RPLNG |
|--|------------------------------|----------|-------------------|------|---|------|-------|------|-------|
| CONVERT TO IBM TAPE FORMAT AND VERIFY | ECOS TL | COMPUTER | DEL.COM. | I | GENERATE STL BUFFER UTILIZATION REPORT | N | Y | N | N |
| CONVERT TO IBM TAPE FORMAT AND VERIFY | PRINTOUT OF ECOS TIMELINE | COMPUTER | DEL.COM. | O | GENERATE ECOS TIMELINE DOCUMENT | N | Y | N | N |
| CONVERT TO IBM TAPE FORMAT AND VERIFY | ECOS TL (EBCDIC) | COMPUTER | DEL.COM. | O | ECOS TAPE BUILD (STS ACTIVITIES) | N | Y | N | N |
| GENERATE ECOS TIMELINE DOCUMENT | PRINTOUT OF ECOS TIMELINE | MANUAL | NONE | I | CONVERT TO IBM FORMAT AND VERIFY | N | Y | N | N |
| GENERATE ECOS TIMELINE DOCUMENT | ECOS TIMELINE DOCUMENT | MANUAL | NONE | O | MISSION SUPPORT PERSONNEL | N | Y | N | N |
| CREATE MMU ALLOCATION FILE | DDU DISPLAYS | COMPUTER | EDT | I | STS ACTIVITIES INPUTS (MSFC) | N | Y | N | N |
| CREATE MMU ALLOCATION FILE | ECOS TL | COMPUTER | EDT | I | GENERATE STL BUFFER UTILIZATION REPORT | N | Y | N | N |
| CREATE MMU ALLOCATION FILE | PRELIM MMU MAP | MANUAL | EDT | I | STS ACTIVITIES INPUTS (MSFC) | N | Y | N | N |
| CREATE MMU ALLOCATION FILE | MMU ALLOCATION | COMPUTER | EDT | O | EVALUATE MMU TAPE MOVEMENT | N | Y | N | N |
| EVALUATE MMU TAPE MOVEMENT | MMU ALLOCATION | COMPUTER | MMUALL | I | CREATE MMU ALLOCATION FILE | N | Y | N | N |
| EVALUATE MMU TAPE MOVEMENT | ECOS TL | COMPUTER | MMUALL | I | GENERATE STL BUFFER UTILIZATION REPORT | N | Y | N | N |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | CYCLES INPUT/OUTPUT DURING | | |
|-------------------------------|----------------------------|----------|-------------------|------|--|----------------------------|-------|------------|
| | | | | | | PREL | BASIC | UPDT RPLNG |
| EVALUATE MMU TAPE MOVEMENT | MMU ALLOCATION PRINTOUT | COMPUTER | MMUALL | O | MISSION SUPPORT PERSONNEL | N | Y | N N |
| EVALUATE MMU TAPE MOVEMENT | MMU OPTIMIZATION REPORT | COMPUTER | MMUALL | O | MMU ENGINEERS FOR EVALUATION | N | Y | N N |
| CREATE COMMAND LIST | COMMANDS FROM PI | MANUAL | EDT | I | P.I. (PRINCIPAL INVESTIGATOR) | N | Y | Y Y |
| CREATE COMMAND LIST | COMMAND.INP | COMPUTER | EDT | O | CHECK COMMAND SYNTAX | N | Y | Y Y |
| CHECK COMMAND SYNTAX | COMMAND.INP | COMPUTER | CHECK | I | CREATE COMMAND LIST | N | Y | Y Y |
| CHECK COMMAND SYNTAX | COMMAND.VFY | COMPUTER | CHECK | O | PRODUCE COMMAND TIMETAGS | N | Y | Y Y |
| PRODUCE COMMAND TIMETAGS | COMMAND.VFY | COMPUTER | MET | I | CHECK COMMAND SYNTAX | N | Y | Y Y |
| PRODUCE COMMAND TIMETAGS | COMMAND.FIN | COMPUTER | MET | O | GENERATE COMMAND TIMELINE | N | Y | Y Y |
| GENERATE COMMAND TIMELINE | DATA FLOW SCHED | COMPUTER | CMDATG | I | GENERATE DATA FLOW REPORTS | N | Y | Y Y |
| GENERATE COMMAND TIMELINE | COMMAND.FIN | COMPUTER | CMDATG | I | PRODUCE COMMAND TIMETAGS | N | Y | Y Y |
| GENERATE COMMAND TIMELINE | ESS TARGET | COMPUTER | CMDATG | I | CREATE ESS TARGET FILE | N | Y | Y Y |
| GENERATE COMMAND TIMELINE | COMMAND T/L | COMPUTER | CMDATG | O | GENERATE POCC CHECKLIST AND COMMAND TIMELINE | N | Y | Y Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SU ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | BASIC UPDT | RPLNG |
|--|-------------------------------------|----------|-------------------|------|--|------|------------|-------|
| CREATE POCC CHECKLIST | MSN SUPPORT PERSONNEL/POH INPUTS | MANUAL | EDT | I | MISSION SUPPORT PERSONNEL AND POH | N | Y | Y |
| CREATE POCC CHECKLIST | ACTIVITY.INP | COMPUTER | EDT | O | CHECK ACTIVITY SYNTAX | N | Y | Y |
| CHECK ACTIVITY SYNTAX | ACTIVITY.INP | COMPUTER | CHECK | I | CREATE POCC CHECKLIST | N | Y | Y |
| CHECK ACTIVITY SYNTAX | ACTIVITY.VFY | COMPUTER | CHECK | O | PRODUCE ACTIVITY TIMETAGS | N | Y | Y |
| PRODUCE ACTIVITY TIMETAGS | ACTIVITY.VFY | COMPUTER | NET | I | CHECK ACTIVITY SYNTAX | N | Y | Y |
| PRODUCE ACTIVITY TIMETAGS | POCC.FIN | COMPUTER | NET | O | GENERATE POCC CHECKLIST AND COMMAND TIMELINE | N | Y | Y |
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | DATA FLOW SCHED | COMPUTER | CG | I | GENERATE DATA FLOW REPORTS | N | Y | Y |
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | ESS TARGET | COMPUTER | CG | I | CREATE ESS TARGET FILE | N | Y | Y |
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | ATT TL | COMPUTER | CG | I | EDIT TO INCORPORATE STS OR OTHER REQMTS (DS) | N | Y | Y |
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | EXP T/L | COMPUTER | CG | I | FINALIZE MISSION TIMELINE | N | Y | Y |

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| ACTIVITY | INPUT/OUTPUT NAME | I/O FORM | SW ASSOC. WITH | TYPE | SOURCE/DESTINATION | PREL | CYCLES INPUT/OUTPUT DURING BASIC UPDT RPLNG |
|--|---------------------------------|----------|-------------------|------|--|------|--|
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | COMMAND T/L | COMPUTER | CG | I | GENERATE COMMAND TIMELINE | N | Y Y Y |
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | POCC.FIN | COMPUTER | CG | I | PRODUCE ACTIVITY TIMETAGS | N | Y Y Y |
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | ATT TL | COMPUTER | CG | I | EDIT CURRENT ATTITUDE T/L TO INCORPORATE STS OR OTHER REQMTS | N | Y Y Y |
| GENERATE POCC CHECKLIST AND COMMAND TIMELINE | POCC CHECKLIST & COMMAND T/L | COMPUTER | CG | O | POCC CADRE | N | Y Y Y |

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COMPUTER INPUT/OUTPUT SUMMARY

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| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|-------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| ACTIVITY.INP | 61440 | 12288 | ACTIVITY CARDS, INITIATOR CARDS, TERMINATION CARDS. |
| ACTIVITY.VFY | 61440 | 12288 | ACTIVITY CARDS, INITIATOR CARDS, TERMINATION CARDS. |
| ANG'LR DIST'CE | 1024000 | 747520 | THIS IS AN ASCII FILE CONTAINING A MATRIX OF THE ANGULAR DISTANCE BETWEEN EACH TARGET AND ALL OTHERS. THERE ARE NO OTHER PARAMETERS ON THE FILE AND THE POSITION IN THE MATRIX CORRESPONDS TO A GIVEN TARGET. |
| ASCN NODE | 409600 | 245760 | THIS IS A LIST-DIRECTED FILE CONTAINING DETAILED ORBITAL EPHEMERIS DATA (SAME AS DETAILED EPHEMERIS FILE) FOR EACH ORBIT AT THE TIME OF ASCENDING NODE. |
| AT PHY CONSTS | 204800 | 102400 | THIS IS AN ON/OFF FILE CONTAINING ON/OFF TIMES WHICH REPRESENT TIME PERIODS WHERE CERTAIN ATMOSPHERIC PHYSICS REQUIREMENTS/CONSTRAINTS HAVE BEEN SATISFIED. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| AT PHY TARGETS | 204800 | 102400 | THIS IS AN ON/OFF FILE CONTAINING ATMOSPHERIC PHYSICS TARGETS AFTER REQUIREMENTS/CONSTRAINTS HAVE BEEN COMBINED (BY UNION, INTERSECTION, OR COMPLEMENT). THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|-------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| ATT TL | 409600 | 102400 | THIS IS AN ON/OFF FILE CONTAINING DATA DEFINING THE ORBITER ATTITUDE TIMELINE. THIS IS A SUBJECT TYPE 16 ON/OFF FILE. |
| ATT TL (NDF) | 10240000 | 1024000 | THIS IS A NAME-DIRECTED FILE CONTAINING DATA DEFINING THE ORBITER ATTITUDE TIMELINE AND ASSOCIATED DATA. INCLUDED ARE ATTITUDE DATA, STATE VECTOR DATA, TARGET DATA, SENSOR DATA, TIME, ATTITUDE RATES DATA, KEYWORDS, AND MANEUVER TIMES. |
| BORB CONSTS | 1024000 | 512000 | THIS IS AN ON/OFF FILE CONTAINING ON/OFF TIMES WHICH REPRESENT TIME PERIODS WHERE CERTAIN PLASMA PHYSICS REQUIREMENTS/CONSTRAINTS HAVE BEEN SATISFIED. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| BORB PAR. | 14336000 | 4096000 | THIS IS A LIST-DIRECTED FILE CONTAINING TIME HISTORIES OF GEOMAGNETIC PARAMETERS. THE DATA DEFINES THE STRENGTH OF THE GEOMAGNETIC FIELD AND THE DIRECTIONS OF THE FIELD AT THE CURRENT ORBITER POSITION. |
| CAND GSTAR | 1024000 | 409600 | THE CANDIDATE GUIDE STAR FILE CONTAINS THE FOLLOWING INFORMATION FOR EACH GUIDE STAR: SCIENCE STAR SEQUENCE NUMBERS AND ID NUMBERS, SCIENCE STAR RIGHT ASCENSION, AND DECLINATION, IPS OPERATION MODE, NUMBER OF GUIDE STARS IN ANNULUS/BONESIGHT REGIONS, GUIDE STAR LOCATION IN SKYMAP CATALOG, AND GUIDE STAR POSITION (ANNULUS OR BONESIGHT). |

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| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|-------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| CASE STO | 204800 | 204800 | ORBIT PARAMETERS STORED IN CASE STORAGE FILES FOR NESP, ASEP, AND/OR TANRAY: RADIUS (KM-NSEP) (M-ASEP), VELOCITY (KM/S-NSEP) (M/S-ASEP), FLIGHT PATH ANGLE (DEG), GEOCENTRIC LATITUDE (DEG), LONGITUDE (DEG), INERTIAL AZIMUTH (DEG), AND TIME (MET-HRS). |
| CEL TAR | 614400 | 368640 | THIS IS A NAME-DIRECTED FILE CONTAINING AN ARRAY DEFINING CANDIDATE CELESTIAL TARGETS. THE ARRAY CONTAINS TARGET NAME, RIGHT ASCENSION, AND DECLINATION. |
| CEL TAR AC/LOS | 10240000 | 1024000 | THIS IS AN ON/OFF FILE CONTAINING CELESTIAL TARGET(S) ACQUISITION AND LOSS TIMES. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| CEL TARGETS | 10240000 | 1024000 | THIS IS AN ON/OFF FILE CONTAINING CELESTIAL TARGETS AFTER REQUIREMENTS/CONSTRAINTS HAVE BEEN COMBINED (BY UNION, INTERSECTION, OR COMPLEMENT). THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| COMM AC/LOS | 1024000 | 737280 | THIS IS AN ON/OFF FILE CONTAINING TDRS AND GROUND STATION COMMUNICATION INFORMATION. INCLUDED ARE A TWO SATELLITE TDRS SYSTEM (S-BAND OR K-BAND, UP OR DOWN) AND 11 GROUND STATIONS. THIS FILE IS A COMBINATION OF SUBJECT TYPES 6,0, AND 1. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|------------------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| COMMAND T/L | 102400 | 51200 | LIST OF ALL COMMANDS BEING EXECUTED BY FLIGHT SOFTWARE. |
| COMMAND.FIN | 102400 | 51200 | ACTIVITY CARDS, INITIATOR CARDS, TERMINATION CARDS. |
| COMMAND.INP | 61440 | 40960 | ACTIVITY CARDS, INITIATOR CARDS, TERMINATION CARDS. |
| COMMAND.VFY | 81440 | 40960 | ACTIVITY CARDS, INITIATOR CARDS, TERMINATION CARDS. |
| COMPARED DATA MGMT CHECKLIST | -0- | 204800 | THE COMPARED DATA MANAGEMENT CHECKLIST CONTAINS DATA INDICATING THE DIFFERENCES OF A CURRENT DATA MGMT CHECKLIST FILE AS COMPARED TO A PREVIOUS FILE. FLAGS INDICATE WHETHER DATA ENTRIES HAVE BEEN REPLACED (R), DELETED (D), OR ARE NEW ENTRIES (N). |
| COO | 409600 | 133120 | THE COOBSERVATION FILE (PI DEVELOPED) IS AN ASCII FILE CONTAINING DATA DEFINING POTENTIAL CELESTIAL OBJECTS TO BE OBSERVED DURING A MISSION. DATA INCLUDES NAME, ID, CLASS, PRIORITY, TIME, CONTINUOUS/NON-CONTINUOUS, CONSTRAINTS, RIGHT ASCENSION, DECLINATION, AND HUT SLIT ANGLE. |
| CREW ACT | 1000000 | 100000 | LISTING OF ON/OFF TIMES FOR EACH ACTIVITY EACH CREWMAN PERFORMS. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|---------------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| DATA FLOW INPUT VARIABLES | -0- | 2048 | THE DATA FLOW INPUT VARIABLES FILE CONTAINS A LIST OF THE VARIABLES COMMON TO MANY OF THE DATA FLOW MODULES. THE BASIC MISSION DESCRIPTION PLUS THE CONFIGURATION OF THE MISSION DATA FLOW ARE DEFINED BY MANIPULATING THE VALUES OF THESE VARIABLES. |
| DATA FLOW MODELS | -0- | 256000 | THE DATA FLOW MODELS FILE CONTAINS DATA WHICH CORRELATES THE EXPERIMENT FO'S AND STEPS IN THE EXPERIMENT TIMELINE FILE TO THE DATA REQUIREMENTS SETS FOR THOSE EXPERIMENTS IN THE DATA FLOW REQMTS FILE. |
| DATA FLOW REQMTS | -0- | 2048 | THE DATA FLOW REQUIREMENTS FILE CONTAINS THE DATA REQUIREMENTS FOR EACH EXPERIMENT. AN EXPERIMENT MAY HAVE MORE THAN ONE SET OF REQMTS SINCE SOME HAVE MULTIPLE FO'S. |
| DATA FLOW SCHED | -0- | 409600 | THE DATA FLOW SCHEDULE FILE CONTAINS DATA DEFINING HDRR RECORD/DUMP TIMES, VIDEO AND ANALOG DOWNLINK TIMES (REALTIME AND DUMPS), ANALOG/VIDEO RECORDERS (1&2) RECORD TIMES, HRM DOWNLINK FORMAT SCHEDULE, TDRS HANDOVER AND/OR ONBOARD RECONFIGURATION POINTS, AND ORBITER TV CASSETTE RECORDERS (1&2) ACTIVITIES. THIS FILE BECOMES PART OF THE MISSION FILE DIRECTORY AND IS THE INTERFACE TO OTHER MISSION PLANNING ORGANIZATIONS. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|------------------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| DATA MANGHT CHECKLIST | -0- | 6144000 | THE DATA MANAGEMENT CHECKLIST FILE CONTAINS DATA WHICH PROVIDES AN OVERVIEW OF ONBOARD RECORDERS, WRN FORMATS, DOWNLINKS, EXP. DEDICATED CHANNELS, AND POCC CONFIGURATION ACTIVITIES. |
| DATA MANGHT CHECKLIST (PREV) | -0- | 6144000 | THE DATA MANAGEMENT CHECKLIST FROM A PREVIOUS CYCLE OR TIME. DATA CONTENT IS THE SAME. |
| DDU DISPLAYS | 20480 | 12288 | EXECUTABLE IMAGE OF ALL MITRA HEX COMMANDS TO DDU. |
| DETAIL EPHEM | 24576000 | 8192000 | THIS IS A LIST-DIRECTED FILE CONTAINING A DETAILED TIME HISTORY OF ORBITAL EPHEMERIS DATA. AT EACH TIME POINT 110 PARAMETERS ARE COMPUTED INCLUDING THE VEHICLE STATE VECTOR IN 3 DIFFERENT COORDINATE SYSTEMS (SPHERICAL, POLAR AND CARTESIAN), AND ORBITAL ELEMENTS. ALSO PROVIDED ARE VARIOUS PARAMETERS DESCRIBING THE GEOMETRY OF ORBITAL FLIGHT. |
| DFA SCHEDULAR CHNDS | -0- | 2048 | THE DFA SCHEDULAR FILE CONTAINS DATA TO ALLOW THE USER AN OPTIONAL INPUT WHICH HE MAY EMPLOY TO MODIFY THE NORMAL SCHEDULING PROCESS. COMMANDS ALLOW THE USER TO OVERRIDE THE BASIC SCHEDULAR LOGIC IF AN UNUSUAL SITUATION IS ENCOUNTERED. |

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| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|---------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| EARTH SHADOW | 40960 | 30720 | THIS IS AN ON/OFF FILE CONTAINING EARTH SHADOW ACQUISITION AND LOSS TIMES. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| EARTH TARGETS | 1024000 | 512000 | THIS IS AN ON/OFF FILE CONTAINING EARTH OBSERVATION TARGETS AFTER REQUIREMENTS/CONSTRAINTS HAVE BEEN COMBINED (BY UNION, INTERSECTION, OR COMPLEMENT). THIS IS A SUBJECT TYPE 0 ON-OFF FILE. |
| ECOS T/L PRINTOUTS | -0- | -0- | PRINTOUT CONTAINS ALL ECOS MASTER AND SUBORDINATE TIMELINES. |
| ECOS TL | 775296 | 455296 | MASTER TIMELINE EVENTS, NET FOR ALL MASTER TIMELINE EVENTS, SUBORDINATE TIMELINE EVENTS, DELTA TIMES FOR ALL SUBORDINATE TIMELINE EVENTS. |
| ECOS TL (EBCDIC) | 775296 | 455296 | MASTER TIMELINE EVENT, NET FOR ALL MASTER TIMELINE EVENTS, SUBORDINATE TIMELINE EVENTS, DELTA TIMES FOR ALL SUBORDINATE TIMELINE EVENTS |
| ESS MODEL | 34000 | 4000 | PRIMARY INPUT FOR ESS. CONTAINS DATA NECESSARY TO SCHEDULE EXPERIMENTS FOR SL MISSIONS. ESS MODEL FILES ARE CREATED AND UPDATED BY VME. SCHEDULING CONTROL DATA MAY ALSO BE CREATED OR UPDATED BY ESS. |
| ESS MODELS PRINTOUT | -0- | -0- | PRINTOUT OF THE ESS MODEL FILE. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|---------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| ESS TARGET | 1652 | 2400 | PROVIDES TARGET AC/LOS TIMES AND PRESET ATTITUDE TIMES WHICH ARE USED BY ESP. CREATED AND EDITED USING TAE. |
| ESS TARGET PRINTOUT | -0- | -0- | PRINTOUT OF THE ESS TARGET FILE. |
| EXP | 1000000 | 100000 | GIVES ALL ON/OFF TIMES OF EACH EXPERIMENT FOR A PARTICULAR MISSION SCHEDULE. |
| EXP T/L | 0 | 1000 | PRIMARY OUTPUT OF ESP. PROVIDES A PERMANENT RECORD OF A SCHEDULE TO USE IN DISPLAY PROGRAMS AND FOR INITIALIZATION. |
| EXP T/L PRINTOUTS | -0- | -0- | PRINTOUT OF THE EXP T/L FILE. |
| EXPERIMENT HEADER | -0- | 102400 | THE EXPERIMENT HEADER FILE CONTAINS DATA REQUIRED TO RUN MANY OF THE REPORT MODULES. THE FILE INCLUDES DEDICATED CHANNELS AND RELATED EXPERIMENT ID'S, ECIO SUBSETS AND RELATED EXPERIMENT ID'S, AND USER ADDRESS SPACES AND RELATED EXPERIMENT ID'S. THE FILE AIDS THE USER IN DEFINING WHAT INFORMATION IS CONTAINED IN THE REPORTS. |
| GND STA AC/LOS | 266240 | 102400 | THIS IS AN ON/OFF FILE CONTAINING GROUND STATION(S) ACQUISITION AND LOSS TIMES. THIS IS A SUBJECT TYPE 1 ON/OFF FILE. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|-------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| GND TRK | 2048000 | 1433600 | THIS IS A LIST-DIRECTED FILE CONTAINING A TIME HISTORY OF DATA DEFINING THE GROUND TRACK OF AN ORBITING VEHICLE. THE FILE CONTAINS MISSION ELAPSE TIME, GEODETIC LATITUDE, LONGITUDE, REV NUMBER, INERTIAL AZIMUTH, ALTITUDE, MANEUVER COUNTER, PERGEE AND APOGEE ALTITUDE, AND BETA ANGLE. |
| GSTAR CAT | 6144000 | 3072000 | THE GSTAR CATALOG FILE CONTAINS THE FOLLOWING TYPES OF INFORMATION: SCIENCE TARGET NAME, NUMBER, RIGHT ASCENSION, DECLINATION, ASTRONOMICAL NAME, CLASS AND SUBCLASS, PRIORITY, OBSERVATION LENGTH, CONTINUOUS/NON-CONTINUOUS, VIEWING TARGET TYPE (DAY, NIGHT, BOTH), ROLL, AND VISUAL MAGNITUDE. FOR GUIDE STARS; TOTAL NUMBER, NUMBER IN BORESIGHT REGION, IPS OPERATION MODE, DATE/TIME TARGET WAS WRITTEN, RIGHT ASCENSION, DECINATION, POSITION ANGLE, ANGULAR DISTANCE BETWEEN GUIDE STAR AND TARGET STAR, VISUAL, ULTRA-VIOLET, AND BLUE MAGNITUDES, GALACTIC LATITUDE AND LONGITUDE, SKYMAP NO., SPECTRAL CLASS, POSITION (BORESIGHT OR ANNULUS). |
| HEMSPR CONSTS | 204800 | 102400 | THIS IS AN ON/OFF FILE CONTAINING ON/OFF TIMES WHICH REPRESENT TIME PERIODS WHERE CERTAIN HEMISPHERE REQUIREMENTS/CONSTRAINTS HAVE BEEN SATISFIED. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |

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| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|----------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| HFS ERROR | -0- | 20480 | THE HFS ERROR FILE CONTAINS ALL THE SCHEDULING ERRORS AND WARNINGS ENCOUNTERED BY THE HRM FORMAT SCHEDULAR MODULE. |
| HRM FORMATS DEFN | -0- | 4096 | THE HRM FORMATS DEFINITION FILE CONTAINS DATA DESCRIBING ALL THE HRM FORMATS DEFINED FOR A MISSION. EACH FORMAT DEFINES A CONFIGURATION OF THE HRM, FORMAT RATES, AND ANY RECORDER CAPABILITIES FOR EACH FORMAT. |
| HRM POSSIBLE FORMATS | -0- | 204800 | THE HRM POSSIBLE FORMATS FILE CONTAINS DATA WHICH DEFINES TDRS COVERAGE AND POSSIBLE HRM FORMATS FOR GIVEN MET'S. |
| ID LIST | 102400 | 40960 | THIS IS AN ASCII FILE CONTAINING ALL THE ID'S OF SCHEDULED TARGETS IN ORDER BY THEIR NUMERICAL VALUE. |
| ID/SEQ LIST | 102400 | 40960 | THIS IS AN ASCII FILE CONTAINING SCHEDULED TARGET ID'S, SEQUENCE NUMBERS AND OTHERS SUPPORTING DATA. |
| INHBT | -0- | 5000 | CONTAINS DEFINITION OF ALL INHIBITS AND THEIR ON/OFF TIMES. |
| IPOL | 204800 | 102400 | THIS IS A LIST-DIRECTED FILE CONTAINING DATA ON EACH POINTING IN TIME ORDER. DATA INCLUDES: TIME, ELEVATION (IPS), CROSS ELEVATION (IPS), ROLL (IPS), OBJECTIVE LOAD NUMBER, RIGHT ASCENSION AND DECLINATION, OBJECTIVE LOAD ROLL, CELESTIAL ROLL, TARGET ID AND NAME. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|--------------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| IPS ROLL ANGLES | 153600 | 102400 | THE IPS ROLL ANGLE FILE CONTAINS THE FOLLOWING TYPES OF DATA: MISSION ELAPSE TIME, STAR NAMES 1 AND 2, BORESIGHT STAR RIGHT ASCENSION AND DECLINATION, RIGHT TRACKER RIGHT ASCENSION AND DECLINATION, IPS ELEVATION, CO-ELEVATION AND ROLL ANGLES, CELESTIAL ROLL ANGLE (ALSO THIS ANGLE PLUS/MINUS 30 DEG), ORBITER ATTITUDE ANGLES, AND ATTITUDE REFERENCE COORDINATE SYSTEM. |
| ITL | 400 | 388 | SUBORDINATE TIMELINE EVENTS, DELTA TIME FROM ACTIVATION OF SUBORDINATE TIMELINES. |
| JOTF | 409600 | 307200 | THIS IS A MITRA HEX FILE CONTAINING DATA FOR THE JOINT OPERATIONS TARGET FILE (ONBOARD DATA SET JOTF). THESE DATA ARE USED BY ONBOARD ECAS TASKS AND INCLUDE: TARGET ID AND NAME, RIGHT ASCENSION, DECLINATION, ROLL, EXPERIMENT SEQUENCE NUMBERS, AND SIMILAR MOVING TARGET DATA. |
| JOTF VERIFICATION REPORT | -0- | -0- | PRINTOUT OF THE JOINT OPERATIONS TARGET FILE (JOTF) DATA SET USED TO VERIFY THE CONTENTS OF THE DATA SET. |
| JSC CAP | 1000000 | 100000 | CONTAINS ALL THE CREW ACTIVITY SCHEDULED AS WELL AS OTHER PERTINENT INFORMATION TO BE USED AS INPUT TO THE JSC CAPS2 SYSTEM. CREATED BY ESP. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|--------------------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| KEYWORD | 2048000 | 1024000 | THIS IS A NAME-DIRECTED FILE CONTAINING AN ARRAY DEFINING KEYWORDS AND ASSOCIATED ATTITUDE DATA. THE ARRAY CONTAINS NUMBER OF KEYWORDS, KEYWORD DESCRIPTION LIST, ATTITUDE DATA LIST, AND KEYWORD LIST. |
| LAUNCH WINDOW/LAUNCH TIME DATA | -0- | -0- | A PLOT (AND SUPPORTING DATA) WHICH DEFINES THE AVAILABLE LAUNCH WINDOW VS DAY OF YEAR. THE PLOT OVERLAYS ALL CONSTRAINTS TO PRODUCE THE WINDOW IMAGES. |
| MASTER FILE | -0- | 512000 | THE MASTER FILE CONTAINS THE OVERALL DATA FLOW PLAN. THIS INCLUDES RECORDER FILL/DUMP PLANS, KU-BAND CHANNELS 2 AND 3 USAGE PROFILE, HRM FORMATS, SATELLITE COVERAGE, DATA RATES, POCC CONFIGURATIONS, AND A SUMMARY OF ANY LOST DATA. |
| MDRPG ERROR | -0- | 20480 | THE MDRPG ERROR FILE LISTS ALL THE ERRORS ENCOUNTERED BY THE MISSION DATA REQUIREMENTS GENERATOR MODULE. |
| MI | 61440 | 20480 | MASTER TIMELINE EVENTS, MET TIMES FOR ALL EVENTS. |
| MISSION WINDOWS | -0- | 512000 | THE MISSION WINDOWS FILE CONTAINS WINDOWS OF OPPORTUNITY FOR DATA DOWNLINKS AND RECORDER DUMPS. |
| MMU ALLOCATION | 4096 | 2048 | NAMES OF ALL FILES ON THE MMU AND A LITTLE INFORMATION ABOUT ALL OF THEM. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|---------------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| MMU ALLOCATION PRINTOUT | -0- | -0- | A PRINTOUT OF THE MMU ALLOCATION FILE WHICH CONTAINS AN ORDERED LIST OF MMU FILES AND SUPPORTING DATA. |
| MMU DAT SET | 40960 | 20480 | THIS IS A MITRA HEX FILE CONTAINING DATA FOR USE BY THE ONBOARD ECOS TIMELINE MAINTENANCE SYSTEM. DATA INCLUDES SUN RISE/SET EVENTS, MOON RISE/SET EVENTS, TDRS ACQUISITION AND LOSS TIMES, SUN RIGHT ASCENSION AND DECLINATION, TIME OF NODAL CROSSING, AND NODAL PERIOD. |
| MMU OPTIMIZATION REPORT | -0- | -0- | SUM OF TAPE MOVEMENT FOR THE MISSION AND AN ORDERED LIST OF MMU FILE NAMES. |
| MOON RISE/SET | 40960 | 30720 | THIS IS AN ON/OFF FILE CONTAINING MOON RISE AND SET TIMES. THIS IS A SUBJECT TYPE 0 ON/OFF FILE |
| MSN IND TARGETS | 409600 | 204800 | THIS IS AN ON/OFF FILE CONTAINING ALL MISSION INDEPENDENT TARGETS. THESE INCLUDE: SUN RISE/SET, MOON RISE/SET, PRELIMINARY TDRS AC/LOS, GROUND STATION AC/LOS, SOUTH ATLANTIC ANOMALY AC/LOS. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| MSN T/L TABLES, PRINTOUTS | -0- | -0- | DETAILED TABULATIONS, PLOTS, AND PRINTOUTS OF THE MISSION TIMELINE. |
| MSN TARGETS | 409600 | 204800 | THIS IS AN ON/OFF FILE CONTAINING ALL THE EXPERIMENT TARGET OPPORTUNITIES (ALL DISCIPLINES REPRESENTED IN THE PAYLOAD COMPLEMENT) FOR THE MISSION. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|-------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| MTL | 2048 | 2040 | MASTER TIMELINE EVENTS. MET TIMES. |
| MVR TL | 204800 | 122880 | THIS IS AN ON/OFF FILE CONTAINING DATA DEFINING THE ORBITER MANEUVER TIMELINE. THIS IS A SUBJECT TYPE 16 ON/OFF FILE. |
| MWG ERROR | -0- | 20480 | THE MWG ERROR FILE LISTS ALL THE SCHEDULING COMMANDS THAT COULD NOT BE HONORED BY THE CREATE MISSION WINDOWS MODULE BECAUSE OF CONFLICTS. |
| NOTES | 1600 | 700 | DATABASE OF PCAP NOTES FOR USE IN GENERATING PCAP CHARTS. |
| NSEP EPIEM | 81920 | 40960 | THIS A LIST-DIRECTED FILE CONTAINING A TIME HISTORY OF VEHICLE POSITION AND VELOCITY, AND ORBITAL ELEMENTS. |
| OBJ LOAD | -0- | 573440 | THIS IS AN ASCII FILE CONTAINING GUIDE STAR OBJECTIVE LOAD DATA WHICH INCLUDES: IPS OPERATING MODE, OBJECTIVE LOAD NO'S, EXPERIMENT NO'S, TARGET ATTITUDE DATA, TRACKER SCAN DATA, SUN POSITION DATA, DIRECTION COSINE MATRIX FOR REFERENCE SYSTEM TRANSFORMATIONS, TRACKER FOV GUIDE STAR AVAILABILITY DATA, GUIDE STAR RIGHT ASCENSION AND DECLINATION, AND TRACKER COUNT RATE DATA. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|----------------------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| OBJ LOAD SUMMARY-SOLAR & STELLAR | -0- | -0- | THE OBJECTIVE LOAD SUMMARY IS A PRINTOUT CONTAINING THE FOLLOWING INFORMATION: OBJECTIVE LOAD NUMBER, IPS PITCH AND YAW OFFSET ANGLES, IPS CELESTIAL ROLL ANGLE, IPS POINTING MODE, RIGHT STAR TRACKER GUIDE STAR NAME, VISUAL MAGNITUDE, RIGHT ASCENSION AND DECLINATION, LEFT STAR TRACKER GUIDE STAR NAME, VISUAL MAGNITUDE, RIGHT ASCENSION AND DECLINATION, BEGIN AND END, DATE/TIME THAT OBJECTIVE LOAD IS VALID. |
| OBS CONSTS | 102400 | 61440 | THIS IS AN ON/OFF FILE CONTAINING ON/OFF TIMES WHICH REPRESENT TIME PERIODS WHERE CERTAIN EARTH OBSERVATION REQUIREMENTS/CONSTRAINTS HAVE BEEN SATISFIED. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| ORBITER OCCULT | 409600 | 358400 | THIS FILE CONTAINS THE ORBITER OCCULTATION DATA RELATIVE TO THE TDRS ANTENNAS. THIS FILE IS OBTAINED FROM JSC. |
| ORS ERROR | -0- | 20480 | THE ORS ERROR FILE LISTS ALL THE SCHEDULING COMMANDS THAT COULD NOT BE HONORED BY THE ONBOARD RECORDER SCHEDULAR MODULE BECAUSE OF CONFLICTS. |
| PBS ERROR | -0- | 20480 | THE PBS ERROR FILE CONTAINS ALL THE SCHEDULING ERRORS AND WARNINGS ENCOUNTERED BY THE PLAYBACK SCHEDULAR MODULE. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|------------------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| PCAP CHARTS | -0- | -0- | DETAILED 1-HR CHARTS PRODUCED ON A HIGH RESOLUTION LASER PRINTER WHICH BECOME PART OF THE PDF. THE PLAN SHOWS THE PROCEDURES THE CREW EXECUTES AND ASSOCIATED NOTES. IT ALSO SHOWS AUXILIARY INFORMATION SUCH AS ORBIT ALTITUDE AND MANEUVERS, SUN/SHADOW, K- AND S-BAND COMMUNICATIONS, HRM FORMATS, RECORDER USAGE, ORBIT REV NUMBER, AND OTHER DATA. |
| PL PHY TARGETS | 1024000 | 512000 | THIS IS AN ON/OFF FILE CONTAINING PLASMA PHYSICS TARGETS AFTER REQUIREMENTS/CONSTRAINTS HAVE BEEN COMBINED (BY UNION, INTERSECTION, OR COMPLEMENT). THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| POCC CHECKLIST & COMMAND T/L | -0- | -0- | PRINTOUT OF THE POCC CHECKLIST AND COMMAND TIMELINE. |
| POCC CONFIG DEFN | -0- | 20480 | THE POCC CONFIGURATION DEFINITION FILE CONTAINS DATA DEFINING EACH PARTICULAR ROUTING CONFIGURATION OF THE DATA STREAMS ON THE GROUND, BOTH REALTIME AND PLAYBACK. |
| POCC POSSIBLE CONFIG | -0- | 102400 | THE POCC POSSIBLE CONFIGURATION FILE PROVIDES ALL POSSIBLE POCC CONFIGURATIONS (BOTH REALTIME AND PLAYBACK) AND ALL THE POSSIBLE USER GROUP CONFIGURATIONS FOR ANY GIVEN TIME. |
| POCC.FIN | 61440 | 12288 | LIST OF ALL POCC UNIQUE ACTIVITIES IN SEQUENTIAL ORDER. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|---------------------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| PREL ATT TL | 102400 | 81920 | THIS IS AN ON/OFF FILE CONTAINING THE PRELIMINARY ATTITUDE TIMELINE. THIS IS A SUBJECT TYPE 16 ON/OFF FILE. |
| PREL TDRS AC/LOS | 3072000 | 2867200 | THIS IS AN ON/OFF FILE CONTAINING PRELIMINARY TDRS ACQUISITION AND LOSS TIMES. THIS IS A SUBJECT TYPE 6 ON/OFF FILE. |
| PRINTOUT OF ECOS TIMELINE | -0- | -0- | PRINTOUT OF THE ECOS TIMELINE TO BE USED IN THE ECOS TIMELINE DOCUMENT DEVELOPMENT. |
| PRINTOUTS OF ATMOS PHYSICS DATA | -0- | -0- | PRINTOUT OF ATMOSPHERIC PHYSICS TARGET OPPORTUNITIES DATA. |
| PRINTOUTS OF CAND GUIDE STARS | -0- | -0- | THE CANDIDATE GUIDE STAR PRINTOUTS CONTAIN DATA FOR GUIDE STAR SELECTION WHICH INCLUDE: STAR NAME AND NUMBER, STAR NO. OF NEXT HIGHER/LOWER RIGHT ASCENSION, STAR NO OF NEXT HIGHER/LOWER DECLINATION, STAR MAGNITUDE AND MAGNITUDE CORRELATION DATA, TIME RANGES WHERE STAR CAN BE USED FOR IPS ALIGNMENT, IPS ROLL ANGLES AT BEGIN/END OF TIME RANGES. |
| PRINTOUTS OF CELESTIAL DATA | -0- | -0- | PRINTOUT OF CELESTIAL TARGET OPPORTUNITIES DATA. |
| PRINTOUTS OF EARTH OBSERV DATA | -0- | -0- | PRINTOUT OF EARTH OBSERVATION TARGET OPPORTUNITIES DATA. |
| PRINTOUTS OF MMU DATA SET | -0- | -0- | PRINTOUTS OF THE MMU DATA SET ARE OUTPUT FOR MANUAL VERIFICATION OF THE DATA SET. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|----------------------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| PRINTOUTS OF MSN IND DATA | -0- | -0- | PRINTOUT OF MISSION INDEPENDENT TARGET OPPORTUNITIES DATA. |
| PRINTOUTS OF PLASMA PHYSICS DATA | -0- | -0- | PRINTOUT OF PLASMA PHYSICS TARGET OPPORTUNITIES DATA. |
| PRINTOUTS OF SOLAR DATA | -0- | -0- | PRINTOUT OF SOLAR TARGET OPPORTUNITIES DATA. |
| PRINTOUTS OF STRAY LIGHT DATA | -0- | -0- | PRINTOUTS OF STRAY LIGHT DATA TO BE USED WHEN SELECTING SOLAR GUIDE STARS. |
| PROC. | 1000 | 500 | DATABASE OF PCAP PROCEDURES FOR USE IN GENERATING PCAP CHARTS. |
| PROCAM | 10240000 | 4096000 | THIS IS A LIST-DIRECTED FILE CONTAINING TIME HISTORIES OF ORBITER ATTITUDES, BODY-REFERENCE POINTING DIRECTIONS FOR VARIOUS ORBIT-RELATED ITEMS (VELOCITY VECTOR, ANGULAR MOMENTUM VECTOR, ETC.), CELESTIAL OBJECTS (SUN, MOON) AND TDRS. |
| PROFILE | -0- | 307200 | THE DATA REQUIREMENTS PROFILE FILE (PROFILE) CONTAINS A PROFILE OF DATA REQUIREMENTS AND POCC RESOURCE USAGE BASED ON THE EXPERIMENT TIMELINE AND THE DATA FLOW REQUIREMENTS SETS. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|-------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| PTS CHARTS | -0- | -0- | 6-HR PLOTS WITH MET & AMET TICK MARKS EVERY 6 MINUTES. THE CHARTS ARE USED FOR ANALYSIS BY MISSION PLANNING ENGINEERS AND FOR MISSION TIMELINE PUBLICATION. ITEMS ON EACH PLOT PAGE INCLUDE: CREW ACTIVITIES; K-BAND, S-BAND, AND GSTDN COMMUNICATIONS COVERAGE; DATA AND VIDEO DOWNLINKS; DATA AND VIDEO RECORDER DUMPS; WATER DUMP AND THRUSTER INHIBITS; ATTITUDE AND MANEUVER PROFILE; RA & DEC OF ORBITER - Z AXIS FOR INERTIAL ATTITUDES; A MOON PHASE DWG; ORBITER REV NO. AND GROUND TRACK PLOT; SUN/SHADOW PERIODS; AND UNATTENDED (NO CREW) PAYLOAD OPERATIONS. PTS CHARTS ARE PART OF THE PDF, THE STANDARD POCC DOCUMENTATION FOR SPACELAB MISSIONS, AND ARE INCLUDED IN THE FLIGHT DEFINITION DOCUMENT. |
| RAD ENVIR | 6144000 | 3072000 | THIS IS A LIST-DIRECTED FILE CONTAINING A TIME HISTORY OF THE RADIATION ENVIRONMENT OF A VEHICLE IN ORBIT. PARAMETERS INCLUDE FLUX DATA, 8-FIELD DATA, AND MCILMAIN PARAMETERS. |
| READP1 | 1024000 | 512000 | THIS IS AN ASCII FILE CONTAINING TARGET DATA AND SUPPORTING DATA USED TO SCHEDULE SCIENCE OBSERVATIONS. TYPES OF DATA INCLUDE TARGET NAME/CLASS/SUBCLASS, REQUESTED VIEWING TIME, CONSTRAINTS DATA, RIGHT ASCENSION AND DECLINATION, TIME GOAL DATA, EARTH SHADOW ACQ/LOSS, AND SAA ACQ/LOSS. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|-------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| RESERVE PERIODS | 204800 | 61440 | THIS IS AN ASCII FILE DEFINING PERIODS OF TIME RESERVED FOR SCIENCE OBSERVATION OR OTHER ACTIVITIES. THE DATA INCLUDES NAME, SCIENCE/NON-SCIENCE, MANEUVER/NO MANEUVER, ON TIME, OFF TIME, CLASS, REQUESTED TIME, RIGHT ASCENSION, AND DECLINATION. |
| SAA AC/LOS | 61440 | 30720 | THIS IS AN ON/OFF FILE CONTAINING SOUTH ATLANTIC ANOMALY (SAA) ACQUISITION AND LOSS TIMES. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| SCIENCE SCHED'LE | 819200 | 409600 | THIS IS AN ON/OFF FILE CONTAINING ON/OFF TIMES OF THE OBSERVATION PERIODS FOR EACH TARGET (SUBJECT). THIS IS A SUBJECT TYPE 3 ON/OFF FILE. |
| SITE AC/LOS | 1024000 | 512000 | THIS IS AN ON/OFF FILE CONTAINING EARTH GROUND SITE ACQUISITION AND LOSS TIMES. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| SITE DEF. | 409600 | 204800 | THIS IS A NAME-DIRECTED FILE CONTAINING AN ARRAY DEFINING GROUND SITE POLYGONS. THE ARRAY CONTAINS THE SITE NAME, NUMBER OF SIDES (VERTICES) FOR EACH SITE, AND GEODETIC LATITUDE AND LONGITUDE OF EACH VERTEX. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|-------------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| SKYMAP DATABSE | 819200 | 409600 | THE SOURCE OF THE STARS FOR DEVELOPING THE SKYMAP DATABASE IS THE "SKYMAP STAR CATALOG" PREPARED FOR GSFC BY CSC IN DECEMBER, 1977. VERSION III OF THIS CATALOG CONTAINS 248727 STARS EQUAL OR GREATER IN BRIGHTNESS THAN 9TH MAGNITUDE. THE STAR POSITIONS ARE GIVEN IN TERMS OF RIGHT ASCENSION AND DECLINATION WITH RESPECT TO EPOCH 2000. THERE ARE 135 DESCRIPTIVE OR QUALIFYING WORDS OF DATA FOR EACH STAR IN THE CATALOG. THESE ARE REDUCED TO 34 WORDS IN THE SKYMAP DATABASE. |
| SOLAR CONSTS | 102400 | 40960 | THIS IS AN ON/OFF FILE CONTAINING ON/OFF TIMES WHICH REPRESENT TIME PERIODS WHERE CERTAIN SOLAR REQUIREMENTS/CONSTRAINTS HAVE BEEN SATISFIED. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| SOLAR TARGETS | 204800 | 102400 | THIS IS AN ON/OFF FILE CONTAINING SOLAR TARGETS AFTER REQUIREMENTS/CONSTRAINTS HAVE BEEN COMBINED (BY UNION, INTERSECTION, OR COMPLEMENT). THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| STATE VECTOR ELECTRONIC | 1618 | 1618 | ELECTRONIC TRANSFER OF STATE VECTOR CONTAINING A COORDINATE SYSTEM DEFINITION, TIME (GMT), VECTOR POSITION (M), AND VECTOR VELOCITY (M/S). |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|-----------------------|------------------|---------|--|
| | MAXIMUM | MINIMUM | |
| STATE VECTOR PRINTOUT | 0 | 0 | PRINTOUT CONTAINS THE FOLLOWING ORBIT PARAMETERS: RADIUS (M), VELOCITY (M/S), FLIGHT PATH ANGLE (DEG), GEOCENTRIC LATITUDE (DEG), LONGITUDE (DEG), INERTIAL AZIMUTH (DEG), AND ORBIT INSERTION TIME (MET-HRS). |
| STL PRINTOUTS | -0- | -0- | PRINTOUTS OF THE ECOS SUBORDINATE TIMELINES. |
| SUB TL | 400 | 388 | SUBORDINATE TIMELINE EVENTS. DELTA TIMES FOR EACH EVENT. |
| SUBCOO | 409600 | 122880 | THE SUBCOO FILE IS AN ASCII FILE CONTAINING A SUBSET OF THE CELESTIAL OBJECTS CONTAINED IN THE COO FILE. THE DATA TYPES ARE THE SAME AS THOSE CONTAINED IN THE COO FILE. |
| SUN AZ/ELV | 409600 | 102400 | THIS IS A LIST-DIRECTED FILE CONTAINING A TIME HISTORY OF THE FOLLOWING DATA: TIME (GMT AND MET), AZIMUTH AND ELEVATIONS OF THE SUN IN THE ATMOS EXPERIMENT COORDINATE SYSTEM, TANGENT RAY HEIGHT, SUN AZIMUTH AND CO-ELEVATION IN THE ORBITER COORDINATE SYSTEM, ALTITUDE, SUN RIGHT ASCENSION AND DECLINATION. |
| SUN RISE/SET | 40960 | 30720 | THIS IS AN ON/OFF FILE CONTAINING SUN RISE AND SET TIMES. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|--------------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| TABULAR REPORTS | -0- | -0- | TABULAR REPORTS AND PLOTS TO BE INCLUDED IN DATA FLOW AND DATA SYSTEMS DOCUMENTATION. |
| TANRAY EPIEM | 409600 | 102400 | THIS IS A LIST-DIRECTED FILE CONTAINING A TIME HISTORY OF ORBITAL ELEMENT PARAMETERS, SUN RISE AND SET EVENTS, AND DATA RELATIVE TO THE TANGENT POINT ON THE ORBITER TO SUN LINE-OF-SIGHT. |
| TARGET ANALYSIS PRINTOUT | -0- | -0- | TARGET ANALYSIS STATISTICAL AND GROUPING DATA THAT AIDS IN BLOCKING OUT EXPERIMENT OPERATION TIMES AND OVERALL SCHEDULING. |
| TDRS AC/LOS | 614400 | 409600 | THIS IS AN ON/OFF FILE CONTAINING TDRS ACQUISITION AND LOSS TIMES. THIS IS A SUBJECT TYPE 6 ON/OFF FILE. |
| TERM AC/LOS | 204800 | 102400 | THIS IS AN ON/OFF FILE CONTAINING THE TIMES OF THE ORBIT TERMINATORS. THE "ON" TIME BEING THE TIME OF DAYLIGHT TO DARKNESS TRANSITION AND THE "OFF" TIME BEING THE TIME OF DARKNESS TO DAYLIGHT TRANSITION. THIS IS A SUBJECT TYPE 0 ON/OFF FILE. |
| TIME GOAL | 12288 | 6144 | THIS IS AN ASCII FILE DEVELOPED BY THE PI DEFINING THE CLASS OF THE OBJECT(S) TO BE VIEWED, THE TIME TO BE SCHEDULED DURING THE DAY, AND THE TIME TO BE SCHEDULED AT NIGHT. |

| INPUT/OUTPUT NAME | FILE SIZE(BYTES) | | INPUT/OUTPUT DESCRIPTION |
|-------------------|------------------|---------|---|
| | MAXIMUM | MINIMUM | |
| VERIF Y ERROR | -0- | 20480 | THE VERIF Y ERROR FILE LIST ALL ERRORS ENCOUNTERED BY THE DATA VERIFICATION MODULE WHEN COMPARING THE SCHEDULES ON THE MASTER FILE TO OTHER DEFINITION AND/OR SCHEDULE FILES. |
| -0- | -0- | -0- | |

TABLE 8
MANUAL INPUT/OUTPUT SUMMARY

MANUAL INPUT OUTPUT SUMMARY DATA

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| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|---------------------------------|--------------------------|----------------------|---|
| ATMOS DESIRED VEHICLE ATTITUDE | VERBAL, WRITTEN | N/A | THE DESIRED VEHICLE ATTITUDE FOR THE ATMOS EXPERIMENT IS DEFINE BY TIME, REFERENCE COORDINATE SYSTEM, PITCH, YAW, AND ROLL. |
| ATTITUDE UPDATES | VERBAL, WRITTEN | N/A | ATTITUDE UPDATES BASED ON ATTITUDE/TDRS ITERATION AND REVIEW CYCLE COMMENTS/INPUTS. |
| BASIC CO-ORBITING REQUIREMENTS | VERBAL, WRITTEN | EFORD | A DETERMINATION OF THE TYPES OF ORBIT OPERATIONS THAT ARE REQUIRED: PROXIMITY OPERATIONS, DEPLOYMENT, RENDEZVOUS, GRAPPLE/CAPTURE. |
| BASIC CREW CYCLE | VERBAL/INFORMAL DOCUMENT | N/A | BASIC SLEEP/ WORK CYCLES OBTAINED FROM JSC. |
| BORB CONSTRAINTS | VERBAL, WRITTEN | N/A | CONSTRAINTS TO CONSIDER WHEN DEVELOPING PLASMA PHYSICS TARGET OPPORTUNITIES INCLUDE: B AZ, B ELEV, B DOT X, L SHELL (TYPICAL). |
| CDMS DICTIONARY | DOCUMENT | N/A | COMMAND AND DATA MANAGEMENT SYSTEM DICTIONARY (PDF DOCUMENT). CONTAINS DISPLAY DEFINITIONS. |
| CDMS SYSTEM DEFINITION | DOCUMENTS | N/A | COLLECTION OF DOCUMENTS THAT DESCRIBE THE CDMS FOR THE FLIGHT IN QUESTION (SPAN,ICD'S,ETC.). |
| CEL TARGET(S) ELEV ANGLE CONSTS | VERBAL, WRITTEN | N/A | ELEVATION ANGLE CONSTRAINTS RELATIVE TO CELESTIAL OBJECT(S) VIEWING PERIODS. |
| CELESTIAL TARGETS | VERBAL, WRITTEN | N/A | CELESTIAL TARGET DATA (NUMBER, NAME, RIGHT ASCENSION, AND DECLINATION) ARE INPUT MANUALLY USING THE STAR PROGRAM TO CREATE A CANDIDATE CELESTIAL TARGET FILE. |
| COMMANDS FROM PI | INFORMAL DOCUMENTS | N/A | CHRONOLOGICAL LISTING OF ALL COMMANDS TO THE PI'S EXPERIMENT. |

| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|-------------------------------|--------------------------|----------------------------------|--|
| COOLING LOAD REQMTS | INFORMAL DOCUMENT | N/A | HEAT REJECTION PROFILES FOR THE INTEGRATED PAYLOAD. |
| CREW H/O CYCLE | VERBAL, WRITTEN | N/A | CREW H/O TIMES, DURATIONS |
| CREW PROCEDURES DOCUMENT | DOCUMENT | PDF | PDF DOCUMENT GIVING A COMPLETE DESCRIPTION OF ALL PAYLOAD CREW PROCEDURES. |
| CREW SHIFT TIMES | VERBAL/INFORMAL DOCUMENT | N/A | CREW SHIFT TIMES. COMES OUT OF BASIC CREW CYCLE INPUT FROM JSC. |
| DATA FLOW ANALYSIS FDD INPUTS | VERBAL, WRITTEN | FDD | DATA FLOW ANALYSIS INPUTS TO THE FLIGHT DEFINITION DOCUMENT. |
| EARTH OBSERV SUN ELEV CONSTS | VERBAL, WRITTEN | N/A | SUN ELEVATION ANGLE CONSTRAINTS RELATIVE TO EARTH OBSERVATION PERIODS. |
| ECOS T/L PRINTOUTS | PRINTOUT | N/A | CONTENTS OF ALL ECOS MASTER AND SUBORDINATE TIMELINES |
| ECOS TIMELINE DOCUMENT | DOCUMENT | ECOS TIMELINE DOCUMENT | CONTENTS OF ALL ECOS MASTER AND SUBORDINATE TIMELINES PLUS SOME BOILERPLATE PAGES |
| ERD'S | DOCUMENT | EXPERIMENT REQUIREMENTS DOCUMENT | FUNCTIONAL OBJECTIVES (FO'S), EQUIPMENT IDENTIFICATION, OPERATIONAL FUNCTIONAL FLOW, STRUCTURAL/MECHANICAL, POINTING & STABILITY PARAMETERS, ALIGNMENT/CO-ALIGNMENT, ORBITAL REQUIREMENTS AND CONSTRAINTS, ELECTRICAL REQUIREMENTS, THERMAL CONTROL/ FUIID REQUIREMENTS, DATA SYSTEM REQUIREMENTS DOCUMENT, FLIGHT SW - SUMMARY OF EXP. COMPUTER SW REQMTS, PHYSICAL CONFIGURATION, MISSION OPERATIONS SUPPORT, POCC REQMTS, SL DATA PROCESSING FACILITY (SLDPF), OTHER DATA PRODUCTS. |

MANUAL INPUT OUTPUT SUMMARY DATA

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| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|----------------------------|----------|----------------------------|--|
| ESS MODELS PRINTOUT | PRINTOUT | N/A | PRINTOUT OF THE ESS MODEL FILE. |
| ESS TARGET PRINTOUTS | PRINTOUT | N/A | PRINTOUT OF THE ESS TARGET FILE. |
| EXP T/L PRINTOUTS | PRINTOUT | N/A | PRINTOUT OF THE EXPERIMENT TIMELINE FILE. |
| FLIGHT DEFINITION DOCUMENT | DOCUMENT | FLIGHT DEFINITION DOCUMENT | SUMMARY - MISSION OVERVIEW, LAUNCH WINDOW, ORBIT DEFN, ATTITUDE DATA, PAYLOAD ON-ORBIT ACTIVITIES, PAYLOAD TIMELINE SUMMARY (PTS CHARTS), ORBITER CREW PAYLOAD SUPPORT REQMTS, CONTAMINATION AVOIDANCE REQMTS, RESOURCE UTILIZATION, CREW TIME USAGE, ELECTRICAL POWER & ENGERGY USAGE, ECOS USAGE, ECAS USAGE; ORBITAL MECHANICS AREA - NOMINAL FLIGHT PROFILE, LAUNCH DATE/TIME, ORBIT DATA REV BY REV, GROUND TRACKS & ORBITAL LIGHTING DATA, COMMUNICATIONS OPPORTUNITIES, CONSTRAINTS, K-BAND, S-BAND & GROUND COVERAGE. DETAILED T/L DATA; APP B - COMPOSITE T/L TABLES, INDIVIDUAL EXP T/L TABLES, CREW & SYSTEM T/L TABLE. |
| FO'S | WRITTEN | ERD'S, FO INPUT SHEETS | FUNCTIONAL OBJECTIVES DEFINITION, RESOURCE AND TIME REQMTS, STEP REQMTS, ETC.. |
| FPA INPUTS | DOCUMENT | FLIGHT PLANNING ANNEX | LAUNCH WINDOW DATA, ORBITAL PARAMETERS DATA, PAYLOAD ELECTRICAL POWER REQMTS - TOTAL SUBSYSTEM PAYLOAD OVER CHARTS/TABLES, RESISTIVE VS CONSTANT POWER USAGE, ESSENTIAL PWR REQMTS, COOLING LOAD REQUIREMENTS, ORITER EQUIPMENT USAGE, FLIGHT ACTIVITY REQUIREMENTS, CREW ACTIVITY REQUIREMENTS, MODEL SUMMARY, PCAP TAPE, ORBITER CREW PAYLOAD SUPPORT, ATTITUDE DATA, EXP. POINTING REQMTS, OBJECT AVOIDANCE DATA, CONTAMINATION AVOIDANCE, PHOTO REQMTS, TV REQUIREMENTS, TV SCIENCE DATA/SPONSOR, VCR REQUIREMENTS. |

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| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|----------------------------|-----------------|----------------------|---|
| GROUND SITE POLYGONS | VERBAL, WRITTEN | N/A | GROUND SITE POLYGONS ARE DEFINED BY NAME OF SITE, NUMBER OF SIDES (VERTICES) FOR EACH SITE, AND GEODETTIC LATITUDE AND LONGITUDE FOR EACH VERTEX. |
| GUIDE STAR CHOICES | VERBAL, WRITTEN | N/A | A LIST OF GUIDE STARS TO BE USED FOR SOLAR OBJECTIVE LOAD GENERATION. THE LIST IS BY NAME AND NUMBER. |
| HDRR DUMP TIMES | VERBAL, WRITTEN | N/A | HDRR DUMP TIMES FROM THE ONBOARD RECORDER PLAYBACK SCHEDULE. |
| INPUTS FROM PI'S | VERBAL, WRITTEN | N/A | INPUTS FROM EXPERIMENT PI'S RELATIVE TO ECOS TIMELINES (STL'S,MTL'S). |
| INPUTS TO MISSION PLANNING | VERBAL, WRITTEN | N/A | DATA PROVIDED FOR MISSION PLANNING PERSONNEL RELATIVE TO CO-ORBITING TARGETS REQMTS/CONSTRAINTS TO BE USED IN DEVELOPING MISSION PROFILE AND EXPERIMENT OPERATION: BLOCK OUT OPERATIONAL TIME PERIODS, PROVIDE TARGET PERIODS FOR FO'S, IDENTIFY MAJOR EVENTS, PROVIDE A MANEUVER OR ATTITUDE TIMELINE. |
| INPUTS/UPDATES TO IPRD | VERBAL, WRITTEN | IPRD | INITIAL INPUTS OR UPDATES MADE TO THE IPRD AS A RESULT OF THE PAYLOAD DATA COLLECTION EFFORT AND ASSOCIATED ANALYSES. |
| INPUTS/UPDATES TO O&IA | VERBAL, WRITTEN | O&IA | INITIAL INPUTS OR UPDATES MADE TO THE O&IA AS A RESULT OF THE PAYLOAD DATA COLLECTION EFFORT AND ASSOCIATED ANALYSES. |

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| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|---------------------------------|-------------------------|----------------------|---|
| IPRD DOCUMENT | DOCUMENT | N/A | INTEGRATED PYLDS RMTS DOCUMENT DEFINING: CONFIGURATION; MASS PROPERTIES; STRUCTURAL MECHANICAL REQUIS; ELECTRICAL POWER AND ENERGY; THERMAL/ENVIRONMENTAL CONTROL; ON-ORBIT VEHICLE DYNAMICS; POINTING & ALIGNMENT; CDMS; SOFTWARE; CREW SYSTEMS; STOWAGE; GROUND OPS; FLIGHT OPS; MISSION UNIQUE ENVIRONMENTS. |
| LATITUDE CONSTRAINTS | VERBAL, WRITTEN | N/A | LATITUDE CONSTRAINTS APPLIED TO HEMISPHERE OPPORTUNITIES SELECTION RELATIVE TO PLASMA PHYSICS TARGET OBSERVATION PERIODS. |
| LAUNCH WINDOW/LAUNCH TIME DATA | VERBAL, WRITTEN | N/A | LAUNCH WINDOW OPENING AND CLOSING (GMT), AND TIME OF LAUNCH (GMT) |
| MGMT AGREEMENT ON GROSS MSN T/L | VERBAL, WRITTEN | N/A | DATA PROVIDING AN AGREEMENT ON THE GROSS MISSION TIMELINE TO BE USED FOR THE MISSION. MISSION PLANNING ITERATIONS TO EXPAND AND/OR REFINE THIS TIMELINE ARE EXPECTED IN ORDER TO RESOLVE PROBLEMS OR CONFLICTS DISCOVERED DURING MORE DETAIL PLANNING ACTIVITIES. |
| MISSION CONFIGURATION | VERBAL/WRITTEN/DOCUMENT | SPAH | LAUNCH TIME, MISSION DURATION, CREW DATA, AND THE TYPES OF RESOURCES AND EQUIPMENT TO BE USED DURING THE MISSION ALONG WITH THEIR AVAILABILITIES. |
| MISSION PROFILE CONSIDERATIONS | VERBAL, WRITTEN | N/A | CONSIDERATIONS IN DEVELOPING A GROSS MISSION TIMELINE INCLUDE: EXP. PRIORITIES, DIFFICULTY IN SCHEDULING EXP. OPERATIONS, SYSTEMS REQMTS, CREW R/O CYCLE, DATA MANAGEMENT, AND MANAGEMENT DIRECTIONS. |
| MISSION T/L ANALYSIS FDD INPUTS | VERBAL, WRITTEN | FDD | MISSION TIMELINE ANALYSIS INPUTS TO THE FLIGHT DEFINITION DOCUMENT.. |

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| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|----------------------------------|------------------|----------------------|--|
| MISSION T/L ANALYSIS FPA INPUTS | VERBAL, WRITTEN | FPA | MISSION TIMELINE ANALYSIS INPUTS TO THE FLIGHT PLANNING ANNEX. |
| MISSION TL TABLES, PRINTOUTS | TABLES, PRINTOUT | N/A | DETAILED TABULATIONS, PLOTS AND PRINTOUT OF THE MISSION TIMELINE. |
| MODIFICATIONS | VERBAL, WRITTEN | N/A | MODIFICATIONS MADE TO DATA FLOW SCHEDULES BASED ON REVIEW AND USER EXPERIENCE. |
| MPE OPS REQMTS | DOCUMENT | ICD | MISSION PECULIAR EQUIPMENT OPERATIONAL REQUIREMENTS. |
| MSN EXP OPPORTUNITIES DATA | VERBAL, WRITTEN | N/A | THIS DATA INCLUDES PRINTOUTS AND/OR PLOTS GENERATED BY THE EXPERIMENT OPPORTUNITIES GENERATION SUBFUNCTION. THESE DATA REPRESENT EXPERIMENT OPERATION PERIODS WHERE REQUIREMENTS AND/OR CONSTRAINTS HAVE BEEN SATISFIED. THESE DATA INCLUDE: MISSION INDEPENDENT, CELESTIAL, ATMOSPHERIC PHYSICS, SOLAR, EARTH OBSERVATION, PLASMA PHYSICS, AND CO-ORBITING TARGET OPPORTUNITIES DATA. |
| MSN SUPPORT PERSONNEL/POH INPUTS | VERBAL, WRITTEN | POCC CHECKLIST | INPUTS TO THE POCC CHECKLIST FROM MISSION SUPPORT PERSONNEL AND EXTRACTED FROM THE POCC OPERATIONS HANDBOOK. |

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| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|-------------------------------|--------------------------|----------------------|---|
| Q&IA | DOCUMENT | Q&IA DOCUMENT | FUNCTIONAL DESCRIPTION; PAYLOAD ELEMENT H/W DESCRIPTION, INSTRUMENT OPERATING MODES, DELIVERABLE ITEMS LIST, SHIPPING INSTRUCTIONS. FLIGHT OPS SUPPORT; FUNCTIONAL OBJECTIVES, POCC SERVICES & DATA PRODUCT REQUIREMENTS, ORBITAL REQMTS & CONSTRAINTS, TRAINING REQMTS. PHYSICAL INTEGRATION; ACTIVITY FLOW, KSC SUPPORT REQUIREMENTS, ON-LINE ACTIVITY REQMTS, PERSONNEL SUPPORT REQMTS, LAUNCH DELAY CONTINGENCY PLANNING, ALTERNATE LANDING SITE CONTINGENCY PLANNING. PROCEDURES INPUTS; KSC PROCEDURES, ON-BOARD FLIGHT PROCEDURES, POCC PROCEDURES, APP A - SPACELAB CDMS REQUIREMENTS FORMS & DATA FORMATS; APP B - MISSION T/L FUNCTIONAL OBJECTIVE REQMTS SHEETS. |
| ORBIT ANALYSIS FDD INPUTS | VERBAL, WRITTEN | FDD | ORBIT ANALYSIS INPUTS TO THE FLIGHT DEFINITION DOCUMENT. |
| ORBIT ANALYSIS FPA INPUTS | VERBAL, WRITTEN | FPA | ORBIT ANALYSIS INPUTS TO THE FLIGHT PLANNING ANNEX. |
| ORBIT DEFINITION PARAMETERS | VERBAL, WRITTEN | N/A | ORBIT ATTITUDE (KM) AND ORBIT INCLINATION (DEG) |
| OTHER PI INPUTS | VERBAL, WRITTEN | N/A | INPUTS MADE BY THE PI'S TO BE INCORPORATED INTO THE PAYLOAD FLIGHT DATA FILE. |
| PAO REQMTS | VERBAL/INFORMAL DOCUMENT | N/A | PUBLIC AFFAIRS OFFICE REQUIREMENTS PERTAINING TO TV PHOTOACTIVITY OF SELECTED EXPERIMENT ACTIVITY. |
| PAYLOAD COMPLEMENT DEFINITION | VERBAL, WRITTEN | FDD | DATA DEFINING THE EXPERIMENTS TO BE FLOWN ON THE MISSION. |
| PAYLOAD SYSTEMS HANDBOOK | DOCUMENT | PFDF | PFDF DOCUMENT CONTAINING CREW PROCEDURES SPECIFIC TO OVERALL PAYLOAD SYSTEMS. |

| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|----------------------------------|-----------------|----------------------------|--|
| PCAP CHARTS | CHARTS | PCAP | DETAILED 1-HR CHARTS PRODUCED ON A HIGH RESOLUTION LASER PRINTER WHICH BECOME PART OF THE PDF. THE PLAN SHOWS THE PROCEDURES THE CREW EXECUTES AND ASSOCIATED NOTES. IT ALSO SHOWS AUXILIARY INFORMATION SUCH AS ORBIT ATTITUDE AND MANEUVERS, SUN/SHADOW, K-AND S-BAND COMMUNICATIONS, HRM FORMAT, RECORDER USAGE, ORBIT REVOLUTION NUMBERS AND OTHER DATA. |
| PCAP DOCUMENT | DOCUMENT | PAYLOAD CREW ACTIVITY PLAN | BOUND TABBED DOCUMENT CONTAINING THE PAYLOAD CREW ACTIVITY PLAN (PCAP) CHARTS AND THE PAYLOAD TIMING SUMMARY (PTS) CHARTS, INCLUDED IN THE PAYLOAD FLIGHT DATA FILE TO ASSIST THE CREW IN ON-BOARD PAYLOAD OPERATIONS. |
| PDF DOCUMENTS | DOCUMENT | PDF | PAYLOAD FLIGHT DATA FILE DOCUMENTS. |
| PI INTERFACE | VERBAL, WRITTEN | N/A | DISCUSSIONS WITH PI'S TO OBTAIN EXPERIMENT REQUIREMENTS AND CONSTRAINTS, CLARIFICATION OF INPUTS, AND COORDINATION OF UPDATES. |
| PI ORBIT TERM REQMTS | VERBAL, WRITTEN | N/A | PI REQMTS/CONSTRAINTS TO CONSIDER WHEN SELECTING ORBIT TERMINATOR TARGETS AS RELATED TO ATMOSPHERIC PHYSICS OBSERVATION PERIODS. |
| PI REQUIREMENTS/CONSTRAINTS | VERBAL, WRITTEN | N/A | EVALUATION AND DEVELOPMENT OF PI CO-ORBITING REQUIREMENTS/CONSTRAINTS FOR INPUT: INSTRUMENT REQUIREMENTS, LIGHTING REQUIREMENTS, LINE-OF-SIGHT POINTING, DATA/COMMUNICATION REQUIREMENTS. |
| PL PHYSICS DESIRED VEH. ATTITUDE | VERBAL, WRITTEN | N/A | THE DESIRED VEHICLE ATTITUDE WHEN COMPUTING ORIENTATION AND STRENGTH OF THE MAGNETIC FIELD FOR PLASMA TARGETS. |

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| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|-------------------------------|-------------------|----------------------|---|
| PL/EXP CONSTRAINTS | VERBAL, WRITTEN | ERD'S, O&IA | TYPICAL PAYLOAD/EXPERIMENT CONSTRAINTS TO CONSIDER WHEN DETERMINING LAUNCH WINDOW AND LAUNCH TIME INCLUDE: BETA ANGLE, COMM COINCIDENCES, SUN OR MOON CONSTRAINTS, ORBIT OR EARTH DARKNESS, TARGET VIEWING, PAYLOAD DEPLOYMENT |
| PRELIM MMU MAP | INFORMAL DOCUMENT | N/A | NAMES OF ALL FILES ON MMU |
| PRINTOUT OF ECOS TIMELINE | PRINTOUT | N/A | PRINTOUT OF THE ECOS TIMELINE TO BE USED IN THE ECOS TIMELINE DOCUMENT DEVELOPMENT. |
| PRINTOUTS OF CAND GUIDE STARS | PRINTOUT | N/A | THE CANDIDATE GUIDE STAR PRINTOUTS CONTAIN DATA FOR GUIDE STAR SELECTION WHICH INCLUDE: STAR NAME AND NUMBER, STAR NO. OF NEXT HIGHER/LOWER RIGHT ASCENSION, STAR NO. OF THE NEXT HIGHER/LOWER DECLINATION, STAR MAGNITUDE AND MAGNITUDE CORRELATION DATA, TIME RANGES WHERE STAR CAN BE USED FOR IPS ALIGNMENT, IPS ROLL ANGLES AT BEGINNING/END OF TIME RANGES. |
| PRINTOUTS OF STRAY LIGHT DATA | PRINTOUT | N/A | PRINTOUT OF STRAY LIGHT DATA TO BE USED WHEN SELECTING SOLAR GUIDE STARS. |

| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|-------------------------------|-----------------|----------------------|---|
| PTS CHARTS | CHARTS | PCAP | 6-HOUR PLOTS WITH MET & AMET TICK MARKS EVERY 6 MINUTES. THE CHARTS ARE USED FOR ANALYSIS BY MISSION PLANNING ENGINEERS AND FOR MISSION TIMELINE PUBLICATION. ITEMS ON EACH PLOT PAGE INCLUDE CREW ACTIVITIES; S-BAND, K-BAND, AND GSTON COMMUNICATION COVERAGES; DATA & VIDEO DOWNLINKS; DATA & VIDEO RECORDER DUMPS; WATER DUMP & THRUSTER FIRE INHIBITS; ATTITUDE & MANEUVER PROFILE; RADEC OF ORBITER - Z AXIS FOR INERTIAL ATTITUDES; A MOON PHASE DWG; ORBITER REVOLUTION #'S AND A GROUND TRACK PLOT; SUN/SHADOW PERIODS; AND UNATTENDED (NO CREW) PAYLOAD OPERATIONS. PTS CHARTS ARE PART OF THE PPDF AND THE STANDARD POCC DOCUMENTATION FOR SPACELAB MISSION, AND ARE INCLUDED IN THE FLIGHT DEFINITION DOCUMENT. |
| RADIATION CONSTRAINTS | VERBAL, WRITTEN | N/A | RADIATION ENVIRONMENT CONSTRAINTS RELATIVE TO EXPERIMENT OPERATIONS (MEV, FLUX, PARTICLE/C2) WHEN DEFINING THE SOUTH ATLANTIC ANOMALY PERIODS. |
| RESERVE PERIOD CONSIDERATIONS | VERBAL, WRITTEN | N/A | THE FACTORS TO CONSIDER WHEN DEVELOPING A RESERVE PERIOD FILE ARE: 1) TIME FOR TARGET, 2) CREW H/O CYCLES, 3) SYSTEMS REQUIREMENTS, 4) PAO ACTIVITIES, 5) MOVING TARGETS. |
| REVIEW CYCLE UPDATES | VERBAL, WRITTEN | N/A | UPDATES MADE TO THE MISSION TIMELINE AS A RESULT OF ATTITUDE/TORS ITERATIONS, DIVISION MANAGEMENT REVIEWS, JSC REVIEWS, AND ECOS MISSION TIMELINE PERSONNEL REVIEWS. |
| SHIFT TIMES | VERBAL, WRITTEN | N/A | CREW SHIFT TIMES. |
| SL/PL INTERFACE DEFINITION | DOCUMENT | ICD-B | DEFINITION OF THE OPERATIONAL INTERFACES BETWEEN SL SYSTEMS AND P/L SYSTEMS. |

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| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|----------------------------------|--------------------------|----------------------|--|
| SOPG TIMES | VERBAL/INFORMAL DOCUMENT | N/A | TIMES THAT THE SCIENCE OPERATIONS PLANNING GROUP IS TO MEET DURING THE MISSION. |
| SPAH AND SL SYSTEM DOCUMENTATION | DOCUMENT | SPAH | SL CONFIGURATION, OPERATION AND ALLOCATIONS |
| SPECIAL CHECKS | VERBAL, WRITTEN | N/A | SPECIAL CHECKS OF DATA FLOW SCHEDULES IN THE PROCESS OF VERIFICATION. |
| SPECIAL CREW NOTES | VERBAL | N/A | SPECIAL NOTES REQUIRED BY THE CREW FOR INPUT INTO THE PCAP NOTES COLUMN. |
| SPECIAL TARGET | INFORMAL | N/A | TARGETS (EXPERIMENT OPPORTUNITIES), WHICH BECAUSE OF EITHER THE TIMING OF THEIR IDENTIFICATION OR THEIR NATURE ALLOWS THEM TO BE BUILT MORE EFFICIENTLY IN THE CREATE ESS TARGET FILE SUBFUNCTION RATHER THAN THE EXPERIMENT OPPORTUNITIES GENERATION SUBFUNCTION. |
| SPECIAL TIMELINE NOTES | INFORMAL PAPER | N/A | SPECIAL NOTES FROM TIMELINE ANALYSIS ENGINEERS REQUIRED TO BE INPUT IN THE PCAP NOTES COLUMN. |
| STATE VECTOR FROM JSC | VERBAL, WRITTEN | N/A | STATE VECTOR DEFINITION AS FOLLOWS: RADIUS (NMI) VELOCITY (FT/SEC), FLIGHT PATH ANGLE (DEG), GEOCENTRIC LATITUDE (DEG), LONGITUDE (DEG), INERTIAL AZIMUTH (DEG), AND TIME (MET-HRS). |
| STATE VECTOR PRINTOUT | PRINTOUT | N/A | PRINTOUT PROVIDES THE STATE VECTOR DEFINITION AS FOLLOWS: RADIUS (KM), VELOCITY (KM/SEC), FLIGHT PATH ANGLE (DEG), GEOCENTRIC LATITUDE (DEG), LONGITUDE (DEG) INERTIAL AZIMUTH (DEG), AND ORBIT INSERTION TIME (MET-HRS.) |
| STL PRINTOUTS | PRINTOUT | N/A | PRINTOUT OF THE ECOS SUBORDINATE TIMELINES. |

| INPUT/OUTPUT NAME | TYPE | DOCUMENT INCLUDED IN | INPUT/OUTPUT DESCRIPTION |
|--------------------------------|-----------------|----------------------|---|
| STORAGE BOOK | DOCUMENT | PDF | THE STORAGE BOOK CONTAINS A LIST OF ALL EQUIPMENT STORED ONBOARD AND IT'S LOCATION. |
| STS CAPABILITIES DOCUMENTATION | DOCUMENTS | N/A | DOCUMENTATION DESCRIBING STS CAPABILITIES AND STS/PAYLOAD/INTERFACES IN DETAIL (SPA, ICD-A). |
| STS CONSTRAINTS | VERBAL, WRITTEN | STS DOCUMENTATION | TYPICAL STS CONSTRAINTS TO CONSIDER WHEN DETERMINING LAUNCH WINDOW AND LAUNCH TIME INCLUDE: LANDING SITE (NOMINAL AND ABORT), LIGHTING AT LAUNCH, LIGHTING AT LANDING, LAUNCH CAPABILITIES, COMM CONSTRAINTS. |
| STS REQMTS/CONSTRAINTS | VERBAL, WRITTEN | N/A | EVALUATION AND DEVELOPMENT OF STS CONSTRAINTS AS RELATED TO CO-ORBITING OPERATIONS FOR INPUTS: STS OPERATIONS CONSTRAINTS, CREW SAFETY, ORBITER FLYABILITY, SYSTEMS REQUIREMENTS/CONSTRAINTS. |
| SUN ELEV CONSTRAINTS (SOLAR) | VERBAL, WRITTEN | N/A | SUN ELEVATION ANGLE CONSTRAINTS RELATIVE TO SOLAR VIEWING PERIODS. |
| TV, PHOTO OPS HANDBOOK | DOCUMENT | PDF | PDF DOCUMENT CONTAINING CREW PROCEDURES ASSOCIATED WITH ON-ORBIT PAYLOAD TV AND PHOTO OPERATIONS. |
| TV, PHOTO SYSTEM CAPABILITIES | INFORMAL | N/A | DESCRIPTION OF THE NUMBER OF CAMERAS, DOWNLINK, ETC. AVAILABLE FOR PAYLOAD USE. INFORMAL AT FIRST, BUT FINALIZED IN THE ICD-B. |
| UPDATES TO MASTER INPUT FILES | VERBAL, WRITTEN | N/A | UPDATES MADE TO THE MASTER INPUT FILES BASED ON A DESKTOP OPERATIONAL VERIFICATION OF THE ECOS MASTER TIMELINES. |
| UPDATES TO STL'S | VERBAL, WRITTEN | N/A | UPDATES MADE TO THE SUBORDINATE TIMELINES BASED ON A DESKTOP OPERATIONAL VERIFICATION OF THE ECOS STL'S. |